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STATIC RADAR CROSS SECTION OF LIGHT
AIRCRAFT. VOLUME II. CHEROKEE 140
AT L-, S-, AND C-BANDS

Test Group (6585th)

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16. Abstract Static radar cross section (RCS) of a single-engine low-wing Piper PA-28-140 Cherokee (Cherokee 140) was measured at 2700, 2800, and 2900 MHz (S-Band) over a range of aircraft attitudes of $\pm 10^\circ$ pitch, 0° to 45° roll, and also at 1250, 1350, 5000, and 5400 MHz at 0° roll, 0° pitch. Median RCS in circular polarization and in linear polarization was independent of frequency. At S-Band the median RCS varied with roll angle most strongly in the two broadside directions. In those two directions the variation was greatest for linear polarization when the radar viewed the lower surface of the aircraft, and was greatest for circular polarization when the radar viewed the upper surface of the aircraft. Median RCS at S-Band was not affected by pitch angle.					
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SECTION I

1. Introduction

This document is Volume II of a three-volume set (References 1 and 2) on measurement of static radar cross section (RCS) of light aircraft. The measurement program comprised three types of single-engined aircraft:

- a. All-metal high wing
- b. All-metal low wing
- c. Fabric-covered (high wing)

Type (b), represented by a Cherokee 140, is the subject of this report. Measurements on a Cessna 150L, typical of type (a), are presented in Volume I, (Reference 1), and the data on type (c), a Super Cub, are given in Volume III, (Reference 2).

The objective of the program was to measure static RCS of typical light aircraft over a range of frequencies, polarizations, and aspect angles. These data would aid in defining the need for and amount of RCS enhancement on light aircraft to improve their visibility on FAA airport surveillance radars.

The data measurements were made at the Radar Target Scatter Division (RAT SCAT) of the 6585th Test Group located on the Alkali Flats, Holloman Air Force Base, New Mexico.

2. Description of Target

The RCS data presented in this report were measured on a Model PA-28-140 utility aircraft, serial number 28-24324, manufactured by the Piper Aircraft Corporation, Lock Haven, Pennsylvania. This aircraft is popularly designated as a Cherokee 140. An Airworthiness Certificate was issued 4 March 1968, a Certificate of Aircraft Registration was issued 23 October 1969.

The aircraft (Figure 1) was equipped with a VOR navigation antenna on the vertical stabilizer, and a VHF communication antenna on the cabin roof. The aircraft was flown to the RCS test site each Monday morning and was flown out each Friday afternoon for week-end use by the flying club that owned it.

3. Instrumentation

Data measurements were achieved with a long-pulse radar system operating on a ground plane range. Figure 2 illustrates the components of

the measurement range. The radar transmitter, receiver, control console, and data recorders were housed in a mobile van. The antennas were supported on a mobile tower. Rotation of the target in azimuth was controlled from the van through underground electrical conduit that also returned azimuth synchro signals to the van. A mobile test van was used on this program in order to operate over a range length not available on the RAT SCAT fixed ranges. Sections 1 and 2 of Appendix A and Table A-1 of Appendix A summarize the characteristics of RAT SCAT facilities and equipment. This information applies to the mobile-mounted equipment used for this program as well as to the fixed installations.

The RCS measurements presented in this report were obtained on a ground plane range wherein the target is measured near to the ground. The ground is present as a scattering object and in such a circumstance coupling between orthogonal components of the transmitted and received fields may exist other than the coupling introduced by the target. This coupling (or depolarizing) can be a problem only for measurement in circular polarization. References 3 and 4 describe the theoretical and experimental studies made on the subject specifically for the RAT SCAT range. It was shown that circular polarization measurements can be made on a ground plane range. The size of the useful target region is approximately the same when circular polarization is used as that obtained when linear polarization is used. The amplitude curvature in the vertical plane is the limiting factor in both cases. Standardized calibration procedures used at RAT SCAT assure the accuracy of RCS measurements using circular polarization.

The primary calibration standard for linear polarization was a 26.6 inch diameter precision aluminum sphere and was used at all measurement frequencies. The calibration standard for circular polarization was a ninety-degree dihedral corner with square faces 0.85 meters in each dimension. This corner was calibrated against the precision sphere, making use of a 45° dipole to transfer from linear to circular polarization. The secondary reference standard was a trihedral corner placed approximately at mid-range (see Figure 2). The RCS level of this secondary reference standard was used to set the decibel scale on all recorded RCS patterns and was utilized for both linear and circular polarization.

The radar antennas were parabolic dishes with dipole feeds for linear polarization and planar spiral feeds for circular polarization. Circularity was better than 0.7 db at all frequencies.

4. Procedure

Data on this program were obtained at seven frequencies and three polarizations over a range of roll and pitch angles as summarized in Table I.

Table I - RCS Measurement Matrix

Freq. (MHz)	Polarization	Roll Angle (Degrees)	Pitch Angle (Degrees)
1250	VV,HH,RR	0	0
1350	VV,HH,RR	0	0
2700	VV,--,RR	0,5,10,20,30,45	-10,-5,0,+5,+10
2800	VV,HH,RR	0,5,10,20,30,45	-10,-5,0,+5,+10
2900	VV,--,RR	0,5,10,20,30,45	-10,-5,0,+5,+10
5000	VV,HH,--	0	0
5400	VV,HH,--	0	0

The first letter designating polarization defines the polarization of the transmit antenna, the second letter defines polarization of the receive antenna. The nomenclature follows common practice in which only the spatial orientation of the electric field vector is specified: V for vertical, H for horizontal, R for right-hand circular.

The aircraft was supported at the azimuth turntable on two dielectric columns. The two columns were expanded polystyrene fabricated in vertical wedge sections to form a cone with circular cross section. On top of each column was transition section of expanded polystyrene pre-shaped to fit the aircraft fuselage and support the target on the columns at the desired roll and pitch attitude. The aircraft was lifted by a mobile crane (Figure 4) and lowered onto the transition sections on top of each column. The aircraft was held in place on the transitions by lengths of parachute cord tied to the fuselage and wings and anchored to winches on the azimuth turntable. In Figure 4 the aircraft has been secured in place and the lifting straps from the crane are being detached from the aircraft. Figure 5 shows the aircraft mounted for RCS measurement at 20° roll, 0° pitch. A more dramatic view is Figure 6, showing the aircraft at 45° roll and 10° negative pitch. The polystyrene collar visible around the aft fuselage in each Figure is the bearing structure for the aft lifting strap from the crane. Target RCS measured at S-Band with and without the collar in place gave identical results. Therefore the collar was left in place at all times to facilitate attaching the aircraft to the crane in case surface wind velocity suddenly increased or became gusty.

The aircraft was set accurately to the desired roll and pitch angle by adjusting tension in the tie-down lines. Roll angle and pitch angle were measured at the propeller hub. The propeller was rotated to an index line, the spinner was removed and a protractor was placed on the propeller hub. Roll angle was measured by sighting a transit on the protractor. Pitch angle was measured on the front of the propeller hub with an inclinometer. Pitch angle thus was measured with respect to the axis of the engine crankshaft. Both roll and pitch were set to an accuracy of ± 0.5 degrees.

Target height was 14 feet at L- and C-Bands, 16 feet at S-Band. Target height is the vertical distance between ground level and a horizontal line (the pitch axis) through the half-height of the fuselage midway between aircraft nose and tail, measured with the aircraft in level flight attitude. Figure 3 illustrates the relation between target height and the axes of pitch and yaw.

A target being measured on a ground plane range is in an electromagnetic field that is the vector sum of (1) the wave energy that travels directly from antenna to target, (2) the wave energy reflected from the surface of the earth, and (3) the wave energy that travels along the surface of the earth. The vector sum produces an interference pattern. Antenna height is adjusted so that the target is located in the first lobe of the interference pattern described by the equation:

$$H_a = \frac{\lambda R}{4H_t}$$

where H_a is antenna height, H_t is target height, λ is wavelength, and R is range length. In order to obtain accurate RCS measurements the following steps are required.

- (a) Adjust antenna height and pointing direction for best field uniformity across the volume occupied by the target.
- (b) Tilt the azimuth turntable so the target rotates in the plane of the antenna beam.
- (c) Minimize reflections from target supports, tie-downs and turntable.
- (d) Calibrate the range at all frequencies and polarizations.

The range parameters used on the program after the above steps were carried out at each frequency are summarized in Table II.

Measurement of RCS began with the three frequencies in S-Band: 2700 MHz, 2800 MHz and 2900 MHz. Because of the large effort required to mount the target for measurement it was prudent to record as many RCS patterns as possible for each mounting of the target. For that reason four antennas were used: one pair for linear polarization, the other pair for circular polarization. The antenna pairs were switched to the radar transmitter and receiver as required. That arrangement allowed measurement of seven RCS patterns for each roll/pitch attitude of the target. If weather conditions remained favorable upon completion of such a pattern group the target was set to the next roll/pitch attitude and RCS measurement went on, continuing until dark. A post-test calibration was then made. If weather became unfavorable (wind velocity above 10 knots, or rain) the target was removed from the turntable and the post-test calibration was made as quickly as possible. In any event the aircraft was not measured in wind velocity above 10 knots or after dark, and the range was calibrated each day before and after a sequence of measurement. The post-test calibration for one day was not used as the pre-test calibration for the next sequence of measurement. The RCS measurements taken each day began with a new calibration.

At L-Band (1250 MHz and 1350 MHz) one pair of antennas was used instead of two pair because only one target roll/pitch attitude was measured (see Table I). The two linear patterns were measured at each frequency followed by the post-test calibration. The linear feeds were replaced by the circular feeds, and a new calibration was made at the two frequencies. The required patterns in circular polarization were then measured and were followed by a post-test calibration.

5. Results

The RCS patterns presented in this report are copies of the original recorded data. Each pattern includes a calibration reference level from which the decibel scale was labelled. Each pattern also is marked with azimuth angle (horizontal scale at the bottom), and is identified as to pitch and roll angle, all of which are defined in Appendix B.

The patterns represent considerable amount of data, containing as they do the dependence of RCS on all combinations of frequency, polarization, roll angle, and pitch angle. Representative relationships, making use of median RCS to simplify the data, are shown in Figures 7 through 10. Median RCS was computed for a 10 degree azimuth increment with a 5 degree overlap.

With the aircraft at zero degrees roll and zero degrees pitch the median RCS in linear polarization and circular polarization was independent of frequency over the range of measurement. In linear polarization the

median RCS was the same for VV as for HH, while a single measurement (Run 527 at 2800 MHz, 0° roll, 0° pitch) indicated a median RCS for VH polarization lower by about 10 db.

The measured RCS patterns show the affect of frequency only for the zero degrees roll, zero degrees pitch attitude of the aircraft. It can be noted that the ratio of fuselage length (7.1m) to wavelength is between 30 and 128 over the measured frequency range, indicating that RCS is in the geometric optics region. From this it can be inferred that median RCS may be independent of frequency at all attitudes of the aircraft.

Figure 7 indicates the affect of positive roll angle (at zero degrees pitch) on median RCS at 2800 MHz, VV polarization. Contours of equal RCS in dbsm are plotted in the azimuth-roll plane. Vertical lines through the plot indicate the affect of roll angle at a given azimuth direction. Variation of RCS with roll (at 0° pitch) was greatest in the 270° broadside direction in which the radar viewed the underside of the aircraft as the aircraft rolled.

In circular polarization the reverse was true--the 90° broadside direction in which the radar viewed the upper surface of the aircraft as the aircraft rolled had the greater RCS variation with roll angle. Figure 8 shows the affect of roll angle (at zero degrees pitch) on median RCS.

The affect of pitch angle (at zero degrees roll) is illustrated in Figure 9 for VV polarization at 2800 MHz. Vertical lines through the graph show the affect of pitch angle for fixed azimuth directions. The variation over the ± 10 degree range of pitch angle is less than 5 db, a trivial amount. The same can be said of circular polarization shown in Figure 10.

6. References

- (1) AFSWC-TR-73-46, Volume I, Static RCS of Light Aircraft, Cessna 150L at L-, S-, C-Band, Dec 1973.
- (2) AFSWC-TR-73-46, Volume III, Static RCS of Light Aircraft, Super Cub at L-, S-, C-Band, Dec 1973.
- (3) RADC-TDR-63-484, An Analysis of the Polarization Capabilities of a Ground Plane Cross Section Range, October 1963.
- (4) RADC-TDR-64-380, Experimental Results of Circular Polarization and Scattering Matrix Measurements, June 1964.

Table II Test Range Parameters for Cherokee 140 RCS Measurements

Freq. (MHz)	Target Ht. (ft)	Antenna Ht.	Antenna Diameter (ft)	Range (ft)	Table Tilt (Min)
1250	14	12'-6"	10	1150	42
1350	14	12'-6"	10	1150	42
2700	16	12'-6"	6	2100	26
2800	16	12'-6"	6	2100	26
2900	16	12'-6"	6	2100	26
5000	14	6'-5"	4	2100	23
5400	14	5'-10"	4	2100	23

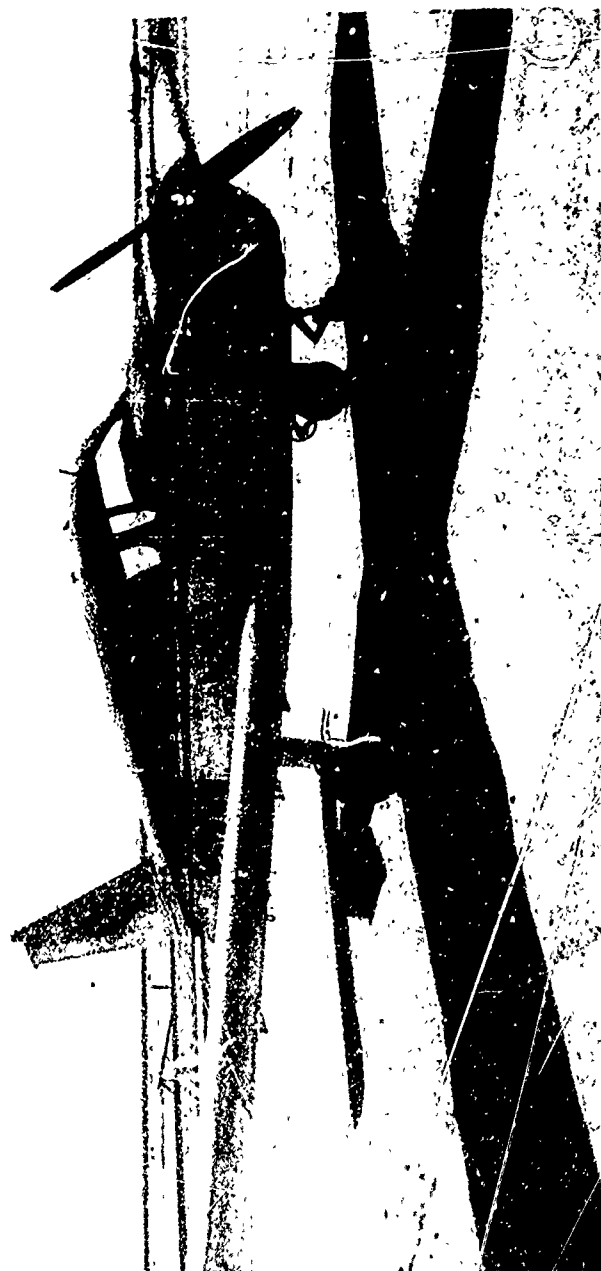


Figure 1 Cherokee 140 Light Aircraft

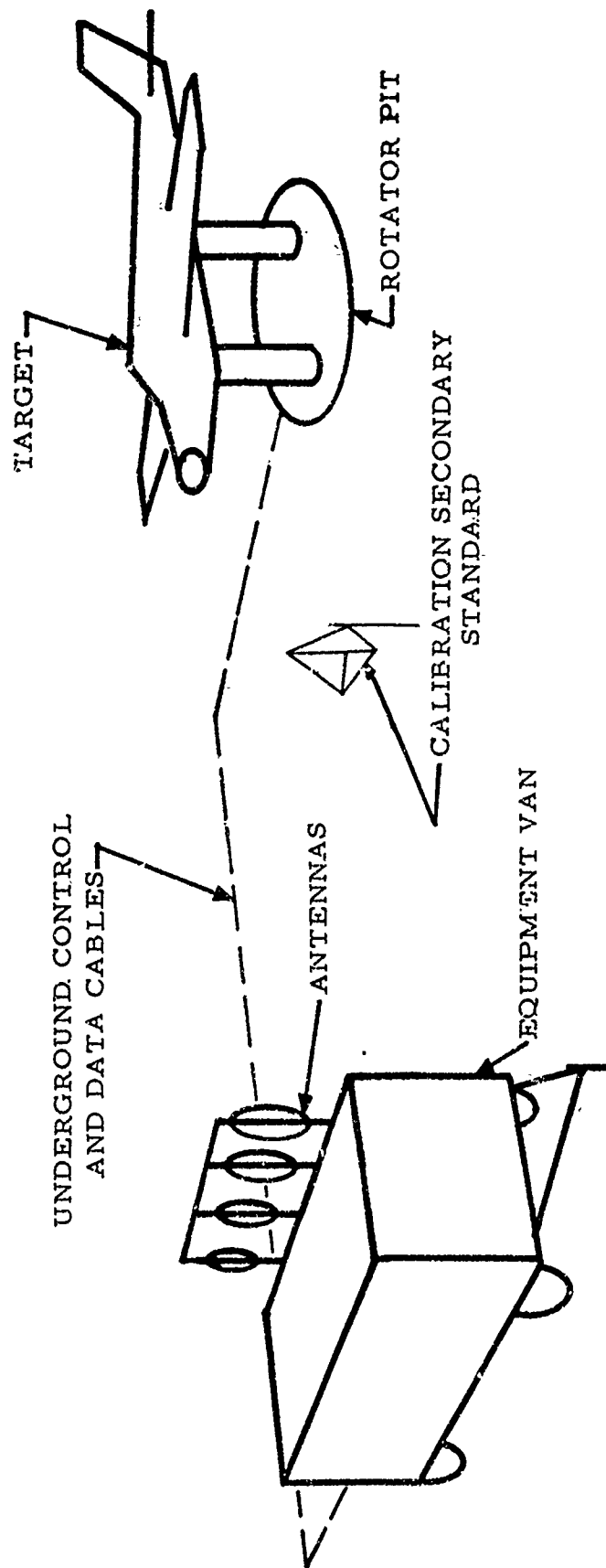


Figure 2 Elements of the Radar Measurement Range

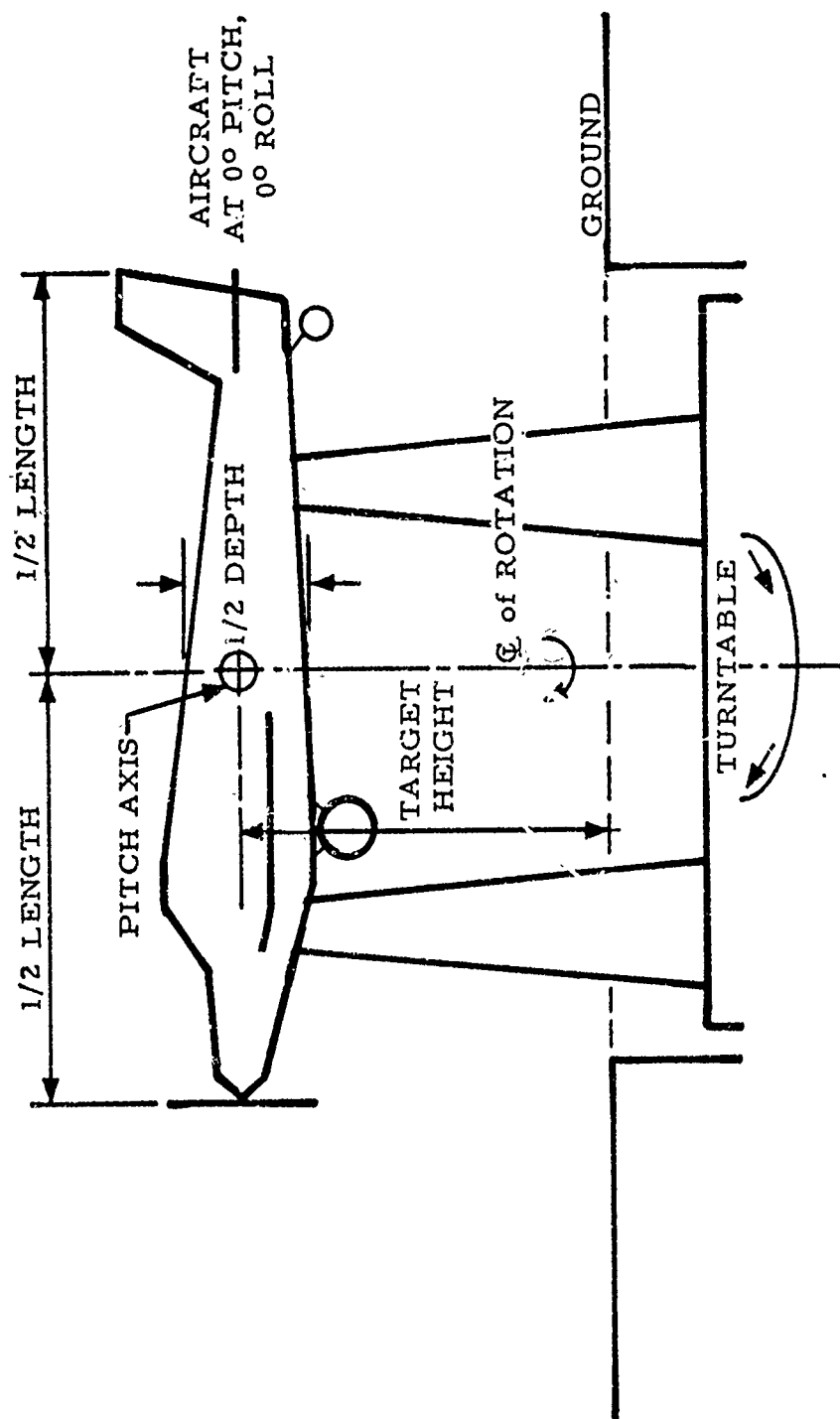


Figure 3 Aircraft Mounting Geometry Showing Location of Pitch Axis, Yaw Axis and Designation of Target Height

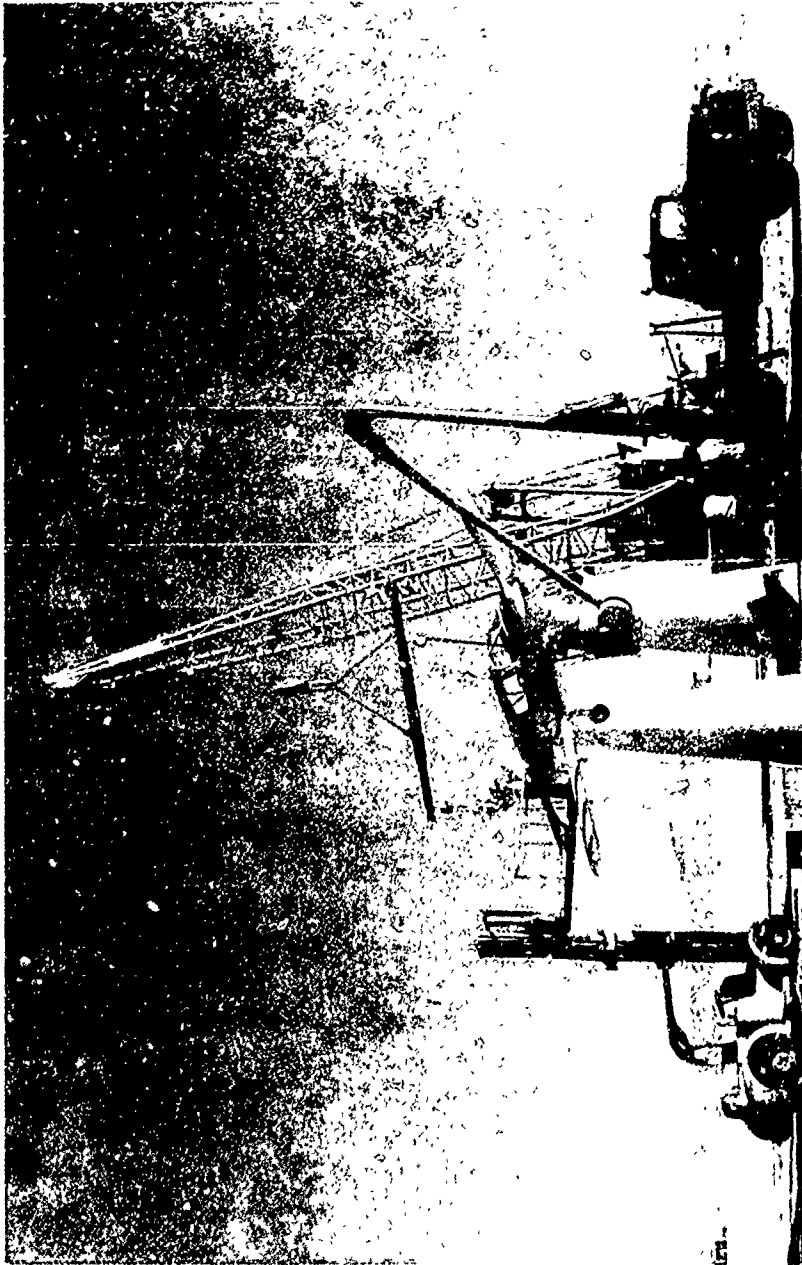


Figure 4 Cherokee 140 being Mounted for RCS Measurement
at 20° Roll, 0° Pitch

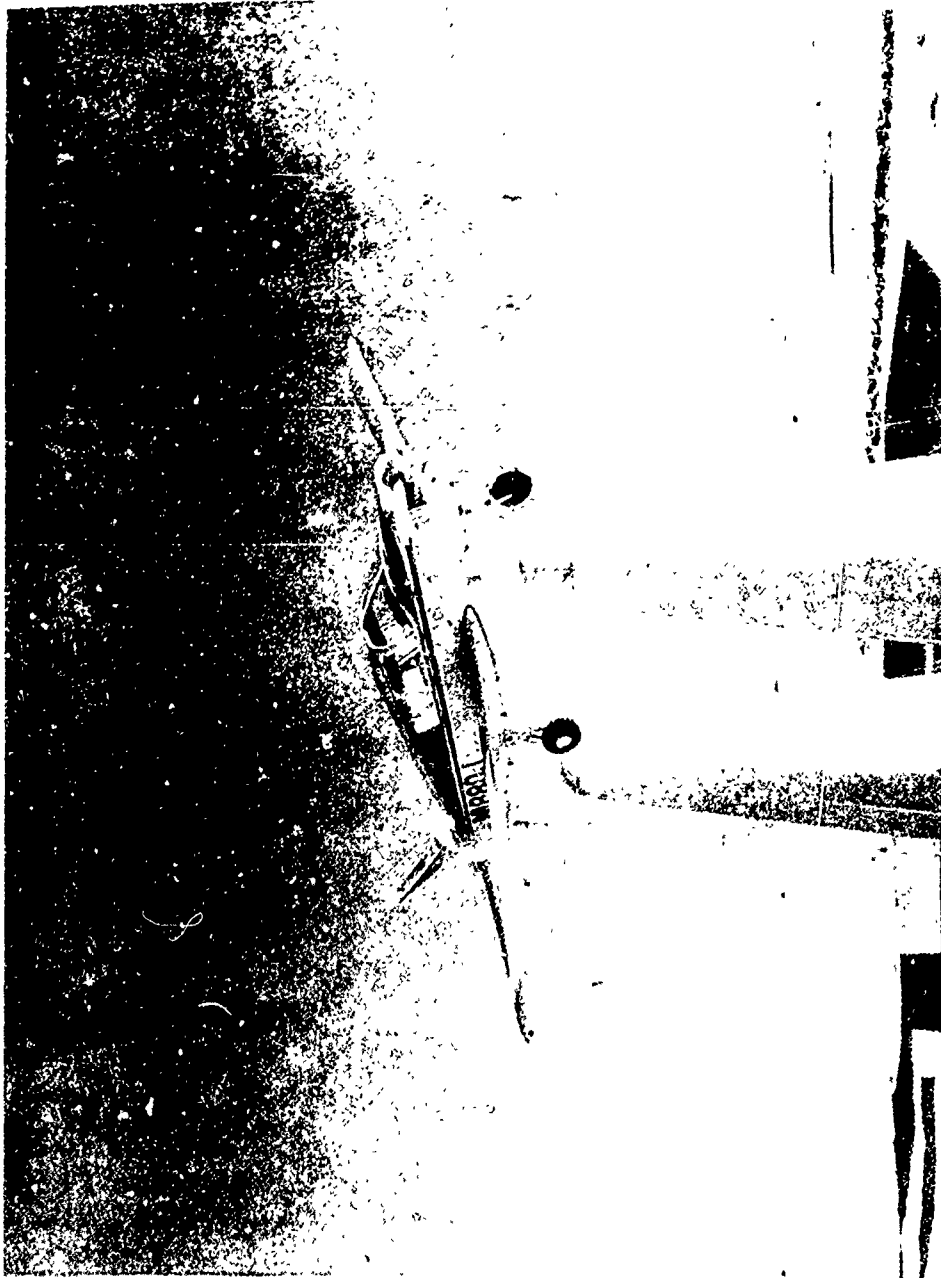


Figure 5 Cherokee 140 Ready for Test at 20° Roll, 0° Pitch



Figure 6 Cherokee 140 Mounted for RCS Measurement
at 45° Roll, -10° Pitch

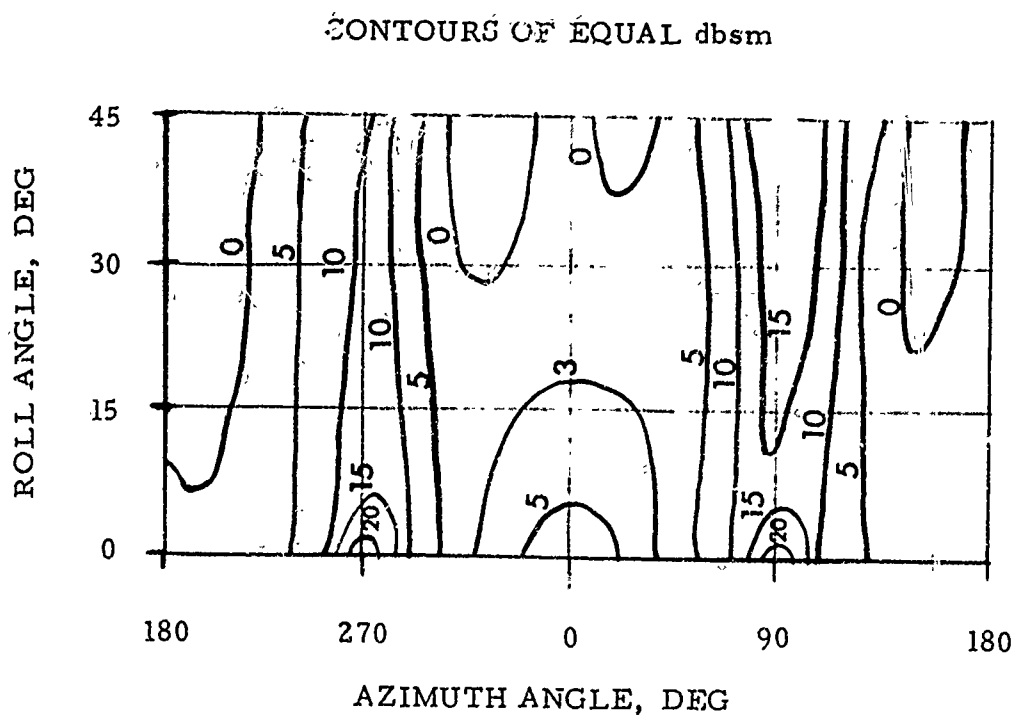


Figure 7 RCS vs Roll Angle at 0° Pitch Angle
for 2800MHz, VV Polarization

CONTOURS OF EQUAL dbsm

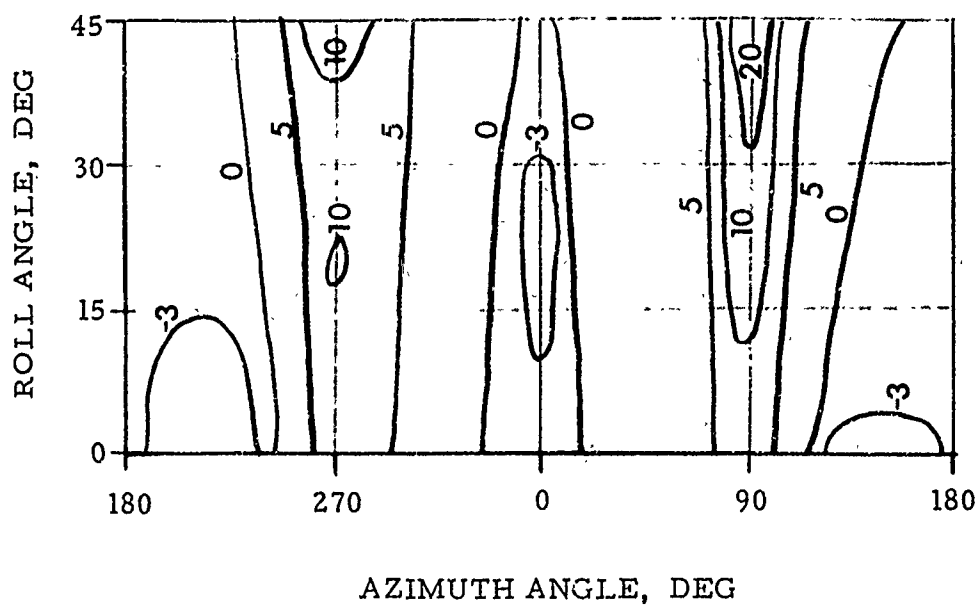


Figure 8 RCS vs. Roll Angle at 0° Pitch Angle for 2800 MHz, RR Polarization

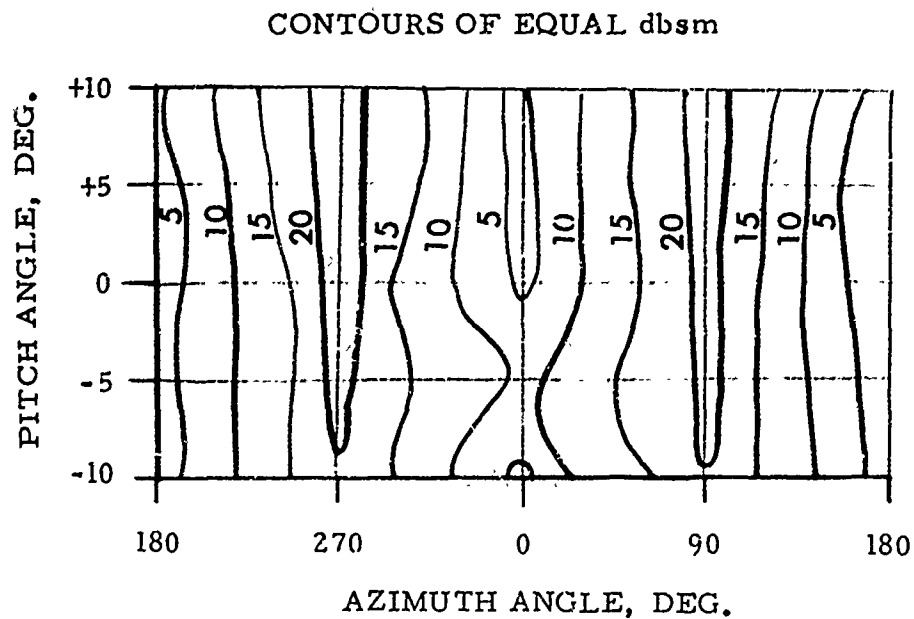


Figure 9 RCS vs. Pitch Angle at 0° Roll Angle for 2800 MHz,
VV Polarization

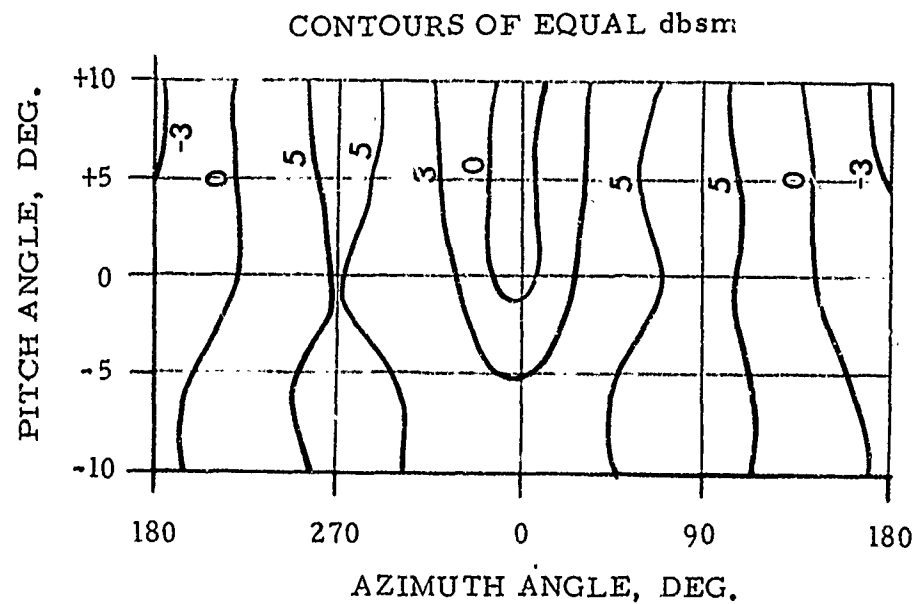


Figure 10 RCS vs. Pitch Angle at 0° Roll Angle for 2800 MHz,
RR Polarization

CONTROL NUMBER			73-01	Table III		DATA PLOT INDEX		Sheet 1
PAGE NO.	RUN	FREQ (MHz)	POLARIZATION	PITCH ANGLE	ROLL ANGLE	TARGET CONFIGURATION AND REMARKS		
25	573	2700	VV	-10	0	Cherokee 140 Airplane		
26	560	2703	VV	- 5	0	Cherokee 140 Airplane		
27	528	2700	VV	0	0	Cherokee 140 Airplane		
28	530	2700	VV	+ 5	0	Cherokee 140 Airplane		
29	543	2700	VV	+10	0	Cherokee 140 Airplane		
30	574	2700	VV	-10	5	Cherokee 140 Airplane		
31	595	2700	VV	- 5	5	Cherokee 140 Airplane		
32	608	2700	VV	0	5	Cherokee 140 Airplane		
33	609	2700	VV	+ 5	5	Cherokee 140 Airplane		
34	643	2700	VV	+10	5	Cherokee 140 Airplane		
35	706	2700	VV	-10	10	Cherokee 140 Airplane		
36	679	2700	VV	- 5	10	Cherokee 140 Airplane		
37	672	2700	VV	0	10	Cherokee 140 Airplane		
38	652	2700	VV	+ 5	10	Cherokee 140 Airplane		
39	645	2700	VV	+10	10	Cherokee 140 Airplane		
40	707	2700	VV	-10	20	Cherokee 140 Airplane		
41	735	2700	VV	- 5	20	Cherokee 140 Airplane		
42	756	2700	VV	0	20	Cherokee 140 Airplane		
43	769	2700	VV	+ 5	20	Cherokee 140 Airplane		
44	770	2700	VV	+10	20	Cherokee 140 Airplane		
45	784	2700	VV	-10	30	Cherokee 140 Airplane		
46	786	2700	VV	- 5	30	Cherokee 140 Airplane		
47	813	2700	VV	0	30	Cherokee 140 Airplane		
48	814	2700	VV	+ 5	30	Cherokee 140 Airplane		
49	827	2700	VV	+10	30	Cherokee 140 Airplane		
50	877	2700	VV	-10	45	Cherokee 140 Airplane		
51	864	2700	VV	- 5	45	Cherokee 140 Airplane		
52	843	2700	VV	0	45	Cherokee 140 Airplane		
53	836	2700	VV	+ 5	45	Cherokee 140 Airplane		
54	828	2700	VV	+10	45	Cherokee 140 Airplane		

CONTROL NUMBER		73-01		Table III		DATA PLCT INDEX		Sheet 2	
PAGE NO.	RUN	FREQ (MHz)	POLARIZATION	PITCH ANGLE	ROLL ANGLE	TARGET CONFIGURATION AND REMARKS			
55	572	2700	RR	-10	0	Cherokee 140 Airplane			
56	561	2700	RR	- 5	0	Cherokee 140 Airplane			
57	529	2700	RR	0	0	Cherokee 140 Airplane			
58	531	2700	RR	+ 5	0	Cherokee 140 Airplane			
59	542	2700	RR	+10	0	Cherokee 140 Airplane			
60	575	2700	RR	-10	5	Cherokee 140 Airplane			
61	596	2700	RR	- 5	5	Cherokee 140 Airplane			
62	607	2700	RR	0	5	Cherokee 140 Airplane			
63	610	2700	RR	+ 5	5	Cherokee 140 Airplane			
64	642	2700	RR	+10	5	Cherokee 140 Airplane			
65	705	2700	RR	-10	10	Cherokee 140 Airplane			
66	680	2700	RR	- 5	10	Cherokee 140 Airplane			
67	673	2700	RR	0	10	Cherokee 140 Airplane			
68	653	2700	RR	+ 5	10	Cherokee 140 Airplane			
69	646	2700	RR	+10	10	Cherokee 140 Airplane			
70	708	2700	RR	-10	20	Cherokee 140 Airplane			
71	736	2700	RR	- 5	20	Cherokee 140 Airplane			
72	757	2700	RR	0	20	Cherokee 140 Airplane			
73	768	2700	RR	+ 5	20	Cherokee 140 Airplane			
74	771	2700	RR	+10	20	Cherokee 140 Airplane			
75	785	2700	RR	-10	30	Cherokee 140 Airplane			
76	787	2700	RR	- 5	30	Cherokee 140 Airplane			
77	812	2700	RR	0	30	Cherokee 140 Airplane			
78	815	2700	RR	+ 5	30	Cherokee 140 Airplane			
79	826	2700	RR	+10	30	Cherokee 140 Airplane			
80	876	2700	RR	-10	45	Cherokee 140 Airplane			
81	865	2700	RR	- 5	45	Cherokee 140 Airplane			
82	844	2700	RR	0	45	Cherokee 140 Airplane			
83	837	2700	RR	+ 5	45	Cherokee 140 Airplane			
84	829	2700	RR	+10	45	Cherokee 140 Airplane			

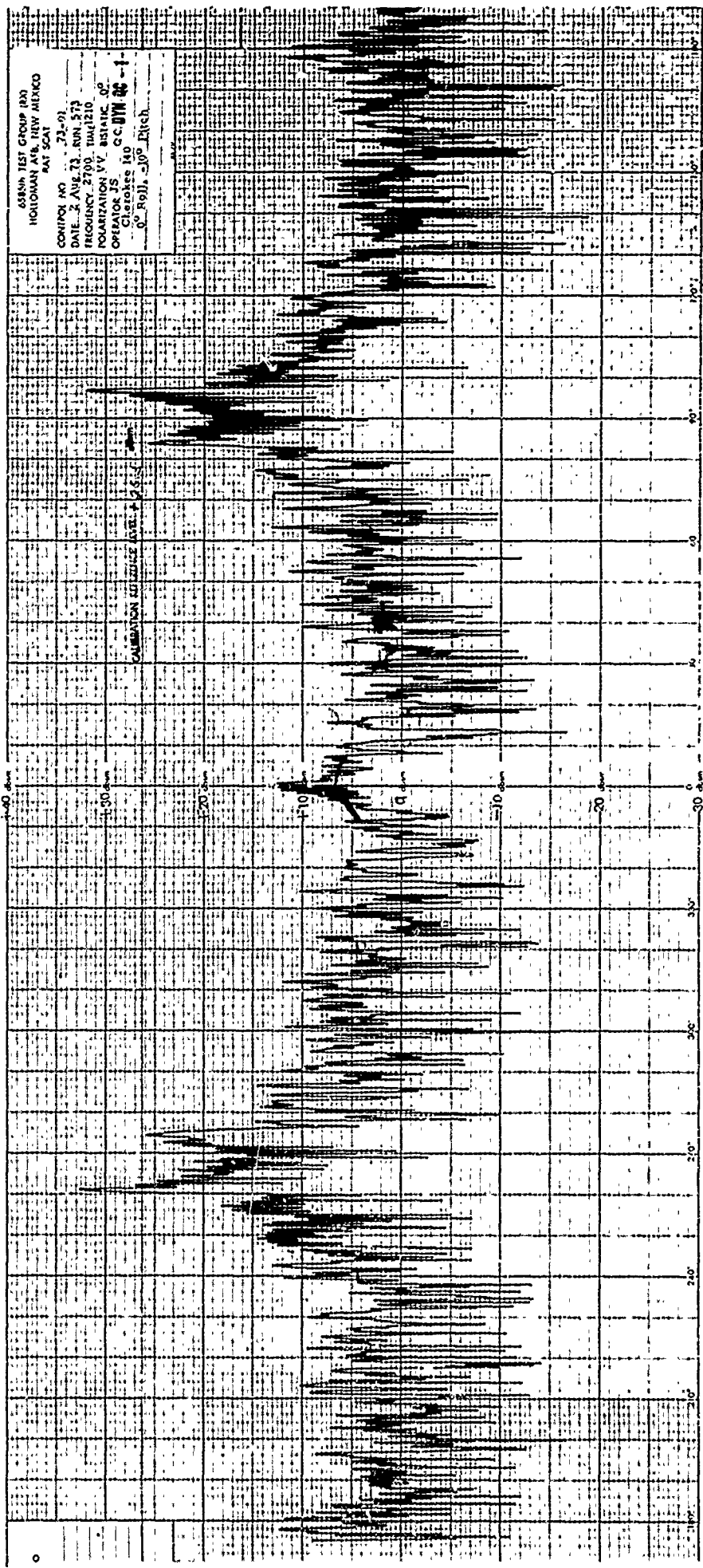
CONTROL NUMBER			73-01	Table III		DATA PLOT INDEX		Sheet 3
PAGE NO.	RUN	FREQ (MHz)	POLARIZATION	PITCH ANGLE	ROLL ANGLE	TARGET CONFIGURATION AND REMARKS		
85	570	2800	V	-10	0	Cherokee 140 Airplane		
86	563	2800	VV	- 5	0	Cherokee 140 Airplane		
87	525	2800	VV	0	0	Cherokee 140 Airplane		
88	533	2800	VV	+ 5	0	Cherokee 140 Airplane		
89	540	2800	VV	+10	0	Cherokee 140 Airplane		
90	577	2800	VV	-10	5	Cherokee 140 Airplane		
91	598	2800	VV	- 5	5	Cherokee 140 Airplane		
92	605	2800	VV	0	5	Cherokee 140 Airplane		
93	612	2800	VV	+ 5	5	Cherokee 140 Airplane		
94	640	2800	VV	+10	5	Cherokee 140 Airplane		
95	704	2800	VV	-10	10	Cherokee 140 Airplane		
96	683	2800	VV	- 5	10	Cherokee 140 Airplane		
97	675	2800	VV	0	10	Cherokee 140 Airplane		
98	670	2800	VV	+ 5	10	Cherokee 140 Airplane		
99	649	2800	VV	+10	10	Cherokee 140 Airplane		
100	724	2800	VV	-10	20	Cherokee 140 Airplane		
101	738	2800	VV	- 5	20	Cherokee 140 Airplane		
102	760	2800	VV	0	20	Cherokee 140 Airplane		
103	765	2800	VV	+ 5	20	Cherokee 140 Airplane		
104	774	2800	VV	+10	20	Cherokee 140 Airplane		
105	783	2800	VV	-10	30	Cherokee 140 Airplane		
106	790	2800	VV	- 5	30	Cherokee 140 Airplane		
107	811	2800	VV	0	30	Cherokee 140 Airplane		
108	816	2800	VV	+ 5	30	Cherokee 140 Airplane		
109	825	2800	VV	+10	30	Cherokee 140 Airplane		
110	875	2800	VV	-10	45	Cherokee 140 Airplane		
111	868	2800	VV	- 5	45	Cherokee 140 Airplane		
112	847	2800	VV	0	45	Cherokee 140 Airplane		
113	840	2800	VV	+ 5	45	Cherokee 140 Airplane		
114	830	2800	VV	+10	45	Cherokee 140 Airplane		

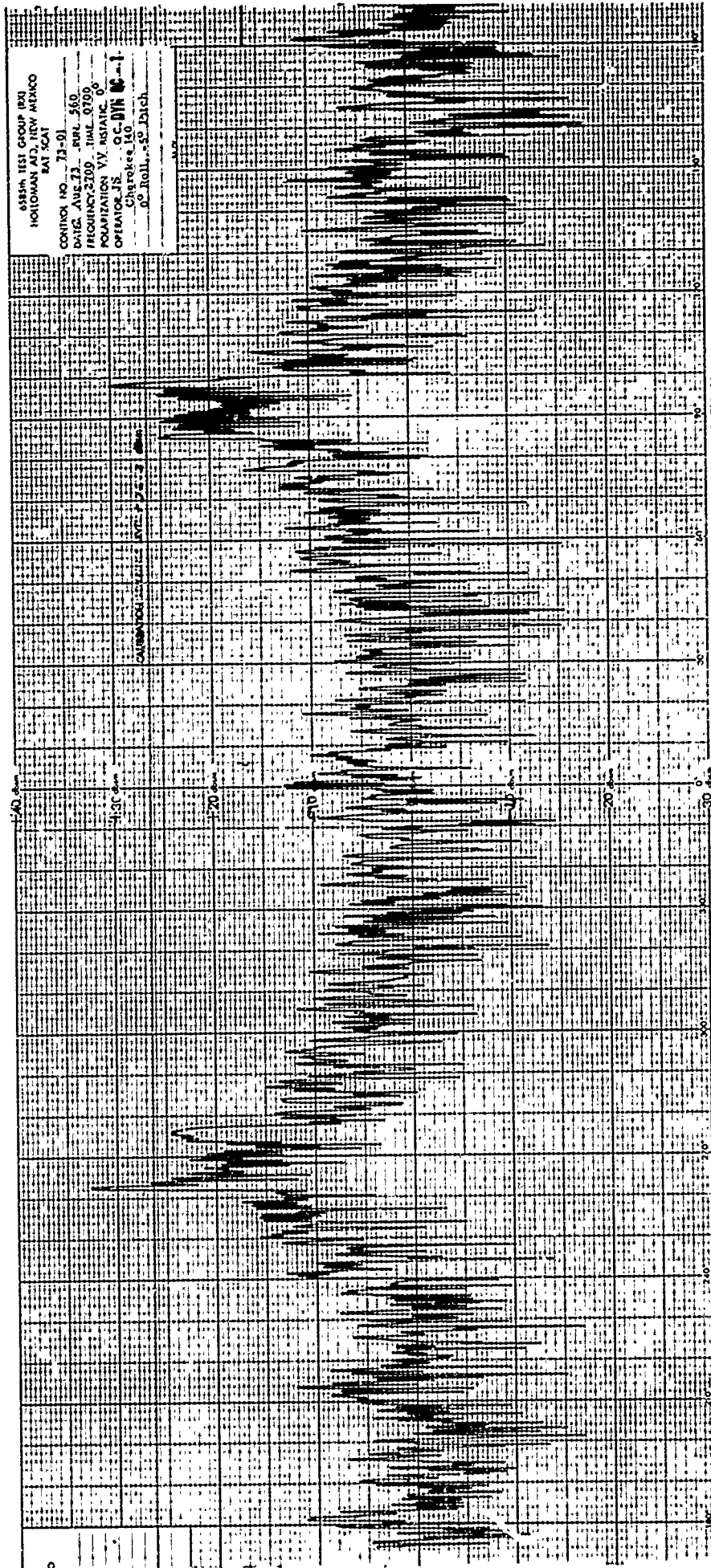
PAGE NO.	RUN	FREQ (MHz)	POLARIZATION	PITCH ANGLE	ROLL ANGLE	TARGET CONFIGURATION AND REMARKS
115	571	2800	HH	-10	0	Cherokee 140 Airplane
116	564	2800	HH	-5	0	Cherokee 140 Airplane
117	526	2800	HH	0	0	Cherokee 140 Airplane
118	534	2800	HH	+5	0	Cherokee 140 Airplane
119	541	2800	HH	+10	0	Cherokee 140 Airplane
120	578	2800	HH	-10	5	Cherokee 140 Airplane
121	599	2800	HH	-5	5	Cherokee 140 Airplane
122	604	2800	HH	0	5	Cherokee 140 Airplane
123	613	2800	HH	+5	5	Cherokee 140 Airplane
124	641	2800	HH	+10	5	Cherokee 140 Airplane
125	703	2800	HH	-10	10	Cherokee 140 Airplane
126	681	2800	HH	-5	10	Cherokee 140 Airplane
127	676	2800	HH	0	10	Cherokee 140 Airplane
128	671	2800	HH	+5	10	Cherokee 140 Airplane
129	648	2800	HH	+10	10	Cherokee 140 Airplane
130	709	2800	HH	-10	20	Cherokee 140 Airplane
131	739	2800	HH	-5	20	Cherokee 140 Airplane
132	759	2800	HH	0	20	Cherokee 140 Airplane
133	766	2800	HH	+5	20	Cherokee 140 Airplane
134	773	2800	HH	+10	20	Cherokee 140 Airplane
135	781	2800	HH	-10	30	Cherokee 140 Airplane
136	788	2800	HH	-5	30	Cherokee 140 Airplane
137	810	2800	HH	0	30	Cherokee 140 Airplane
138	817	2800	HH	+5	30	Cherokee 140 Airplane
139	824	2800	HH	+10	30	Cherokee 140 Airplane
140	874	2800	HH	-10	45	Cherokee 140 Airplane
141	867	2800	HH	-5	45	Cherokee 140 Airplane
142	845	2800	HH	0	45	Cherokee 140 Airplane
143	838	2800	HH	+5	45	Cherokee 140 Airplane
144	831	2800	HH	+10	45	Cherokee 140 Airplane
145	527	2800	VH	0	0	Cherokee 140 Airplane, Cross polarized pattern

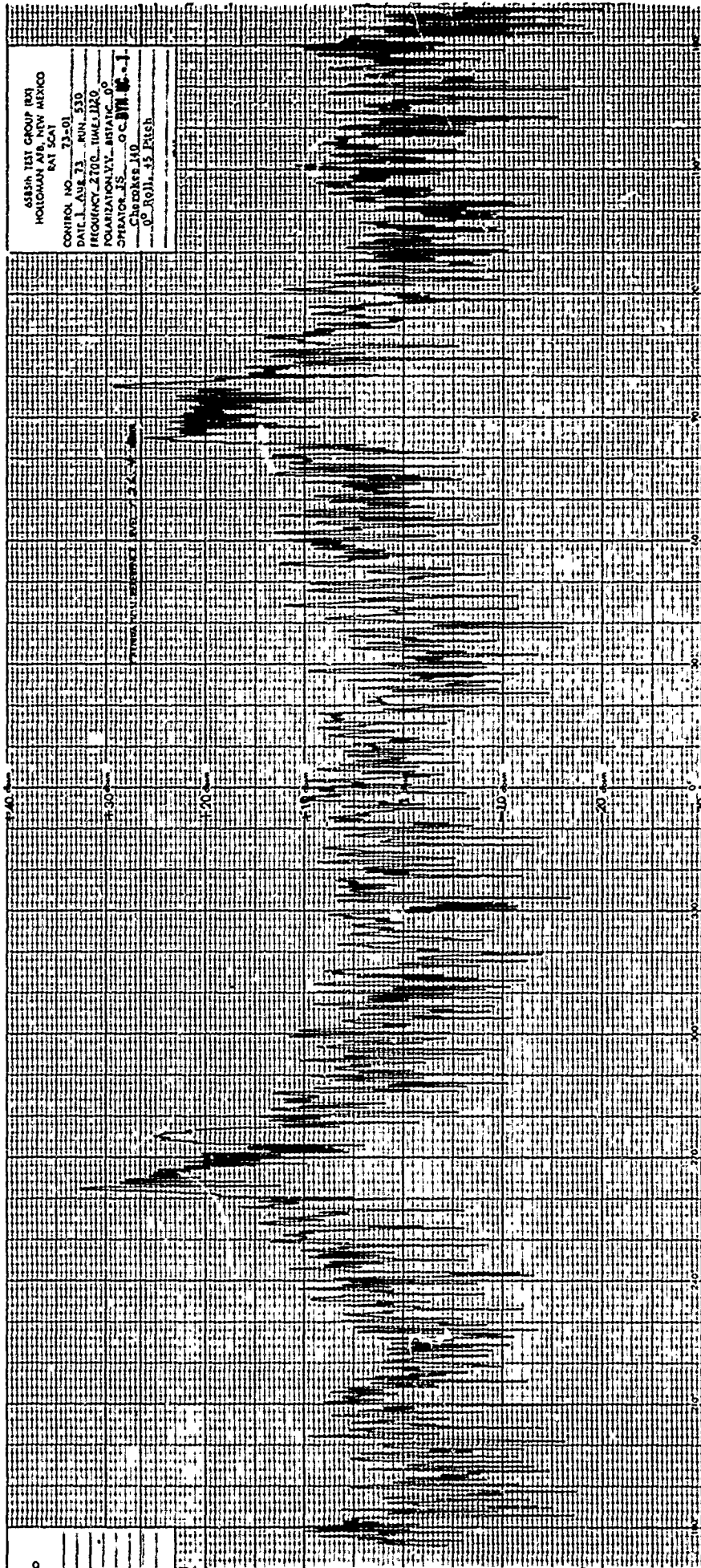
CONTROL NUMBER 73-01		ATA PLOT INDEX		Sheet 5		
Table 111		Table 111		Table 111		
PAGE NO.	RUN	FREQ (MHz)	POLARIZATION	PITCH ANGLE	ROLL ANGLE	TARGET CONFIGURATION AND REMARKS
146	569	2800	RR	-10	0	Cherokee 140 Airplane
147	562	2800	RR	-5	0	Cherokee 140 Airplane
148	512	2800	RR	0	0	Cherokee 140 Airplane
149	532	2800	RR	+5	0	Cherokee 140 Airplane
150	544	2800	RR	+10	0	Cherokee 140 Airplane
151	576	2800	RR	-10	5	Cherokee 140 Airplane
152	597	2800	RR	-5	5	Cherokee 140 Airplane
153	606	2800	RR	0	5	Cherokee 140 Airplane
154	611	2800	RR	+5	5	Cherokee 140 Airplane
155	618	2800	RR	+10	5	Cherokee 140 Airplane
156	702	2800	RR	-10	10	Cherokee 140 Airplane
157	682	2800	RR	-5	10	Cherokee 140 Airplane
158	674	2800	RR	0	10	Cherokee 140 Airplane
159	669	2800	RR	+5	10	Cherokee 140 Airplane
160	647	2800	RR	+10	10	Cherokee 140 Airplane
161	725	2800	RR	-10	20	Cherokee 140 Airplane
162	737	2800	RR	-5	20	Cherokee 140 Airplane
163	758	2800	RR	0	20	Cherokee 140 Airplane
164	767	2800	RR	+5	20	Cherokee 140 Airplane
165	772	2800	RR	+10	20	Cherokee 140 Airplane
166	782	2800	RR	-10	30	Cherokee 140 Airplane
167	789	2800	RR	-5	30	Cherokee 140 Airplane
168	809	2800	RR	0	30	Cherokee 140 Airplane
169	818	2800	RR	+5	30	Cherokee 140 Airplane
170	823	2800	RR	+10	30	Cherokee 140 Airplane
171	873	2800	RR	-10	45	Cherokee 140 Airplane
172	866	2800	RR	-5	45	Cherokee 140 Airplane
173	846	2800	RR	0	45	Cherokee 140 Airplane
174	839	2800	RR	+5	45	Cherokee 140 Airplane
175	825	2800	RR	+10	45	Cherokee 140 Airplane
176	42	2900	RR	N/A	N/A	Cherokee 140 Airplane Background, Columns 90/91 with 0° Roll, 0° Pitch Transitions

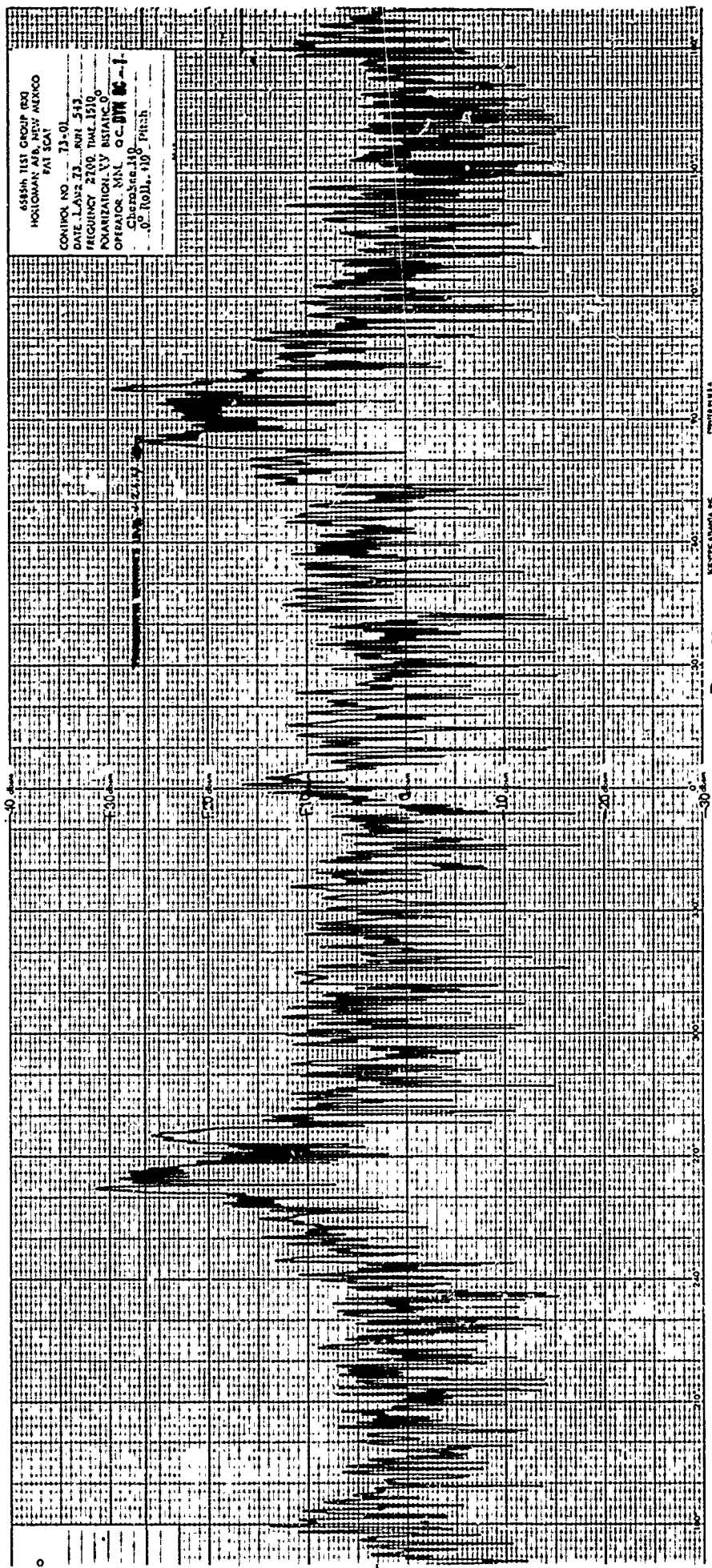
CONTROL NUMBER 73-01		Table III		DATA PLOT INDEX		Sheet 6	
PAGE NO.	RUN	FREQ (MHz)	POLARIZATION	PITCH ANGLE	ROLL ANGLE	TARGET CONFIGURATION AND REMARKS	
177	567	2900	VV	-10	0	Cherokee 140 Airplane	
178	566	2900	VV	-5	0	Cherokee 140 Airplane	
179	511	2900	VV	0	0	Cherokee 140 Airplane	
180	536	2900	VV	+5	0	Cherokee 140 Airplane	
181	537	2900	VV	+10	0	Cherokee 140 Airplane	
182	580	2900	VV	-10	5	Cherokee 140 Airplane	
183	601	2900	VV	-5	5	Cherokee 140 Airplane	
184	602	2900	VV	0	5	Cherokee 140 Airplane	
185	615	2900	VV	+5	5	Cherokee 140 Airplane	
186	616	2900	VV	+10	5	Cherokee 140 Airplane	
187	700	2900	VV	-10	10	Cherokee 140 Airplane	
188	684	2900	VV	-5	10	Cherokee 140 Airplane	
189	678	2900	VV	0	10	Cherokee 140 Airplane	
190	667	2900	VV	+5	10	Cherokee 140 Airplane	
191	651	2900	VV	+10	10	Cherokee 140 Airplane	
192	777	2900	VV	-10	20	Cherokee 140 Airplane	
193	741	2900	VV	-5	20	Cherokee 140 Airplane	
194	762	2900	VV	0	20	Cherokee 140 Airplane	
195	763	2900	VV	+5	20	Cherokee 140 Airplane	
196	776	2900	VV	+10	20	Cherokee 140 Airplane	
197	779	2900	VV	-10	30	Cherokee 140 Airplane	
198	791	2900	VV	-5	30	Cherokee 140 Airplane	
199	807	2900	VV	0	30	Cherokee 140 Airplane	
200	820	2900	VV	+5	30	Cherokee 140 Airplane	
201	821	2900	VV	+10	30	Cherokee 140 Airplane	
202	871	2900	VV	-10	45	Cherokee 140 Airplane	
203	870	2900	VV	-5	45	Cherokee 140 Airplane	
204	848	2900	VV	0	45	Cherokee 140 Airplane	
205	841	2900	VV	+5	45	Cherokee 140 Airplane	
206	834	2900	VV	+10	45	Cherokee 140 Airplane	

CONTROL NUMBER		73-01		Table III		DATA PLOT INDEX		Sheet 7	
PAGE NO.	RUN	FREQ (MHz)	POLARIZATION	PITCH ANGLE	ROLL ANGLE	TARGET CONFIGURATION AND REMARKS			
207	568	2900	RR	-10	0	Cherokee 140 Airplane			
208	565	2900	RR	-5	0	Cherokee 140 Airplane			
209	510	2900	RR	0	0	Cherokee 140 Airplane			
210	535	2900	RR	+5	0	Cherokee 140 Airplane			
211	538	2900	RR	+10	0	Cherokee 140 Airplane			
212	579	2900	RR	-10	5	Cherokee 140 Airplane			
213	600	2900	RR	-5	5	Cherokee 140 Airplane			
214	603	2900	RR	0	5	Cherokee 140 Airplane			
215	614	2900	RR	+5	5	Cherokee 140 Airplane			
216	617	2900	RR	+10	5	Cherokee 140 Airplane			
217	701	2900	RR	-10	10	Cherokee 140 Airplane			
218	685	2900	RR	-5	10	Cherokee 140 Airplane			
219	677	2900	RR	0	10	Cherokee 140 Airplane			
220	668	2900	RR	+5	10	Cherokee 140 Airplane			
221	650	2900	RR	+10	10	Cherokee 140 Airplane			
222	778	2900	RR	-10	20	Cherokee 140 Airplane			
223	740	2900	RR	-5	20	Cherokee 140 Airplane			
224	761	2900	RR	0	20	Cherokee 140 Airplane			
225	764	2900	RR	+5	20	Cherokee 140 Airplane			
226	775	2900	RR	+10	20	Cherokee 140 Airplane			
227	780	2900	RR	-10	30	Cherokee 140 Airplane			
228	792	2900	RR	-5	30	Cherokee 140 Airplane			
229	808	2900	RR	0	30	Cherokee 140 Airplane			
230	819	2900	RR	+5	30	Cherokee 140 Airplane			
231	822	2900	RR	+10	30	Cherokee 140 Airplane			
232	872	2900	RR	-10	45	Cherokee 140 Airplane			
233	869	2900	RR	-5	45	Cherokee 140 Airplane			
234	849	2900	RR	0	45	Cherokee 140 Airplane			
235	842	2900	RR	+5	45	Cherokee 140 Airplane			
236	833	2900	RR	+10	45	Cherokee 140 Airplane			



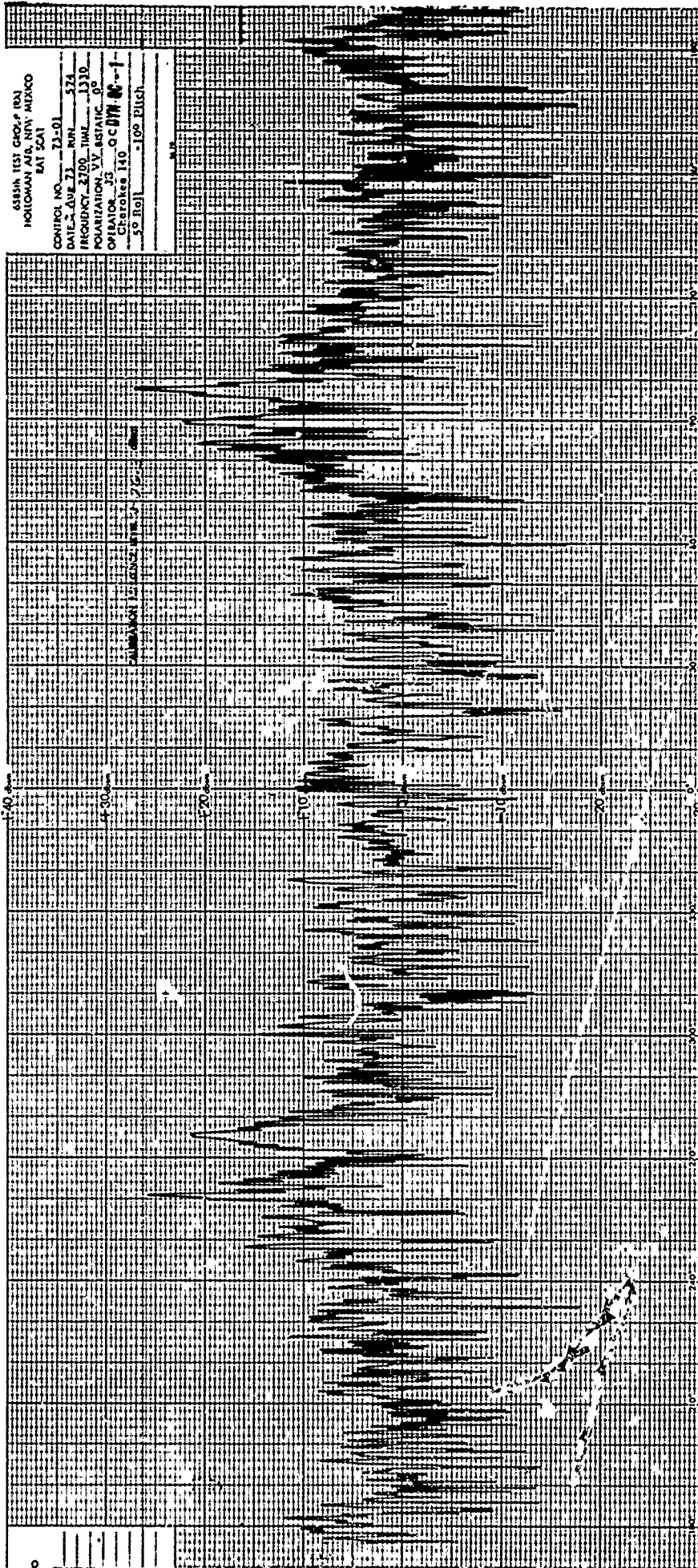


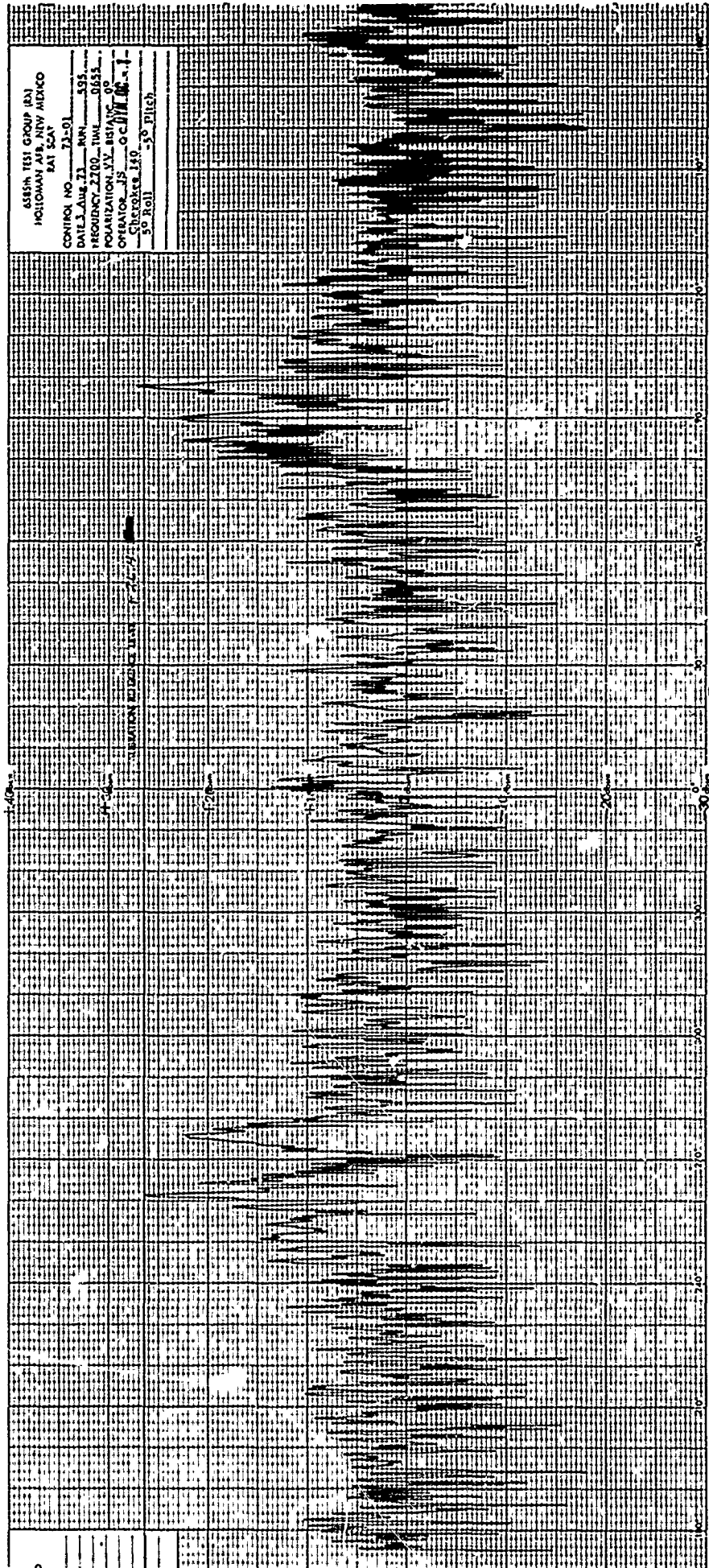




SEISMIC TEST GROUP (R)
HOLCOMB, N.Y., N.Y. MEXICO
LAT 30N

CONTROL NO. 73-01
DATE 2/2/73 RUN 574
FREQUENCY 2700 HZ TIME 1310
POLARIZATION VV, AZIMUTH 0°
OPERATOR JS OGDIN 86-1-1
Cherokee 140
50 Roll - 100 Blotch

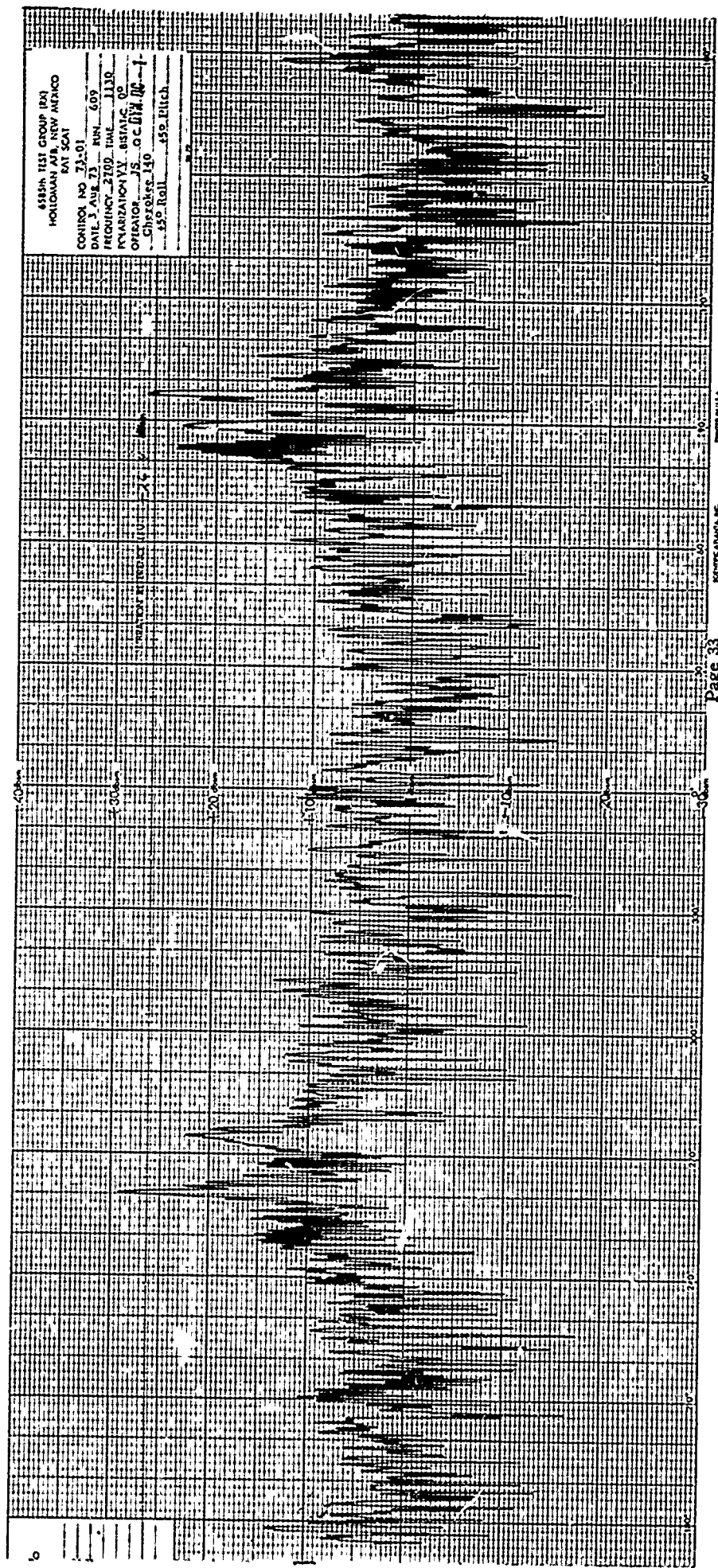




4884H TEL GROUP 100
HOTONAH AT NEW MEXICO
PAT 501

CONTROL NO. 72-01
DATE 1 Aug 73 RUN 608
FREQUENCY 2700 TWT 1010
POLARIZATION XY STATIC 00
OPERATOR JS GCDYN 00-1
Cherokee 110
50 Roll 00 Flush

100m
200m
300m
400m
500m
600m
700m
800m
900m
1000m
1100m
1200m
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4300m
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9000m
9100m
9200m
9300m
9400m
9500m
9600m
9700m
9800m
9900m
10000m



450 Hz TEST GROUP 1001
HOLLOMAN AFB, NEW MEXICO
FAT SCAT
CONTROL NO 73-01
DATE 3 Aug 73 RUN 609
FREQUENCY 2700 Hz TIME 1130
VARIATION VV STATIC 00
OPERATOR JS oc 0174 00-1
Cherokee 140
450 Roll 450 Pitch

ASSSH TEST GROUP (R1)
HOLCOMAN AFB, NEW ALBANY
BAT SCAT

CONTROL NO. 33-01

DATE 7 AUG 73

TIME 0730

FREQUENCY 2700 MHz

POLARIZATION VV

OPERATOR NUT

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10-000000000

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800

900

1000

1100

1200

0

10

20

30

40

50

60

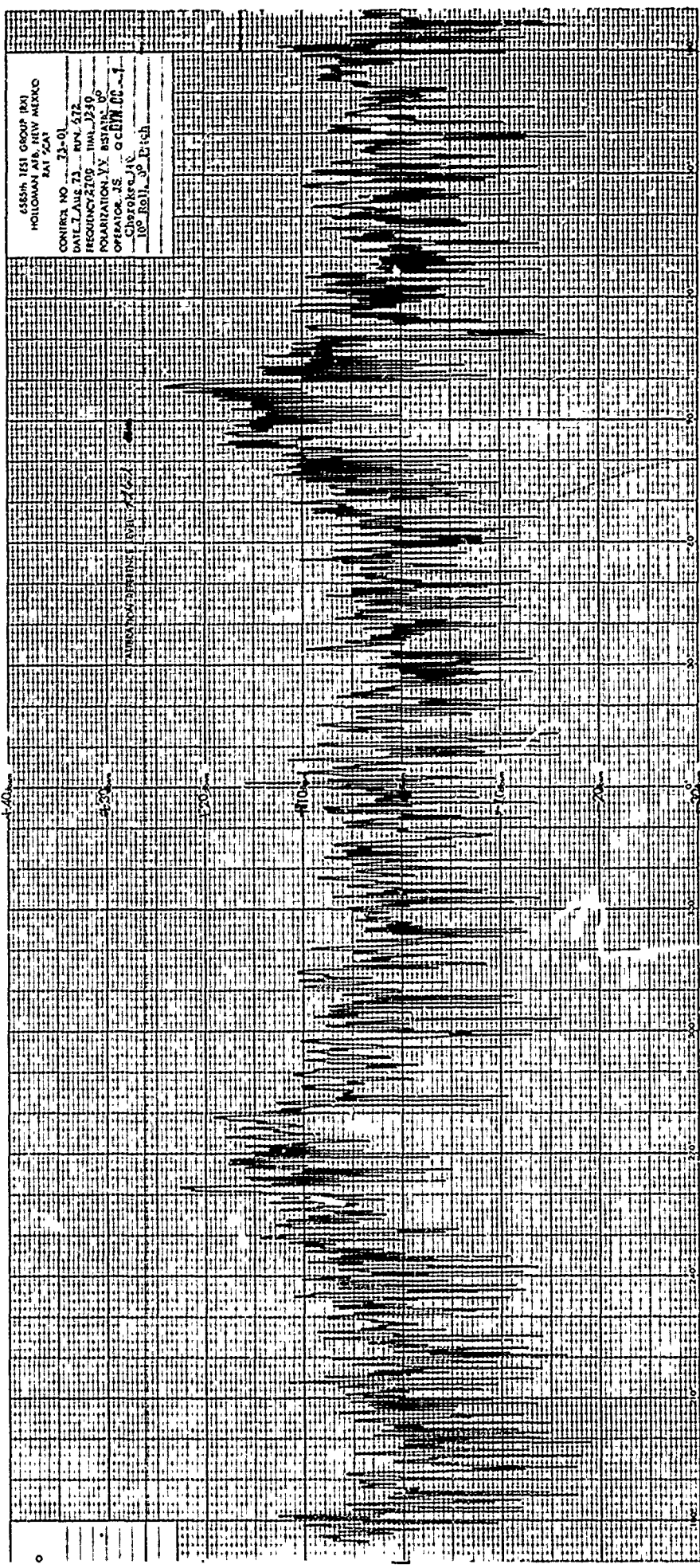
70

80

90

100

110



6385th TEST GROUP (R2)
HOLLISMAN AFB, NEW MEXICO

RAT SCAT

CONTROL NO. 73-01

DATE: AUG 73

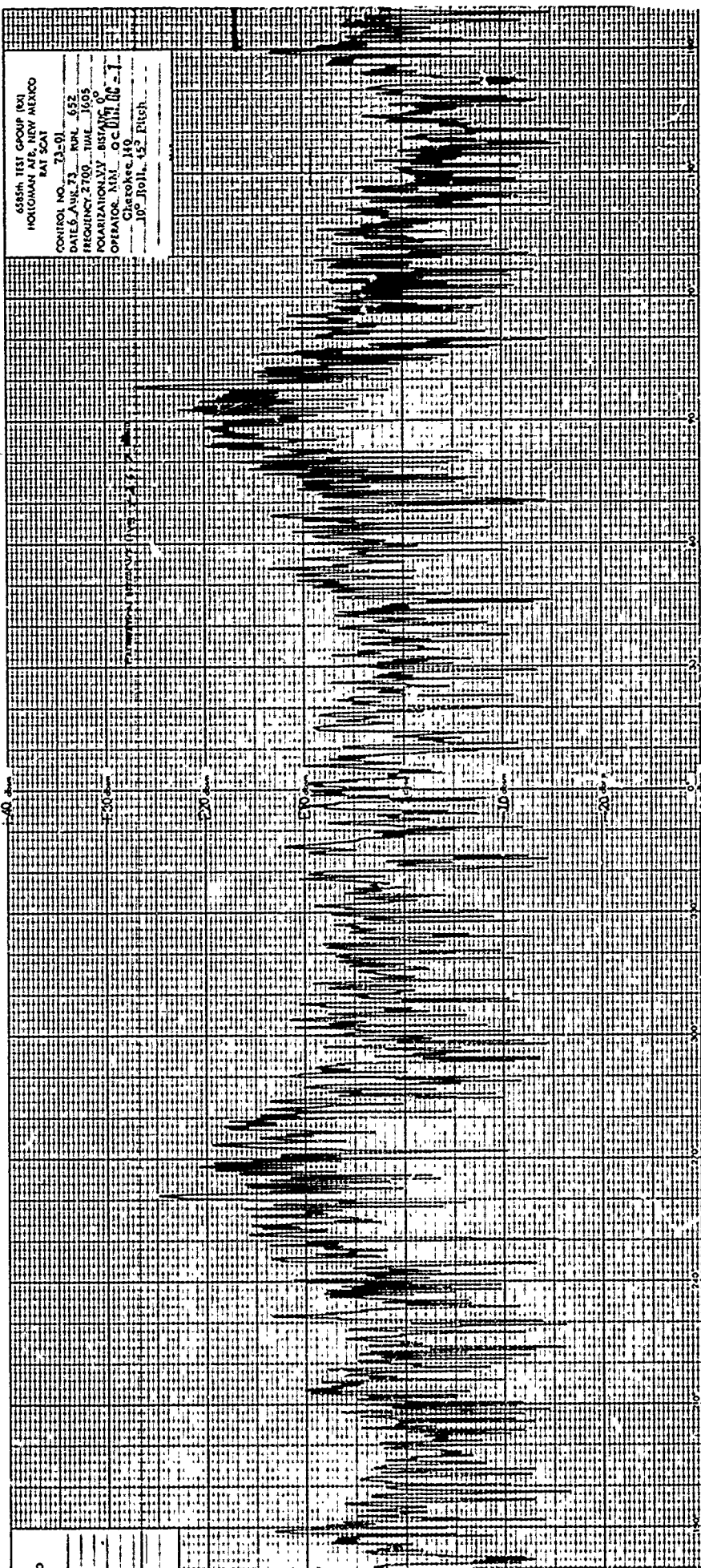
FREQUENCY: 2700

POLARIZATION: 85%

OPERATOR: NIN

Glucose 140

10° Roll, 45° Pitch



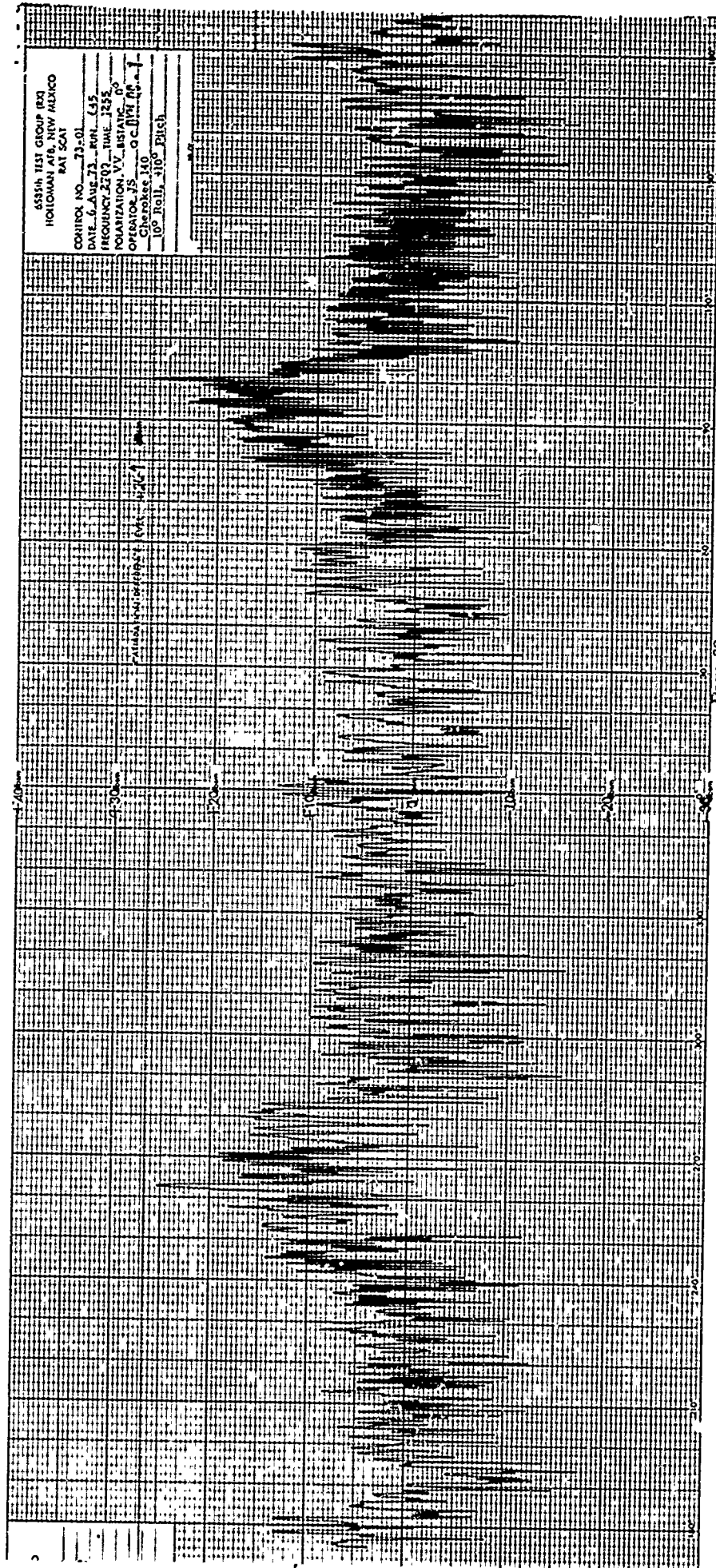
RECEIVED 11/1/73

RECEIVED 11/1/73

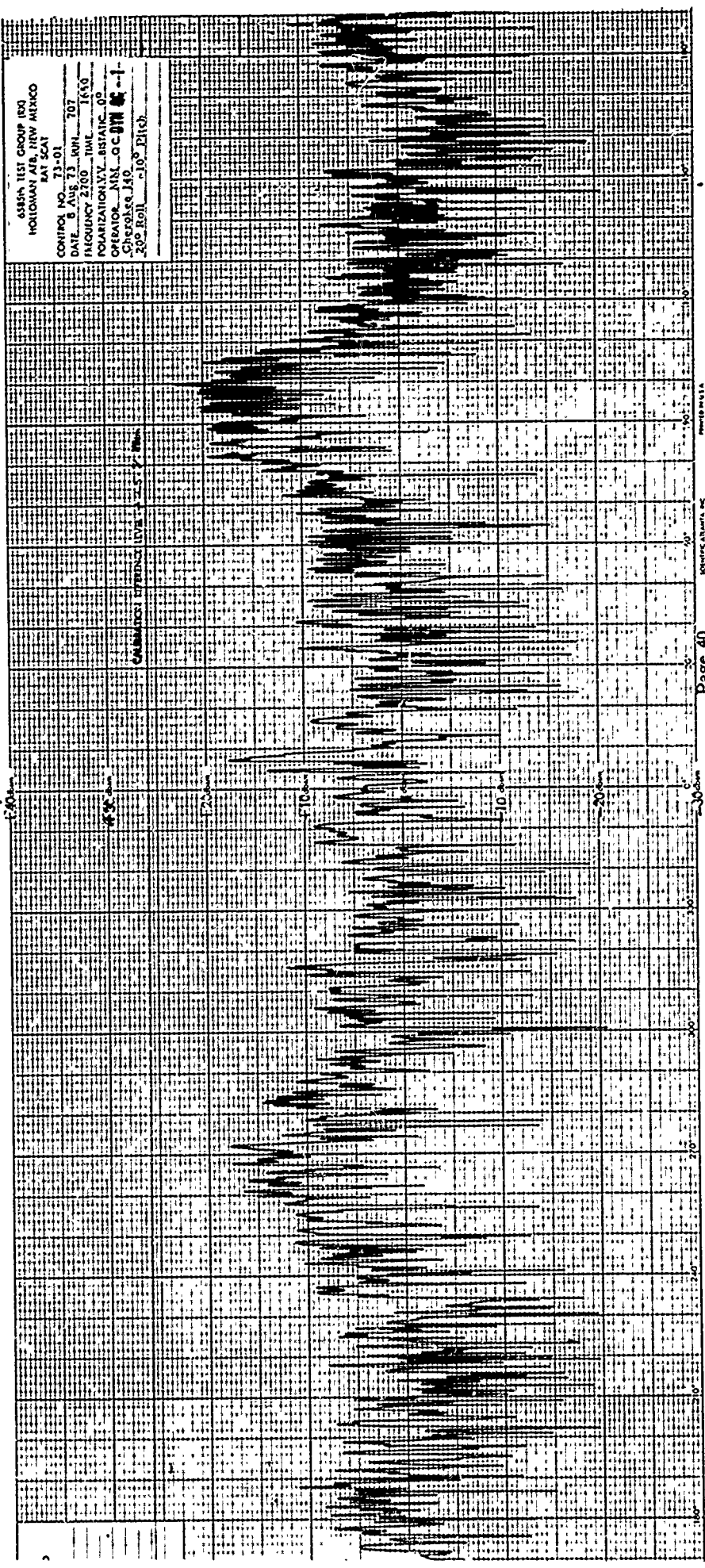
Page 38

655th TEST GROUP BQ
HOLLOMAN AFB, NEW MEXICO
RAT SCAT

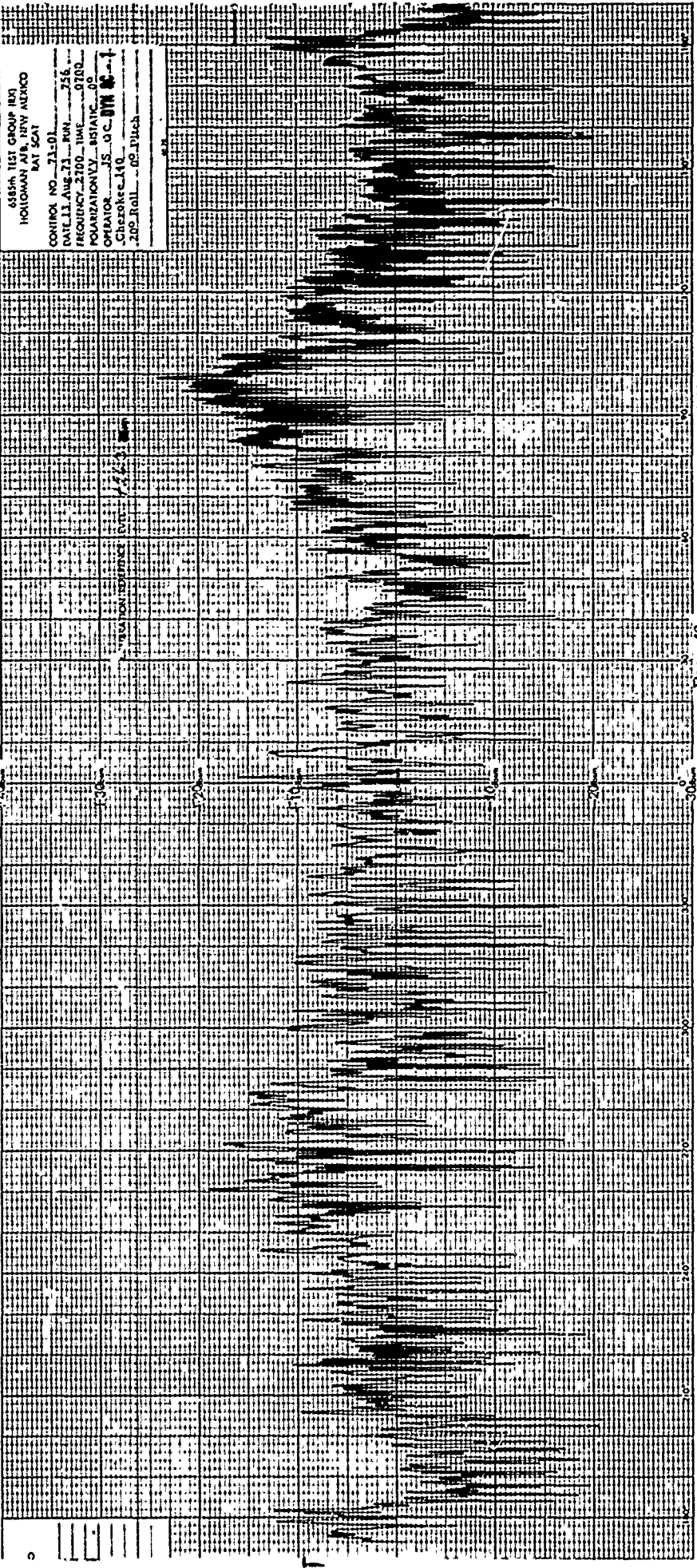
CONTROL NO. 73-01
DATE 6 AUG 73 - RUN 135
FREQUENCY 2707 - TIME 1255
POLARIZATION VV - AZIMUTH 0°
OPERATOR JS - OC 0111 00-1
Cherokee 140
0° Roll, 410° Pitch



6585A TEST GROUP 1000
HOLLAND AFB, NEW MEXICO
EAT SCAT
CONTROL NO. 73-01
DATE 8 AUG 73 KPH 707
FREQUENCY 2700 MHz 1450
POLARIZATION V. HORIZONTAL 0°
OPERATOR MIN. OC 0111 00-1
Chevrolet 140
200 Roll - 10° Pitch



[illegible]



48554 151 GROUP RD
HOLDMAN AFB, NEW MEXICO
BAT SCAT

CONTROL NO. 21-01

DATE 13 AUG 73 RUN 169

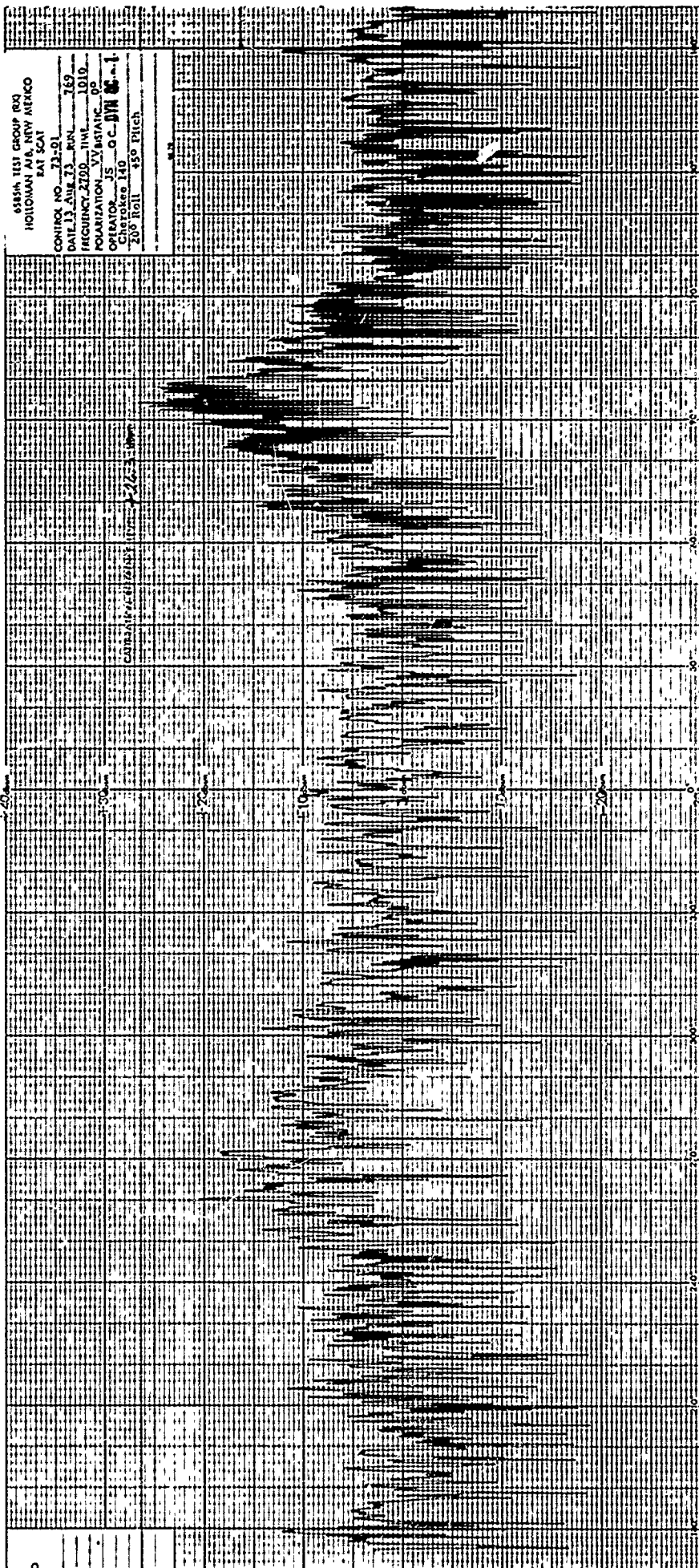
FREQUENCY 2100 TIME 0010

POLARIZATION VV STATIC 0°

OPERATOR JS OC JIN 8-1

Cherokee 140

200 Roll 150 Pitch

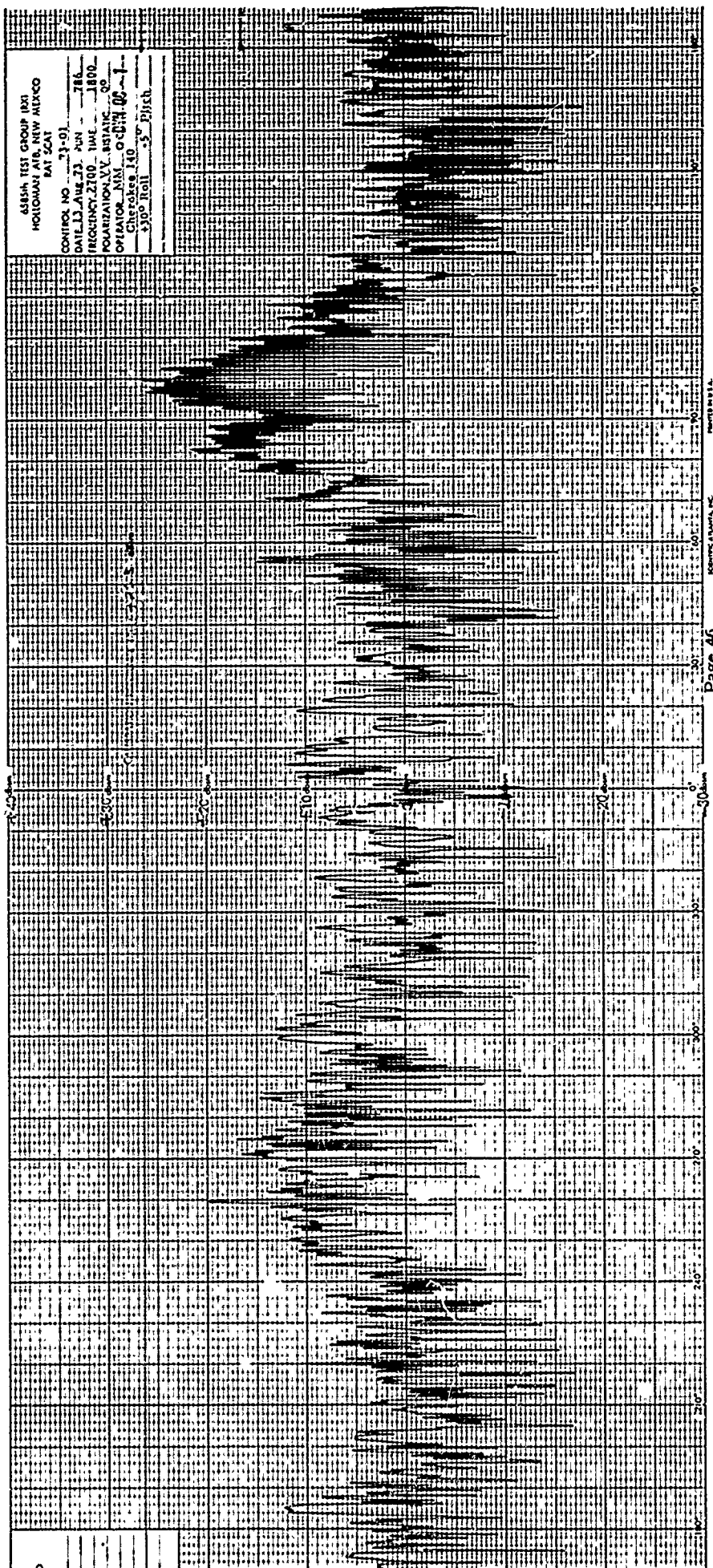


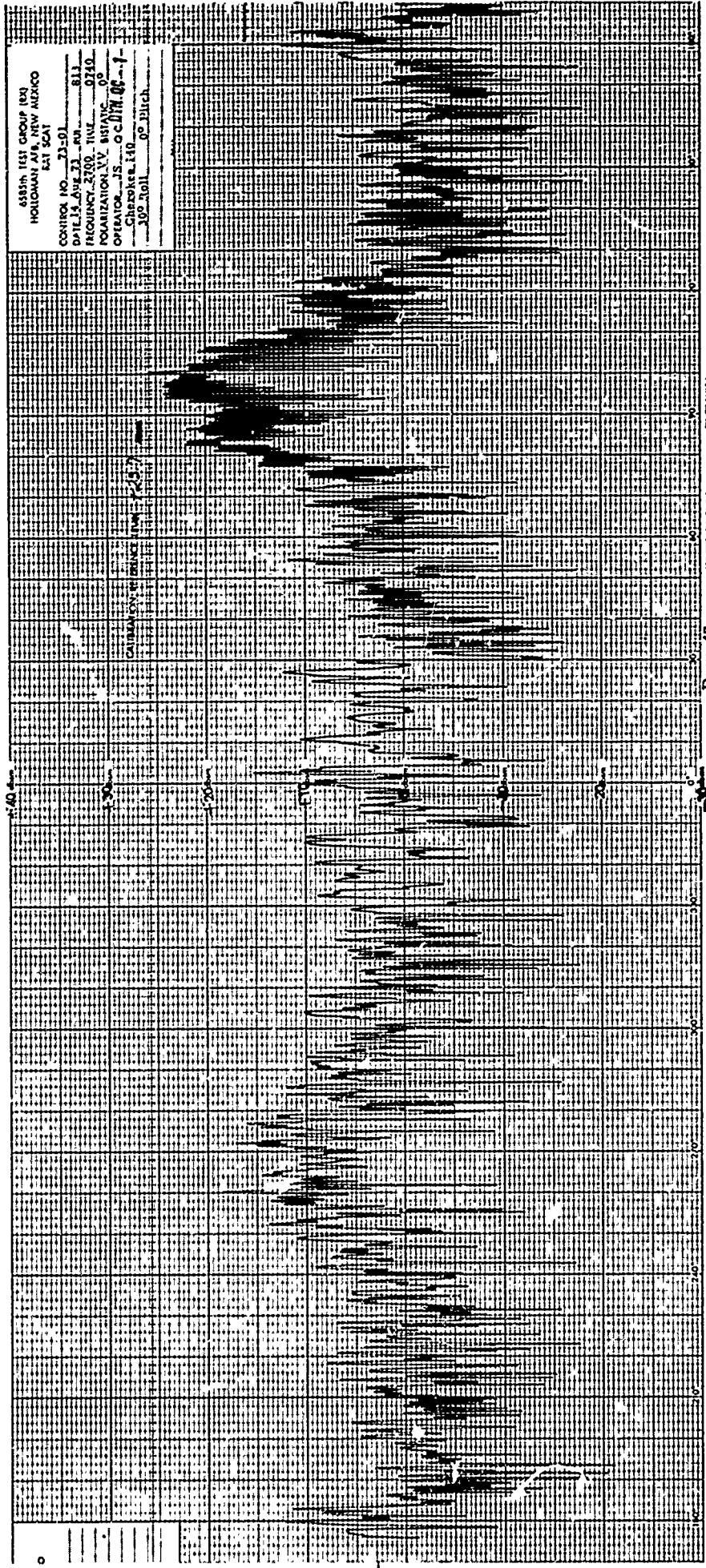
CONTROL NO 73-01
DATE 13 AUG 73 RUN 784
FREQUENCY 2700 TIME 1620
POLARIZATION VV STATIC 00
OPERATOR MM acb/HC-1
Chezokee 130
130° Roll -10° Pitch

2014.01.14. 10:11

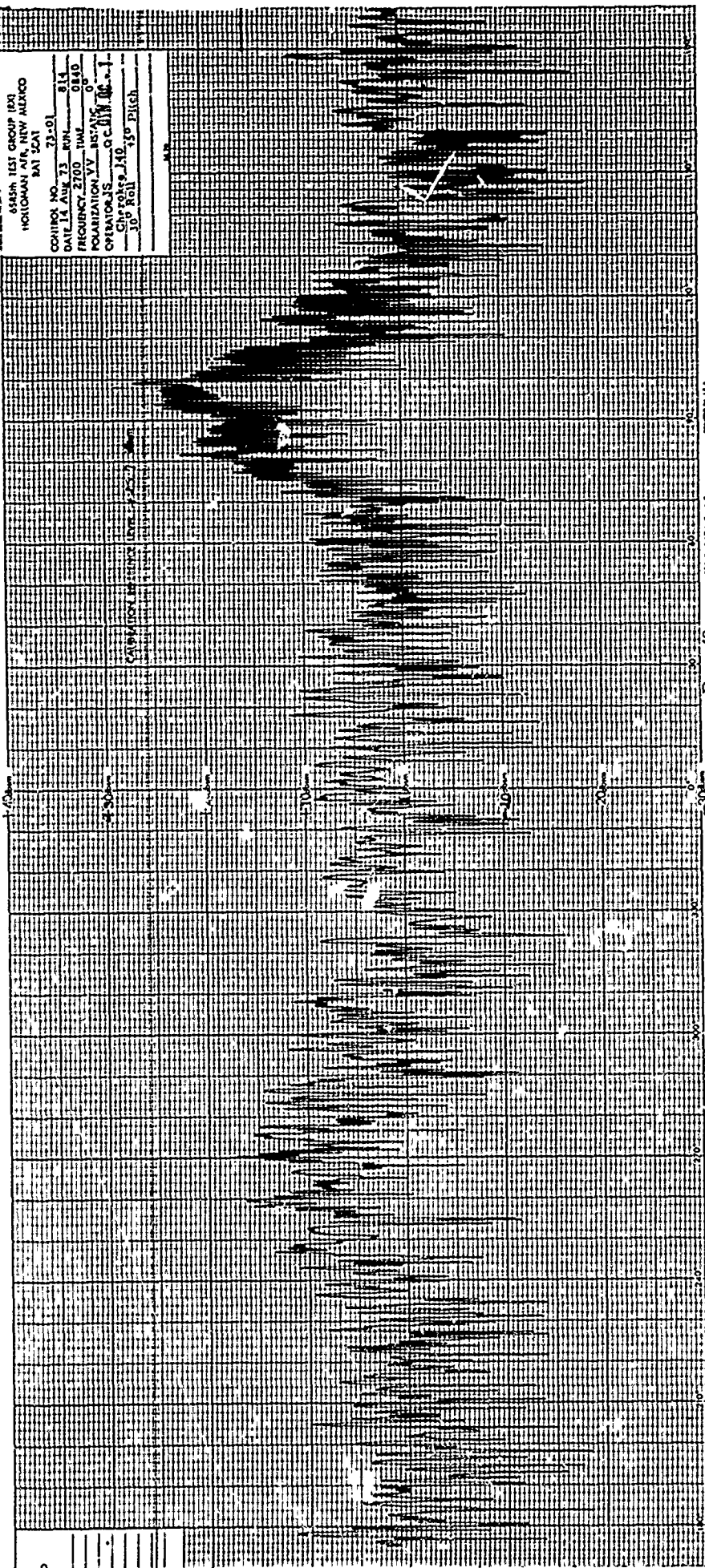
Page 45 WMA DC

CHART NO. 71 70



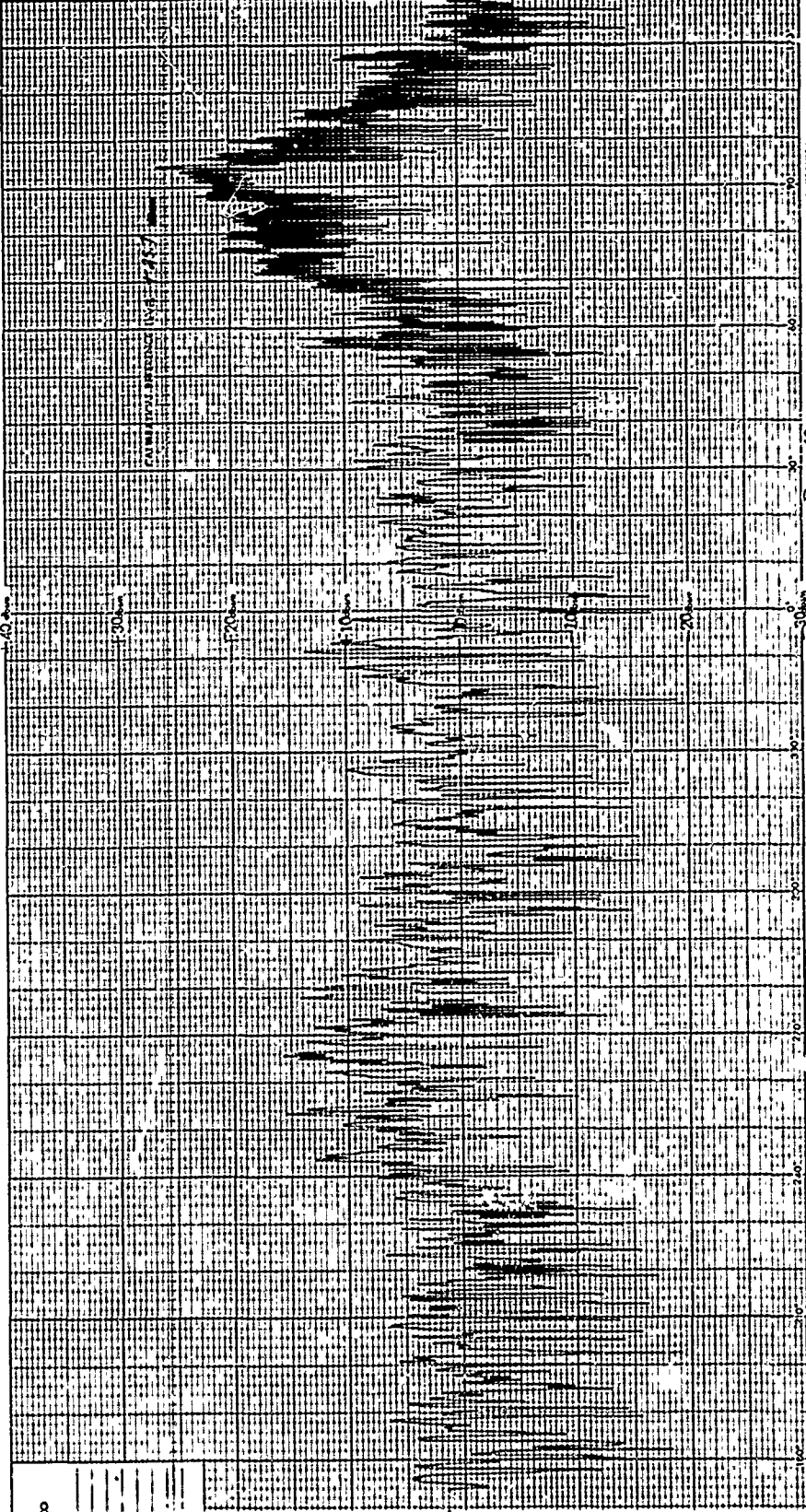


4885A 101 GROUP (N)
HOLDWAY AIR NEW MEXICO
EAT SAT
CONTROL NO. 23-51
DATE 15 AUG 73 PM 811
FREQUENCY 2700 MHz 9740
POLARIZATION VV BISTATIC 0°
OPERATOR JS O C JTH 08-1
Chevrolet 140
100 Roll 0° Bluch



5554A TEST GROUP 103
HOLCOMB, ALA. WITH METHOD
BY SUN

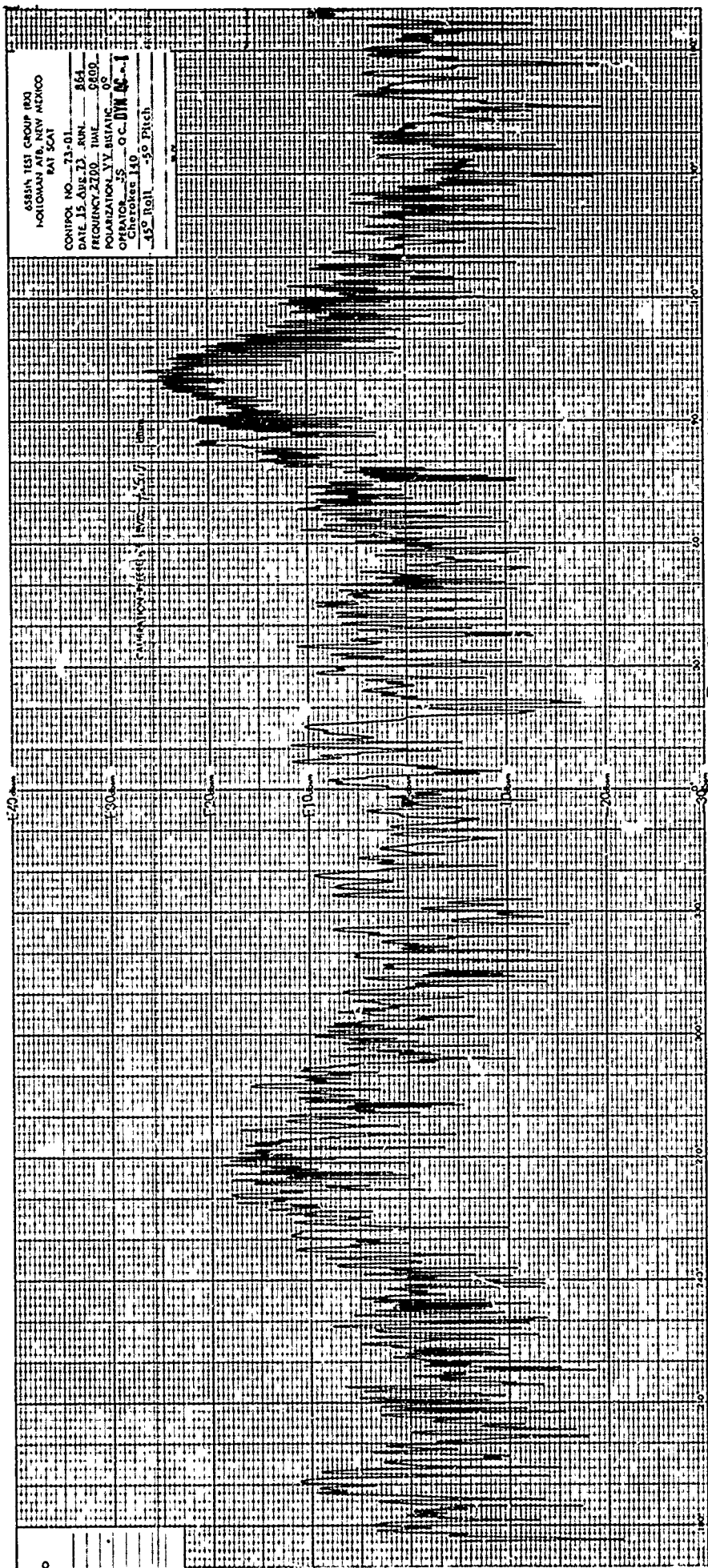
CONTROL NO. 73-01
DATE 14 Aug 73 RUN 927
FREQUENCY 2700 MHz 1150
POLARIZATION VV HORIZONTAL 0°
CREATOR JS O C III 00-1
Cherokee 140
10° Roll 110° Pitch



[illegible]

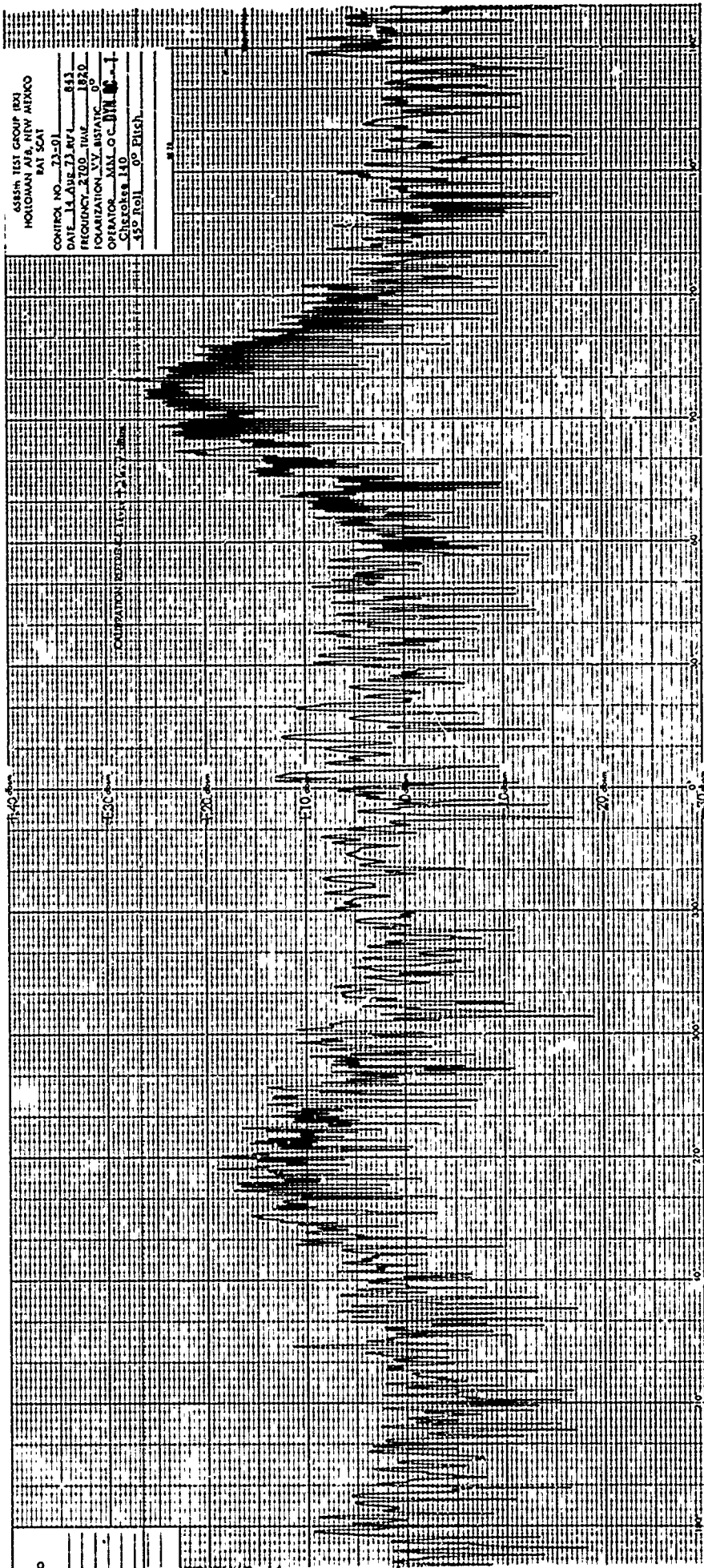
6854A TEST GROUP (R3)
HOLCOMB
PAT 5041

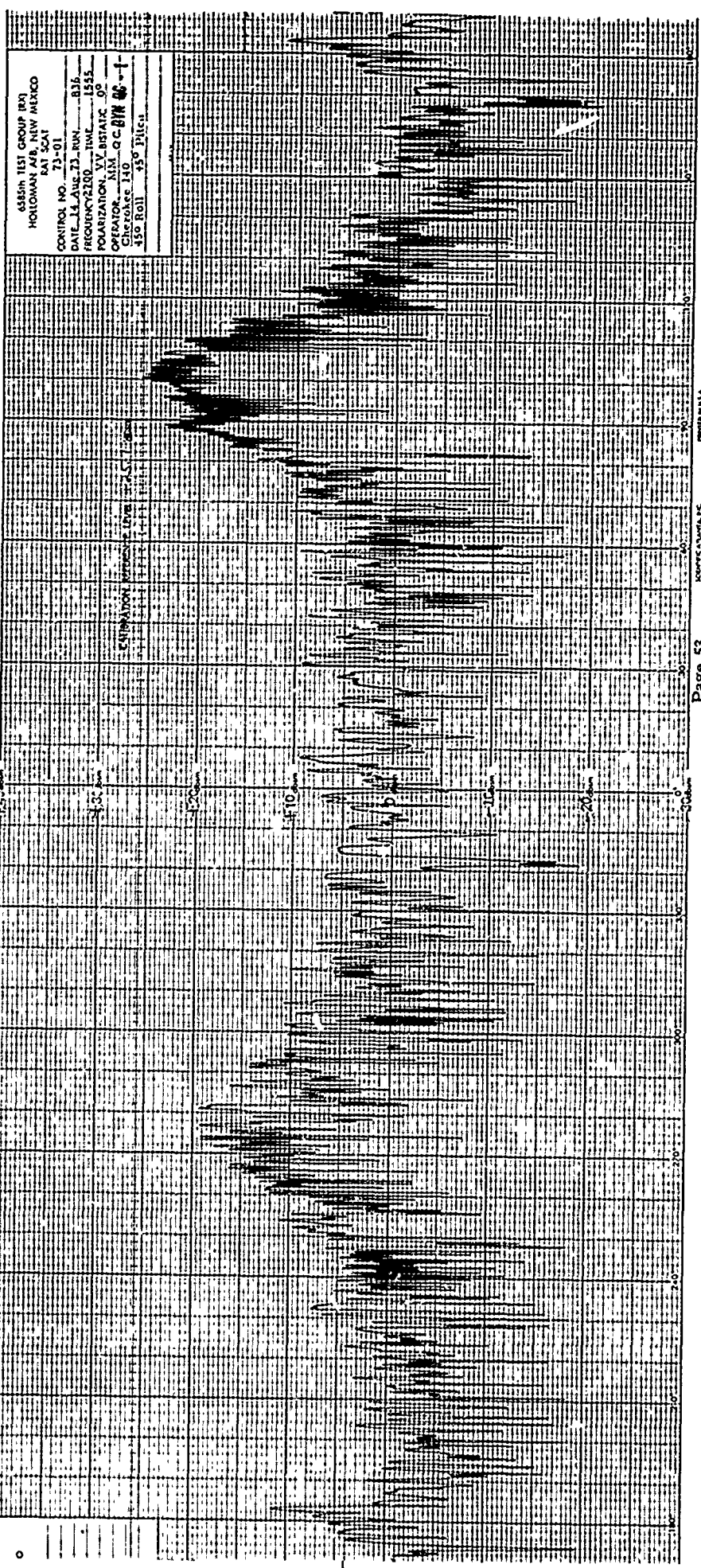
CONTROL NO. 73-01
DATE 15 AUG 73 RUN 864
FREQUENCY 2700 TIME 0800
POLARIZATION YV STATIC 05
OPERATOR JS O.C. DW 40-1
Cherokee 140
45° Roll -50 Pitch

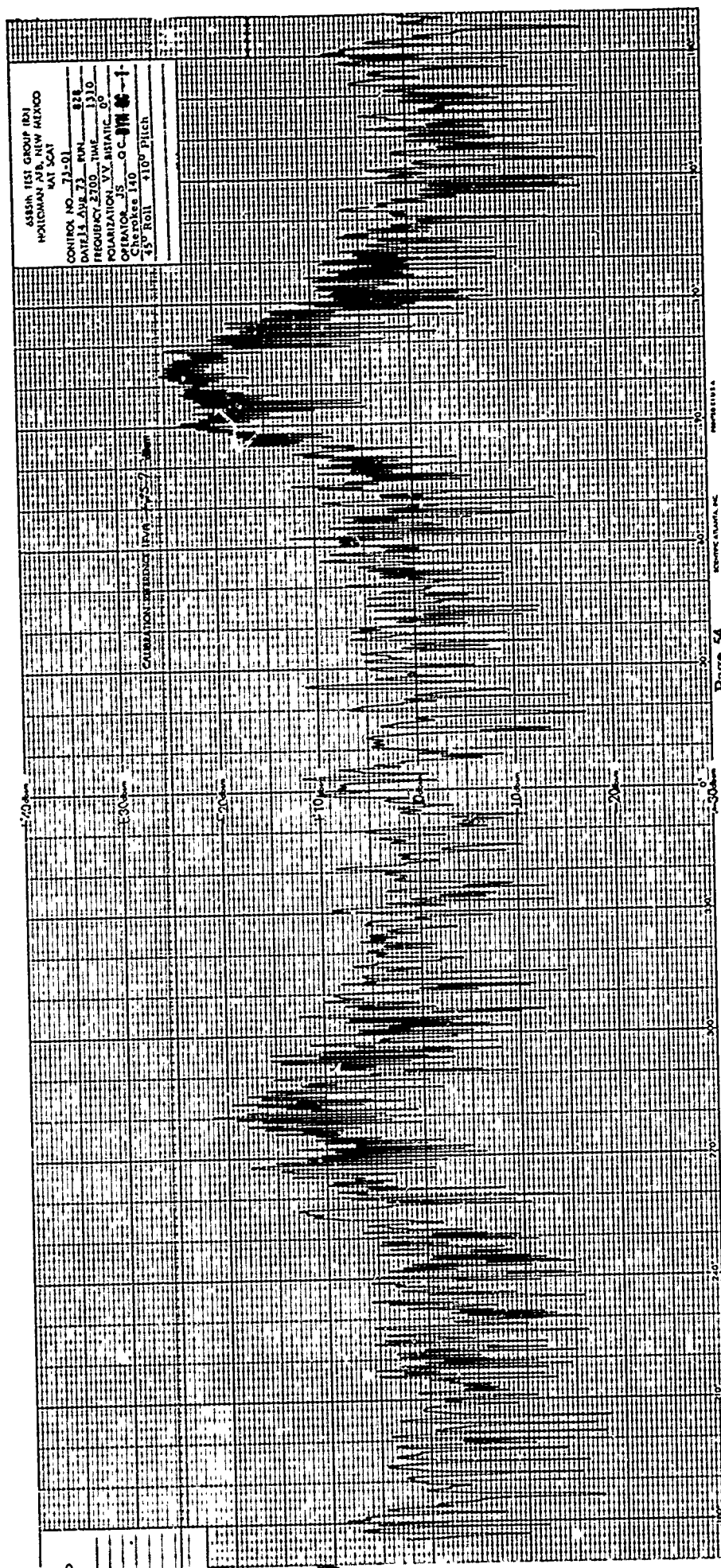


6385th TEST GROUP BDL
HOLLAMAN AFB, NEW MEXICO
BAT SCAT

CONTROL NO. 73-01
DATE 14 APR 73 AV 1 813
FREQUENCY 2700 TIME 1820
Polarization V.V. ANGLE 0°
OPERATOR MM OC DUN MC-1
Character 140
45° Incl 0° Ecliptic

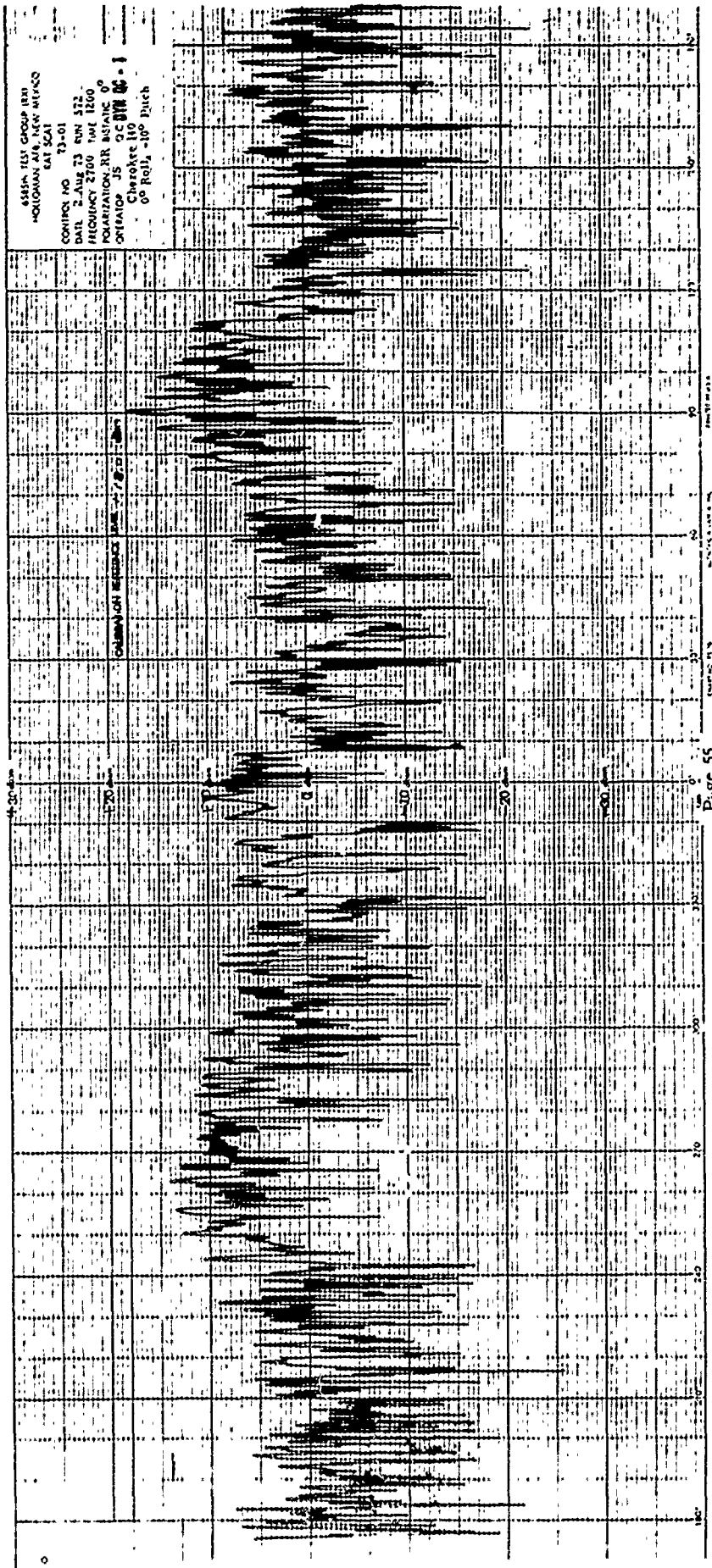


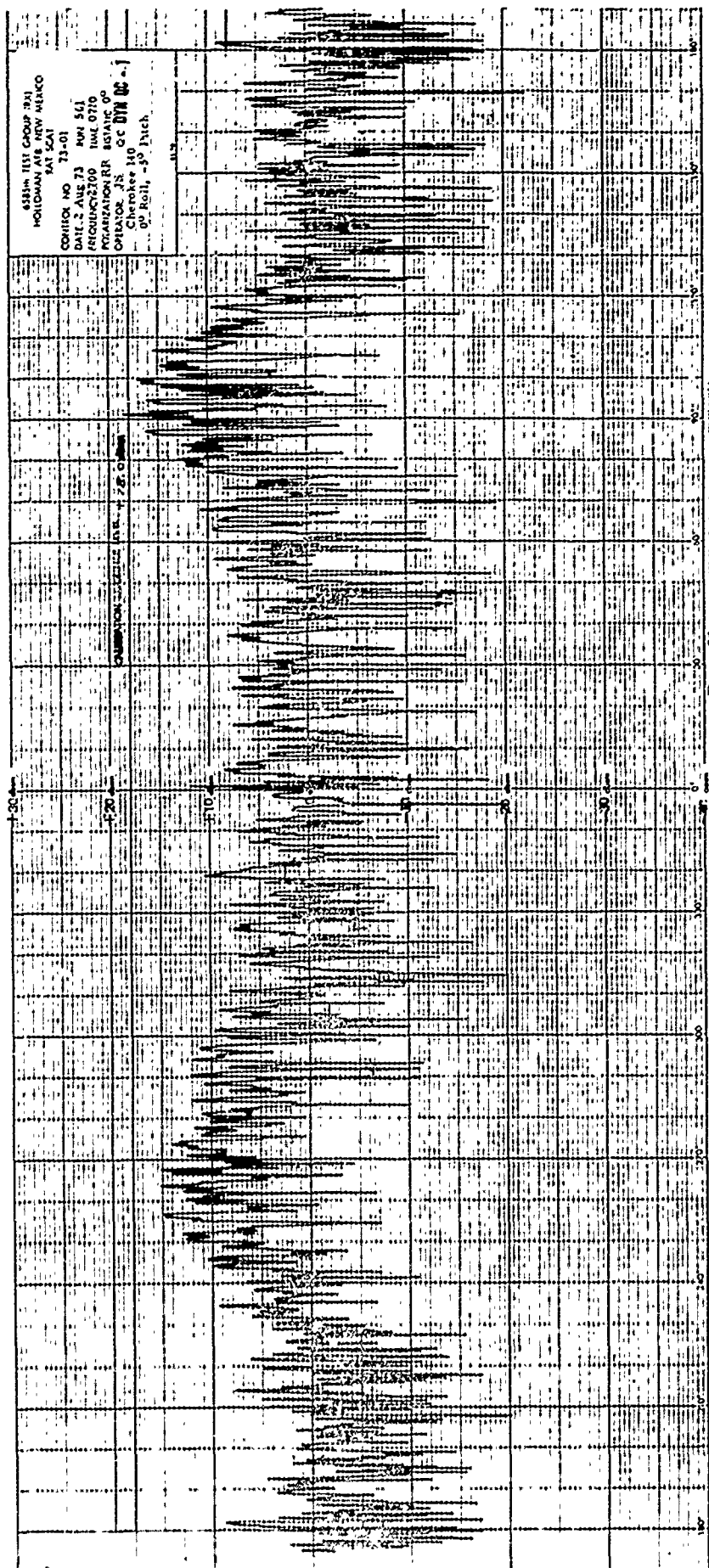


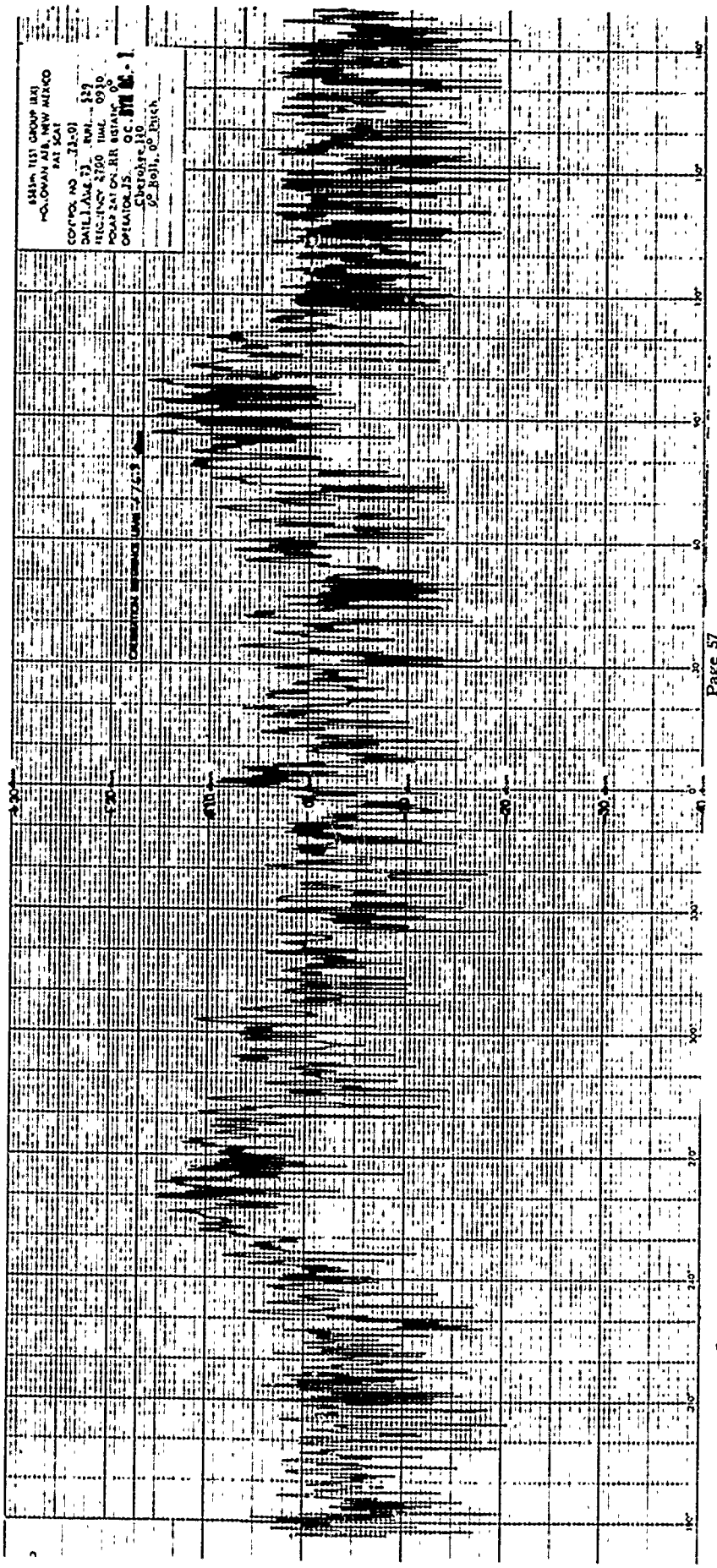


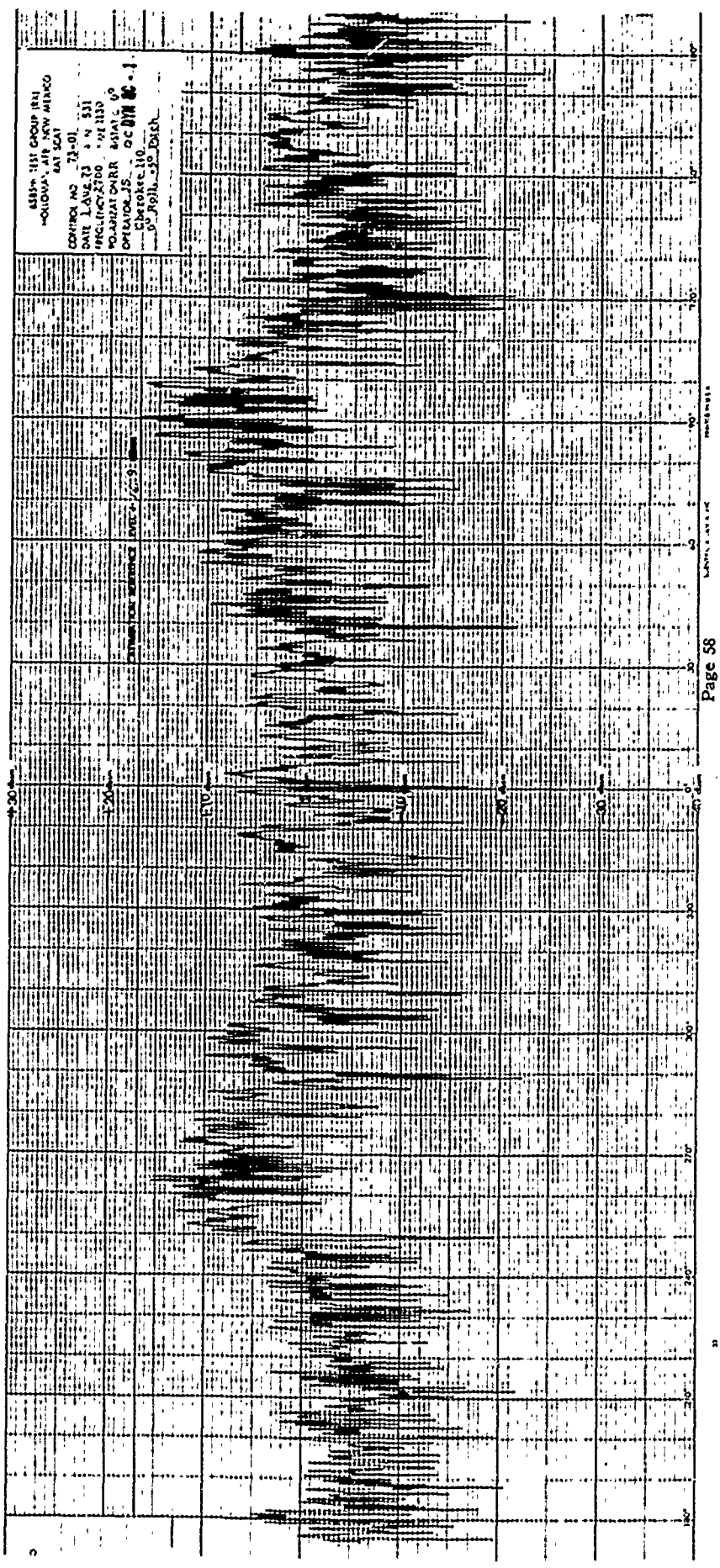
ASSEN TEST GROUP IDJ
HOLCOMAN AFB, NEW ALXCOO
MAT SCAT

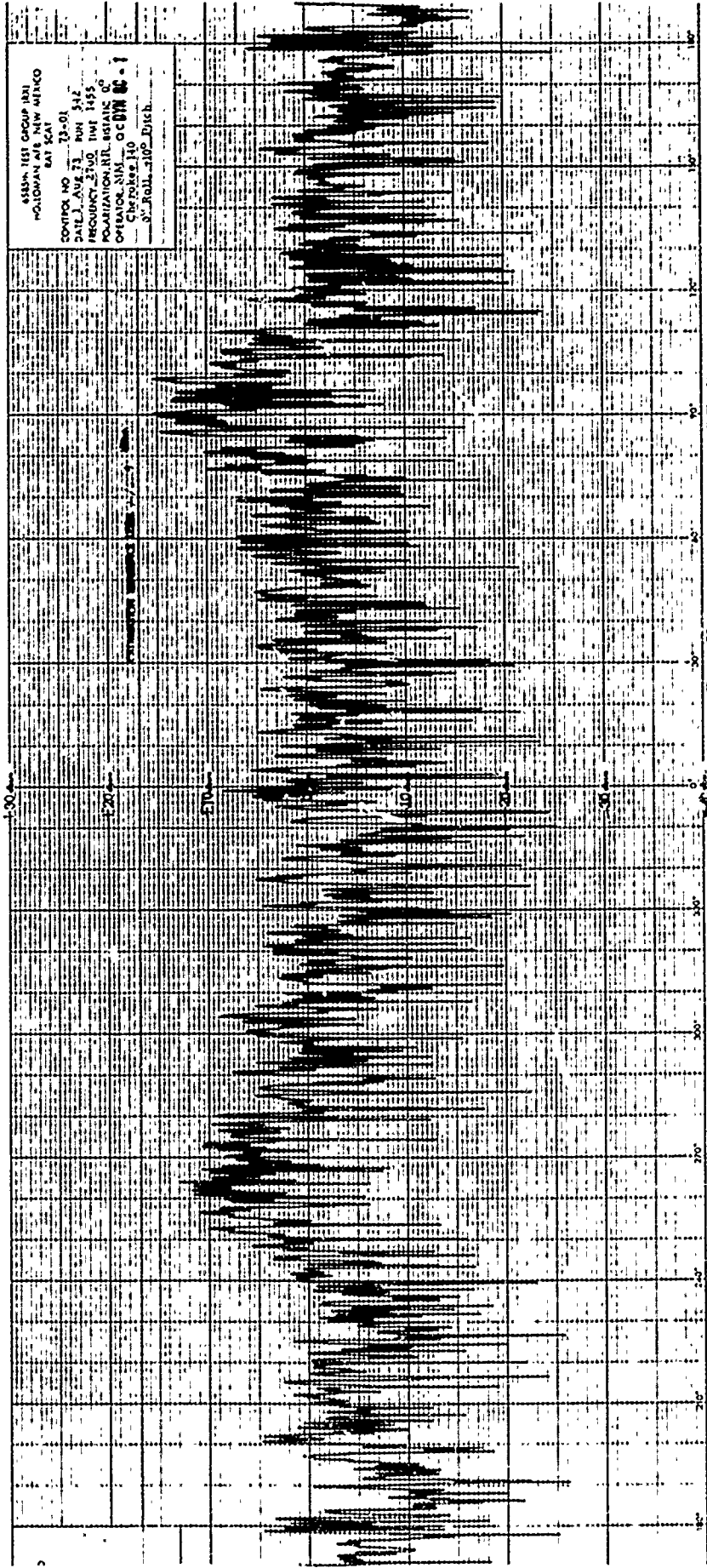
CONTROL NO. 71-01
DATE 11-29-73 RUN 828
FREQUENCY 2700 MHz 1310
POLARIZATION VV BRITISH 00
OPERATOR 35 G.C. 1118-1-1
Circuits 110 G.C. 1118-1-1
45° Roll +10° Pitch

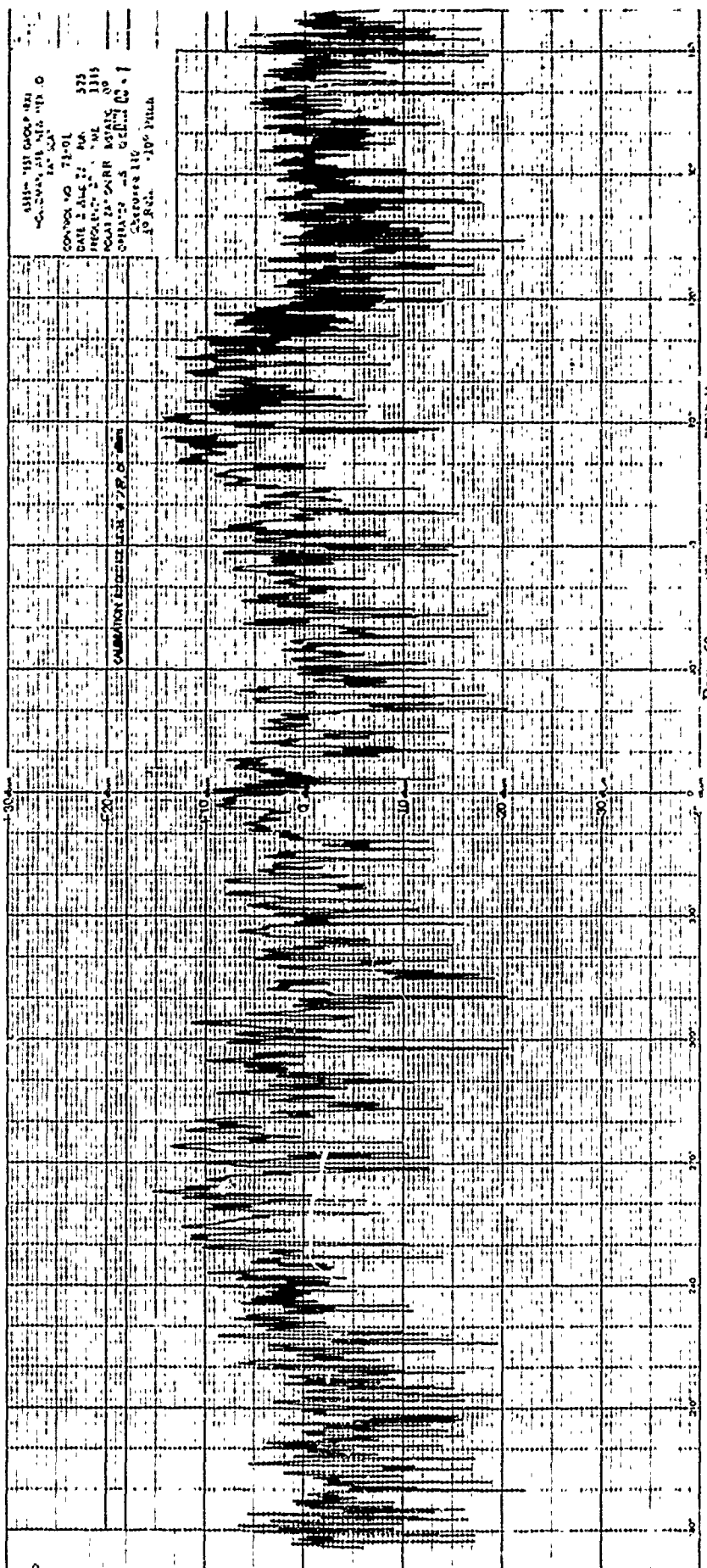








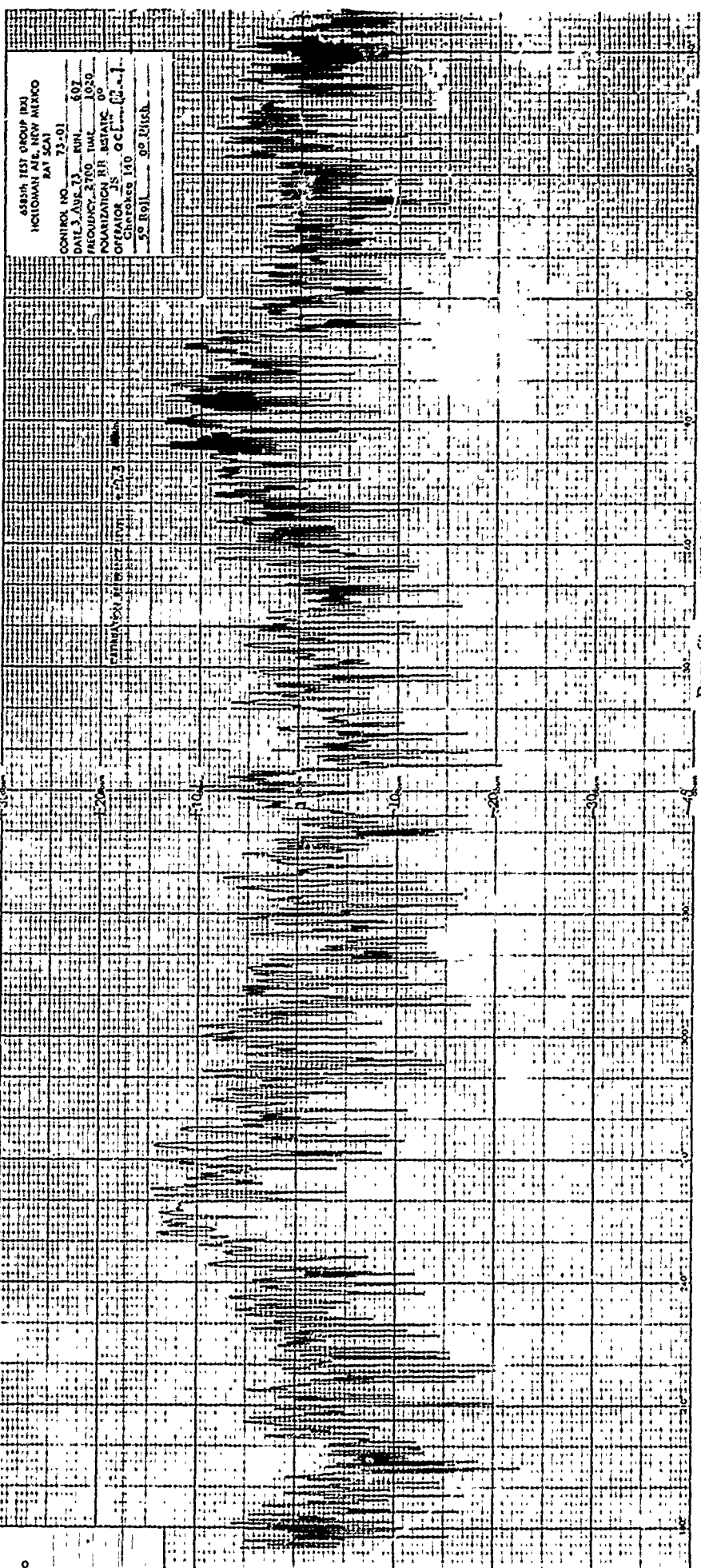




ASSEM TEST GROUP (BX)
HOLLOWAY AIR NEW MEXICO
BAT SCAT

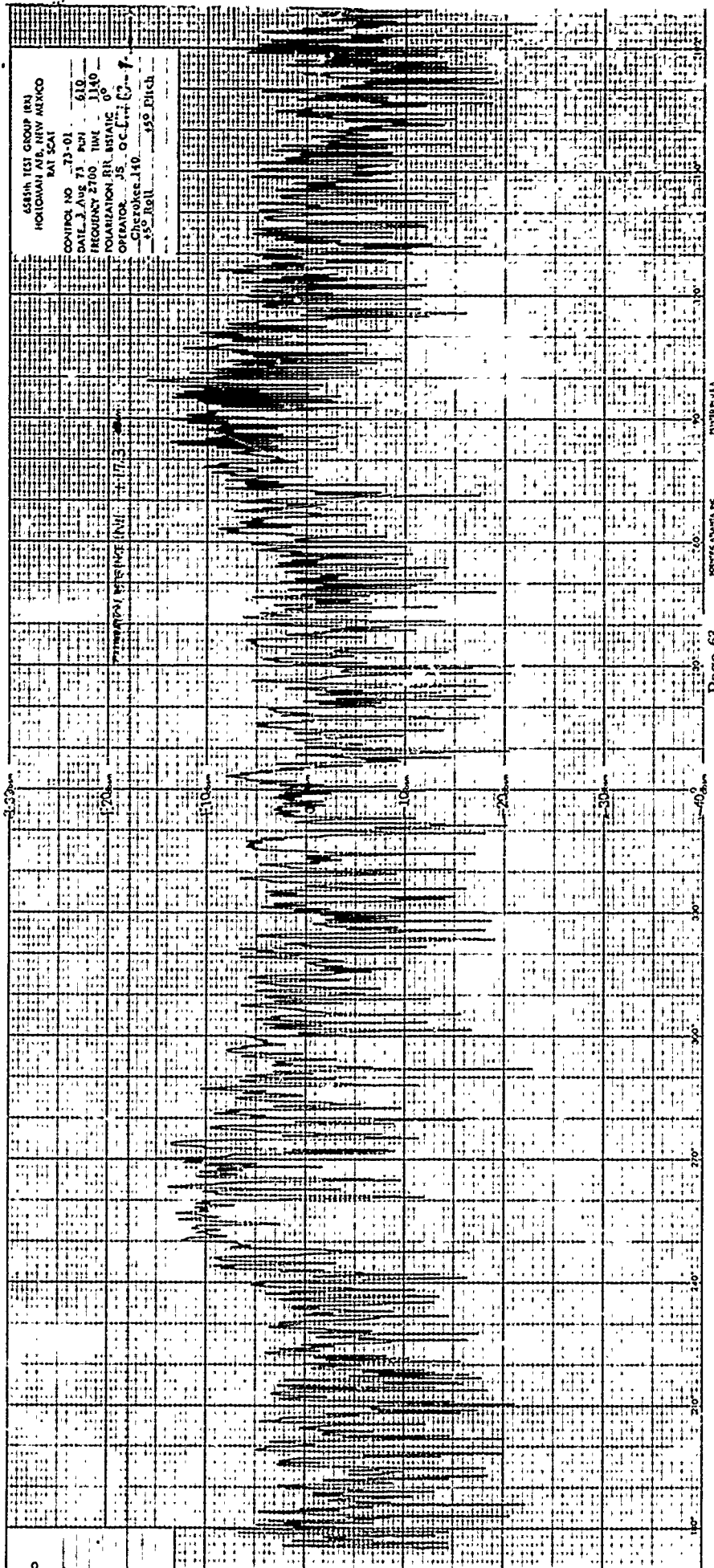
CONTROL NO. 73-01
DATE 1 AUG 71 RPL 570
FREQUENCY 2700 KHZ 0785
MODULATION PER BURST 100
OPERATOR JS GCUH 00-1
50 Roll -90 JUCH

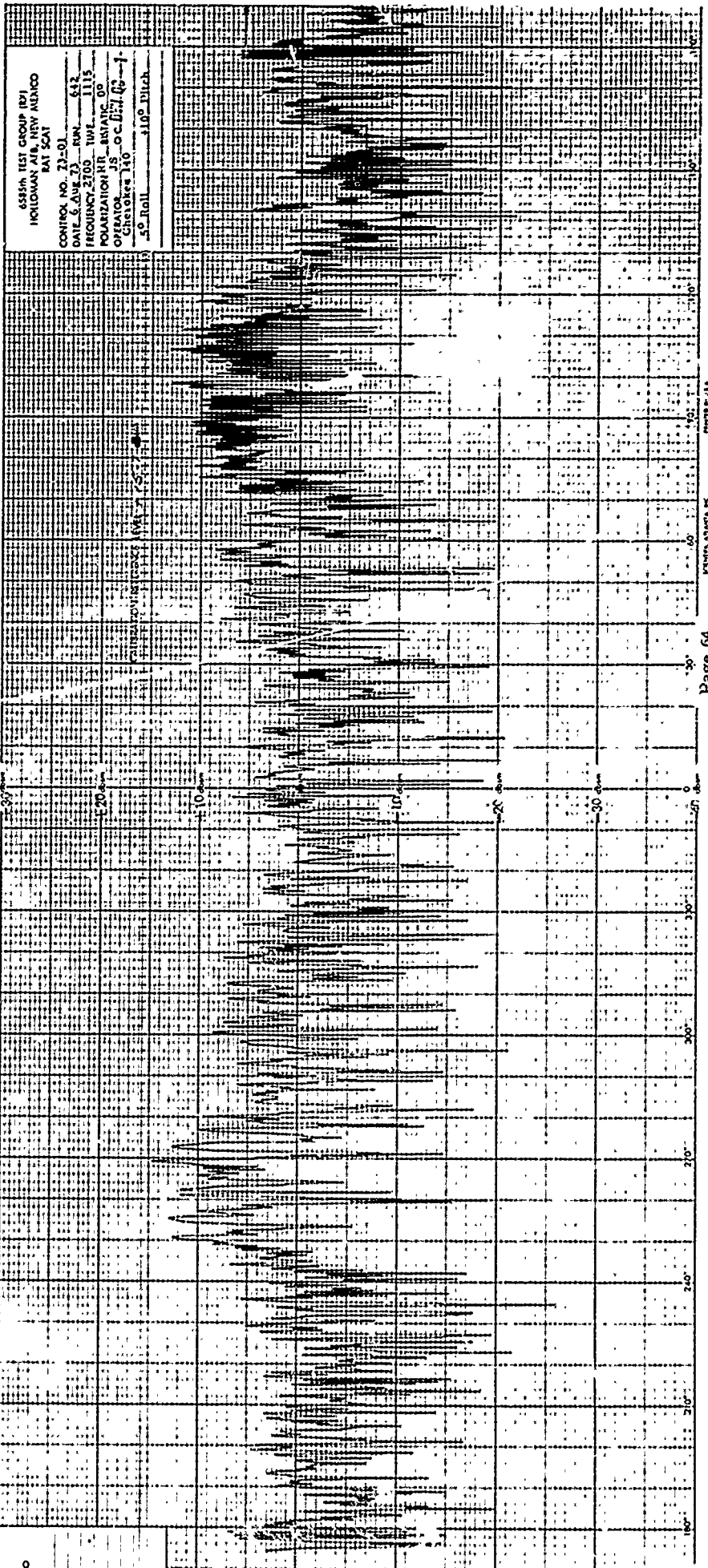
ALTIMETER 10000 FT



45544-107 GROUP (R)
HOLIMAN AIR NEW MEXICO
BAT SOI

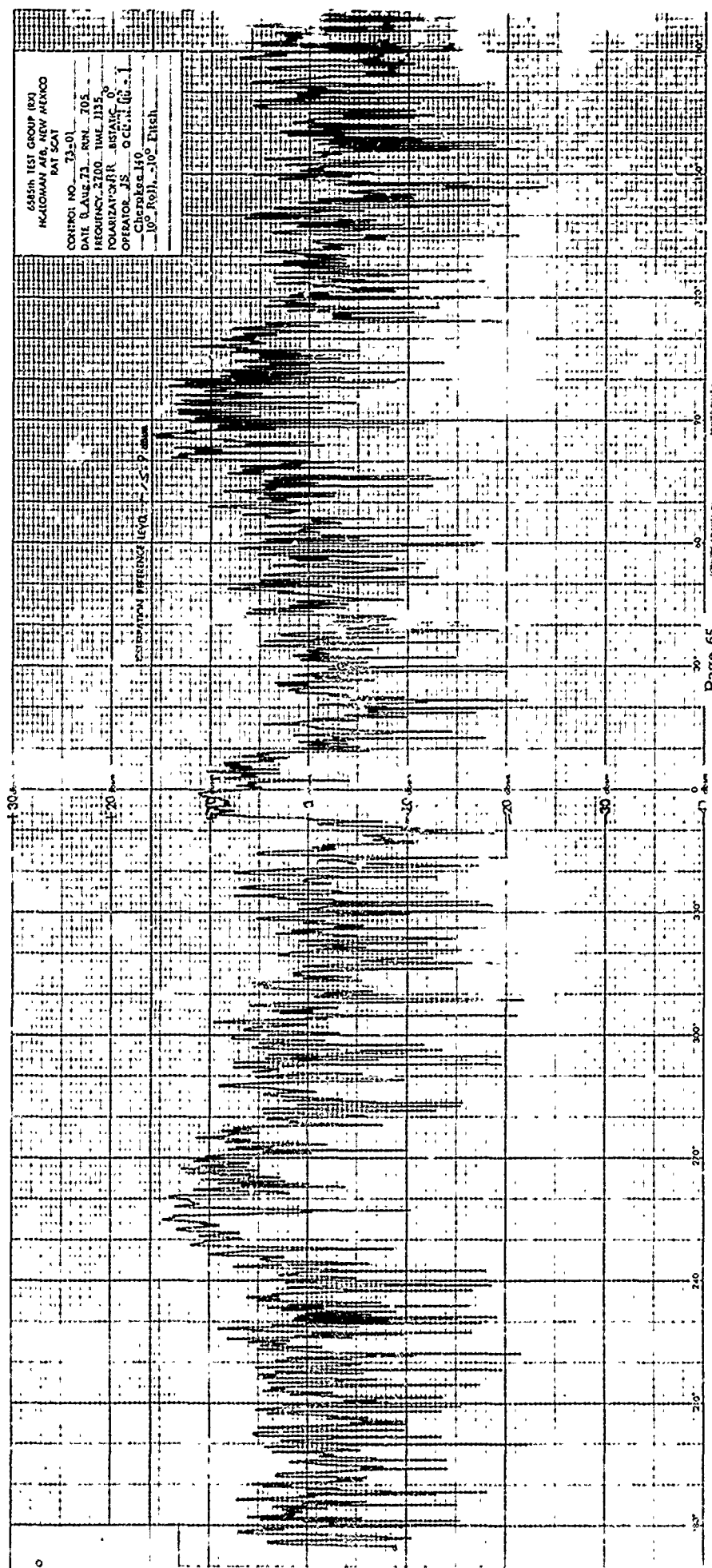
CONTROL NO. 23-01
DATE 3 AUG 73 RUN 607
FREQUENCY 2700 HZ 1020
POLARIZATION AIR BEARING 0°
OPERATOR JS GCL
Character 140
5° Rel 90° Right





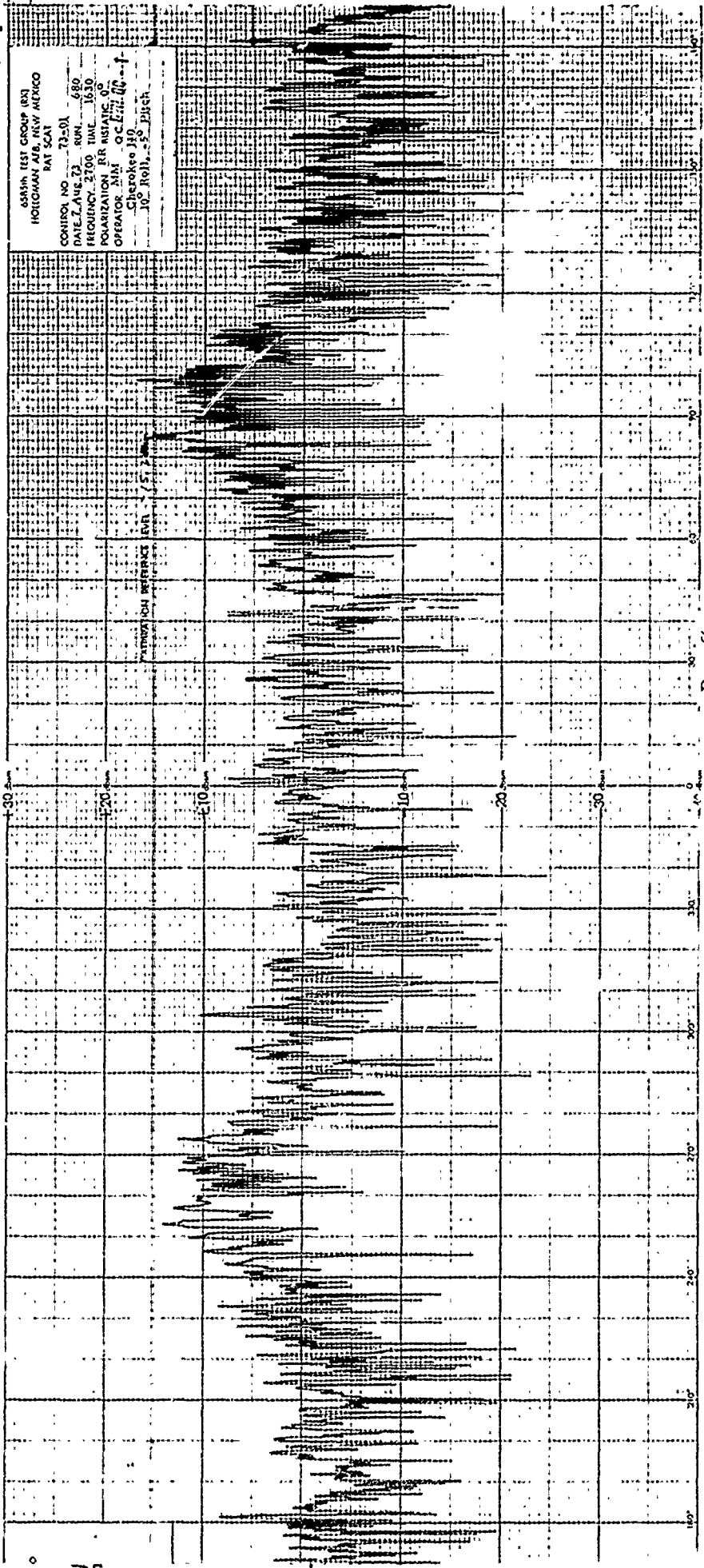
6585th TEST GROUP (RX)
MCLEOMAN AFB, NEW MEXICO
RAT SCAT

CONTROL NO. 73-01
DATE 8 AUG 73 RUN 705
FREQUENCY 2700 TIME 1135
POLARIZATION R BISTATIC 0
OPERATOR JS OCEAN 10-1
Cherokee 140
10° Roll, 10° Pitch



6551th TEST GROUP (EX)
HOLLOMAN AFB, NEW MEXICO
RAT SCAT

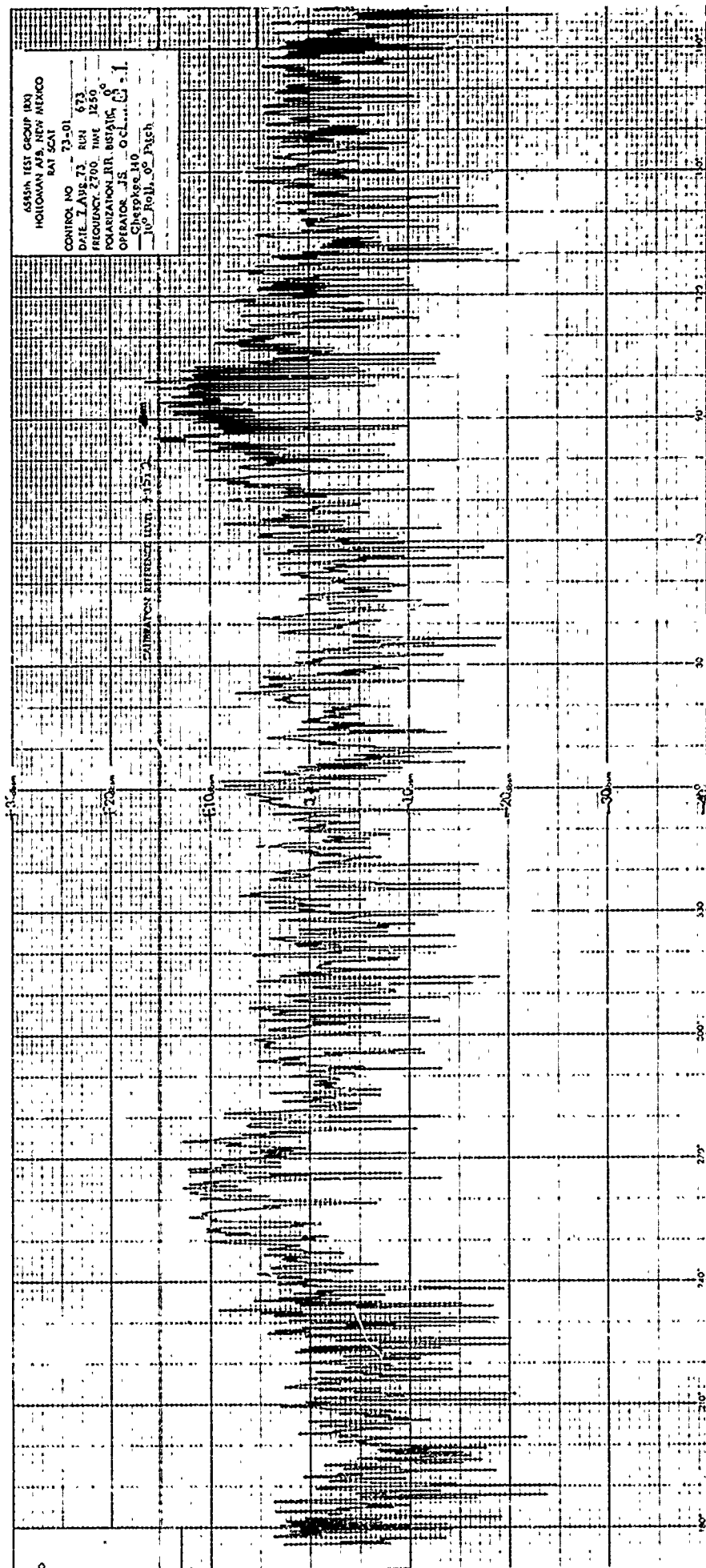
CONTROL NO. 72-01
DATE 7 AUG 72 RUN 680
FREQUENCY 2700 TIME 1630
POLARIZATION RR INSTRUC 00
OPERATOR MM OC 72-01-40-1
Cherokee J40
10° Roll, 5° Pitch



ASSIN TEST GROUP (BX)
HOLLAND AFB, NEW MEXICO
BAT SCAT

CONTROL NO 73-01
DATE LAUN 73 JUN 1973
FREQUENCY 2700 MHz 1250°
POLARIZATION R.R. BUREAU 0.1
OPERATOR JS OCL 11 11
Cherokee MO
10° Roll, 0° Pitch

CUMULATIVE RICHNESS DIV. 1-1512



655TH TEST GROUP (RA)
HOLLAND AFB, NEW MEXICO
BAT SCAT

CONTROL NO. 73-01

DATE 6 Aug 73 RPT 653

FREQUENCY 2700 TIME 1615

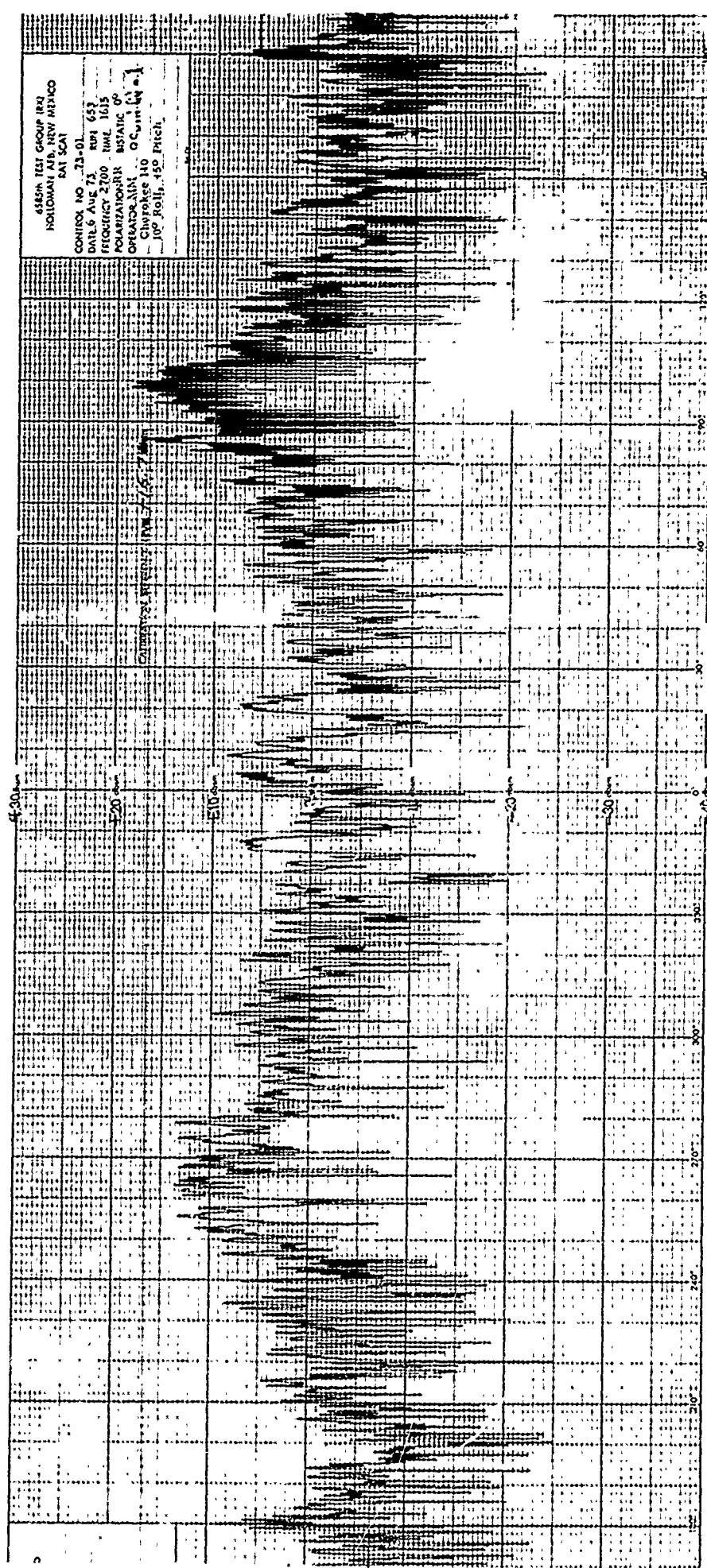
POLARIZATION R-R INSTANT 0°

OPERATOR NIN O. C. Smith

Cherokee 140 0.50 Pitch

100° Roll, 450 Pitch

CUMULATIVE FREQUENCY 1000-2700



ASSN. TEST GROUP 100
HOLDMAN AT NEW MEXICO
BAT 501

CONTROL NO. 73-81

DATE 6 Aug 53. RUN 446

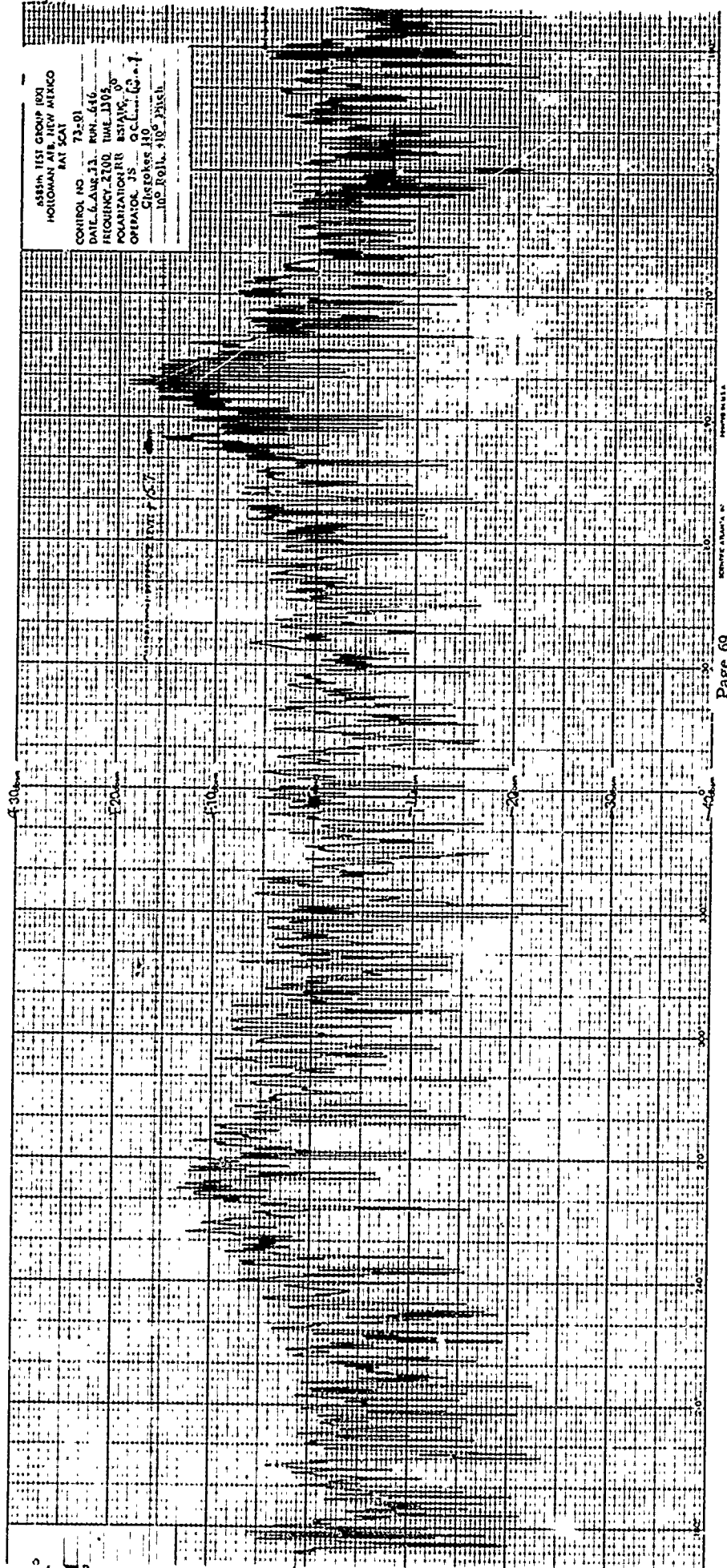
FREQUENCY 2700 TIME 1305

POLARIZATION R.R. 85° 10' 0"

OPERATOR J.S. G.C. 100 100

Class 100 100

100 Roll 100 Pitch



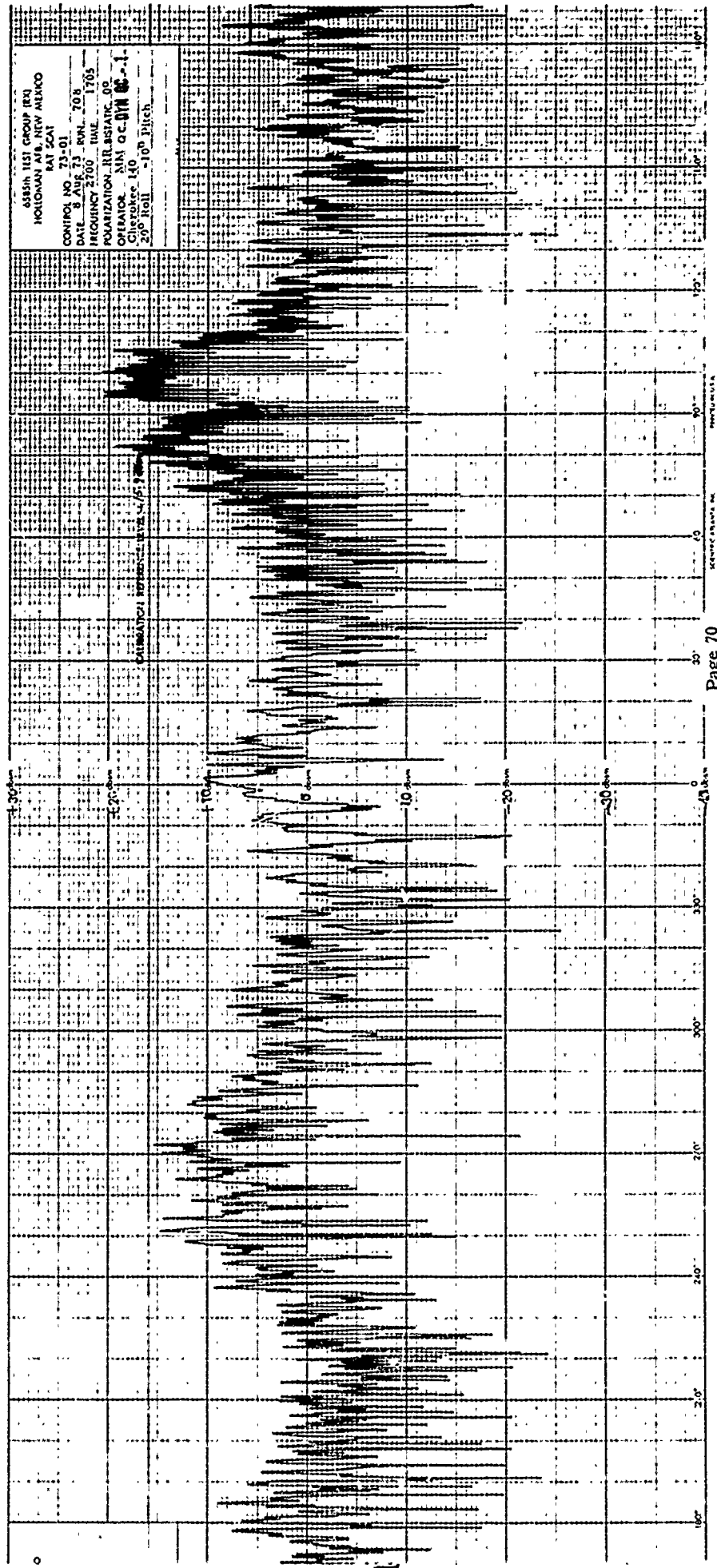
6555th TEST GROUP (EX)
HOLLOMAN AFB, NEW MEXICO
RAT SCAT

CONTROL NO. 73-01
DATE 8 AUG 73 RUN 708
FREQUENCY 2700 TIME 1705

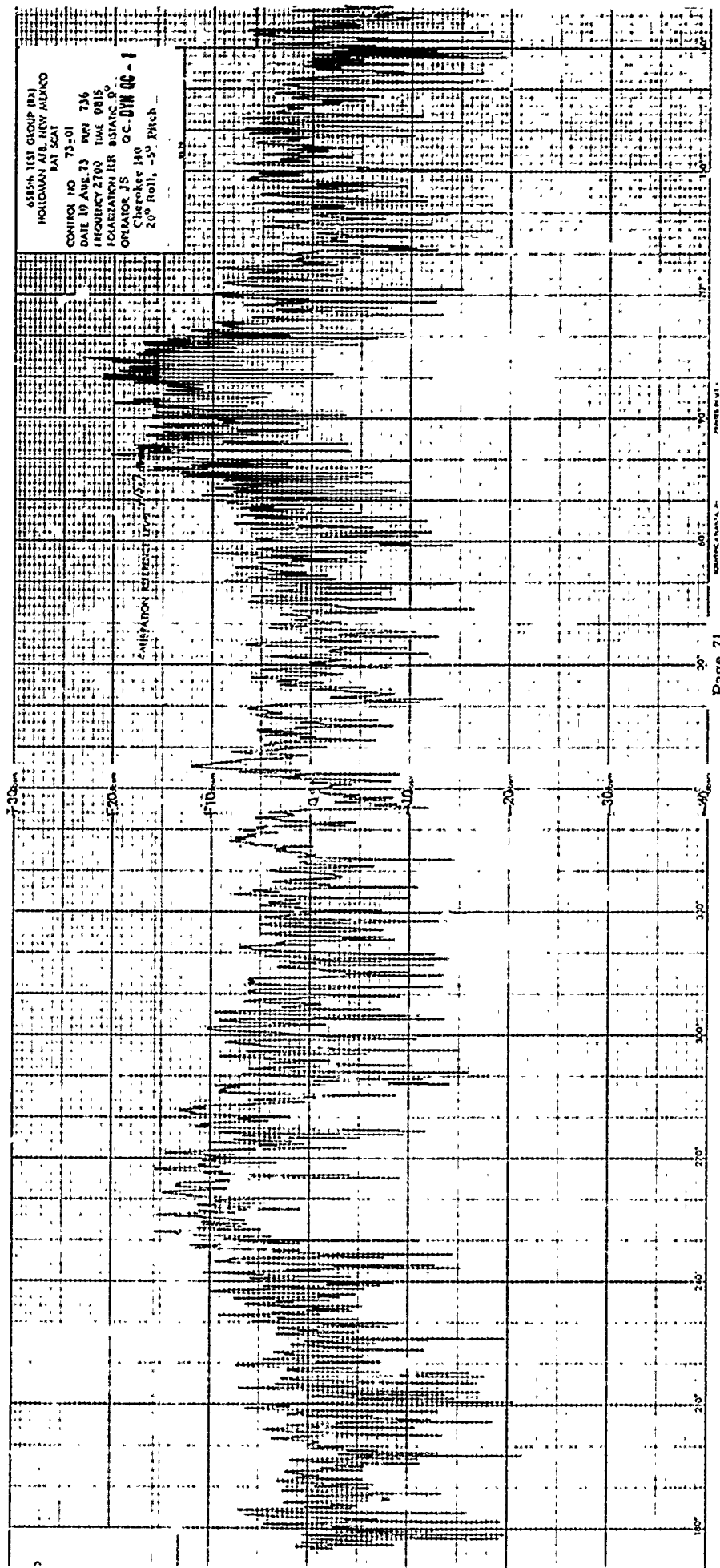
POLARIZATION R.R. 182147C. 00
OPERATOR MM OC 014 00-1

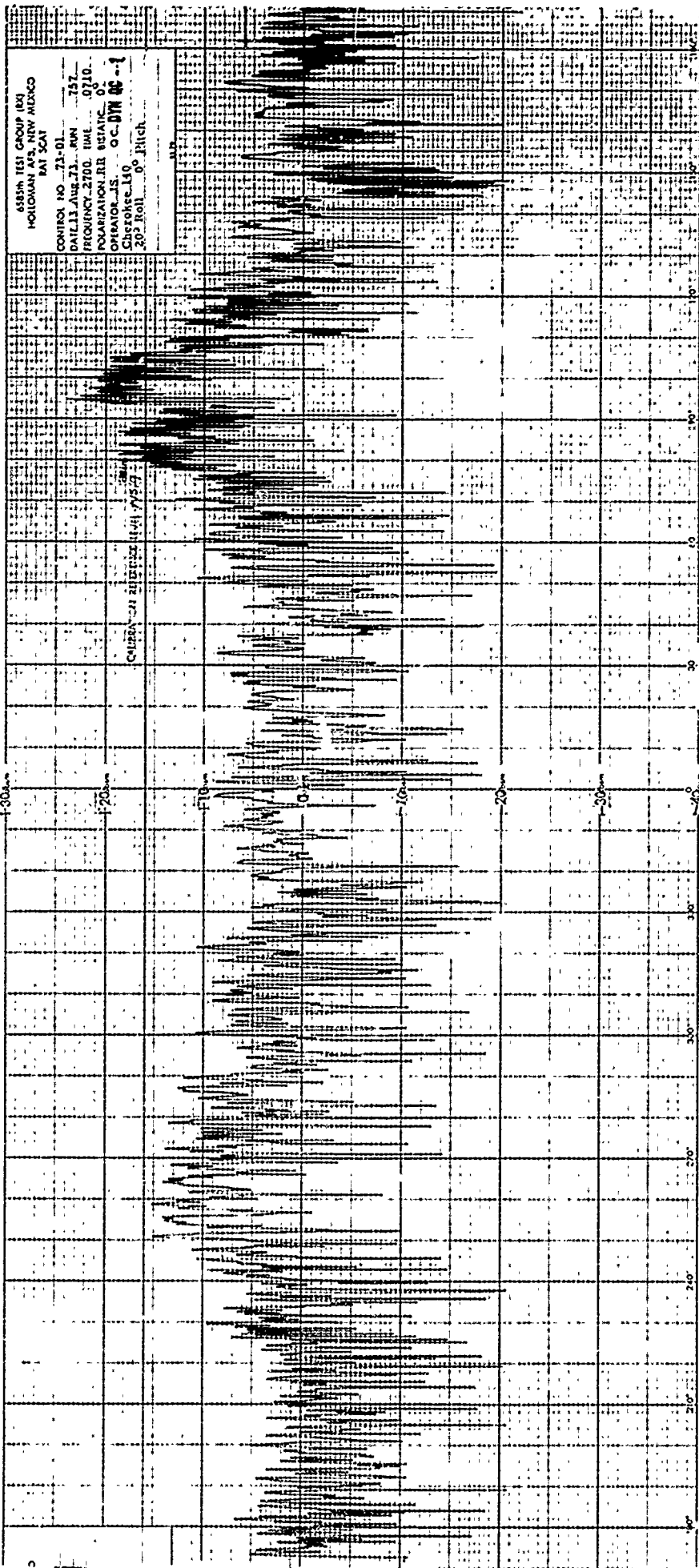
Character 140
20° Roll -10° Pitch

ESTIMATED ROTATION RATE 4.57 deg/sec

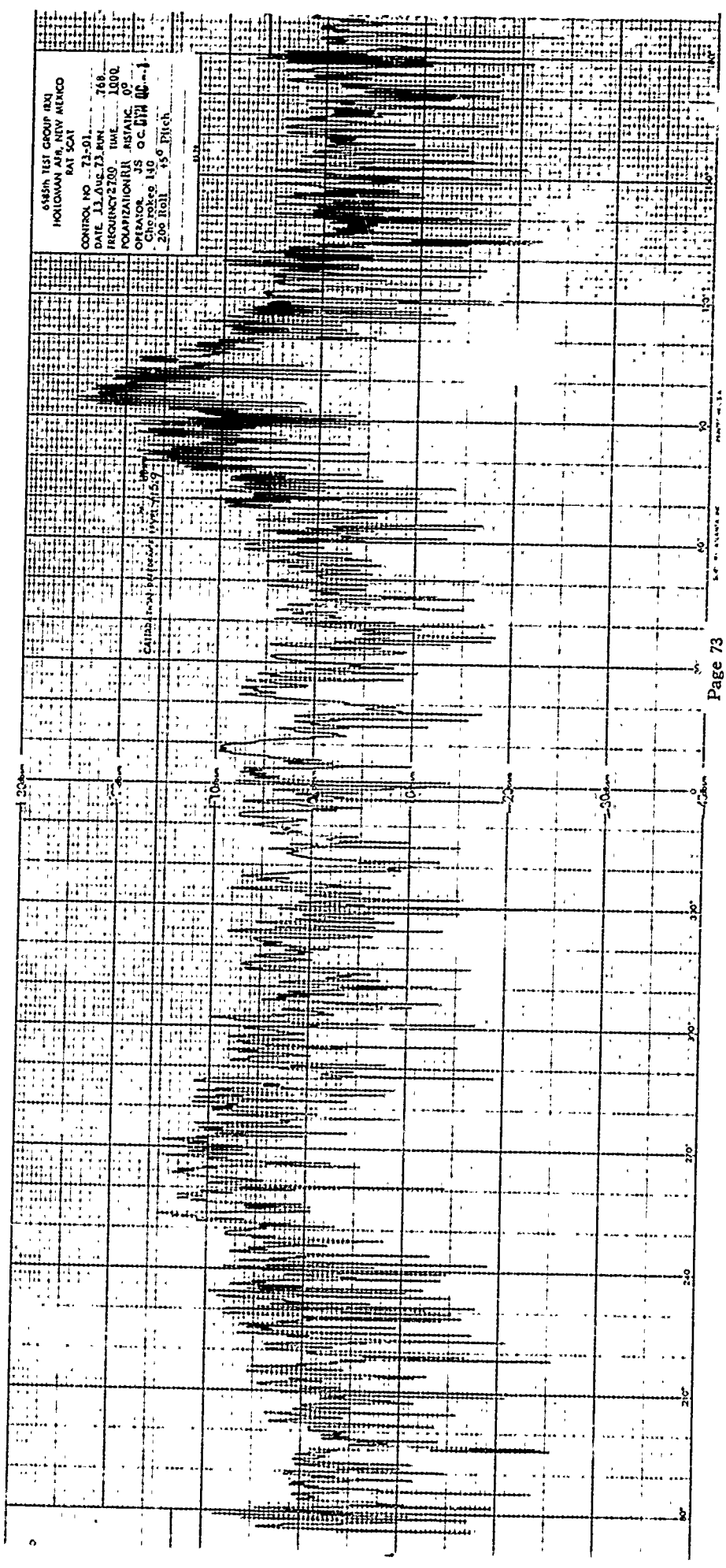


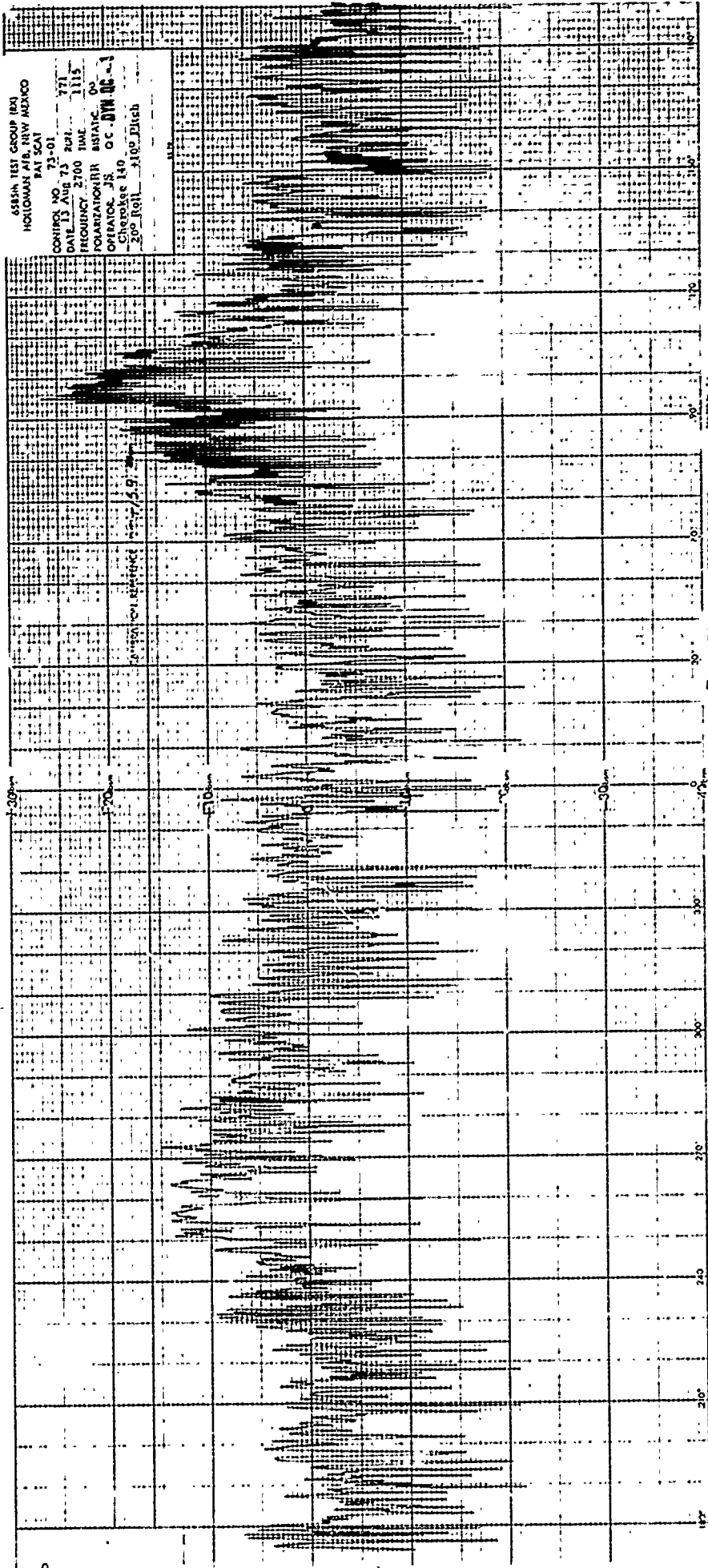
655PM TEST GROUP (BA)
HOLLOMAN AFB, NEW MEXICO
BAT SCAT
CONTROL NO 73-01
DATE 10 AUG 73 PW 736
FREQUENCY 2700 MHz 0815
FORECASTER JER BOSTON 0°
OPERATOR JS OC-DYN 00-1
Character J40
20° Roll, -5° Pitch





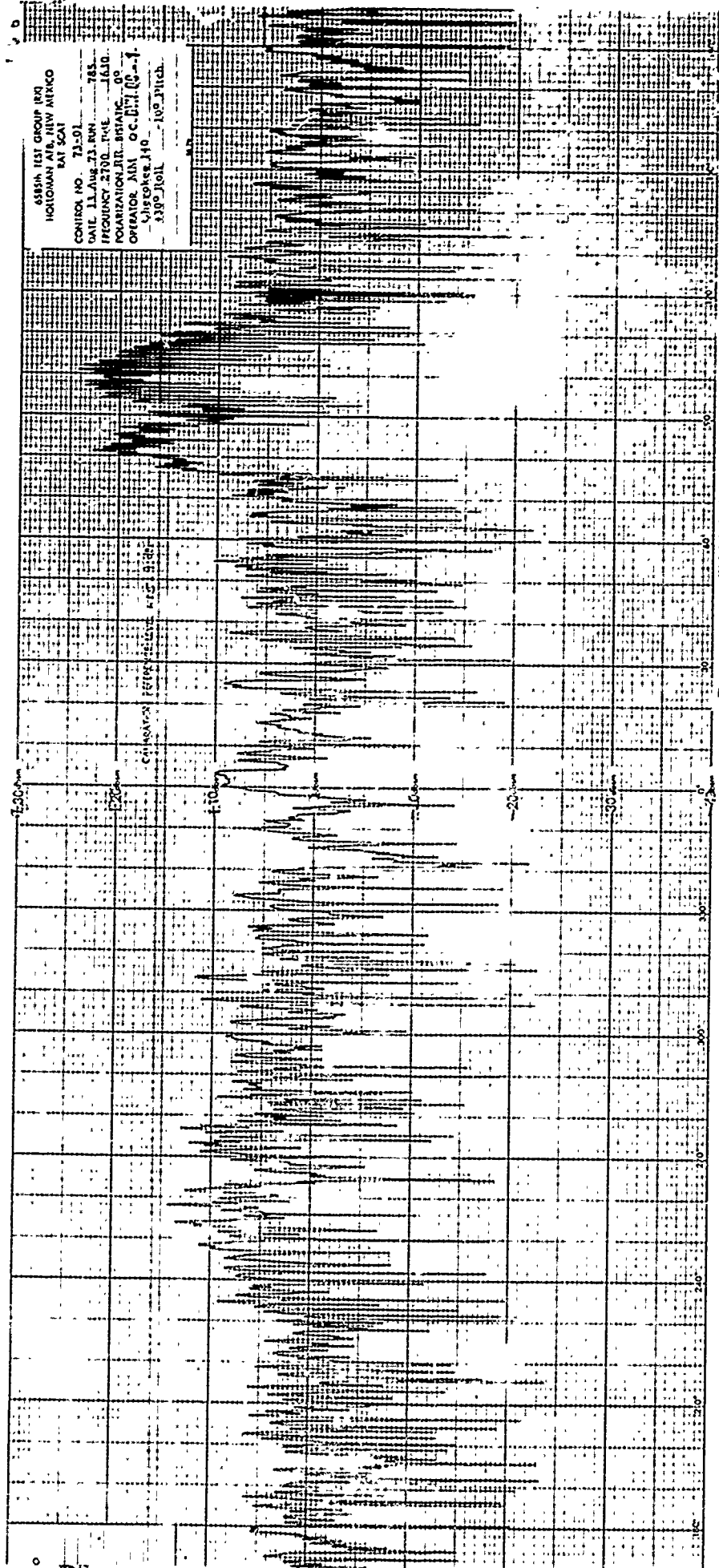
6153th TEST GROUP (BX)
 HOLLOMAN AIR, NEW MEXICO
 RAI 5047
 CONTROL NO. 73-01
 DATE 11 AUG 73 RUN 768
 FREQUENCY 2700 TIME 1000
 POLARIZATION RH ASTATIC 0°
 OPERATOR JS OC 0711 00-1
 Cherokee 140
 200 Roll 15° Pitch





6585A TEST GROUP (HQ)
HOLLOMAN AFB, NEW MEXICO
EAT SCAT

CONTROL NO. 72-01
DATE 11 Aug 73 RUN 785
FREQUENCY 2700 MHz 1610
POLARIZATION R.R. BISTATIC 0°
OPERATOR AIN OC-111 (0-1)
CIRCUITS 140 - 100 PITCH



ASST. TEL. OP. IN CH.
HONOLULU AIR TEL. OFFICE

DATE JUN 73

TIME 12:01

CONTROL NO. 72-01

DATE 12 JUN 73

TIME 12:01

FREQUENCY 2700

WPM 1812

POBARTON RR

OPERATOR M.M.

CHECKER M.M.

3300 Roll

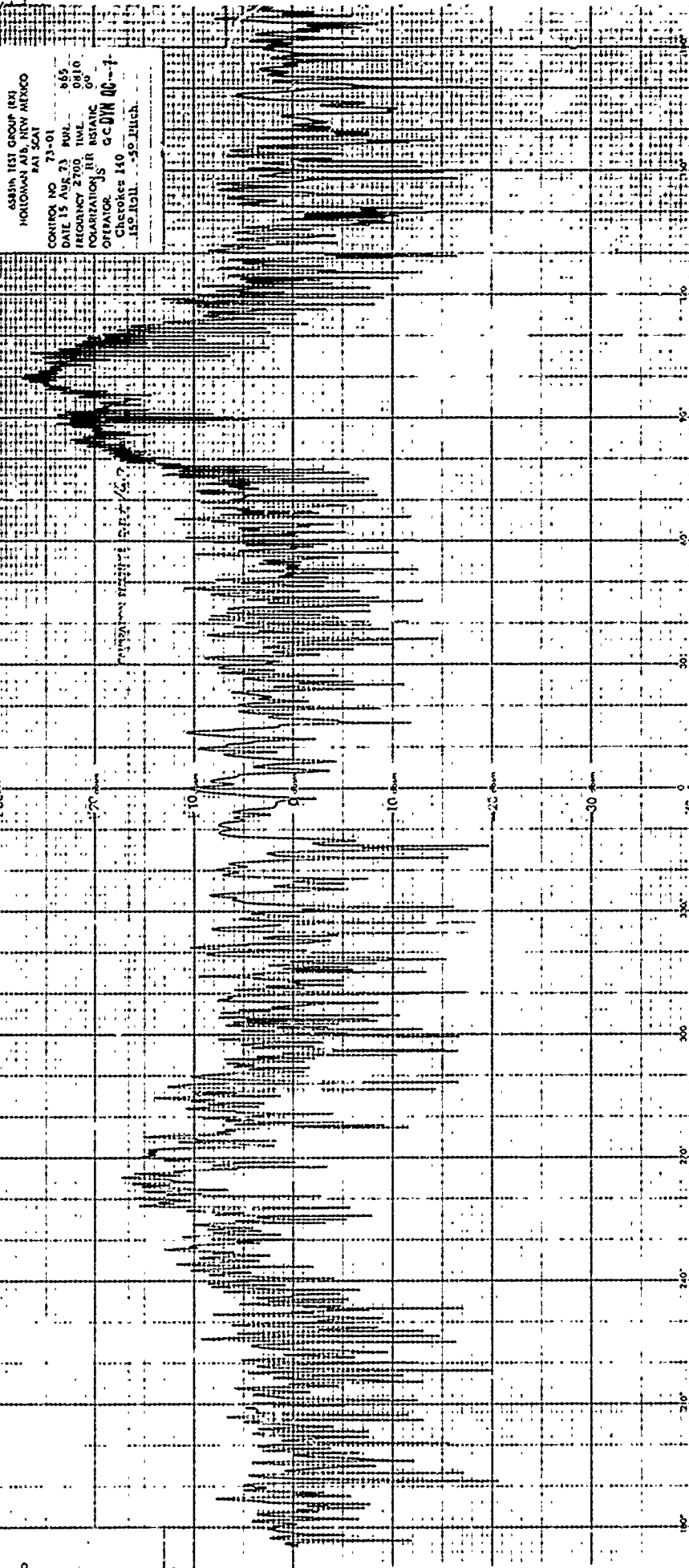
CONTROL NO 73-01
DATE 14 AUG 73 RUN 826
FREQUENCY 2700 MHz 114
POLARIZATION HORIZONTAL 00
OPERATOR JS OCT 1966
Cherokee 150
100 Roll +100 Pitch

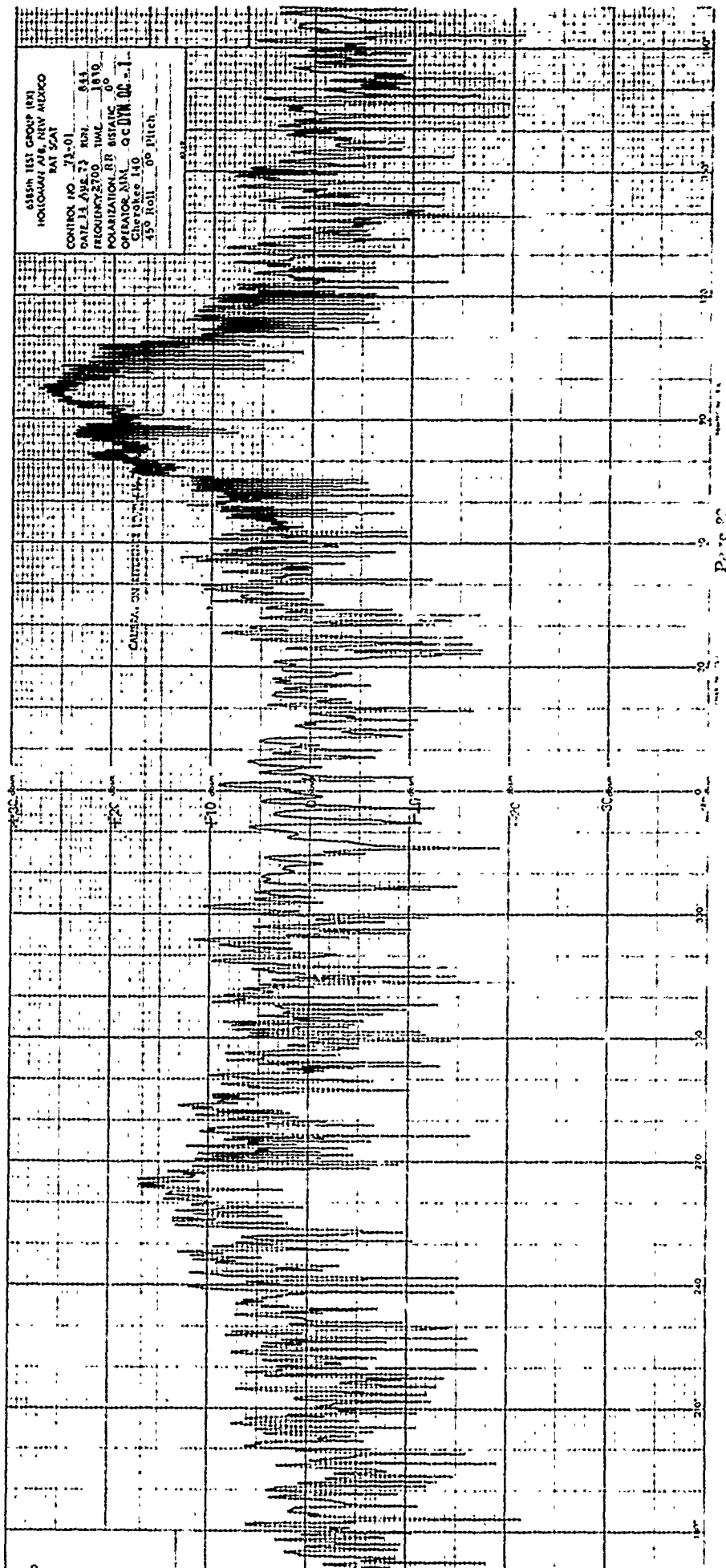
Campbell University

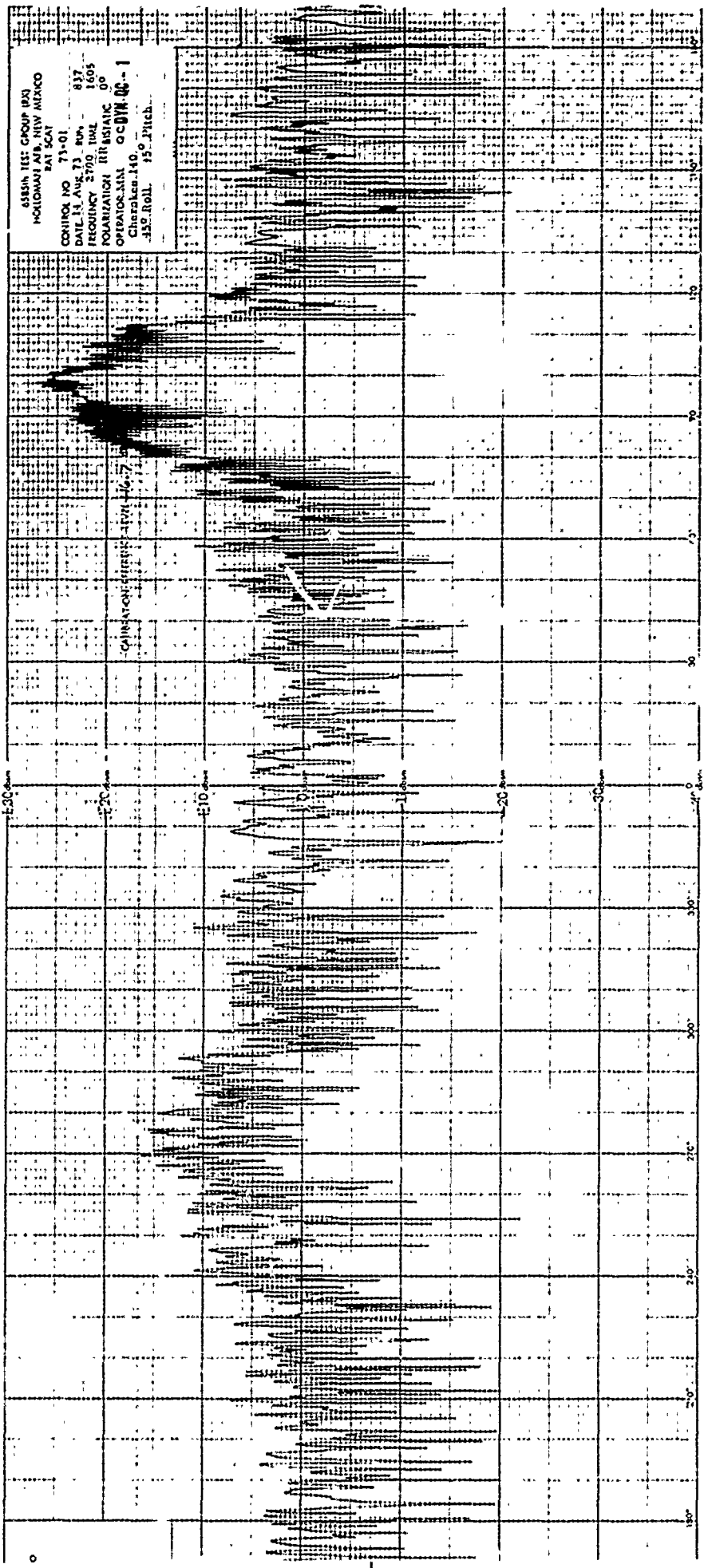
ASLJA TEST GROUP (EX)
HOLDMAN AFB, NEW MEXICO
EAT SCAT

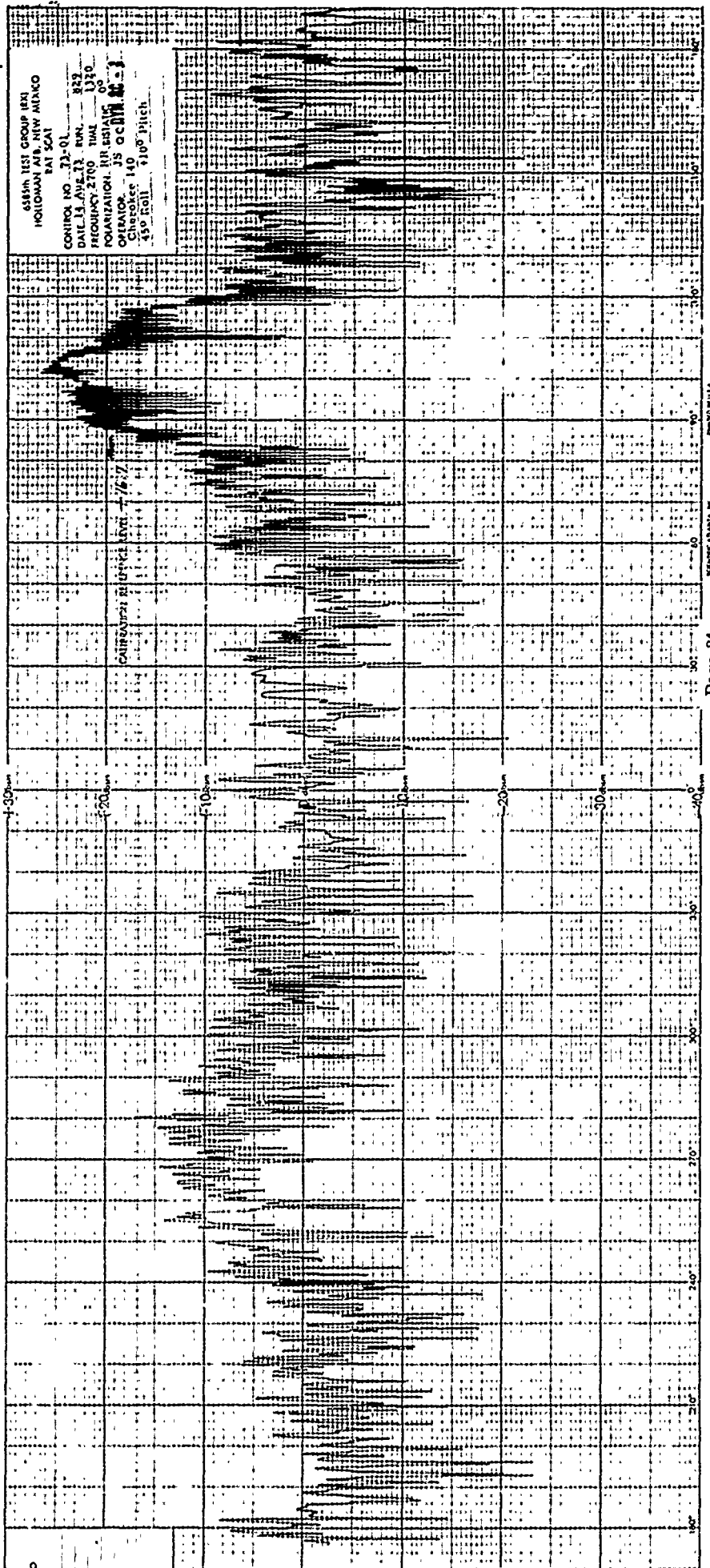
CONTROL NO 73-01
DATE 15 AUG 73 9AM 876
FREQUENCY 2730 TIME 1240
POLARIZATION RR ASTATIC 0°
OPERATOR JS OC DYN 00--1
Cherokee 140
45° Roll ~10° Pitch

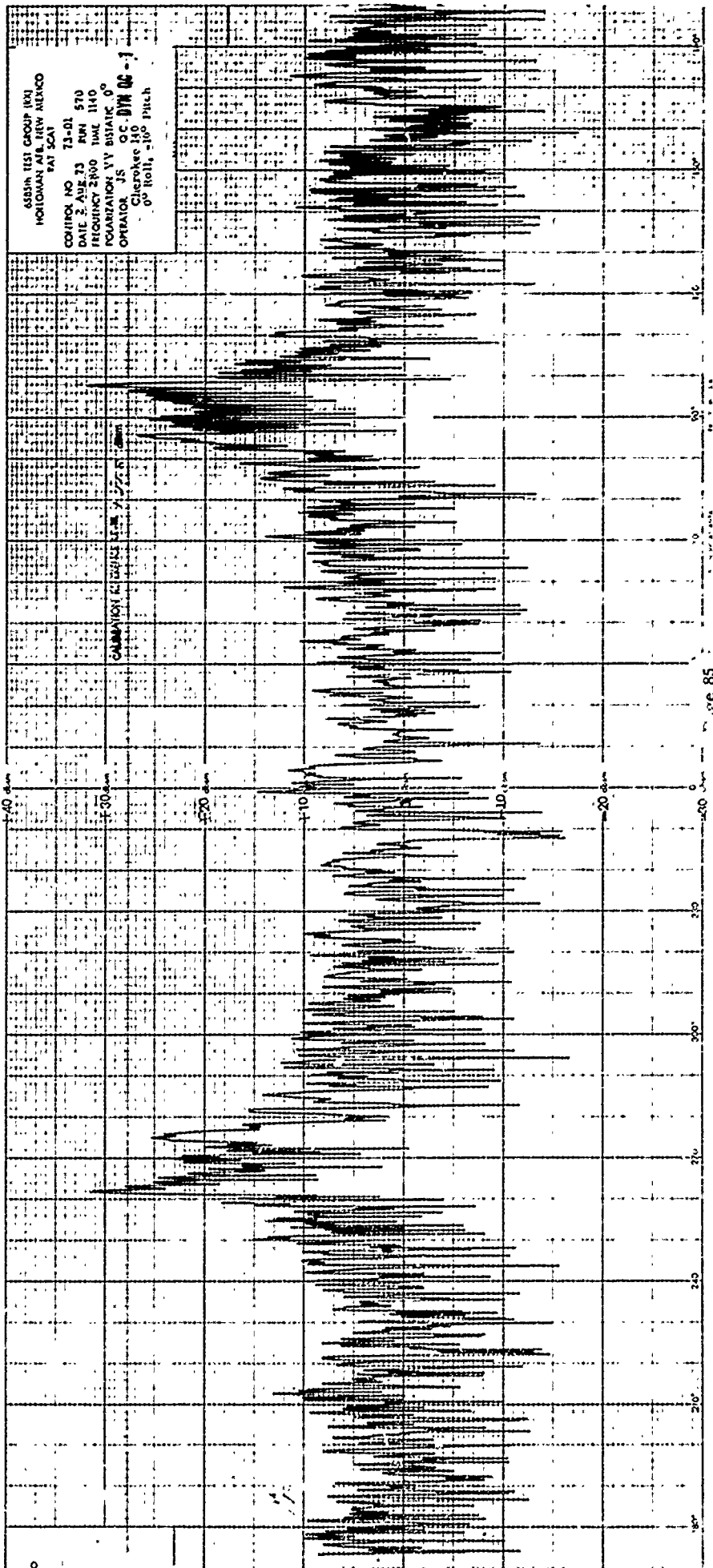
Calibration Scale: 1m, 1/2, 1/4, 1/8, 1/16, 1/32, 1/64, 1/128, 1/256, 1/512, 1/1024, 1/2048, 1/4096, 1/8192, 1/16384, 1/32768, 1/65536, 1/131072, 1/262144, 1/524288, 1/1048576, 1/2097152, 1/4194304, 1/8388608, 1/16777216, 1/33554432, 1/67108864, 1/134217728, 1/268435456, 1/536870912, 1/1073741824, 1/2147483648, 1/4294967296, 1/8589934592, 1/17179869184, 1/34359738368, 1/68719476736, 1/137438953472, 1/274877906944, 1/549755813888, 1/1099511627776, 1/2199023255552, 1/4398046511104, 1/8796093022208, 1/17592186044416, 1/35184372088832, 1/70368744177664, 1/140737488355328, 1/281474976710656, 1/562949953421312, 1/1125899906842624, 1/2251799813685248, 1/4503599627370496, 1/9007199254740992, 1/18014398509481984, 1/36028797018963968, 1/72057594037927936, 1/144115188075855872, 1/288230376151711744, 1/576460752303423488, 1/1152921504606846976, 1/2305843009213693952, 1/4611686018427387904, 1/9223372036854775808, 1/18446744073709551616, 1/36893488147419103232, 1/73786976294838206464, 1/147573952589676412928, 1/295147905179352825856, 1/590295810358705651712, 1/1180591620717411303424, 1/2361183241434822606848, 1/4722366482869645213696, 1/9444732965739290427392, 1/18889465931478580854784, 1/37778931862957161709568, 1/75557863725914323419136, 1/151115727451828646838272, 1/302231454903657293676544, 1/604462909807314587353088, 1/1208925819614629174706176, 1/2417851639229258349412352, 1/4835703278458516698824704, 1/9671406556917033397649408, 1/19342813113834066795298816, 1/38685626227668133590597632, 1/77371252455336267181195264, 1/154742504910672534362390528, 1/309485009821345068724781056, 1/618970019642690137449562112, 1/1237940039285380274899124224, 1/2475880078570760549798248448, 1/4951760157141521099596496896, 1/9903520314283042199192993792, 1/19807040628566084398385987584, 1/39614081257132168796771975168, 1/79228162514264337593543950336, 1/158456325028528675187087900672, 1/316912650057057350374175801344, 1/633825300114114700748351602688, 1/1267650600228229401496703205376, 1/2535301200456458802993406410752, 1/5070602400912917605986812821504, 1/10141204801825835211973625643008, 1/20282409603651670423947251286016, 1/40564819207303340847894502572032, 1/81129638414606681695789005144064, 1/162259276829213363391578010288128, 1/324518553658426726783156020576256, 1/649037107316853453566312041152512, 1/1298074214633706907132624082305024, 1/2596148429267413814265248164610048, 1/5192296858534827628530496329220096, 1/10384593717069655257060992658440192, 1/20769187434139310514121985316880384, 1/41538374868278621028243970633760768, 1/83076749736557242056487941267521536, 1/166153499473114484112975882535043072, 1/332306998946228968225951765070086144, 1/664613997892457936451903530140172288, 1/1329227995784915872903807060280344576, 1/2658455991569831745807614120560689152, 1/5316911983139663491615228241121378304, 1/10633823966279326983230456482242756608, 1/21267647932558653966460912964485513216, 1/42535295865117307932921825928971026432, 1/85070591730234615865843651857942052864, 1/170141183460469231731687303715884105728, 1/340282366920938463463374607431768211456, 1/680564733841876926926749214863536422912, 1/1361129467683753853853498429727072845824, 1/2722258935367507707706996859454145691648, 1/5444517870735015415413993718908291383296, 1/10889035741470030830827987437816582766592, 1/21778071482940061661655974875633165533184, 1/43556142965880123323311949751266331066368, 1/87112285931760246646623899502532662132736, 1/174224571863520493293247799005065324265472, 1/348449143727040986586495598010130648530944, 1/696898287454081973172991196020261297061888, 1/1393796574908163946345982392040522594123776, 1/2787593149816327892691964784081045188247552, 1/5575186299632655785383929568162090376495104, 1/11150372599265311570767859136324180752990208, 1/22300745198530623141535718272648361505980416, 1/44601490397061246283071436545296723011960832, 1/89202980794122492566142873090593446023921664, 1/178405961588244985132285746181186892047843328, 1/35681192317648997026457149

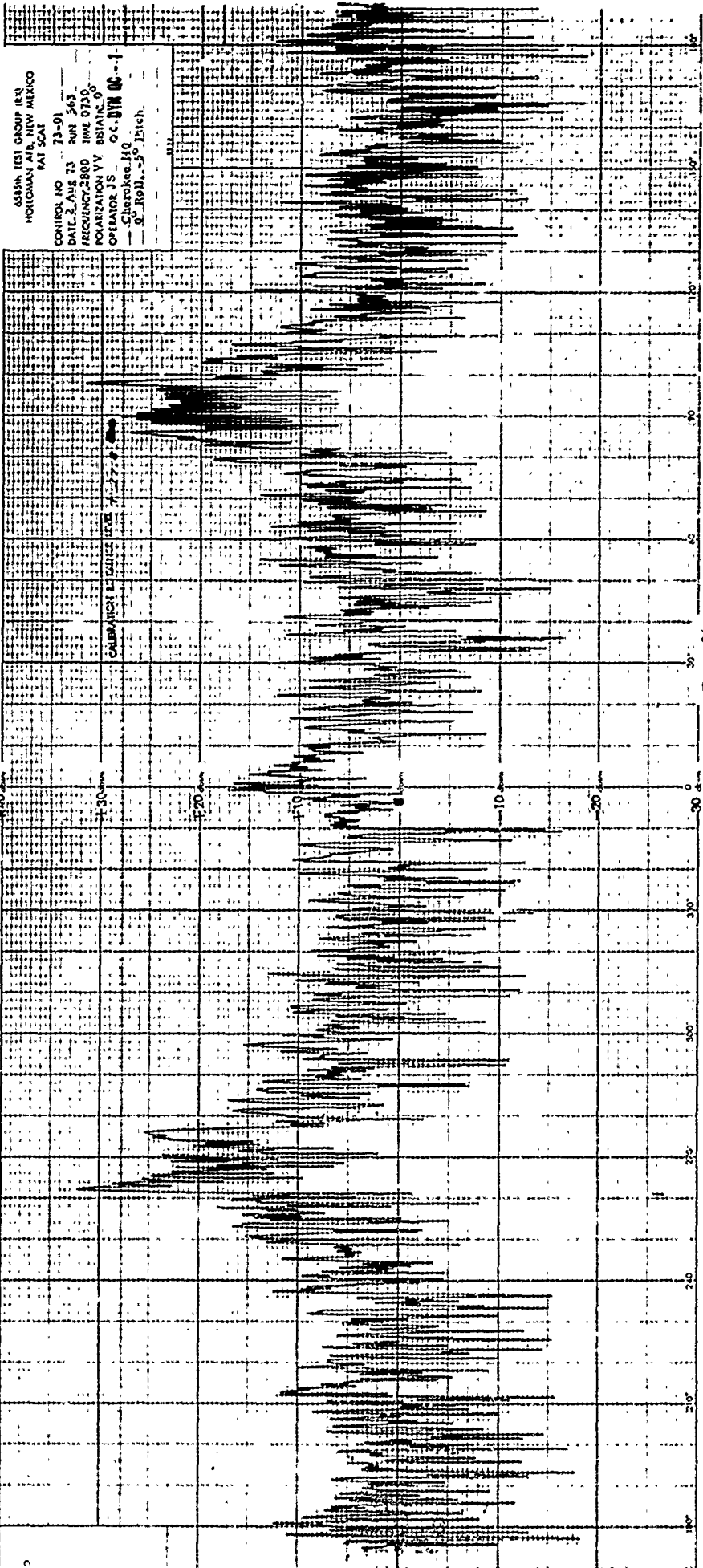


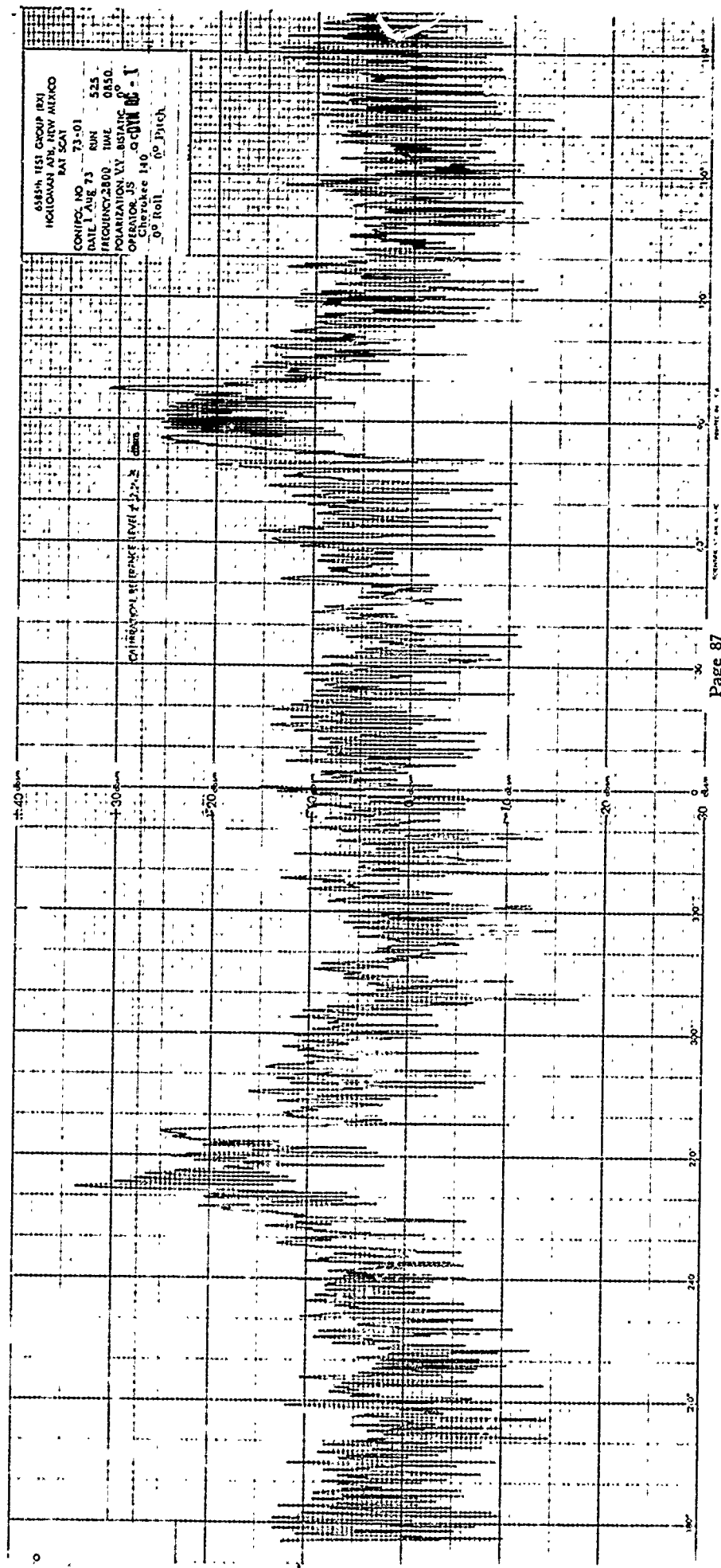




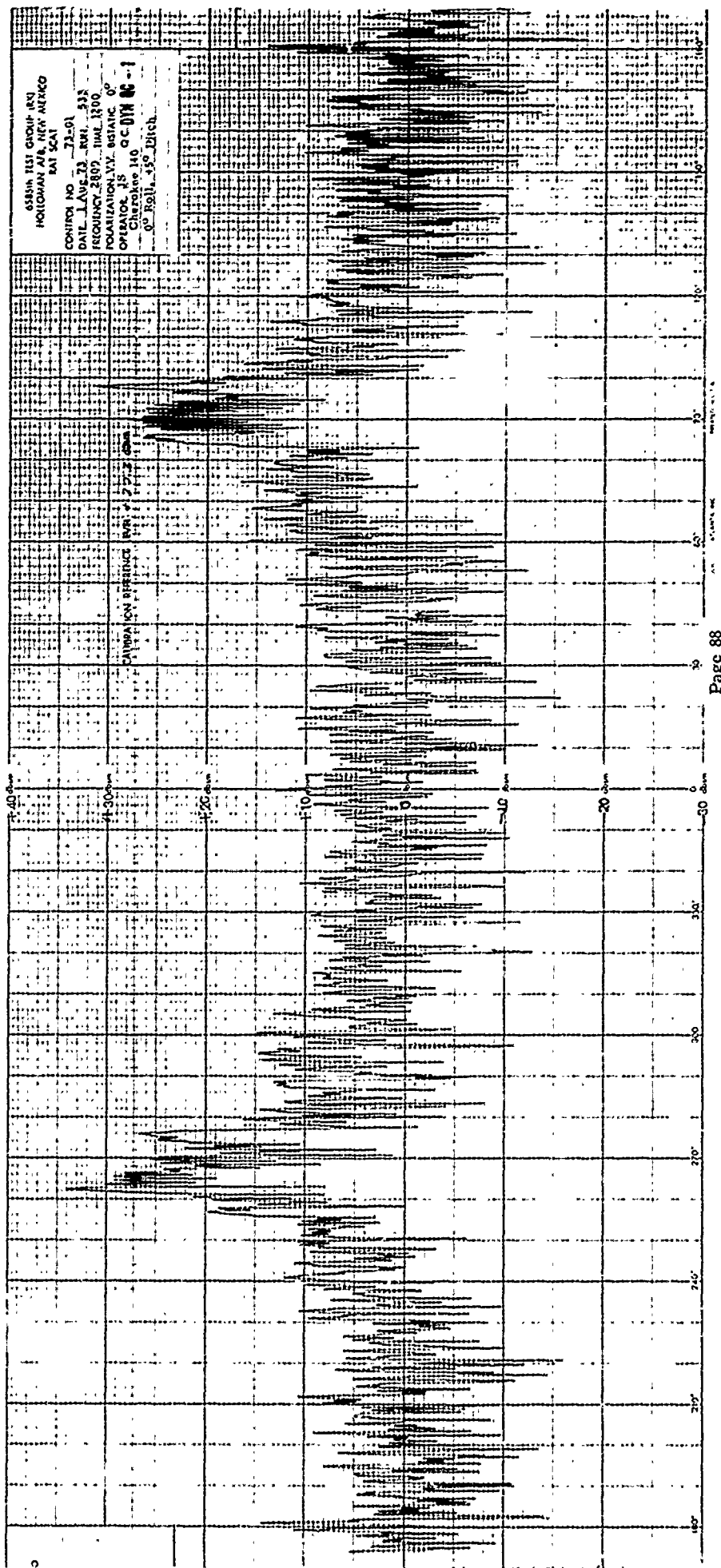


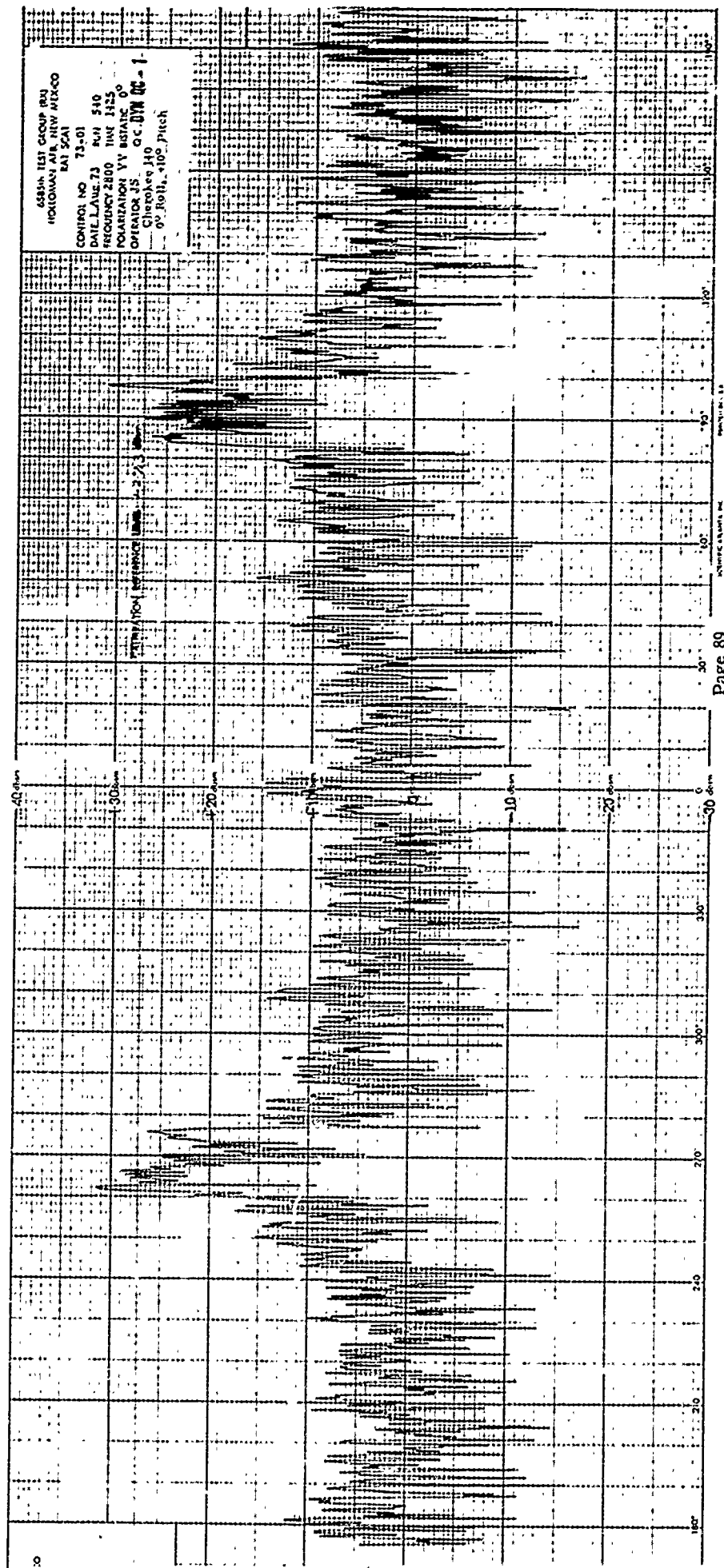




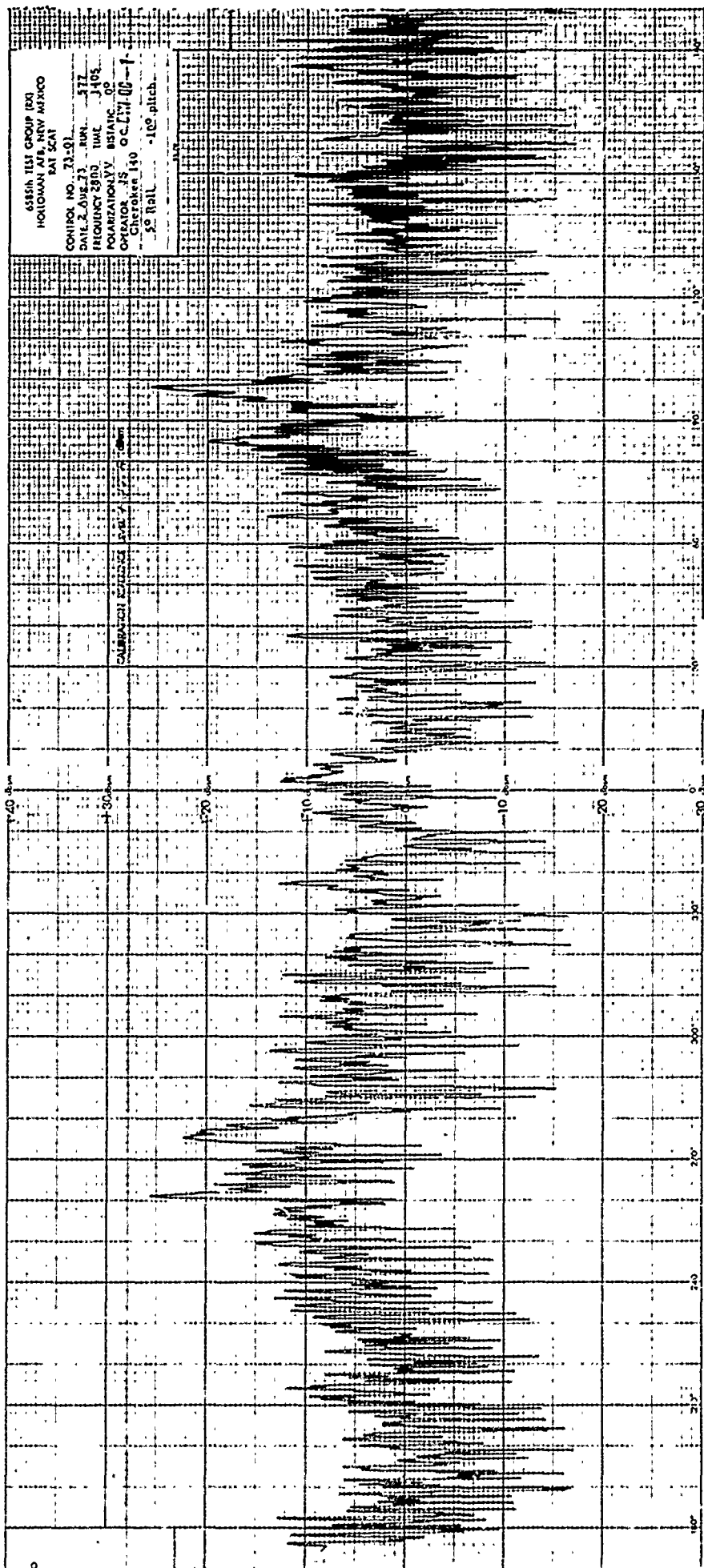


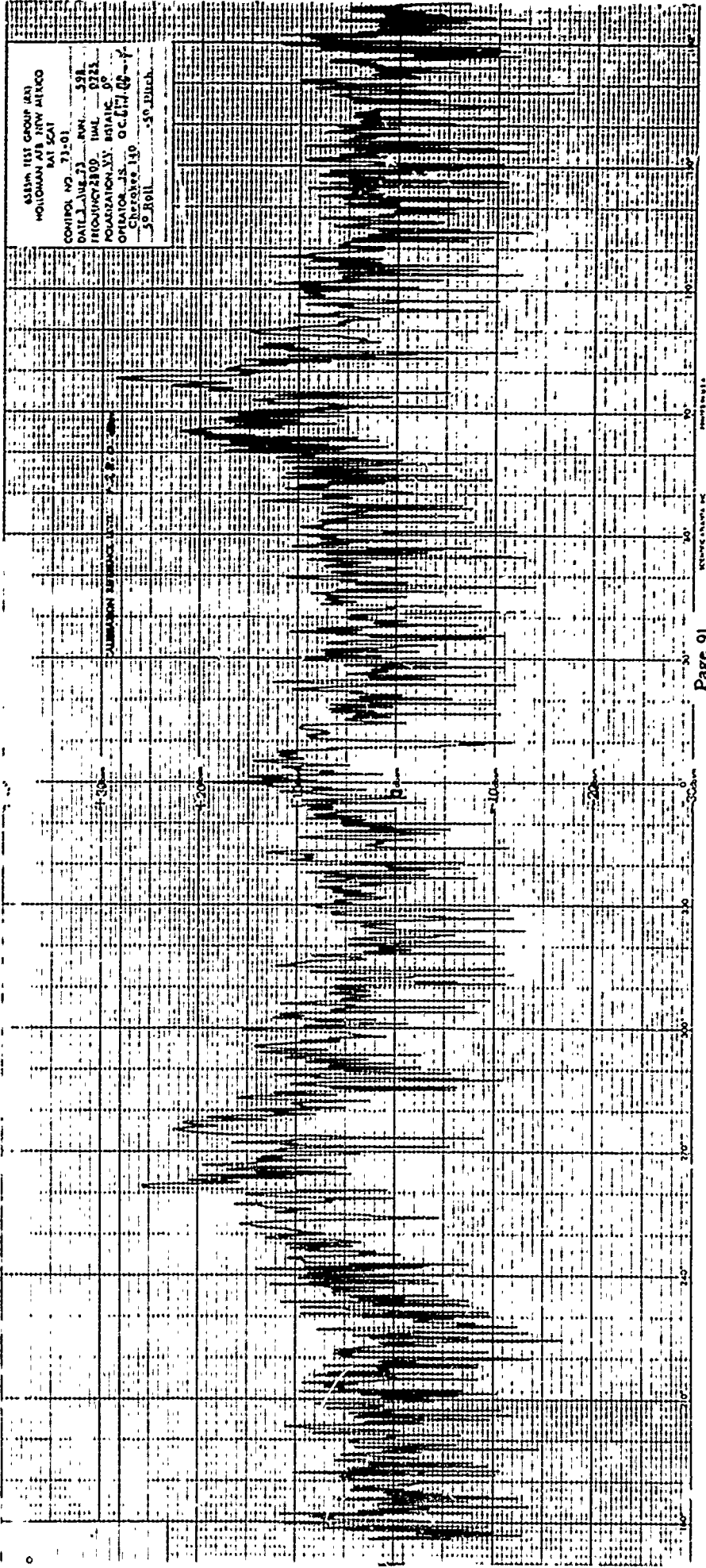
6385th TEST GROUP (B1)
HOLLAMAN AFB, NEW MEXICO
BAT SCAT
CONTRACT NO. 73-01
DATE 1 Aug 73 RUN 525
INQUIRY 2800 TIME 0850
POLARIZATION VY. B51AIC
OPERATOR JS
Channel 140
9° Roll 9° Pitch

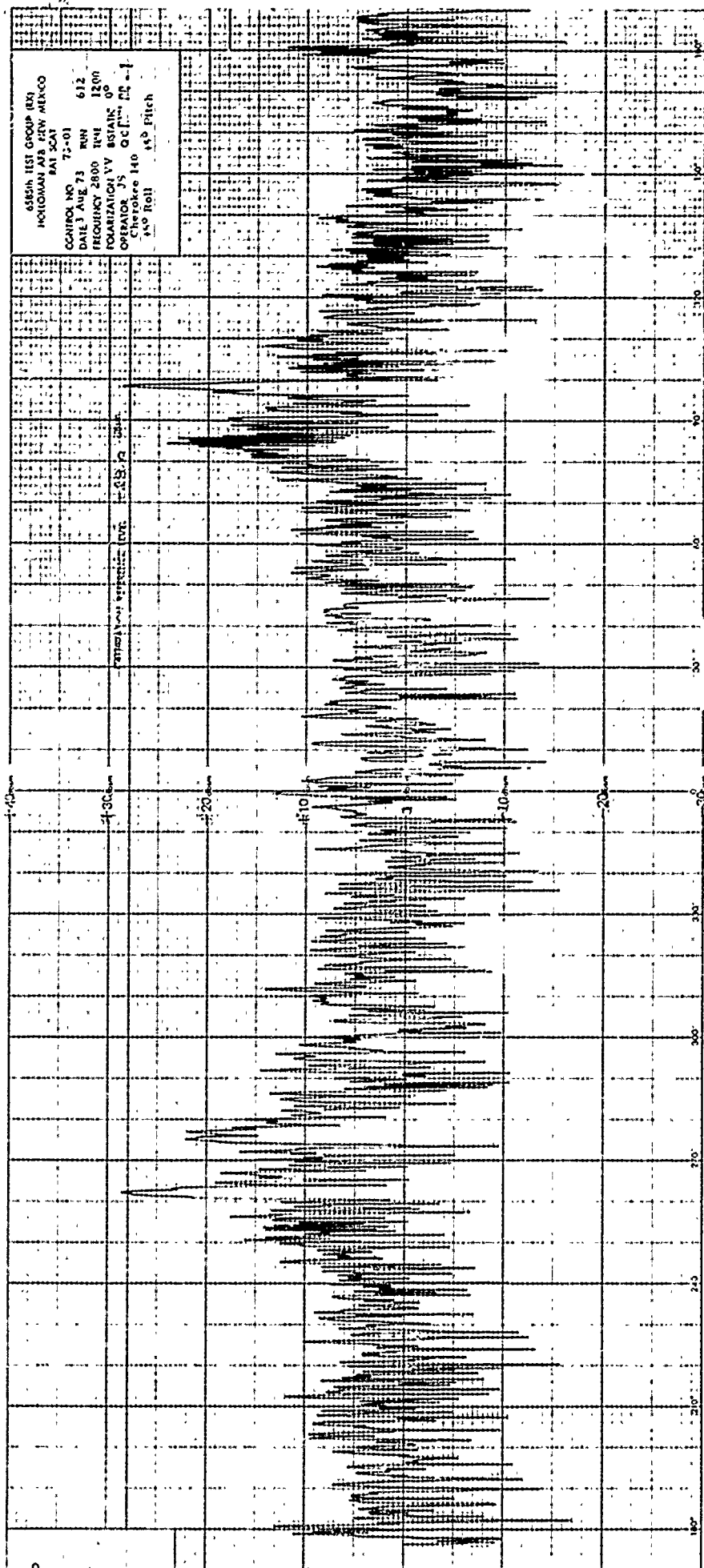


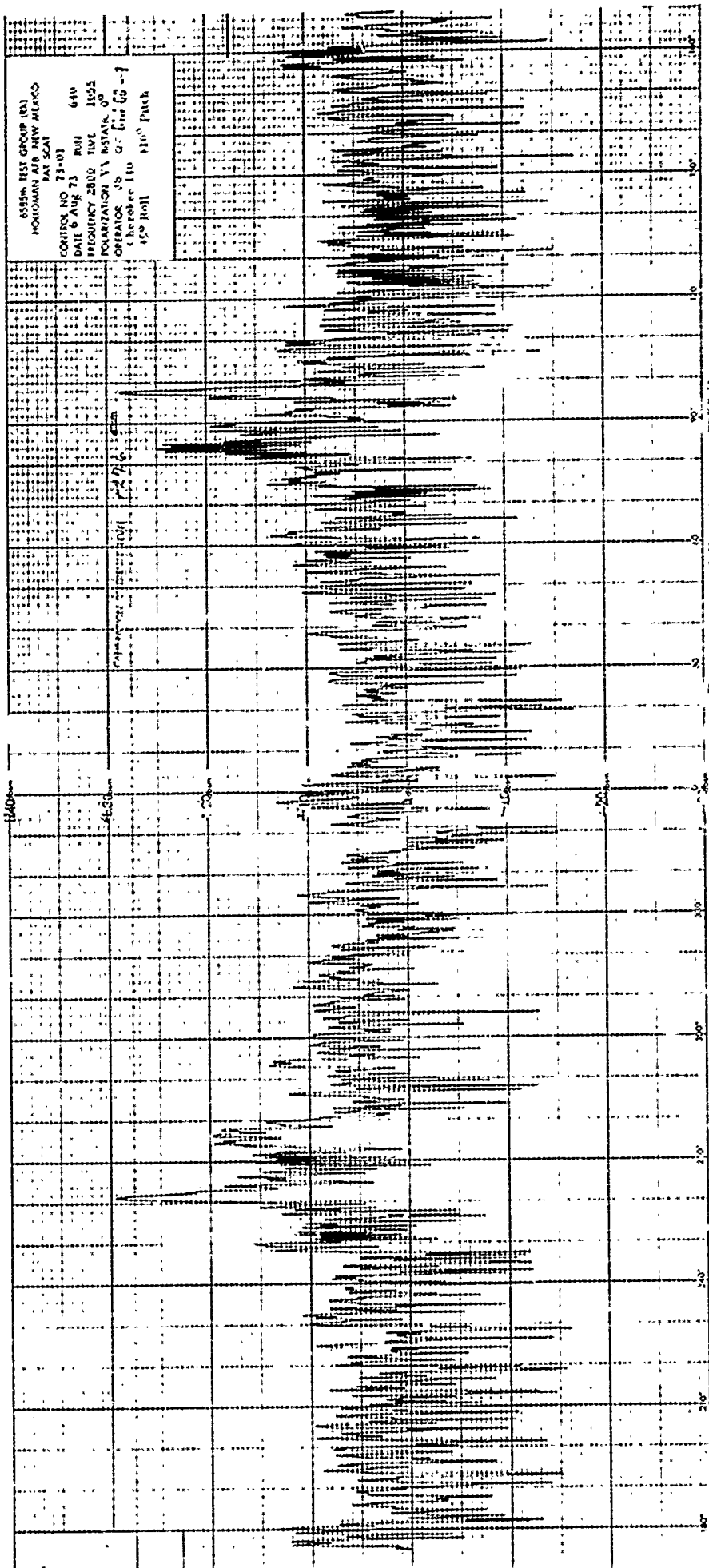


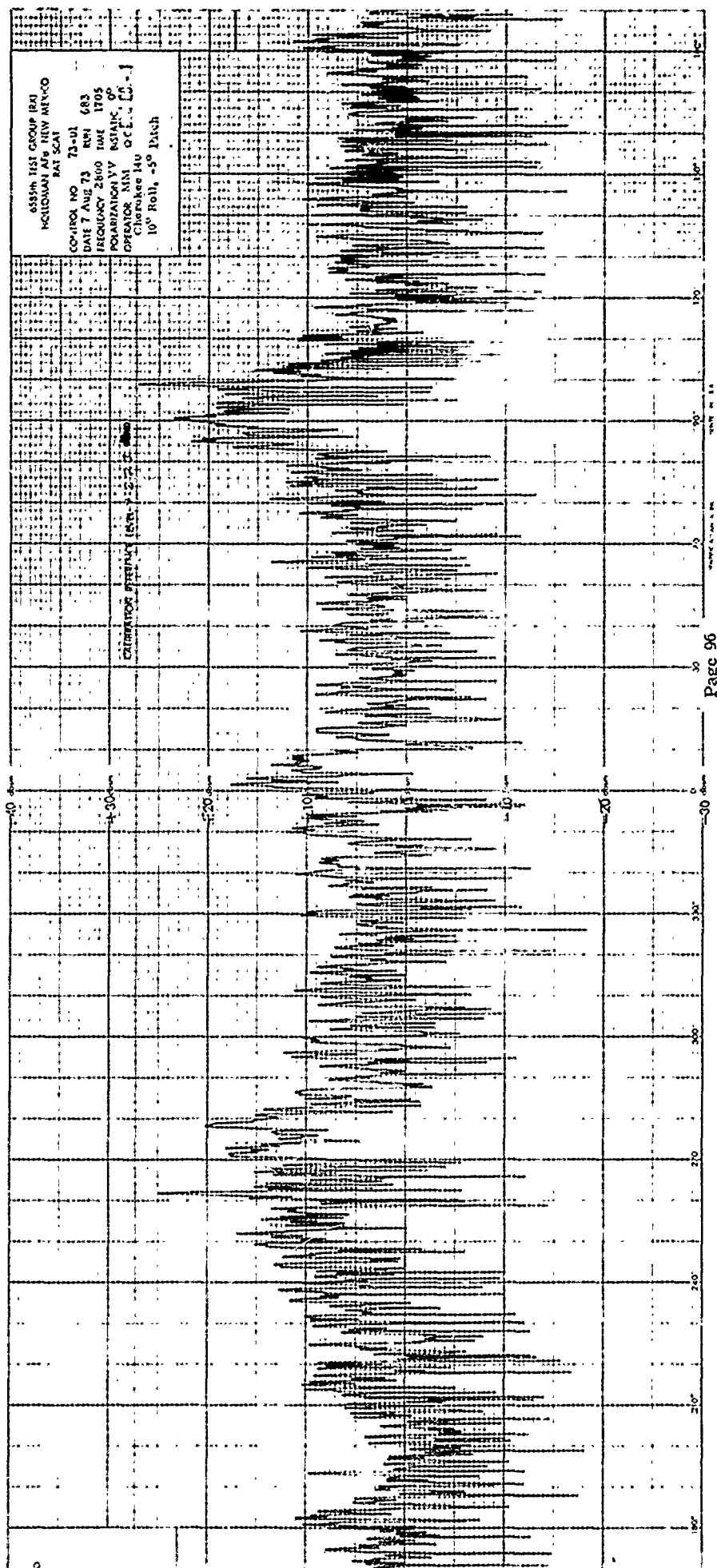
6854A TEST GROUP RU
HOLLOMAN AFB, NEW MEXICO
EAT SCAN
CONTROL NO 73-01
DATE LAUG 73 RUN 510
FREQUENCY 2800 HZ 1155
POLARIZATION VV STATIC 0°
OPERATOR JS OC DYN MC-1
Chertok 140
0° Roll, 10° Pitch

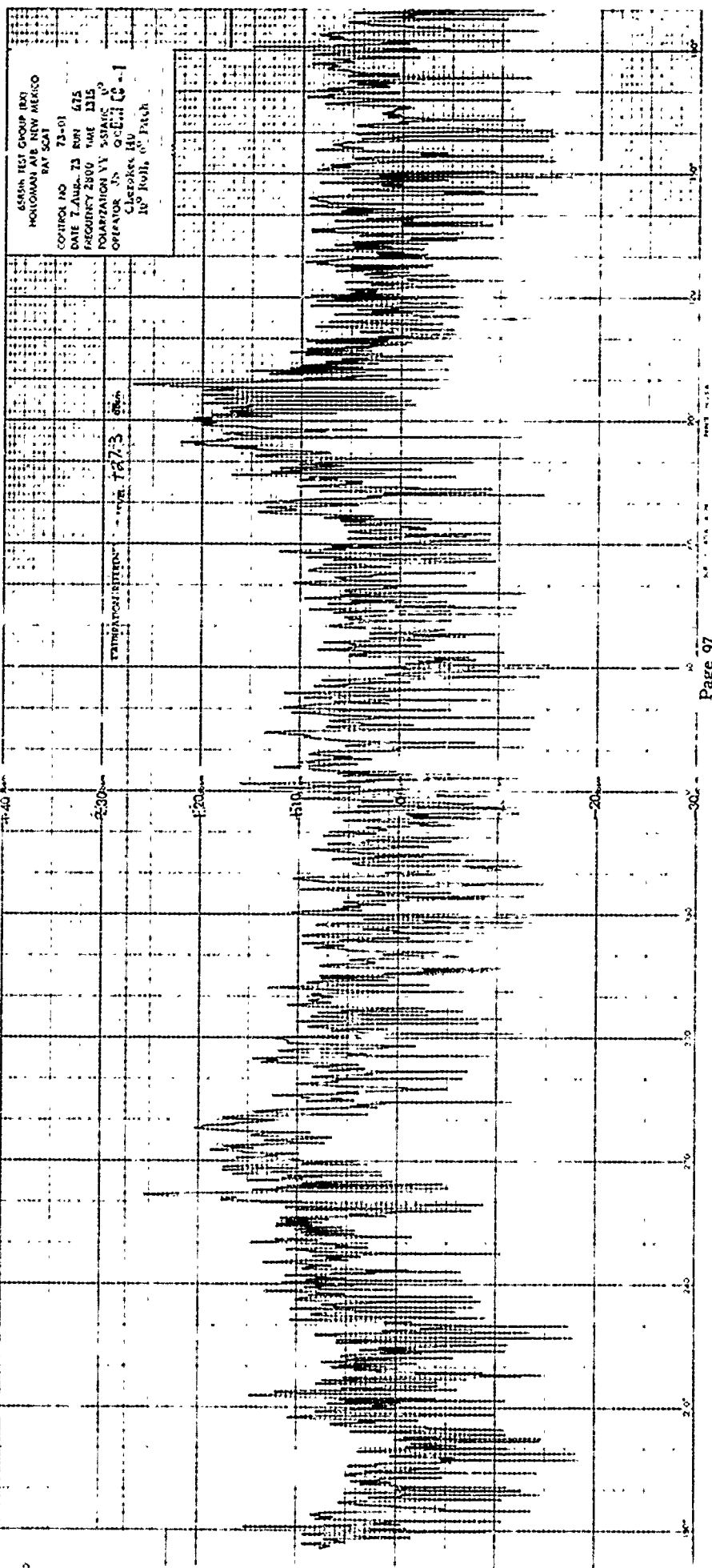


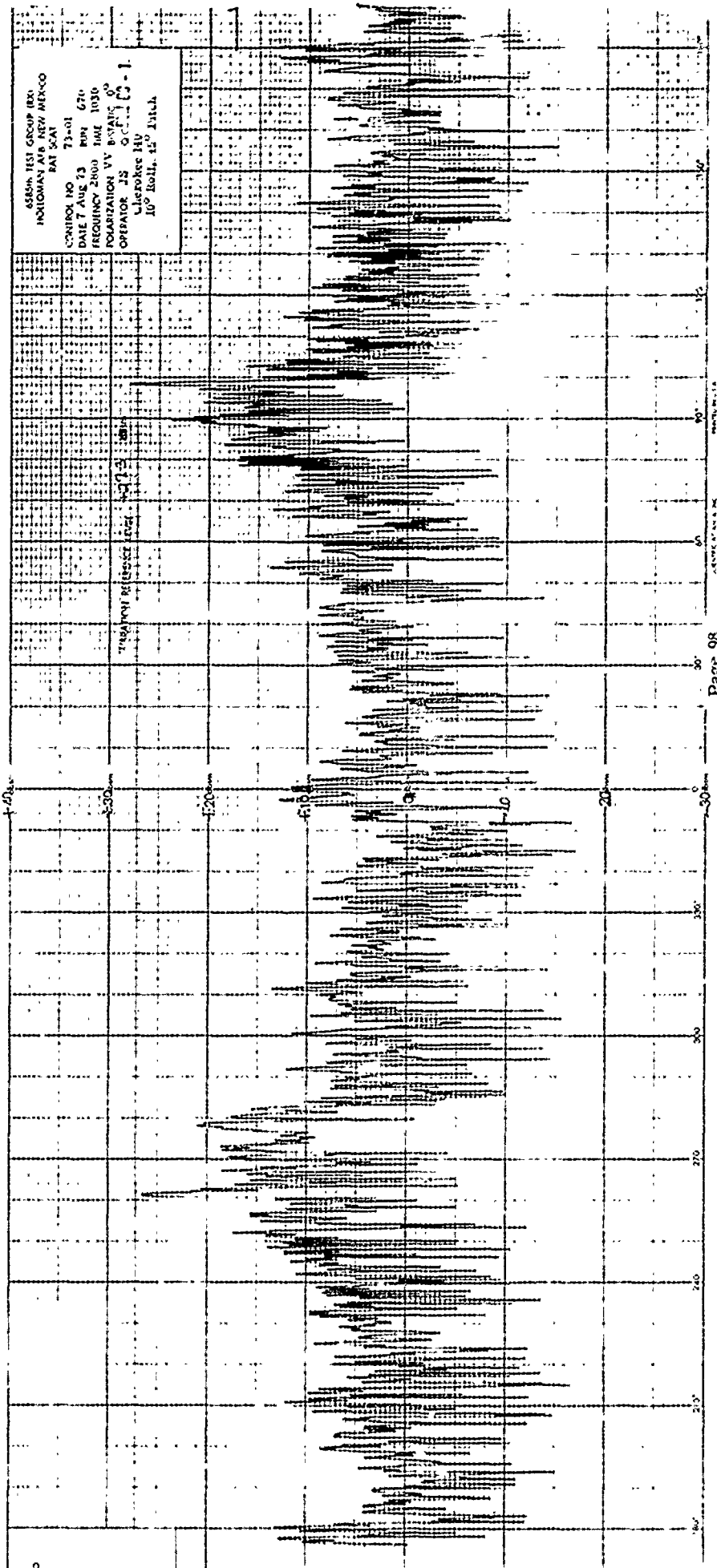


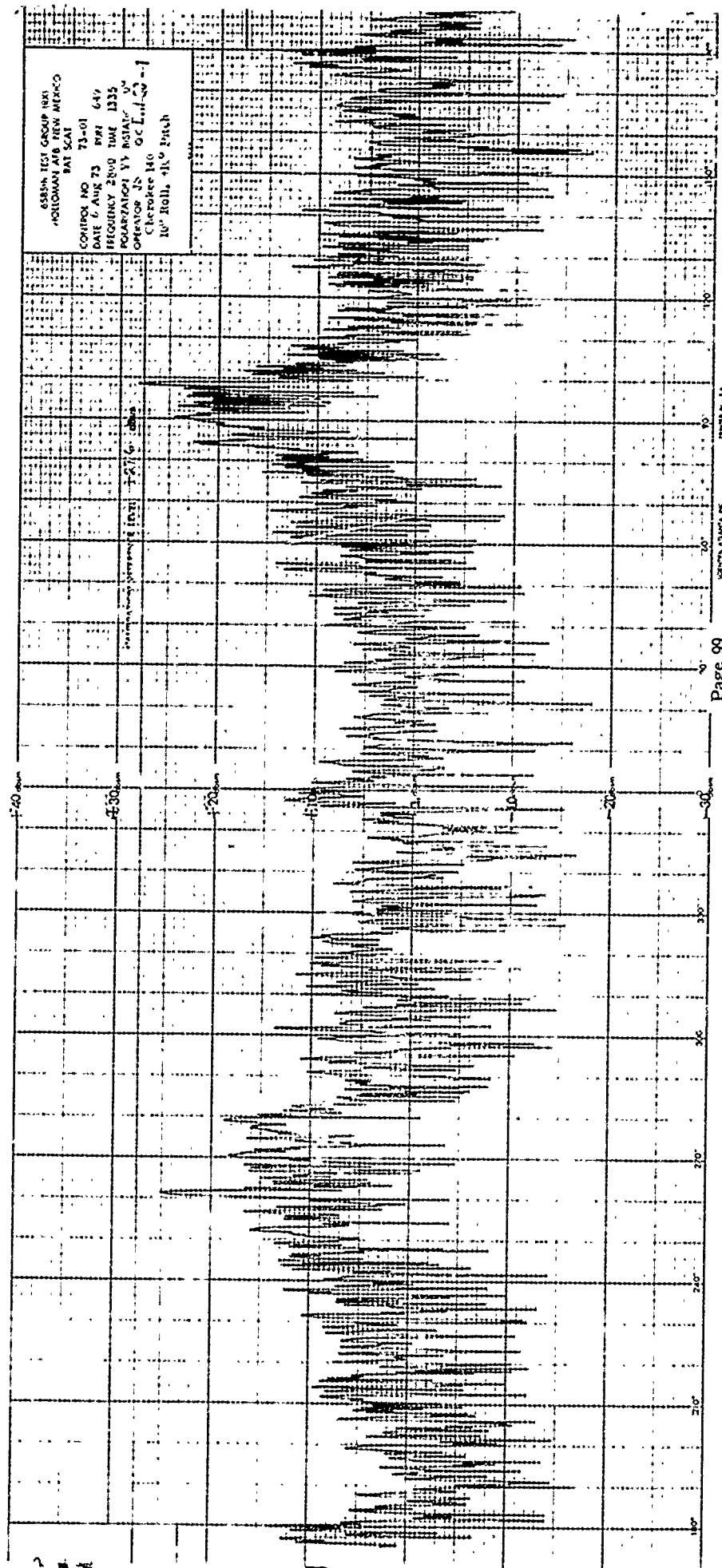




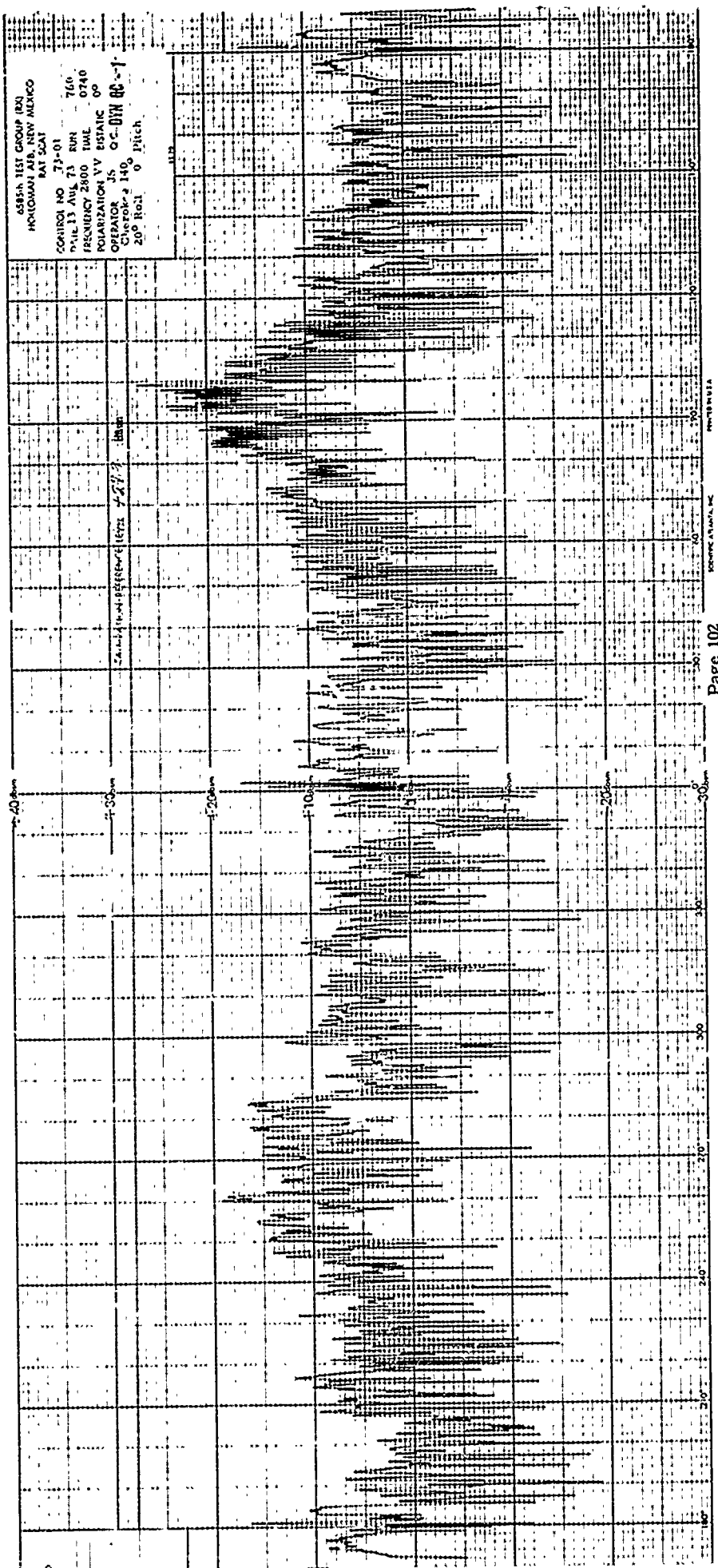


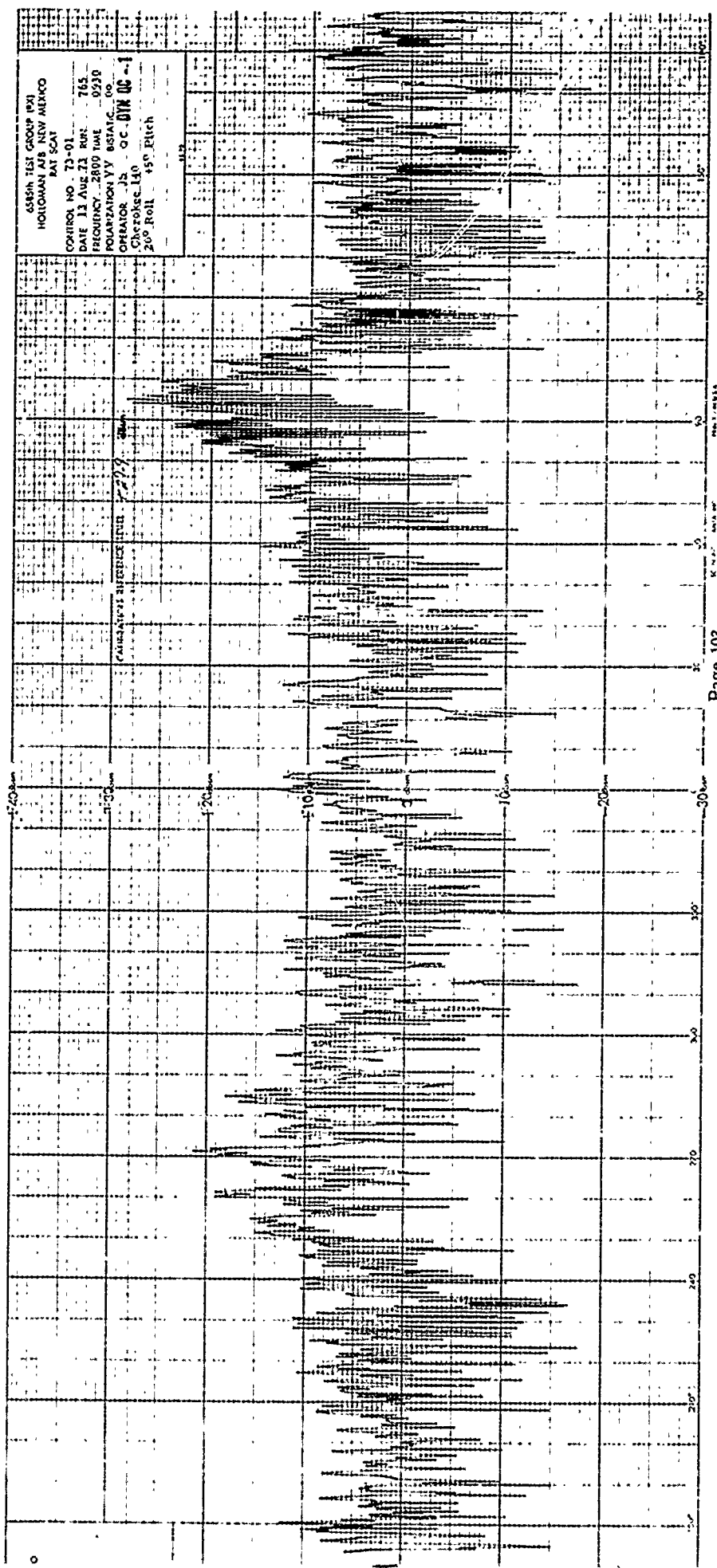


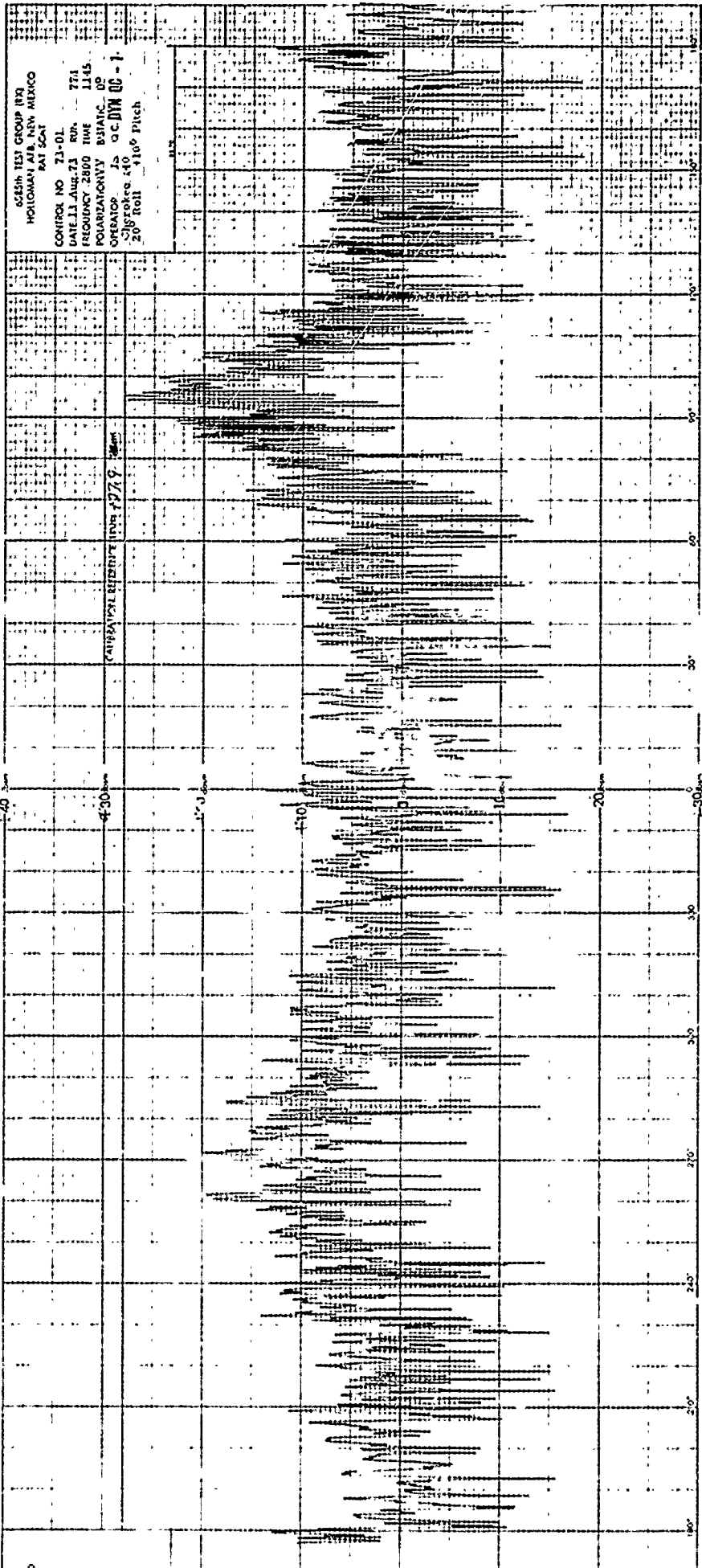




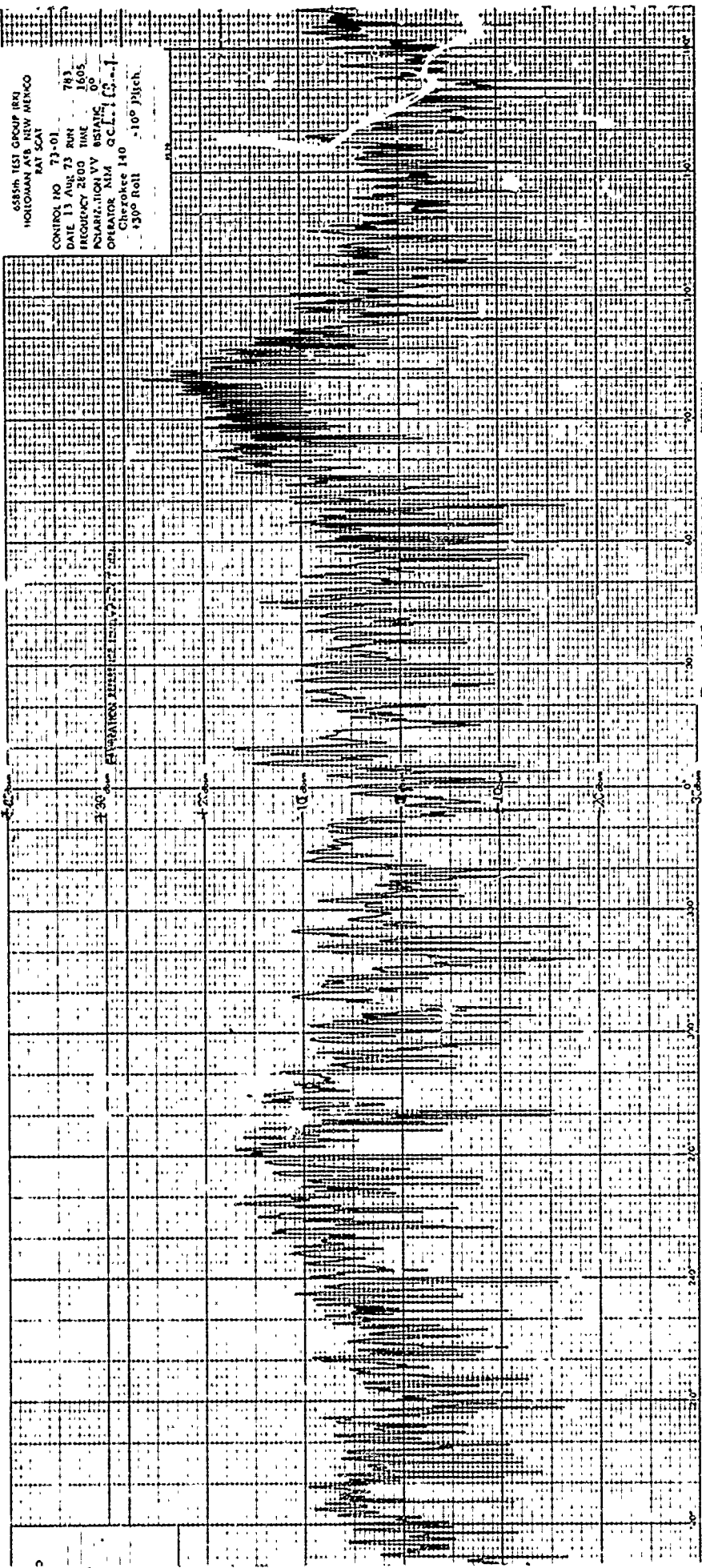
[illegible]

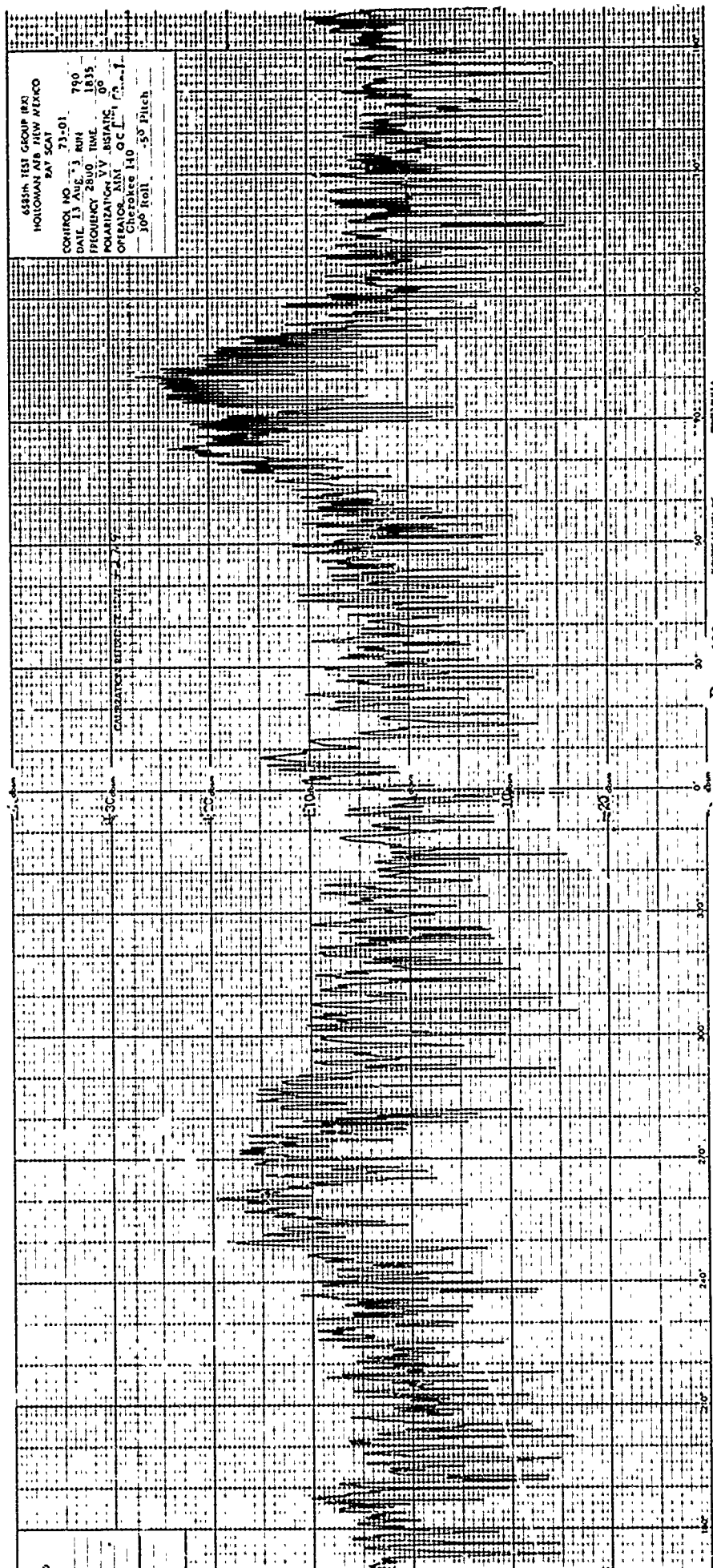


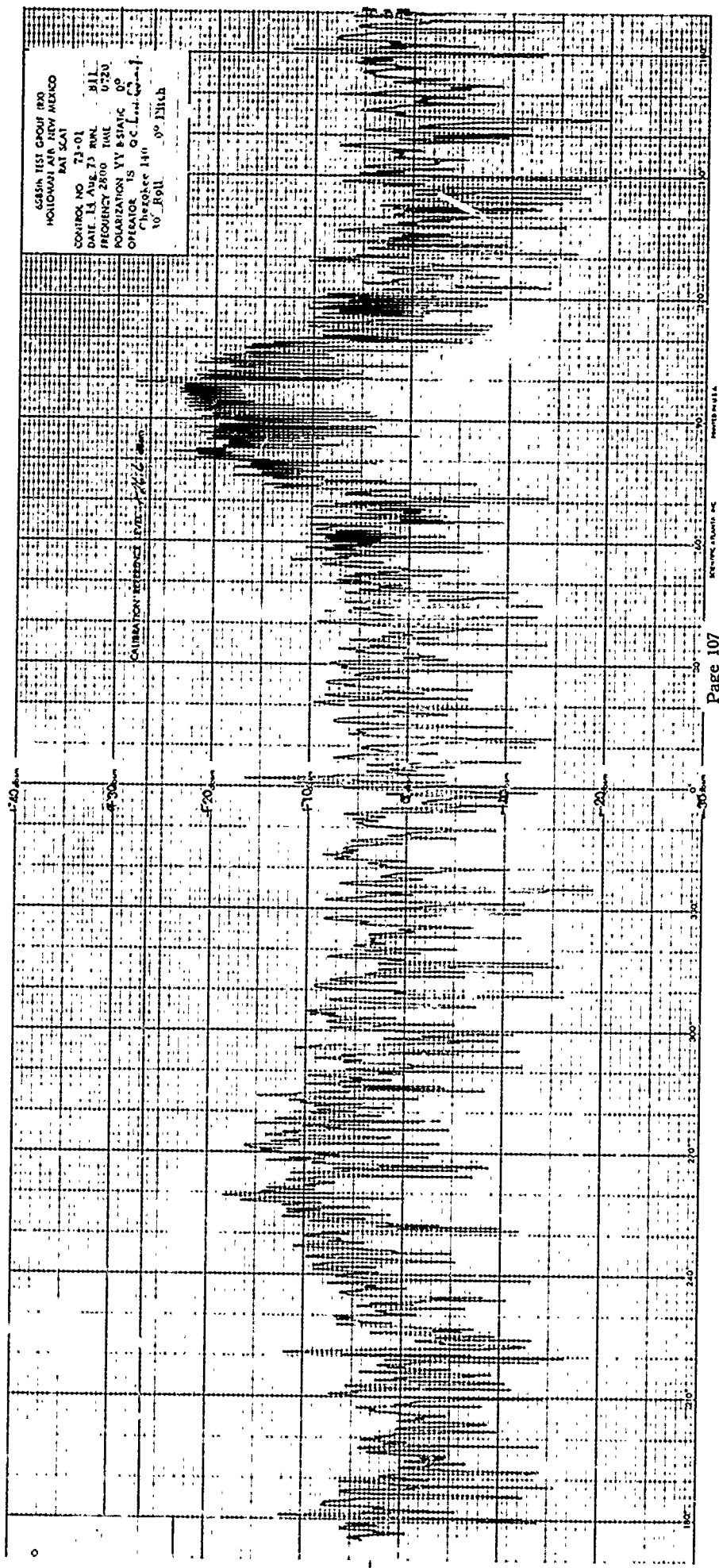




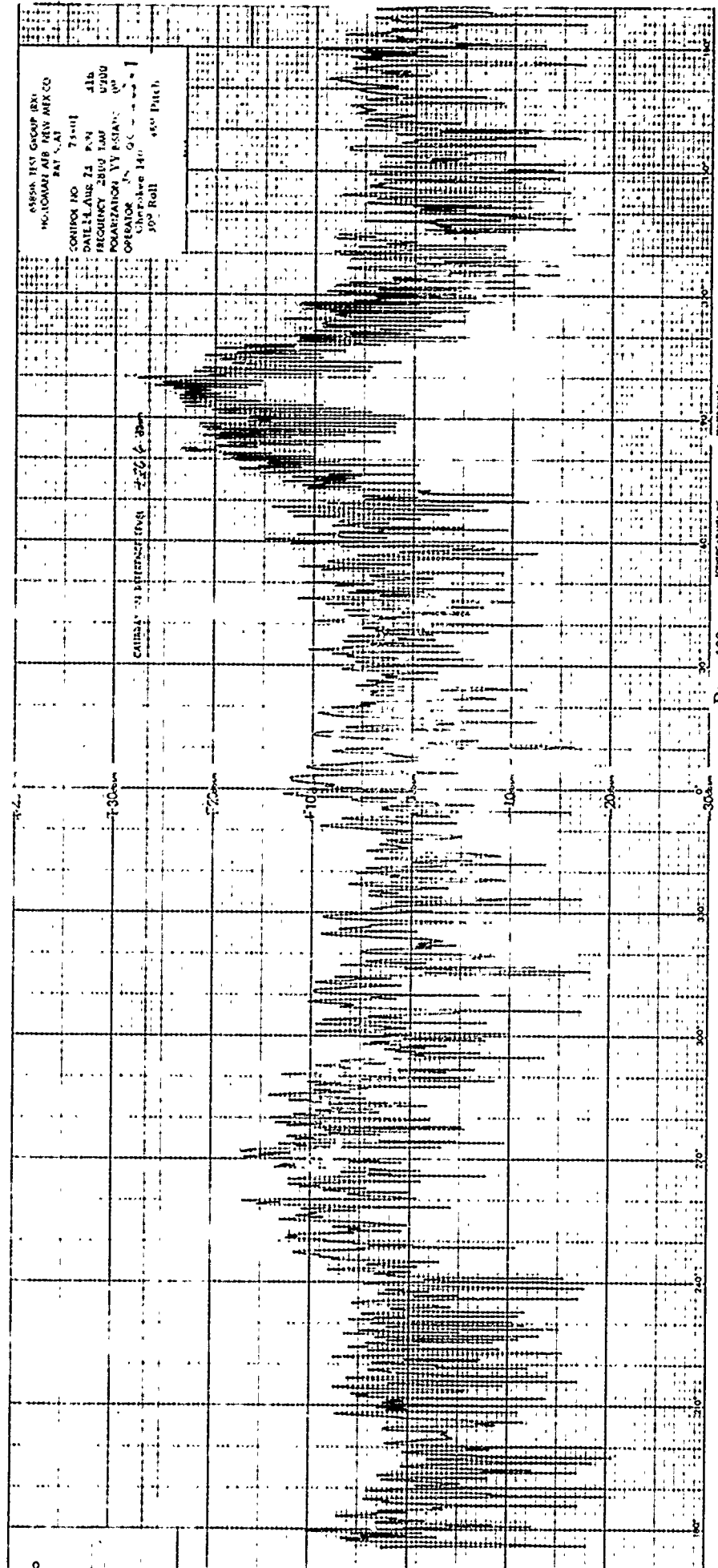
6585th TEST GROUP (RTJ)
 HORTONMAN ATB NEW MEXICO
 DAT SCAT
 CONTROL NO 73-01 783
 DATE 13 AUG 73 RUN 1605
 FREQUENCY 2600 TIME 1605
 POLARIZATION VV BR1415 00
 OPERATOR NIM GCL 15-1
 Cherokee 140
 +30° Roll +10° Pitch

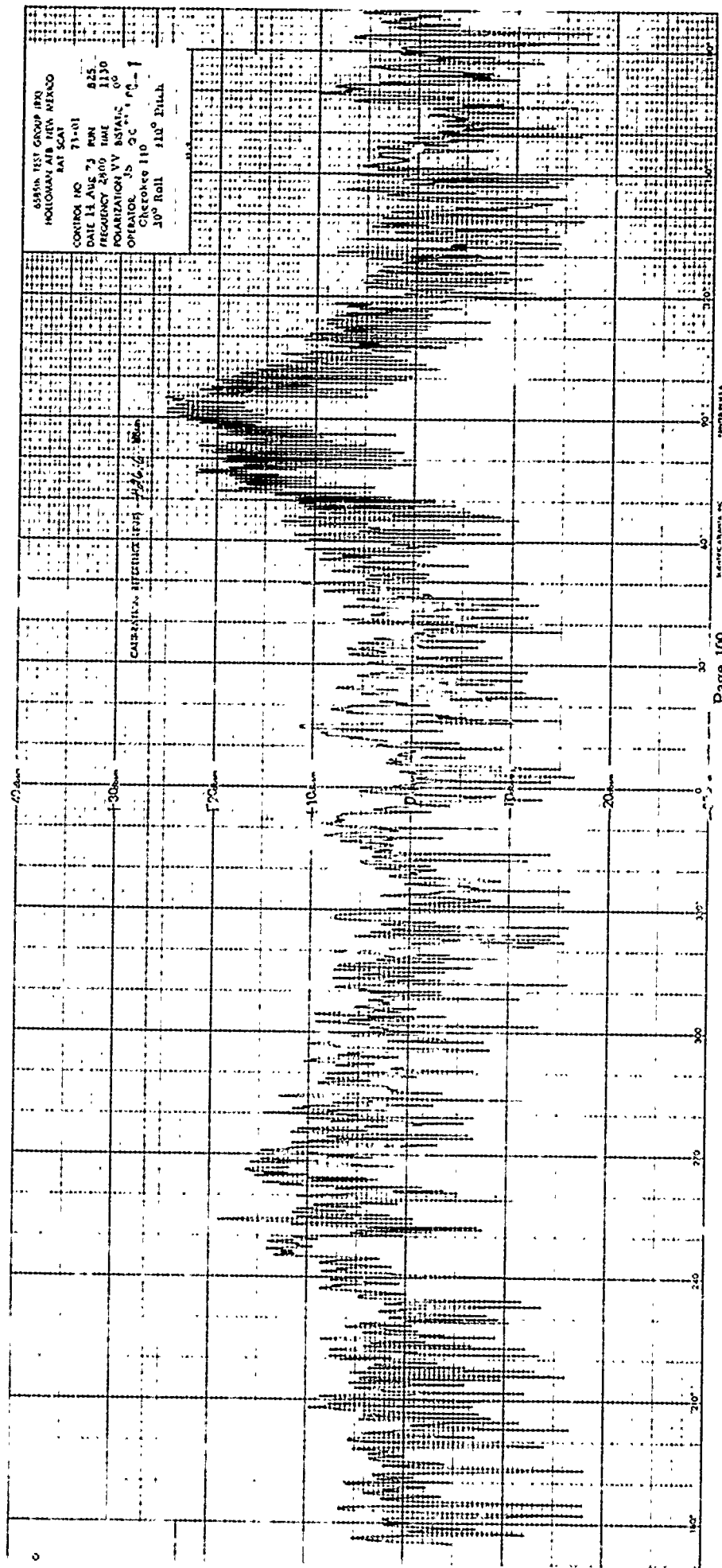


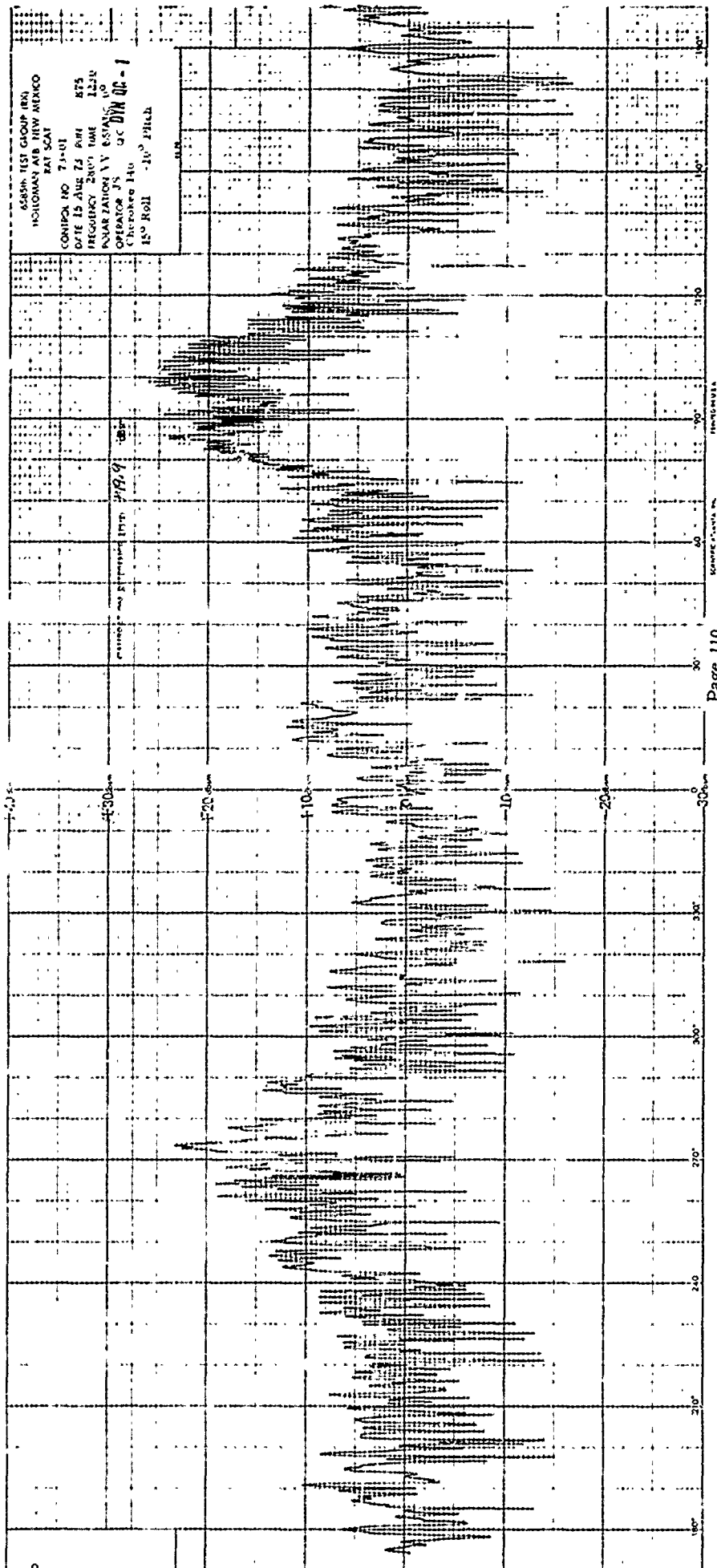


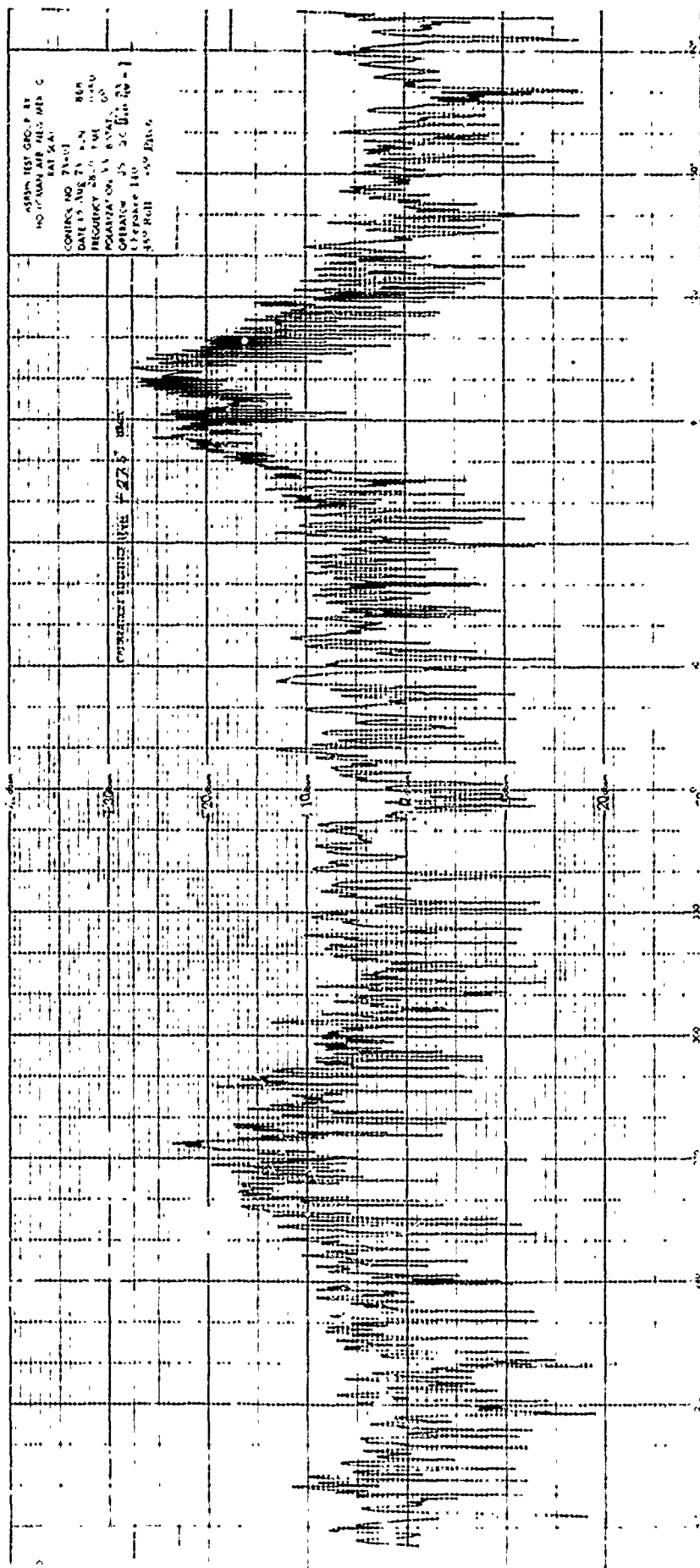


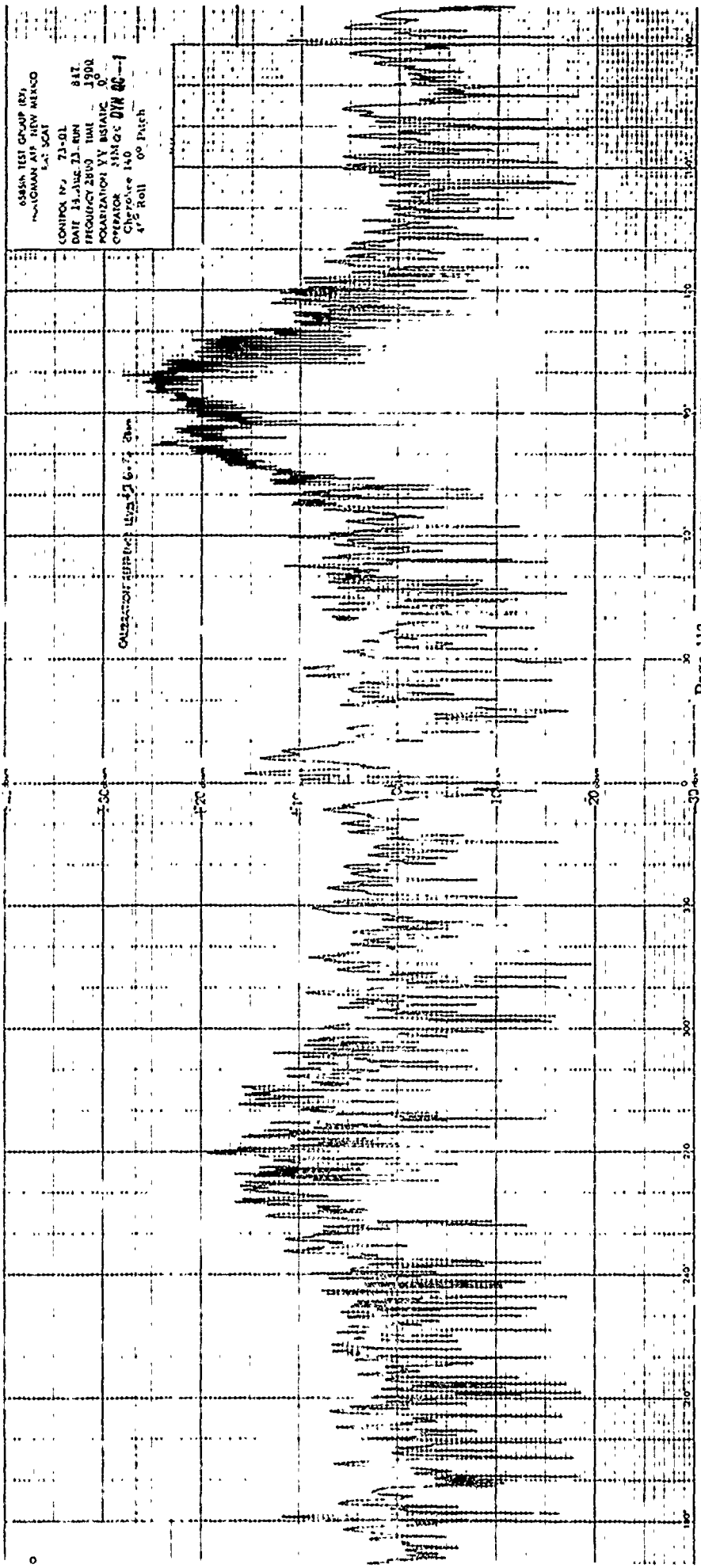
GAUSS TEST GROUP 100
HOLLAND AIR NEW MEXICO
RAT SCAT
CONTROL NO 71-01
DATE 14 AUG 71 RNL 911
FREQUENCY 2800 MHz 0.220
POLARIZATION VV STATIC 0°
OPERATOR IS GC L. L. Conf
Cherokee 141 0° Pitch
10 Roll

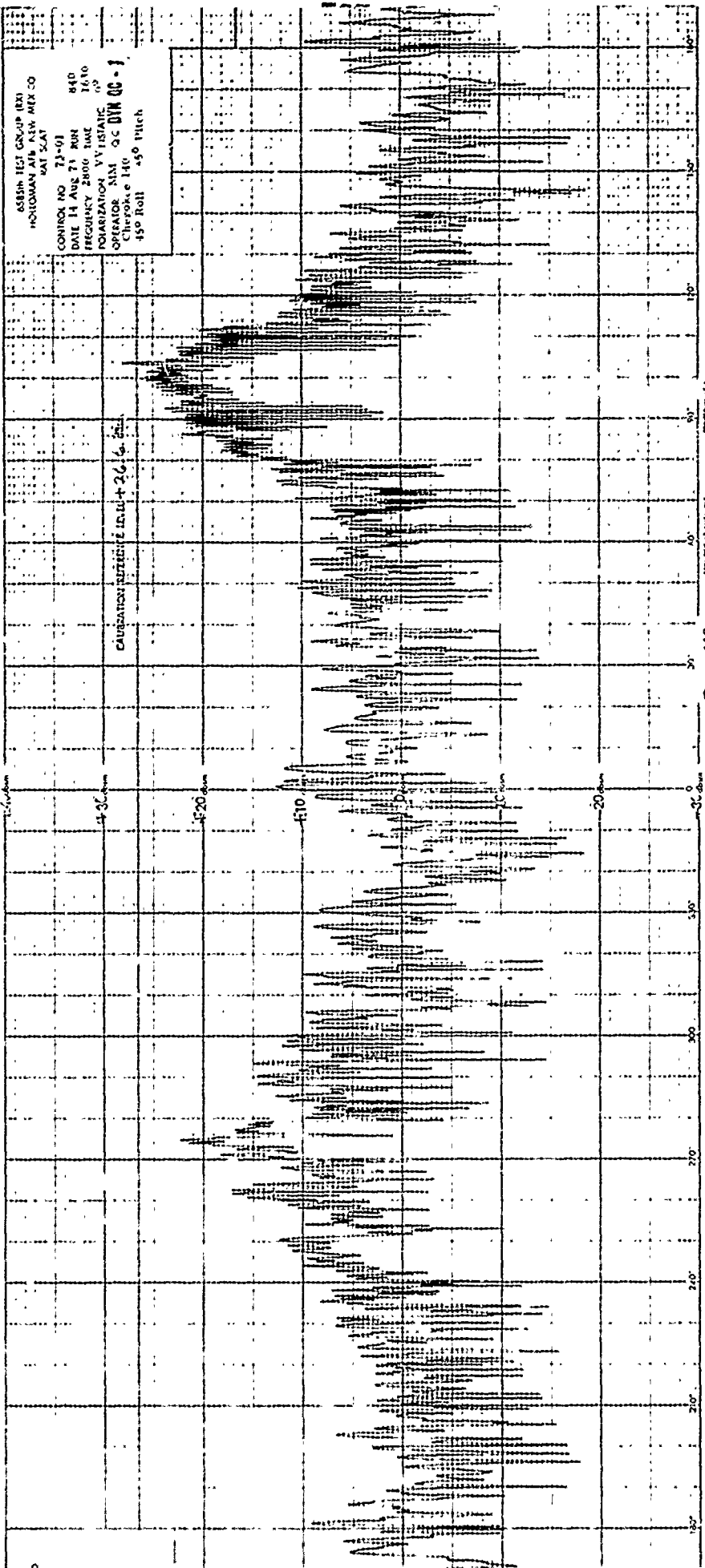


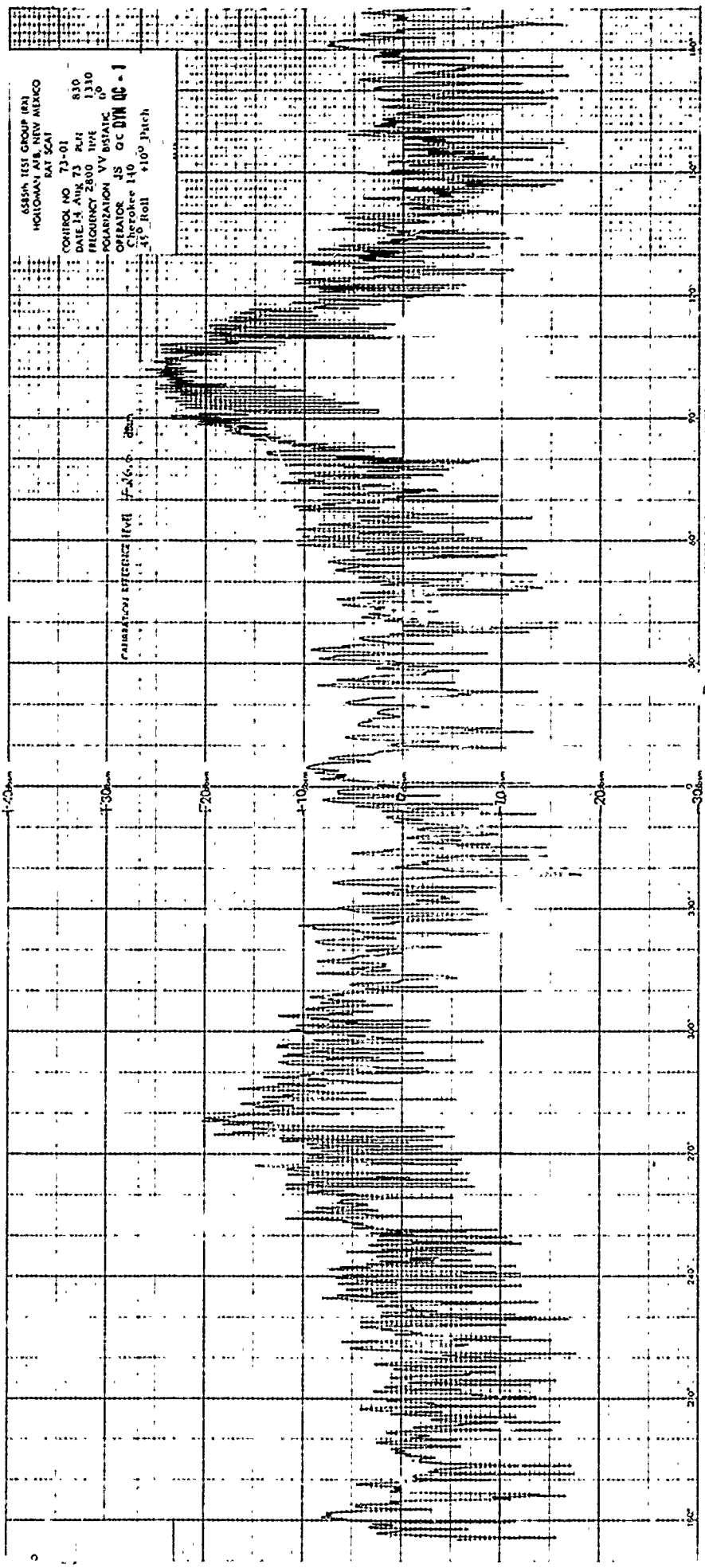


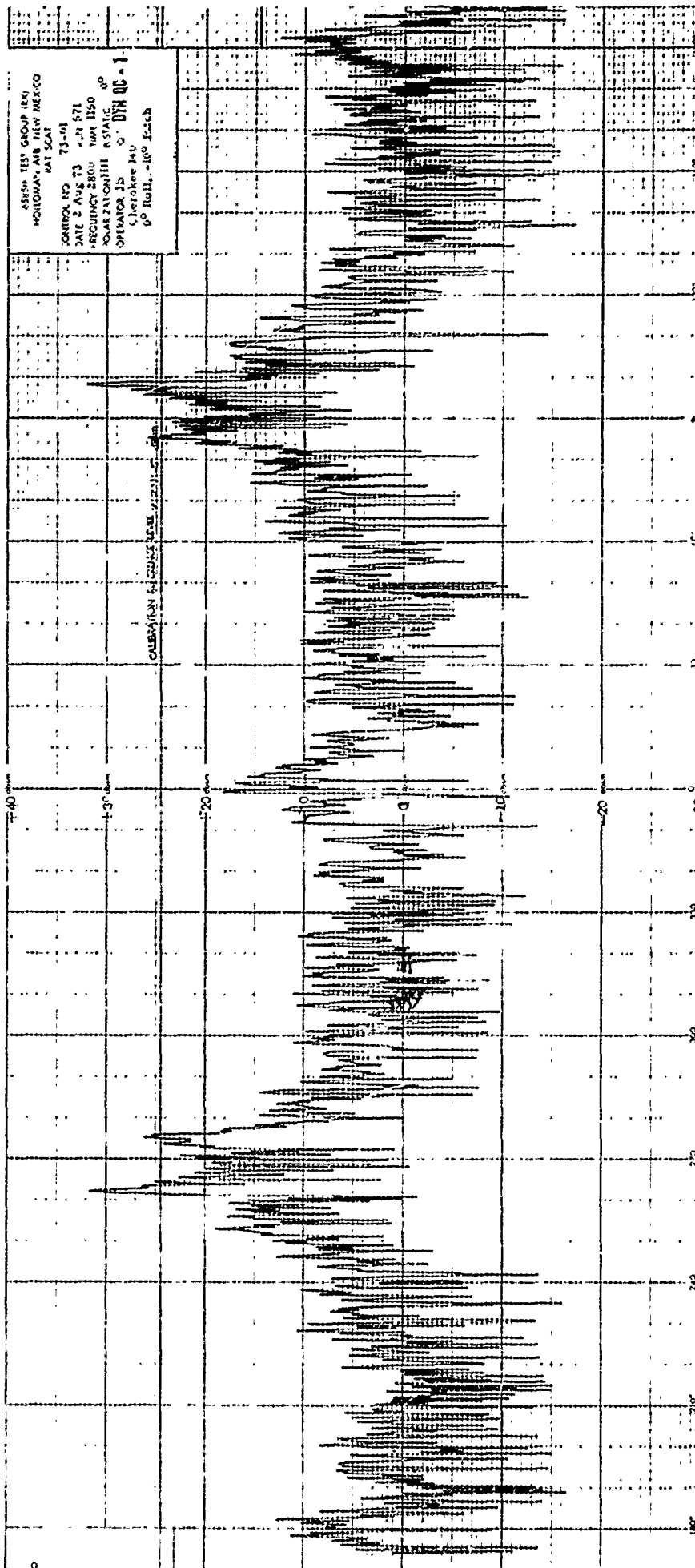


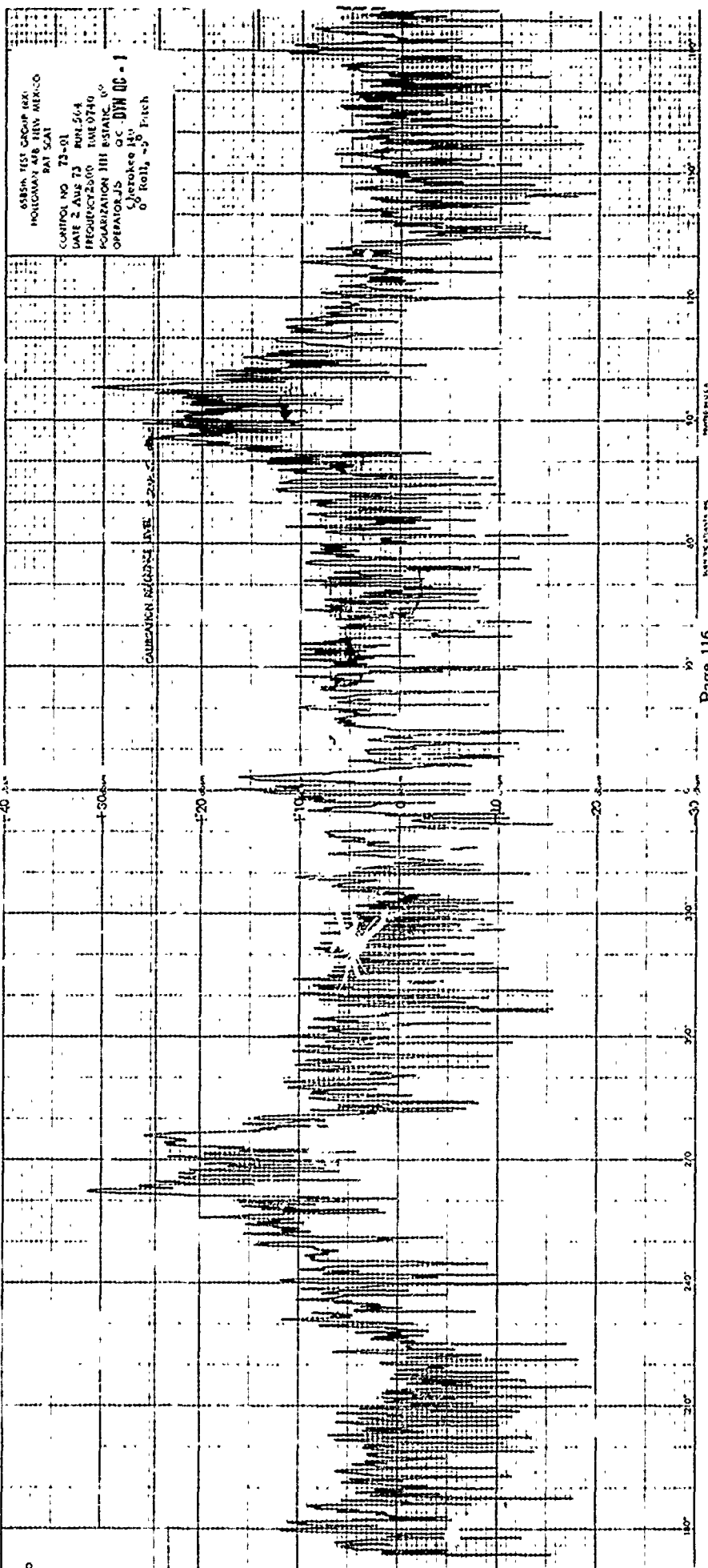












15544 TEST GROUP (23)
HOLTHAM AIR NEW ALKCO

BAT CAT

CONTROL NO 73-01

DATE 1 AUG 73

TIME 1426

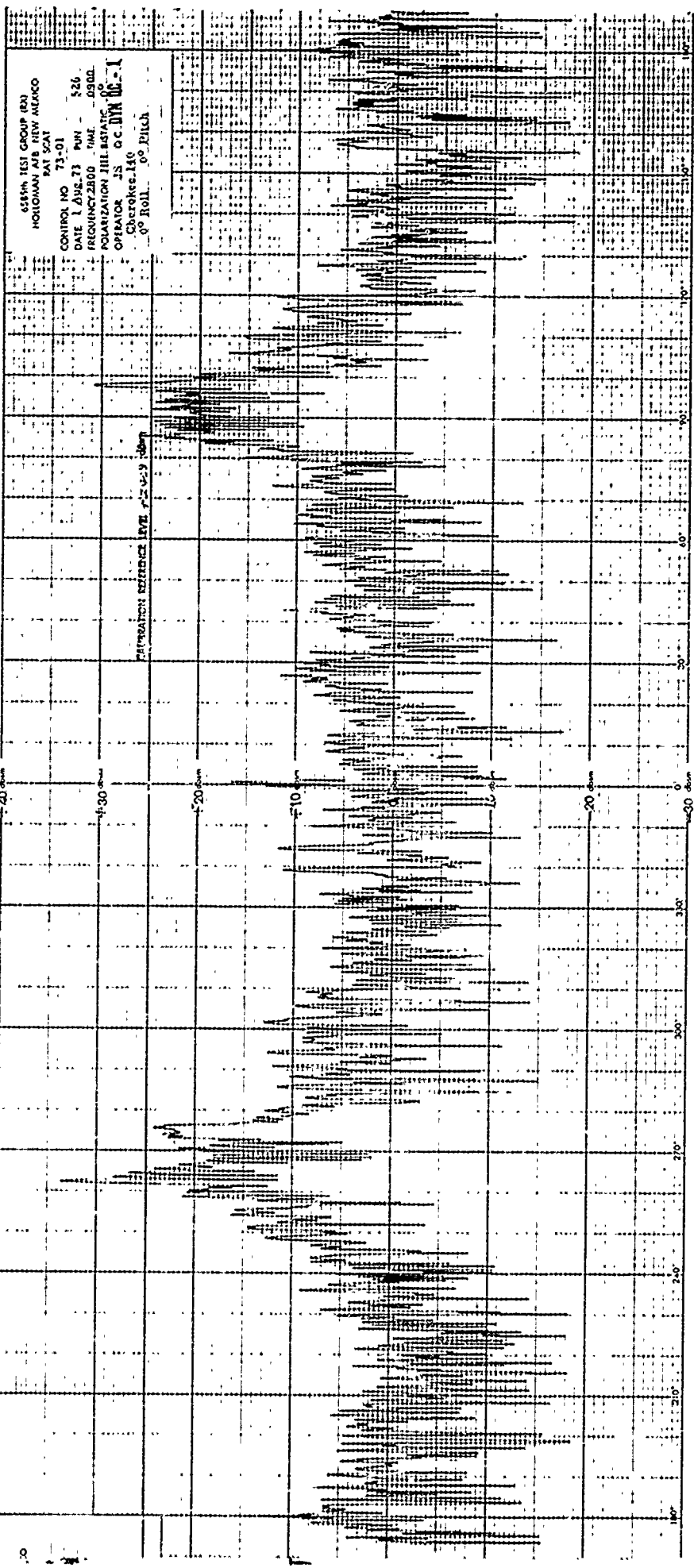
FREQUENCY 2800

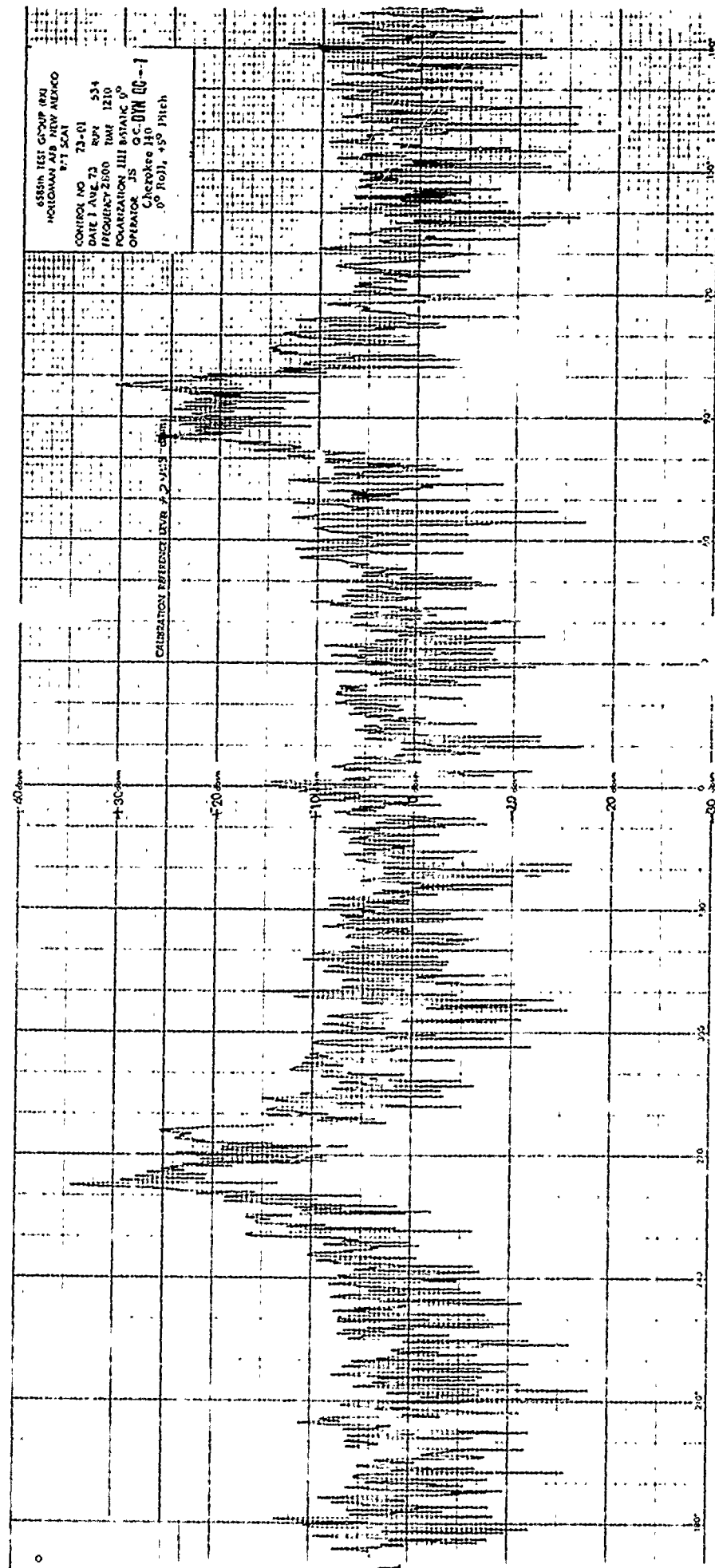
POLARIZATION JHL ESTATE

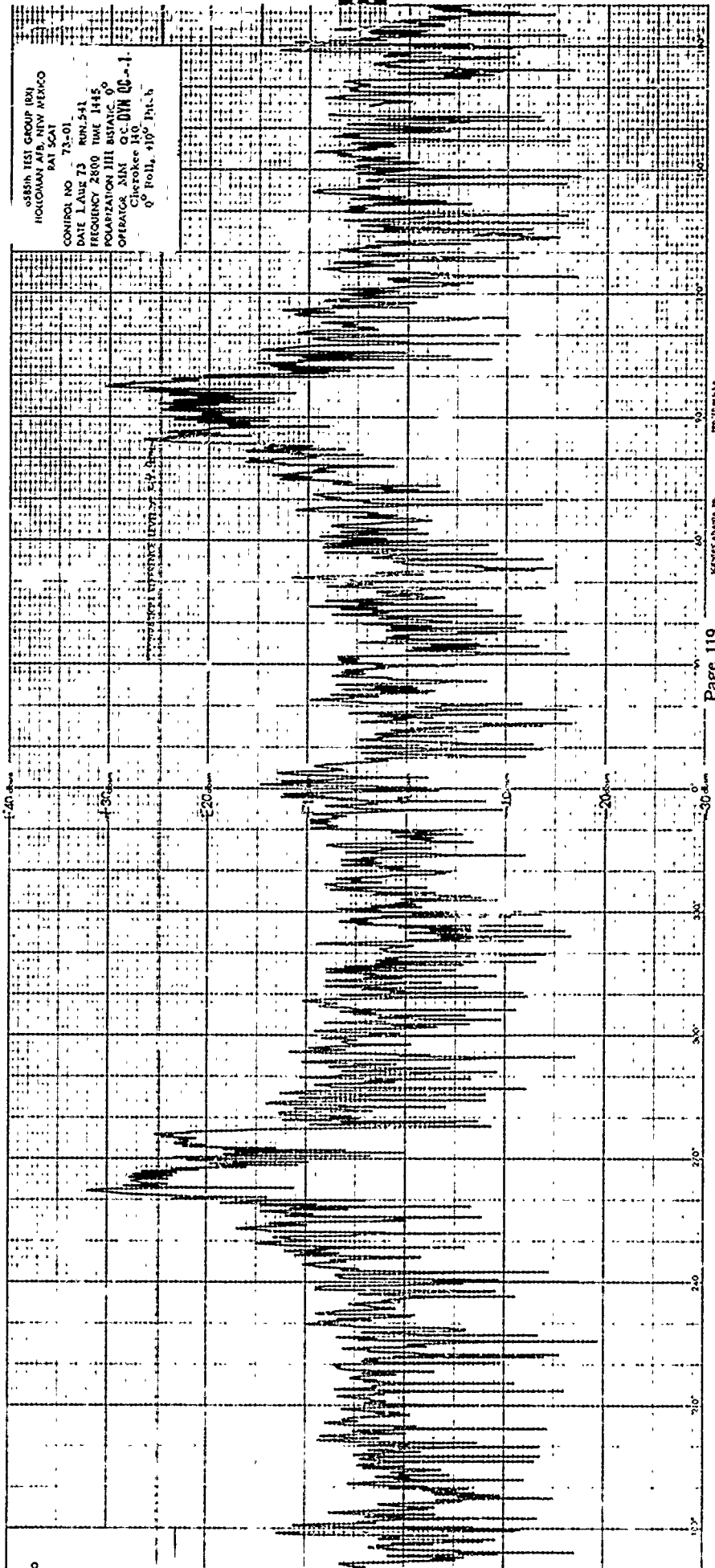
OPERATOR JS GC DRA DE-1

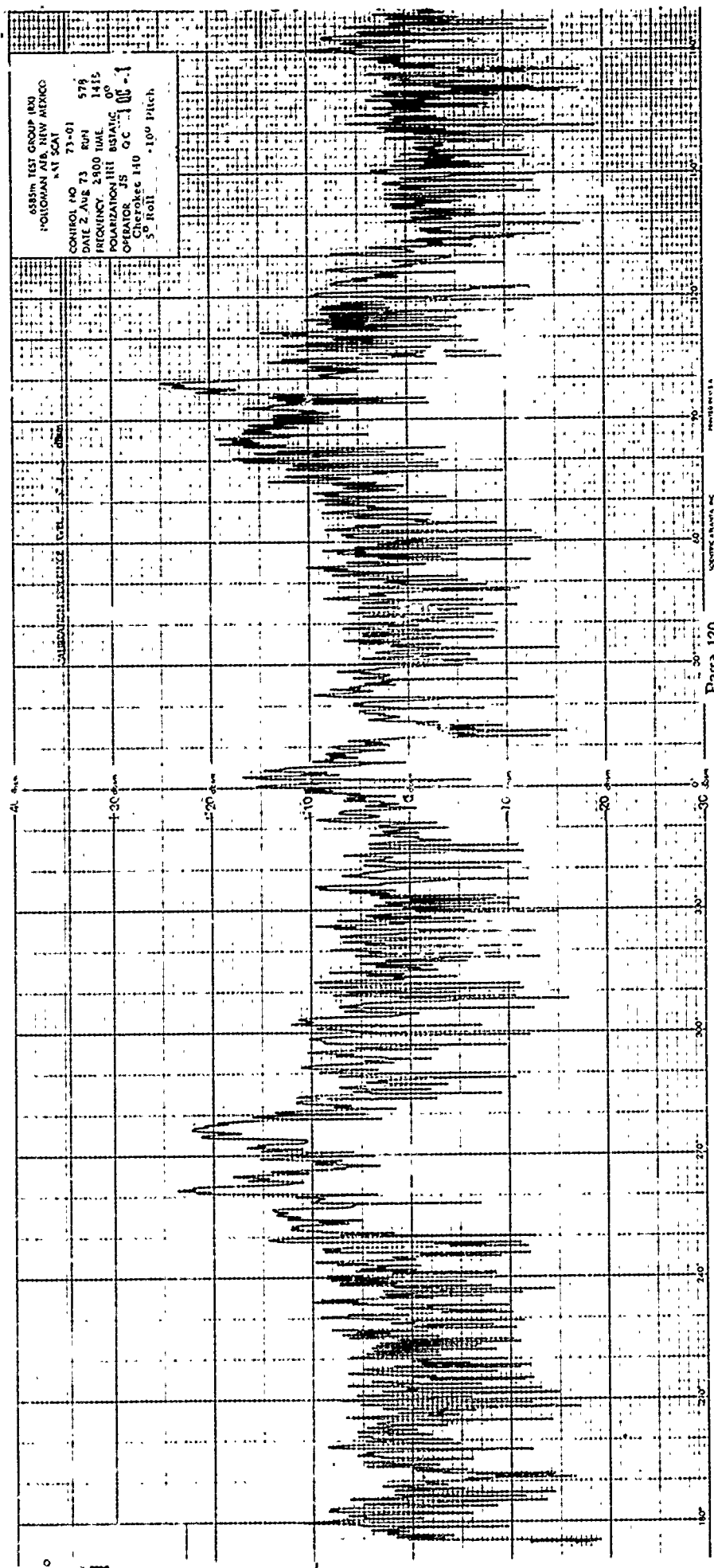
Cherokee 140

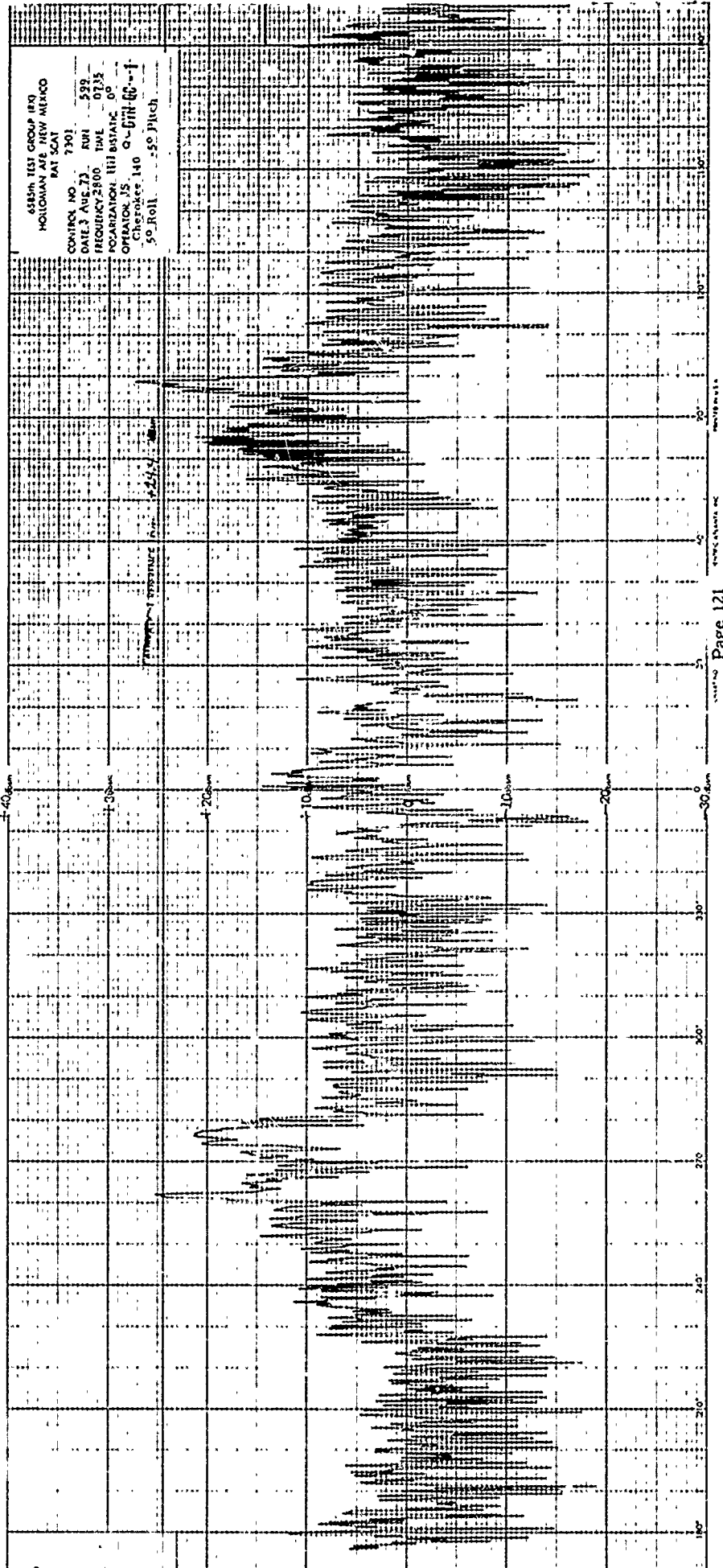
0° Roll 0° Pitch

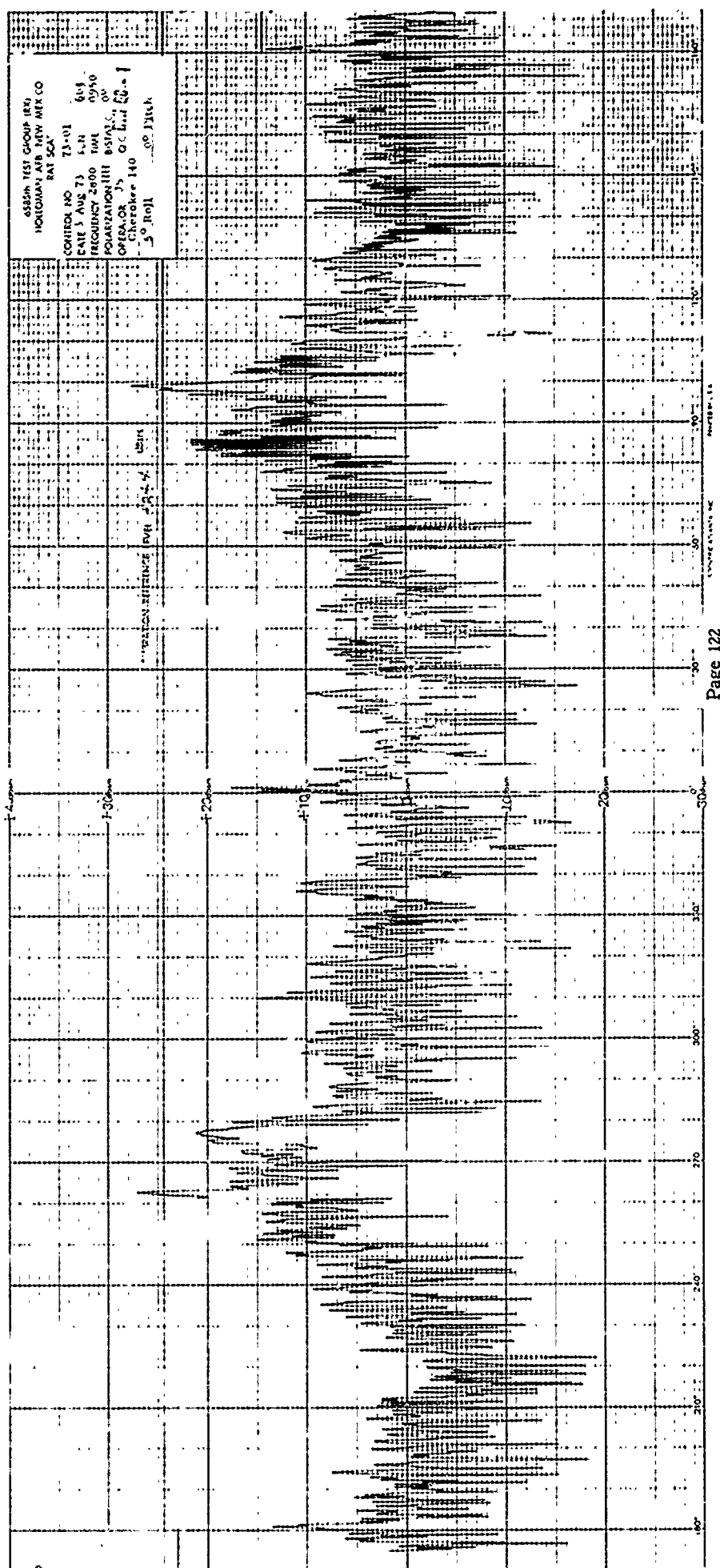


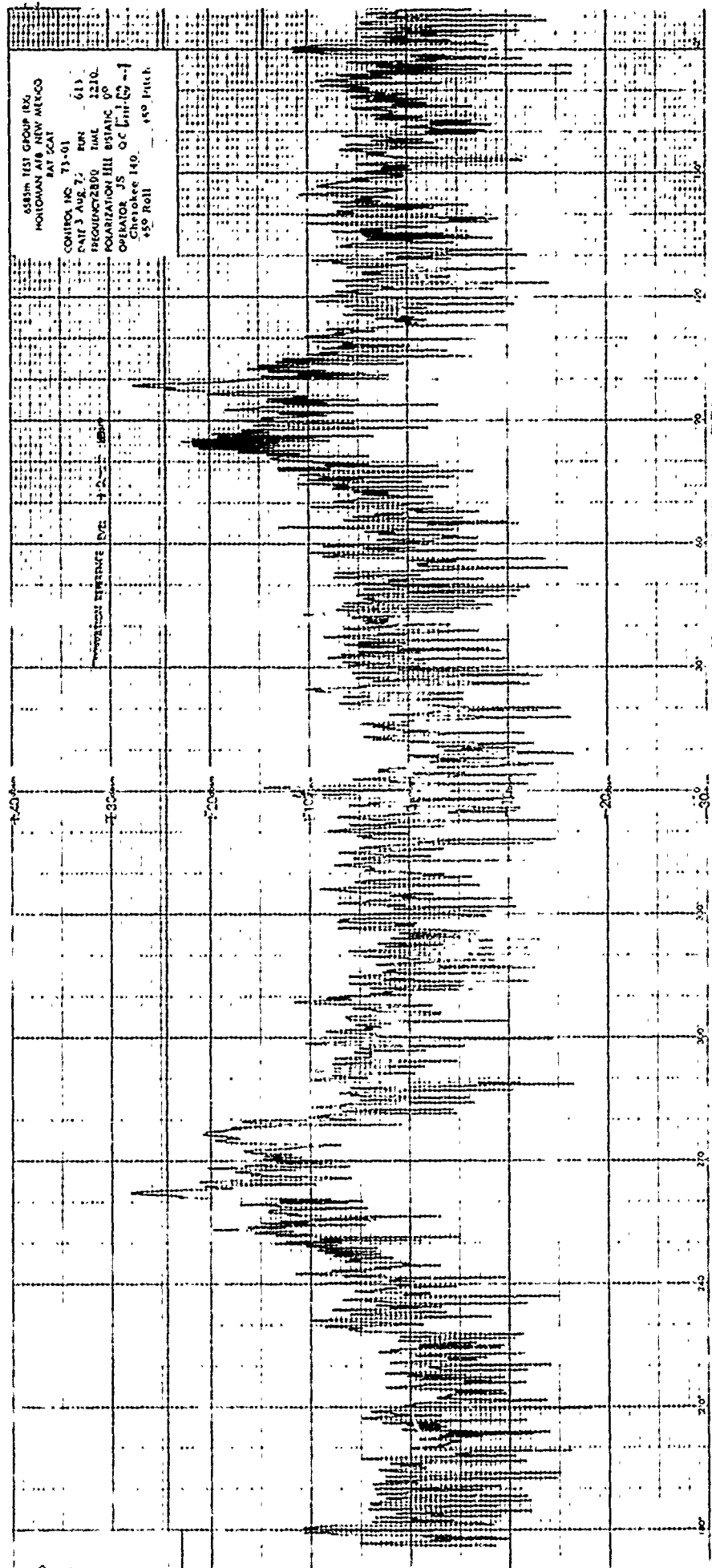


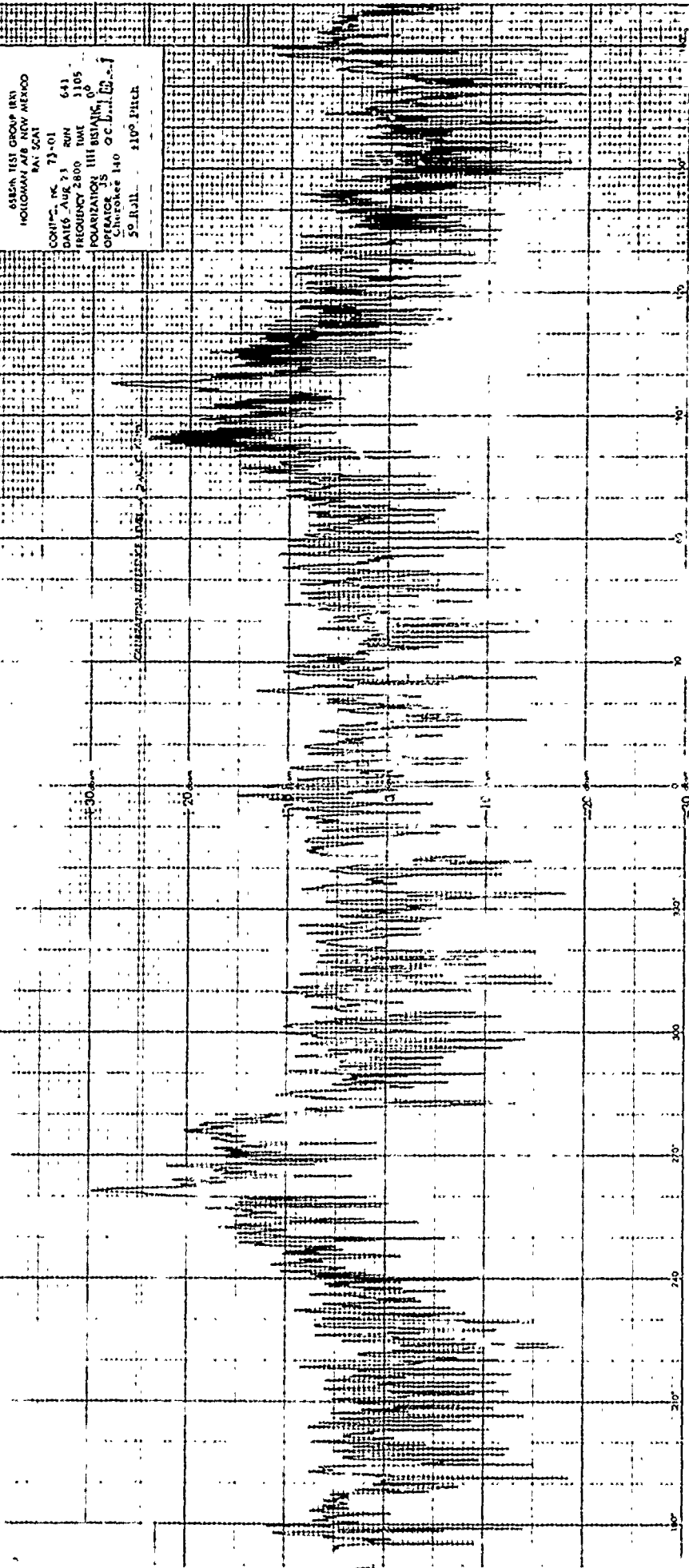


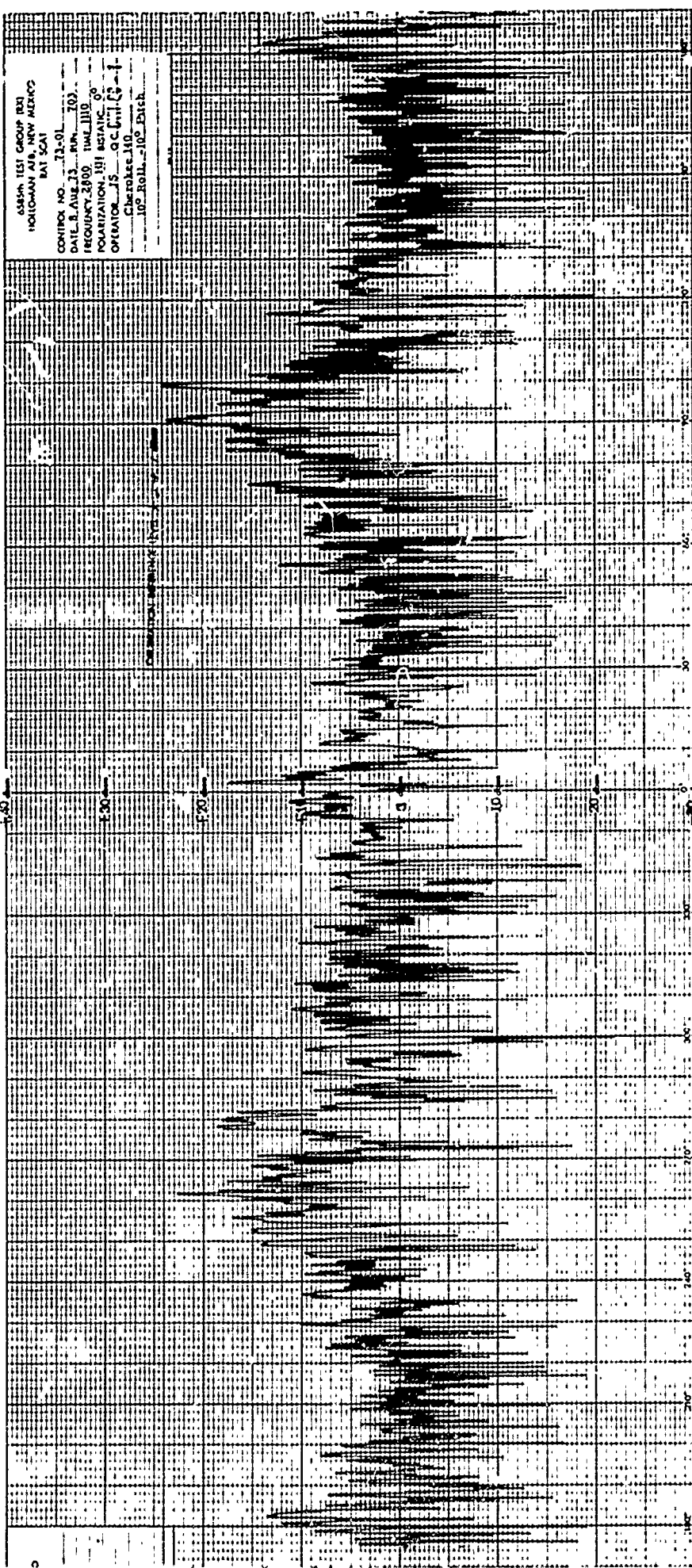






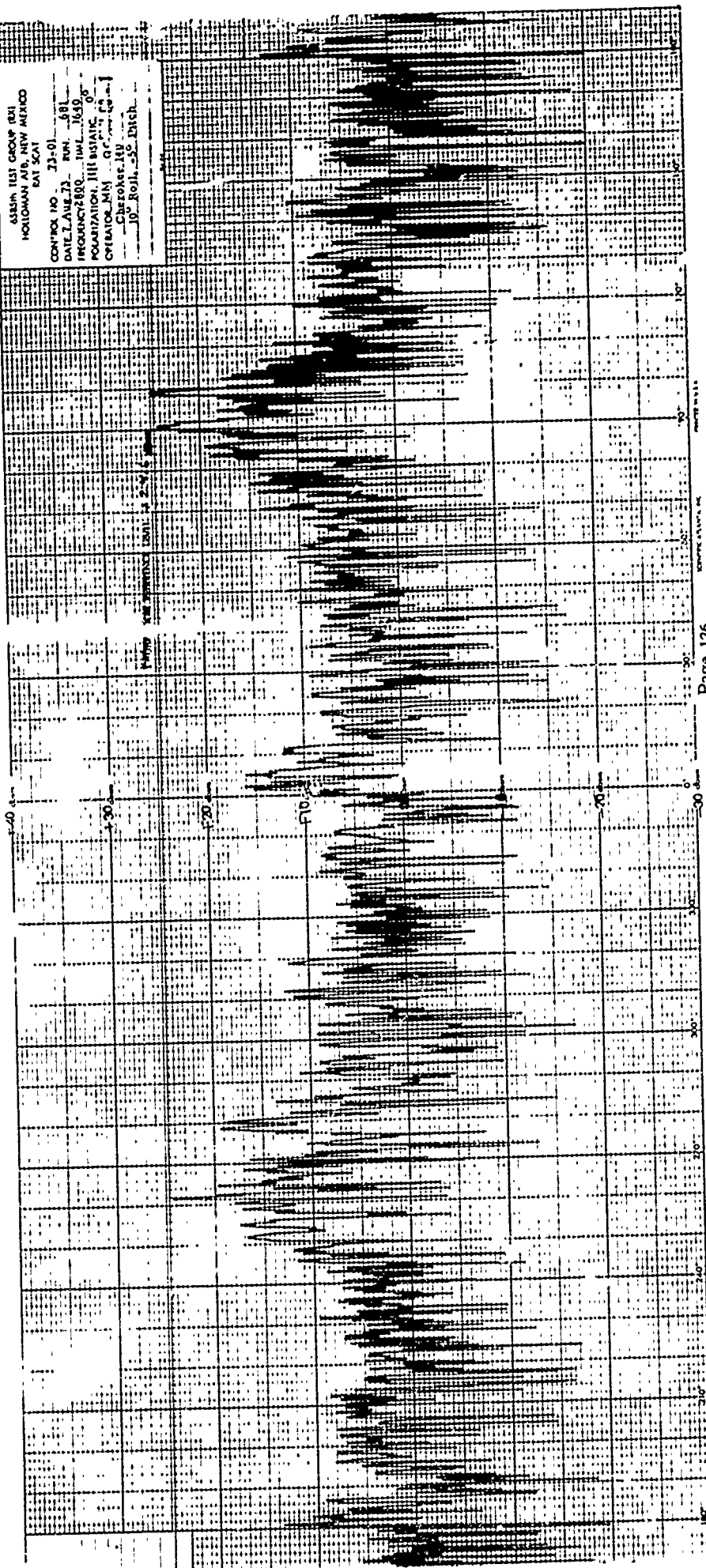


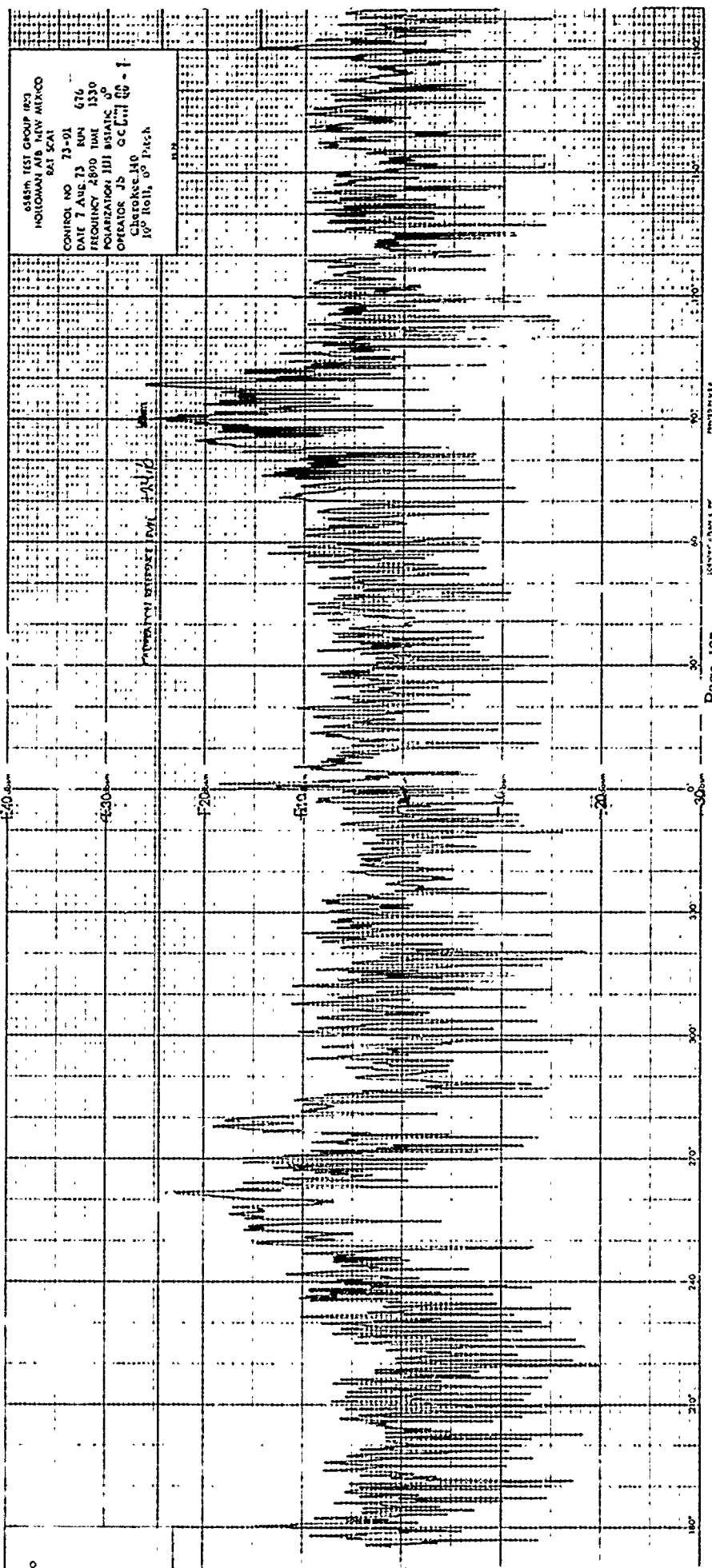


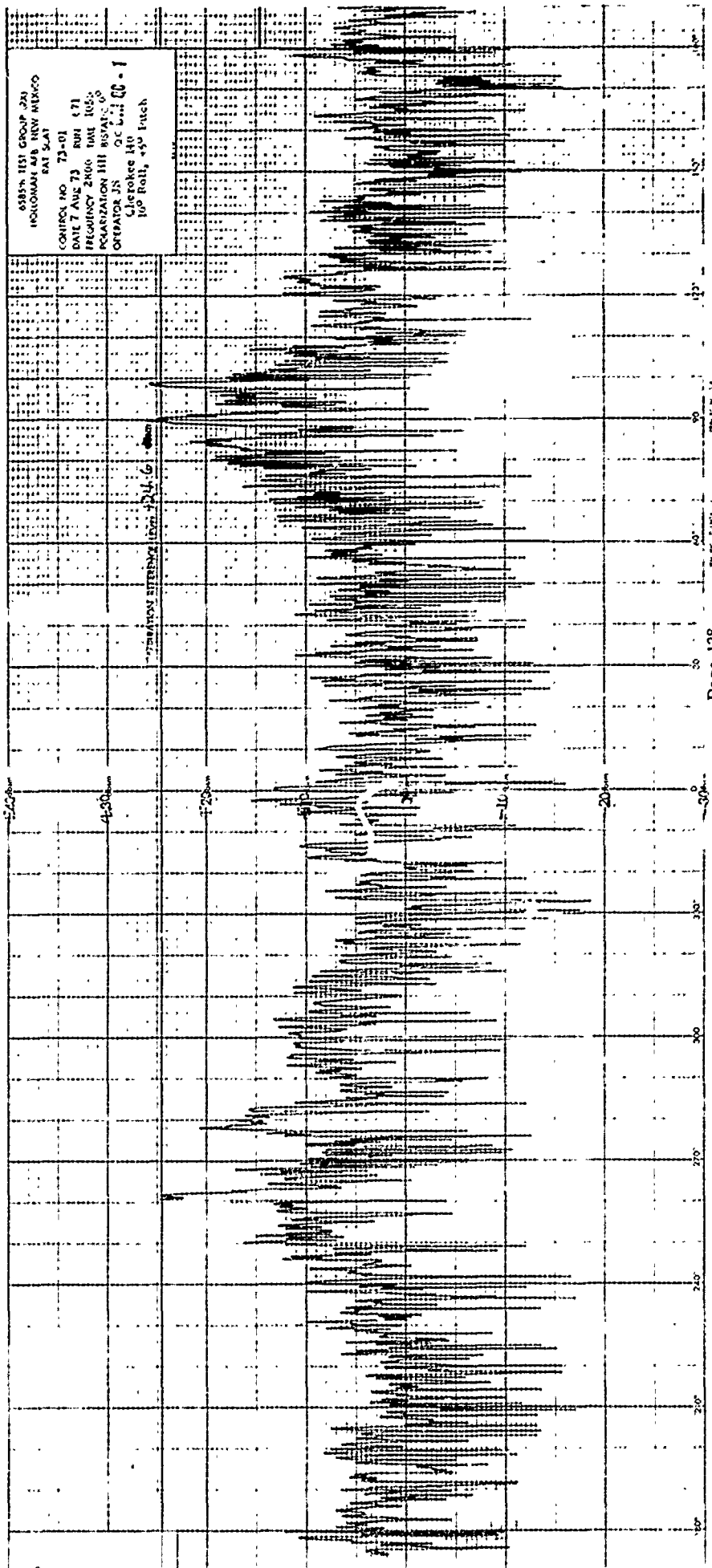


ASSIN TEST GROUP EX1
HOLLAND AFB, NEW MEXICO
BAT SCAT

CONTROL NO. 23-01
DATE 2 AUG 73 RUN 681
REGISTRATION 2400 INCH 1640
REGISTRATION III INSTANT 05
CYCLES/INCH 1000
CLOCKWISE 10
10 800L 55Z ENCL







658th TEST GROUP (BU)
HONOLULU, AIR NEW MEXICO
PAT SCAT

CURTIS, NO 73-01

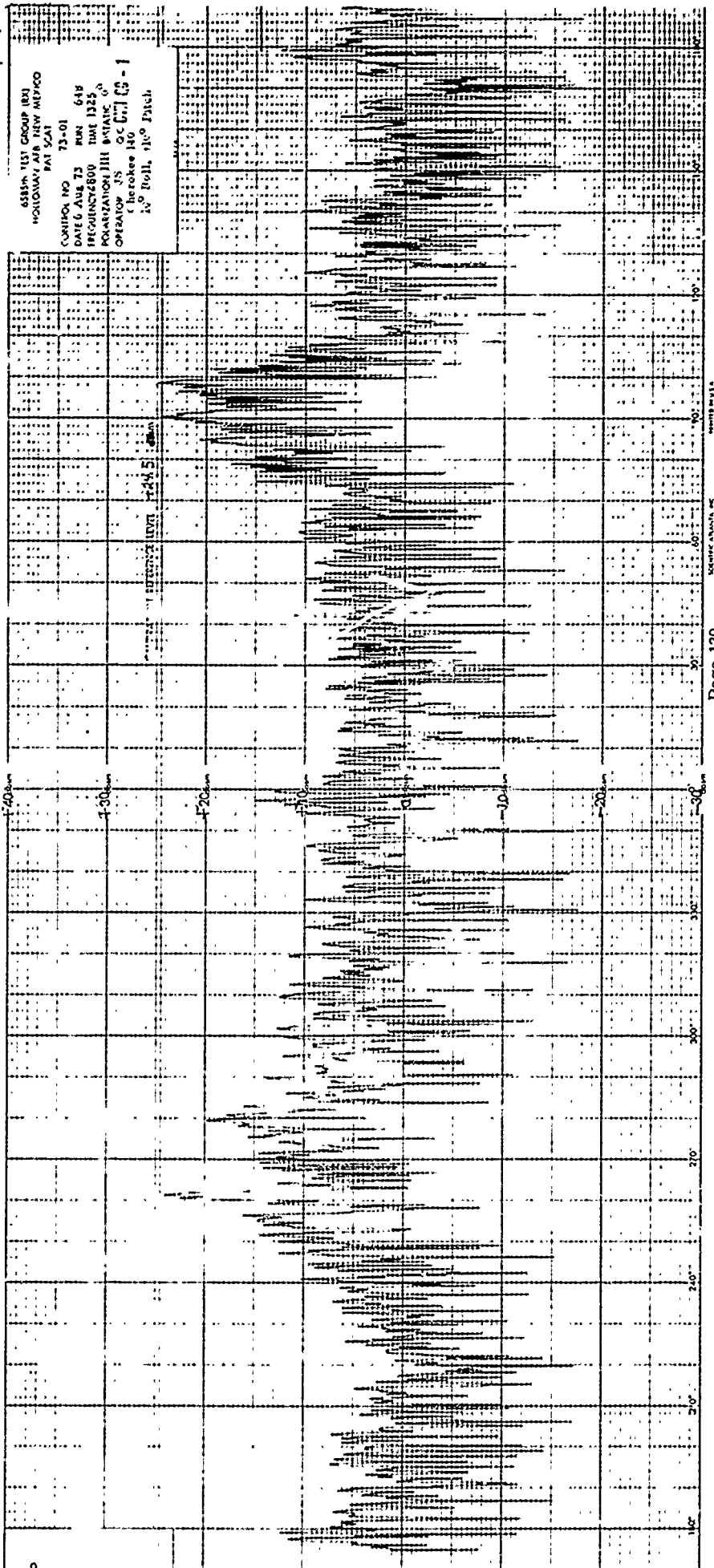
DATE 6 AUG 73 RUN 648

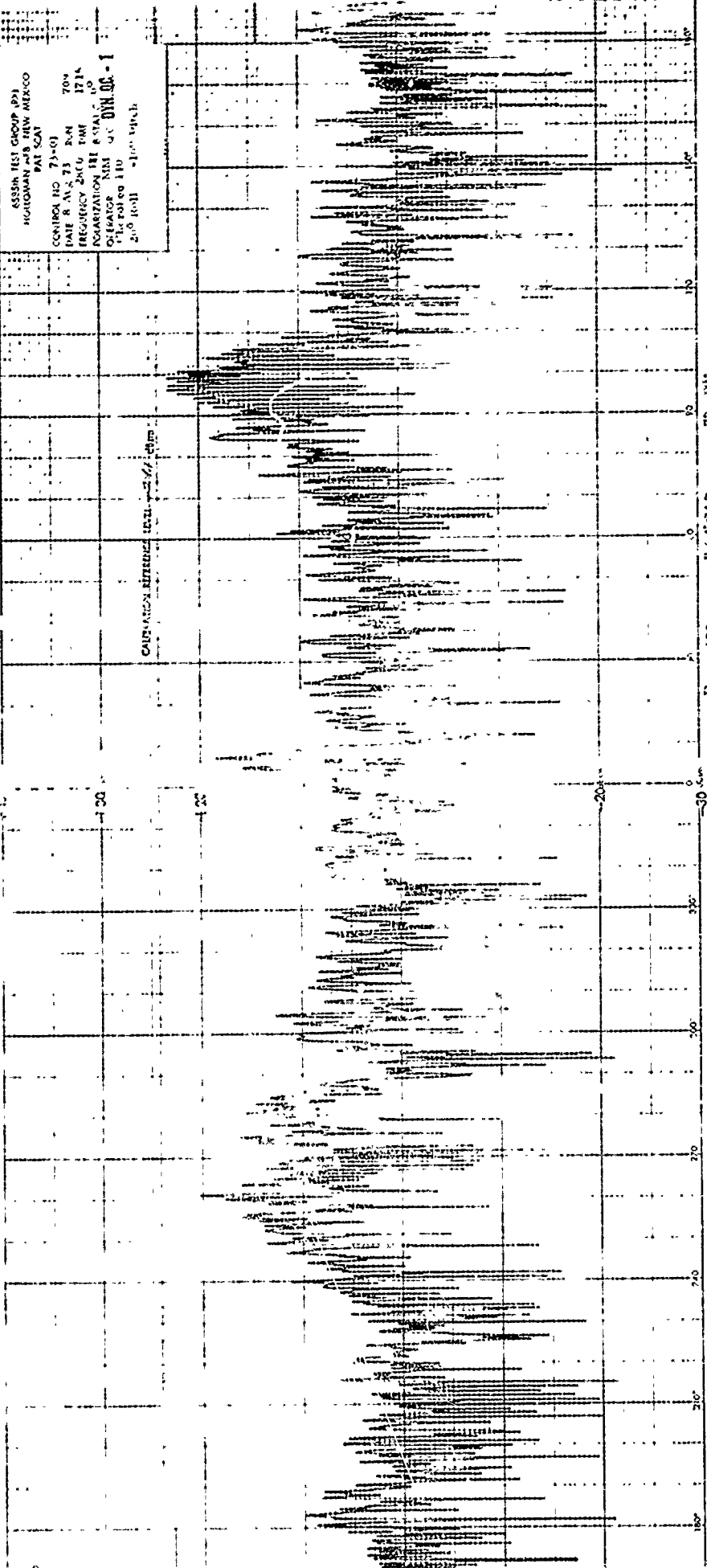
FREQUENCY 2800 THW 1325

POLARIZATION IIII STATIC 0°

OPERATOR JS OC GT 5 - 1

(Berkeley 140 11° 0° Patch)





6438th TEST GROUP PZA,
HOLLAMAN AFB, NEW MEXICO
RAT SCAT

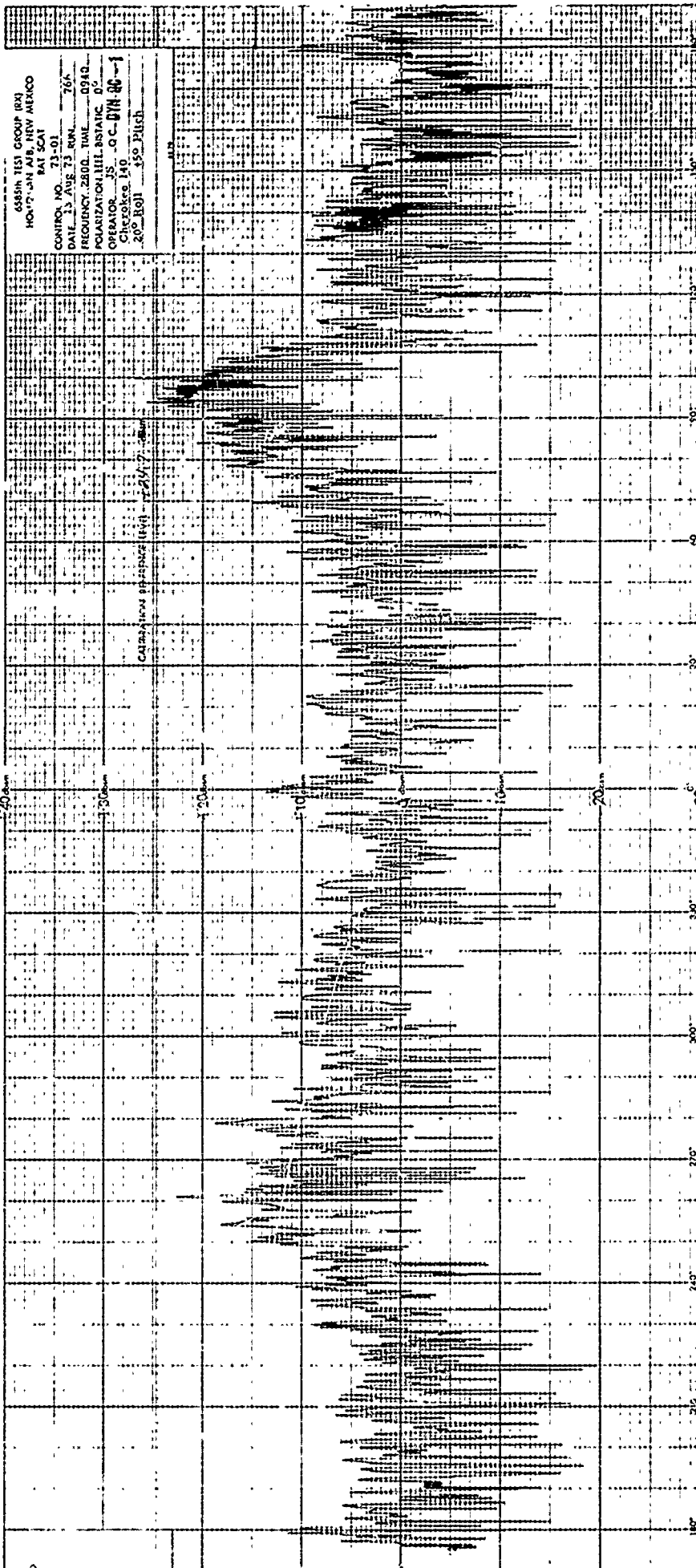
CONTROL NO 73-01 739
DATE 10 AUG 7 RUN TIME 0855
FREQUENCY 2800 MHz
POLARIZATION IIII BYPASS 00
OPERATOR JS OC DYN QC-1
Cherokee 140
260 Tolls -5° Pitch

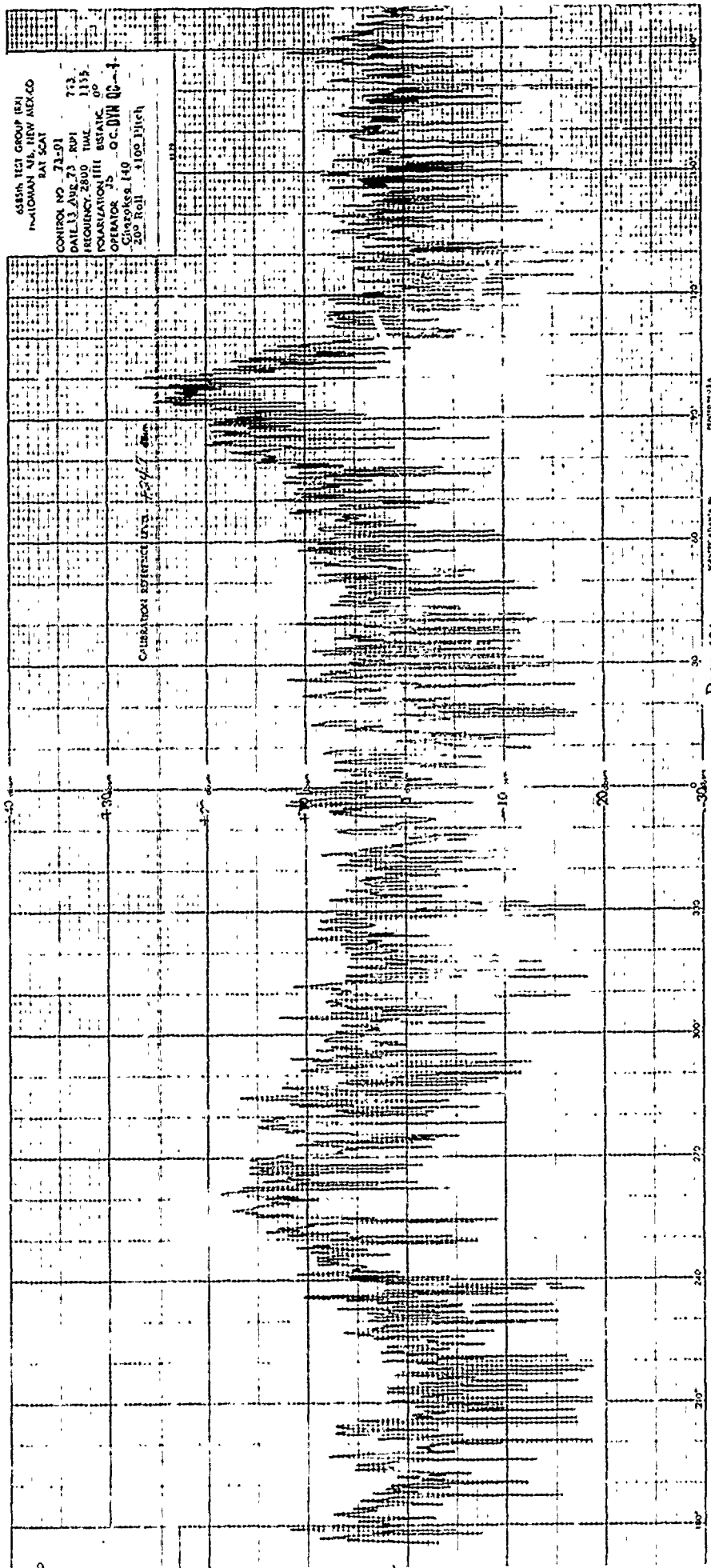
CHIRP/NOISE TEST-02 (Pm-200) 2800 MHz

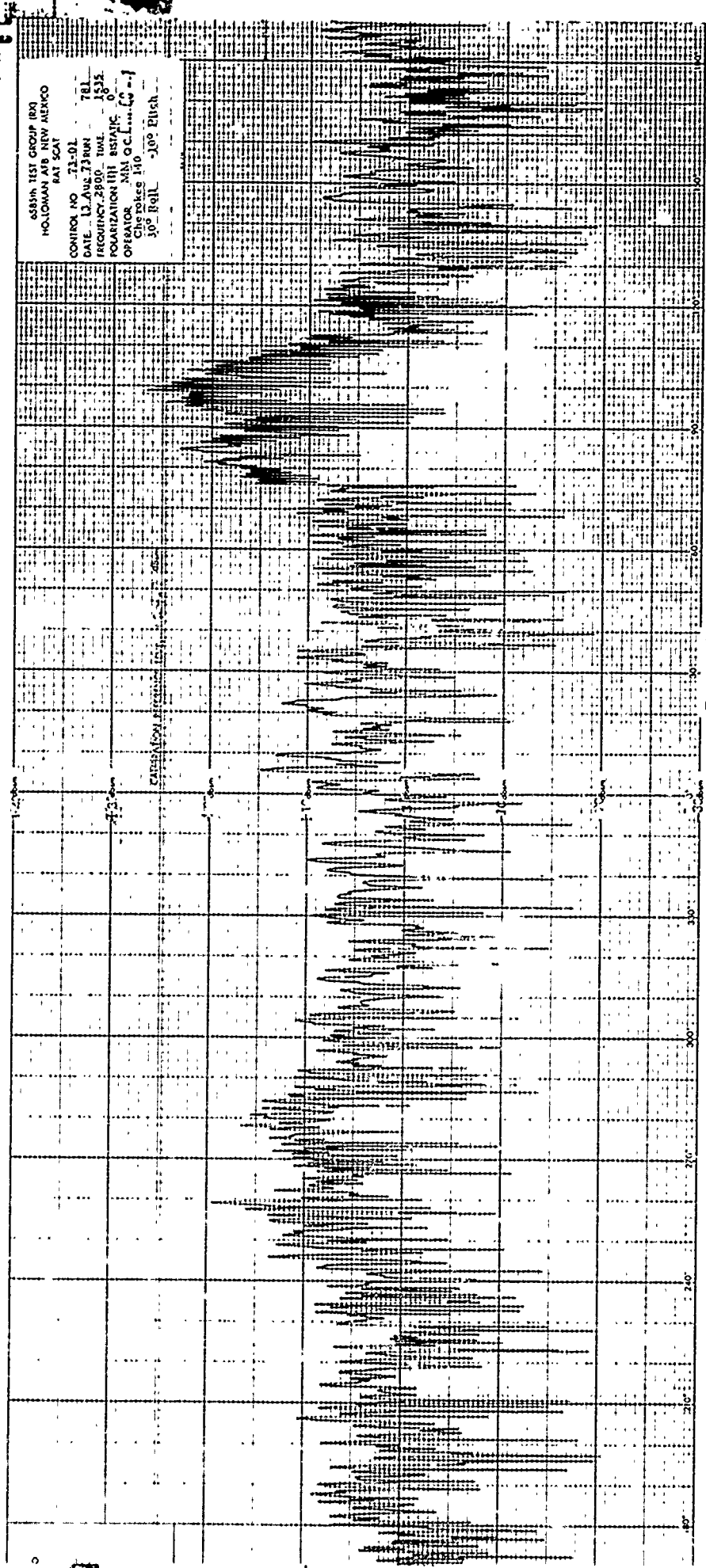
50 MHz
40 MHz
30 MHz
20 MHz

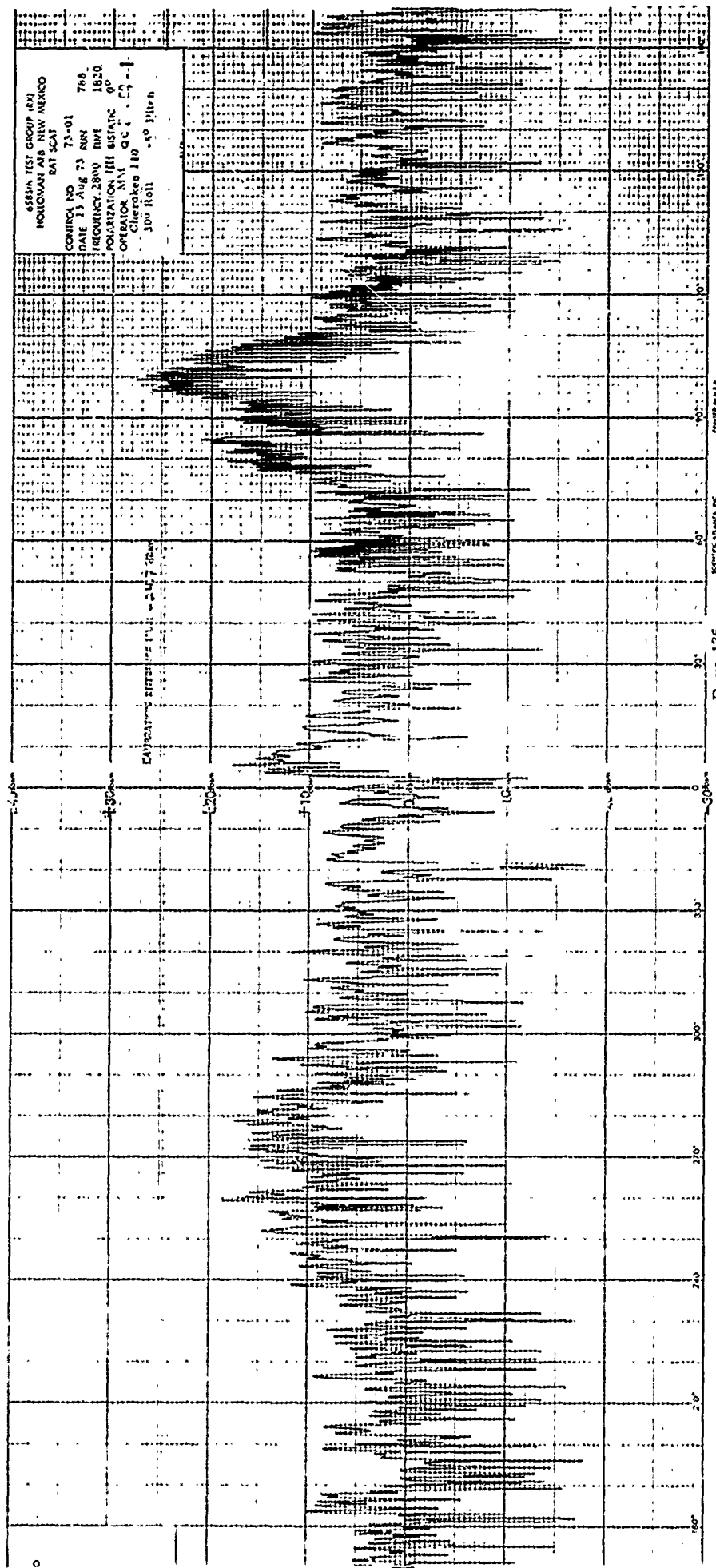
0 10 20 30 40 50 60 70 80 90 100

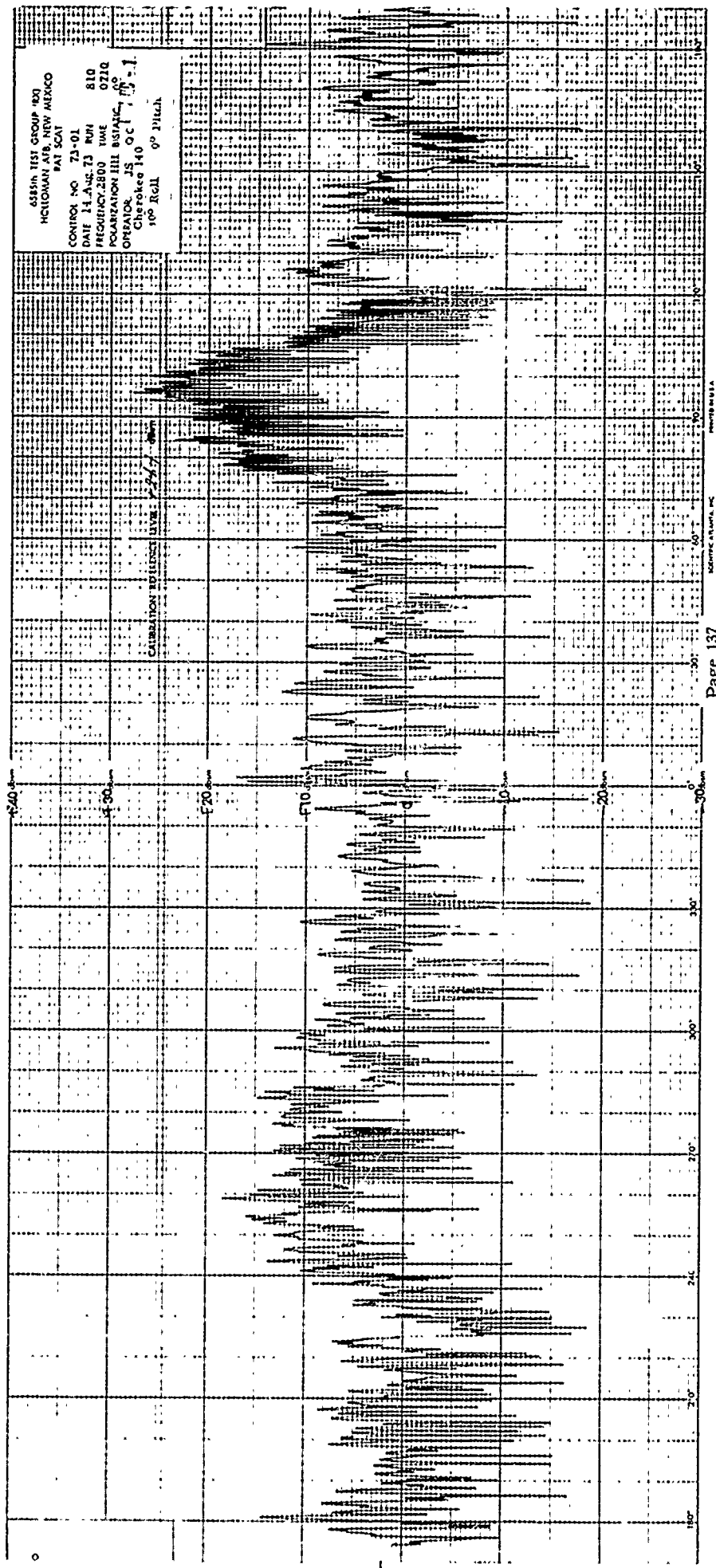
6553N TEST GROUP (BX)
HOLLAND AIR NEW AEXCO
EAT SCAI
CONTROL NO 73-01
DATE 13 AUG 73 ERT
FREQUENCY 2400 TAVC
POLARIZATION III BRTAHC
OPERATOR 149 QC DIV 66 -1
200 Boll 00 Pitch

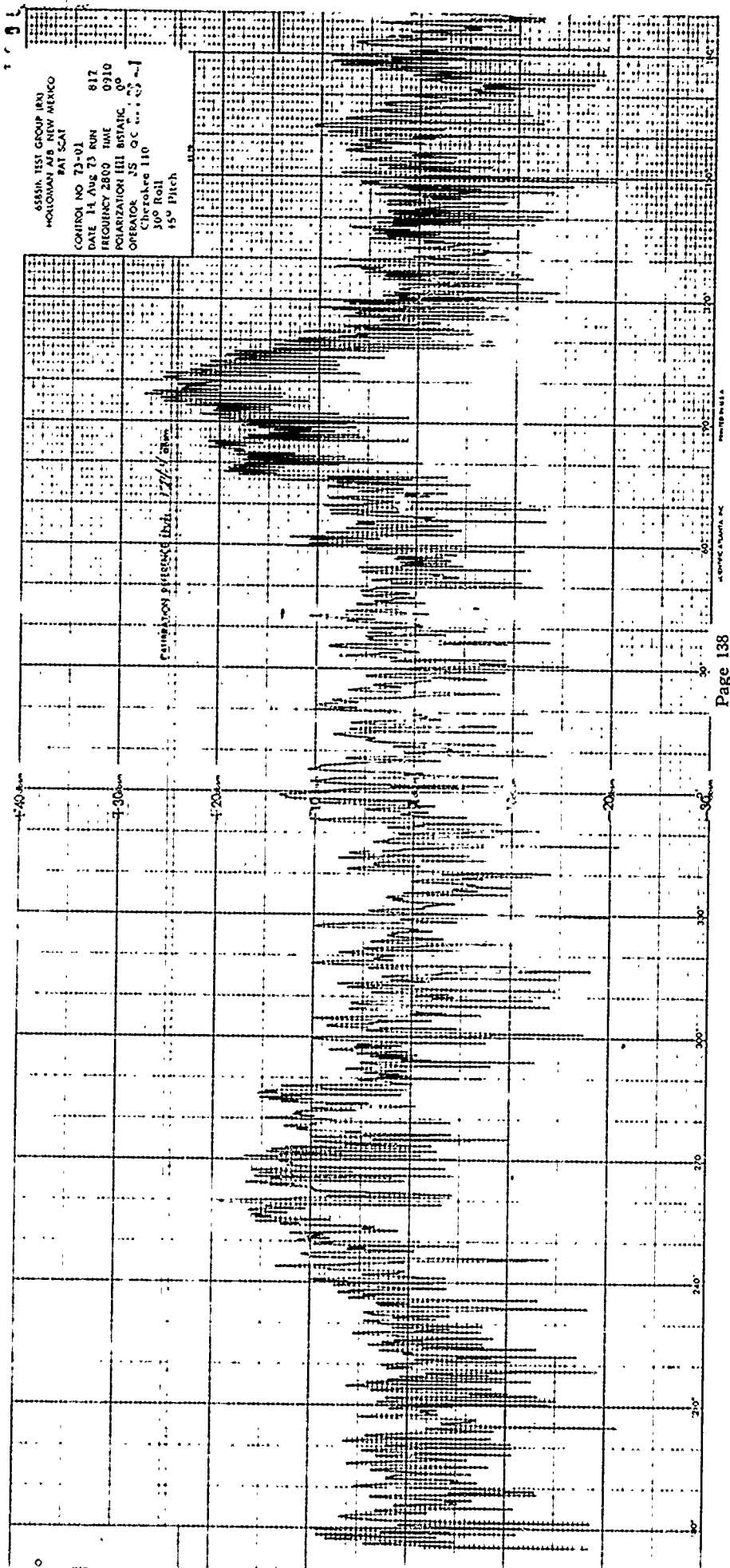






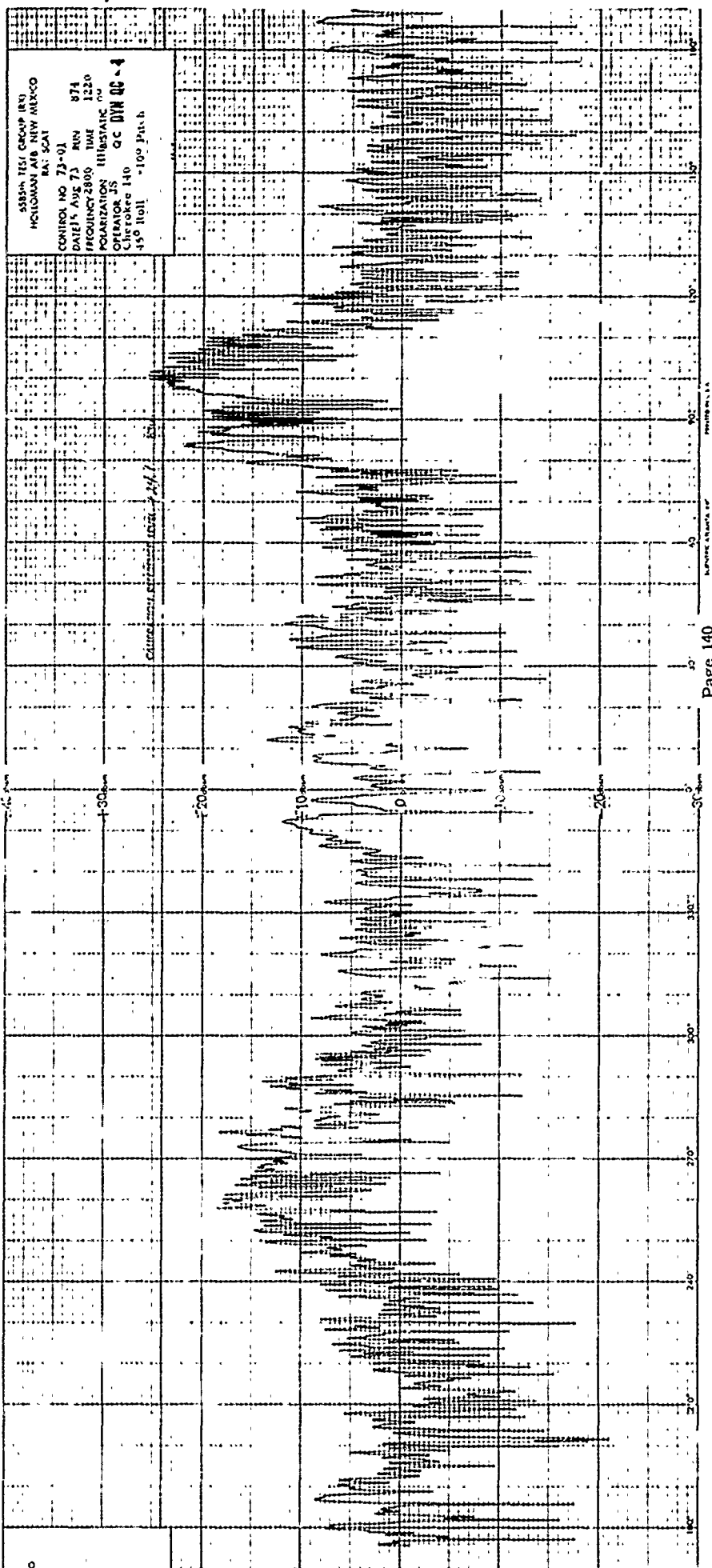


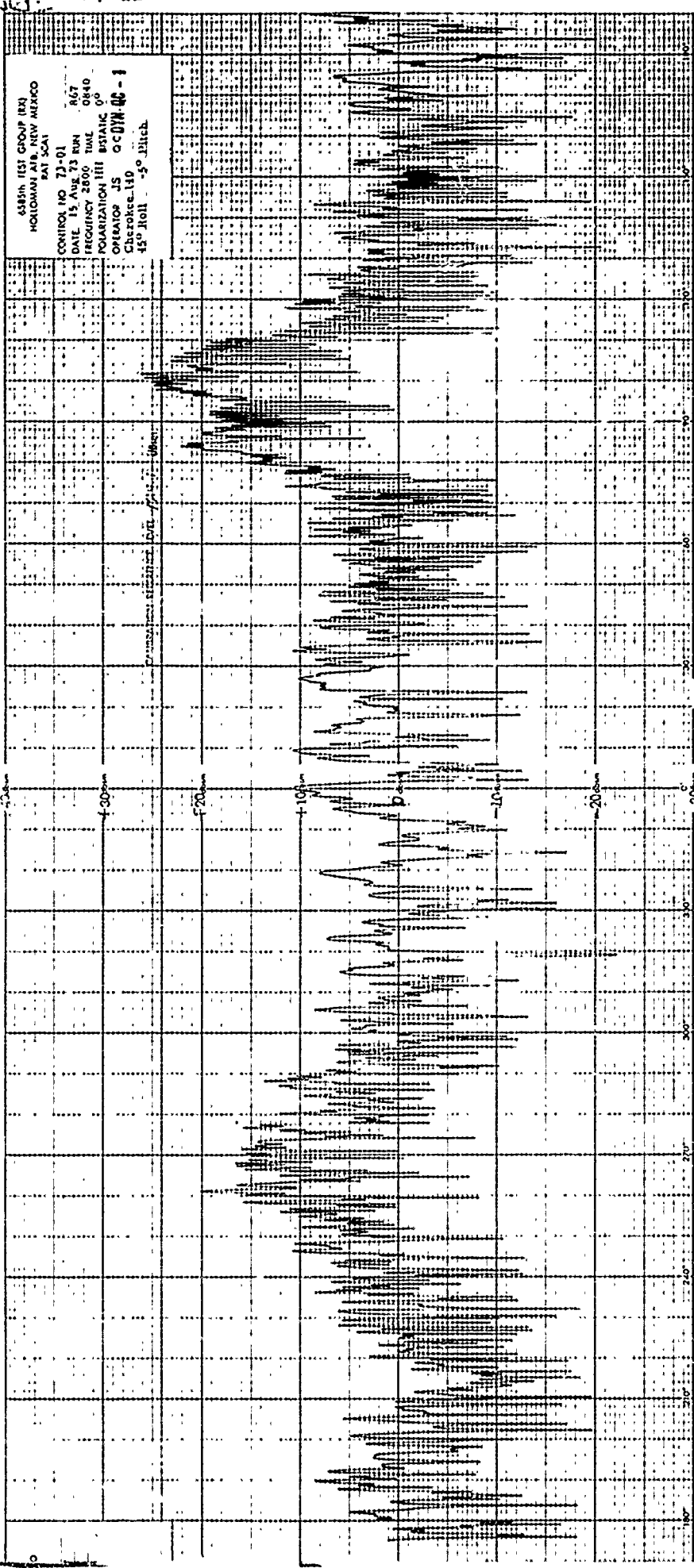


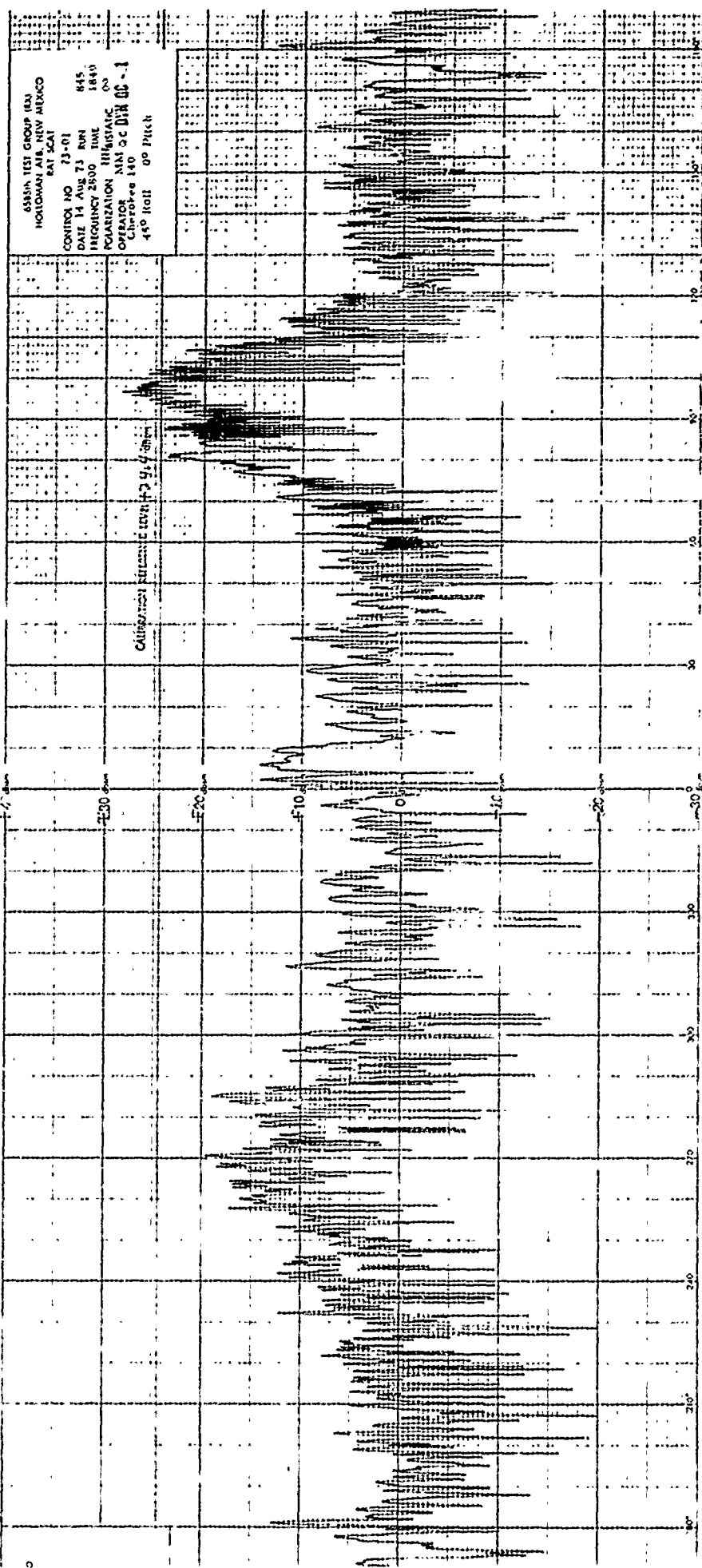


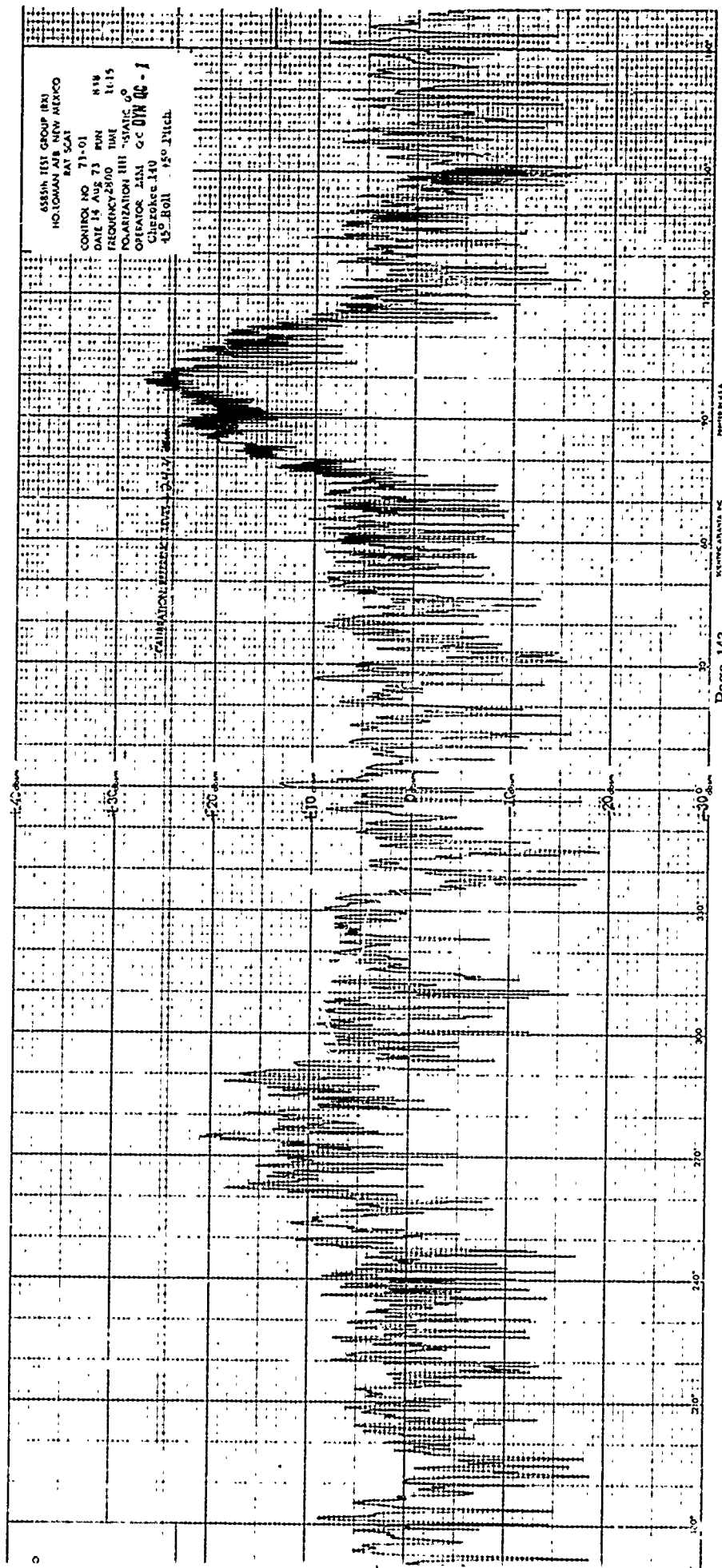
6553K TEST GROUP IRU
 HOLLOMAN AFB NEW MEXICO
 BAT SCAT
 CONTROL NO 73-01 817
 DATE 14 AUG 73 RUN 0910
 FREQUENCY 2500 HNE
 POLARIZATION III BISTATIC
 OPERATOR JS GC 110
 Channel 110
 300 Roll
 15° Pitch

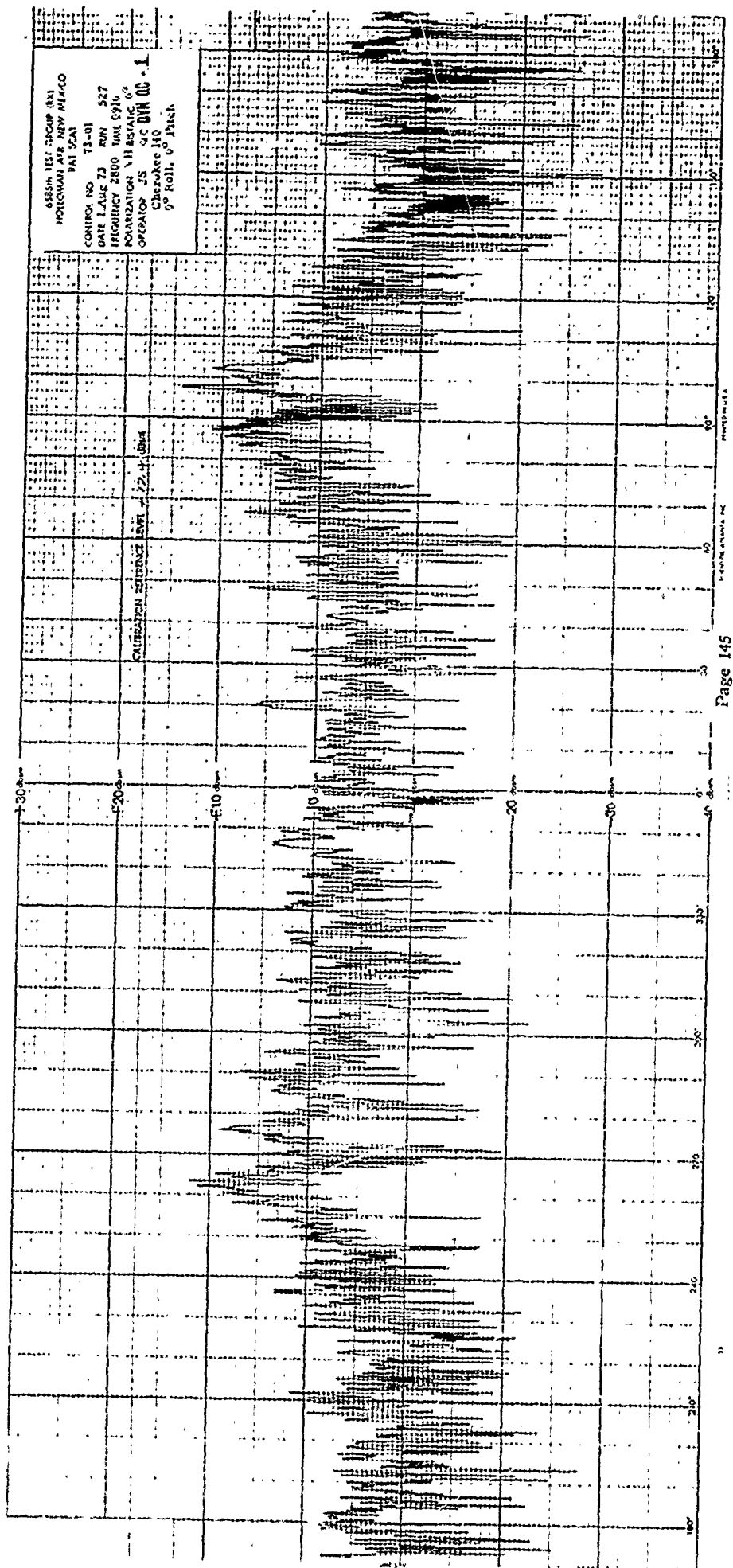
[illegible]

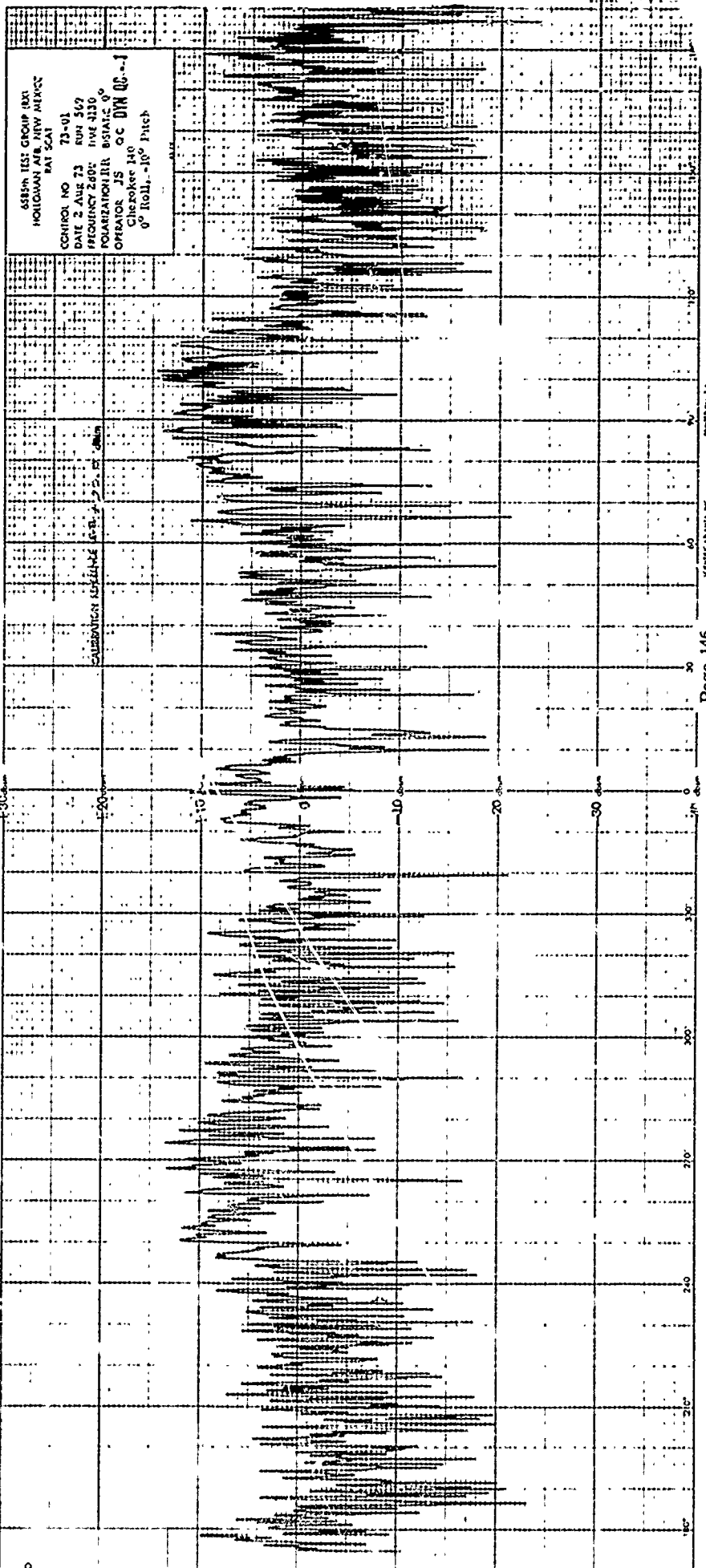












5555A TEST GROUP (20)
HOLLOMAN, NEW MEXICO
BAT SCAT

CONTROL NO 73-01

DATE 2 AUG 73 RUN 562

FREQUENCY 2800 MHz 0720

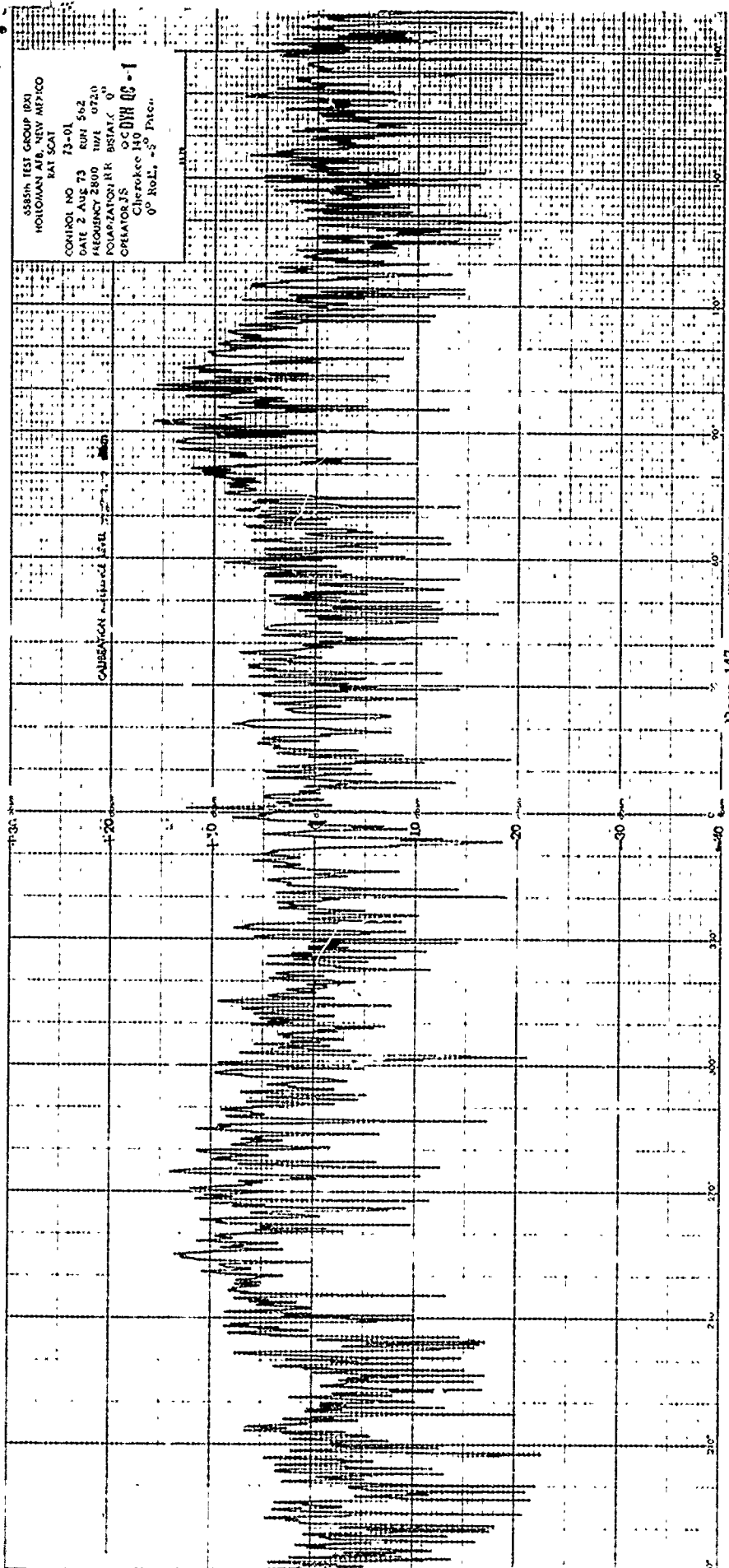
POLARIZATION R R BISTATIC 0°

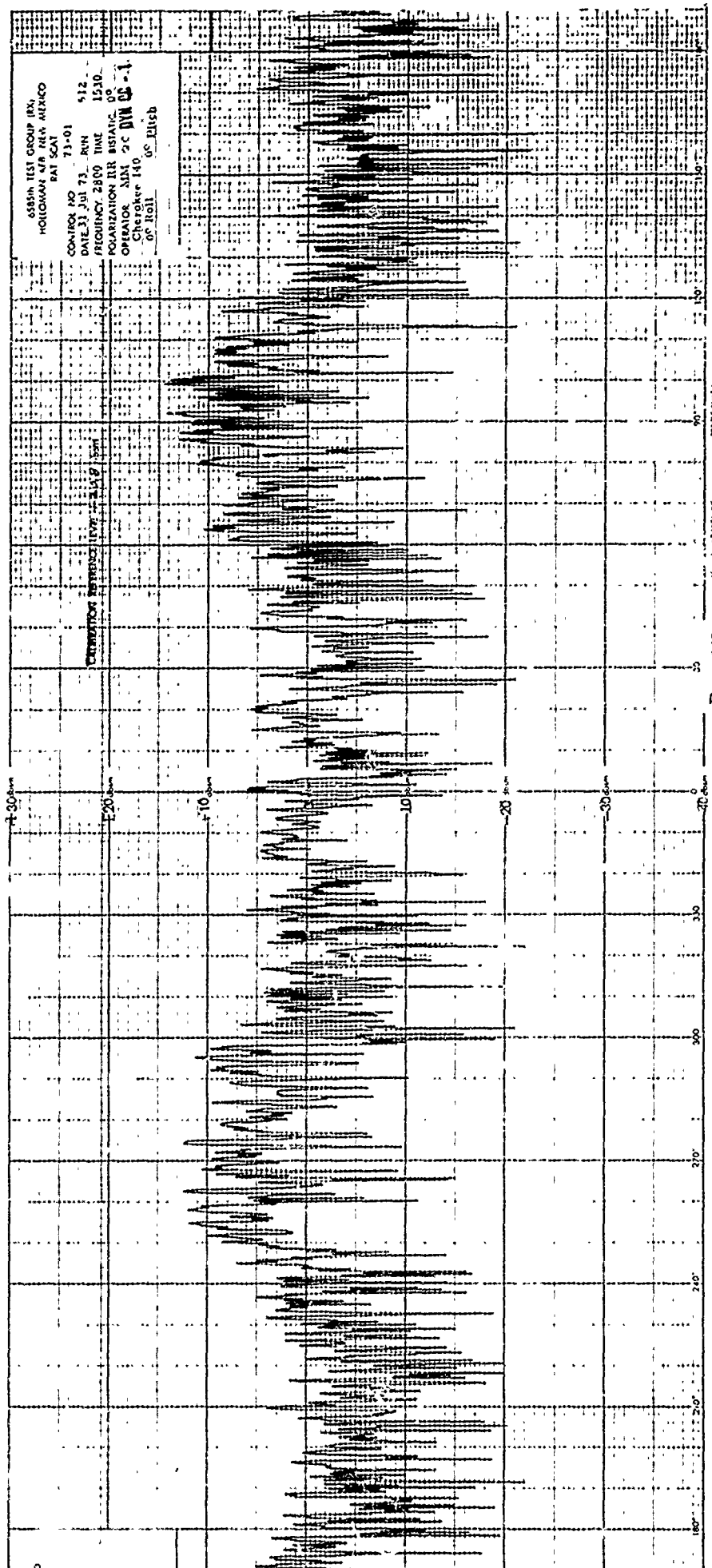
OPERATOR JS OC DIV 05-1

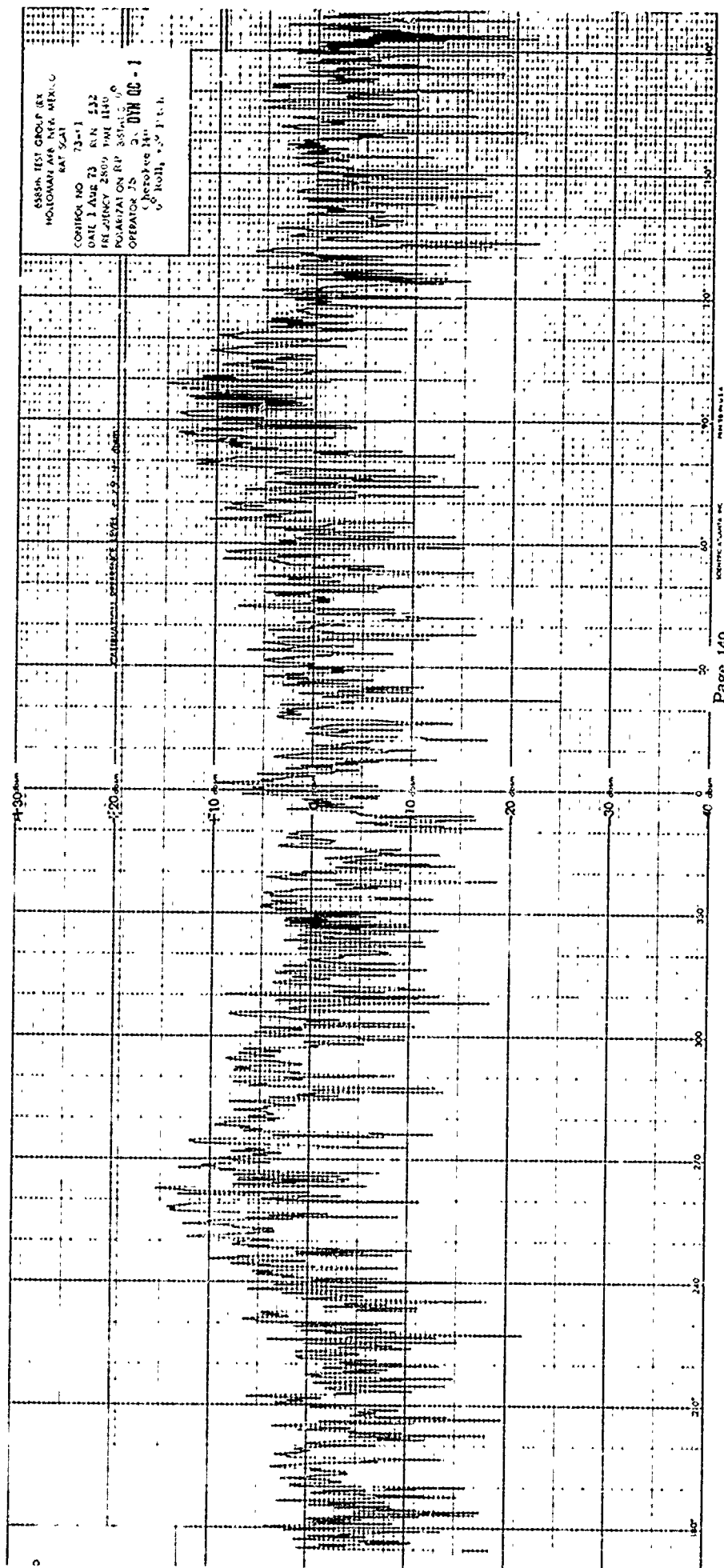
Cherokee 110

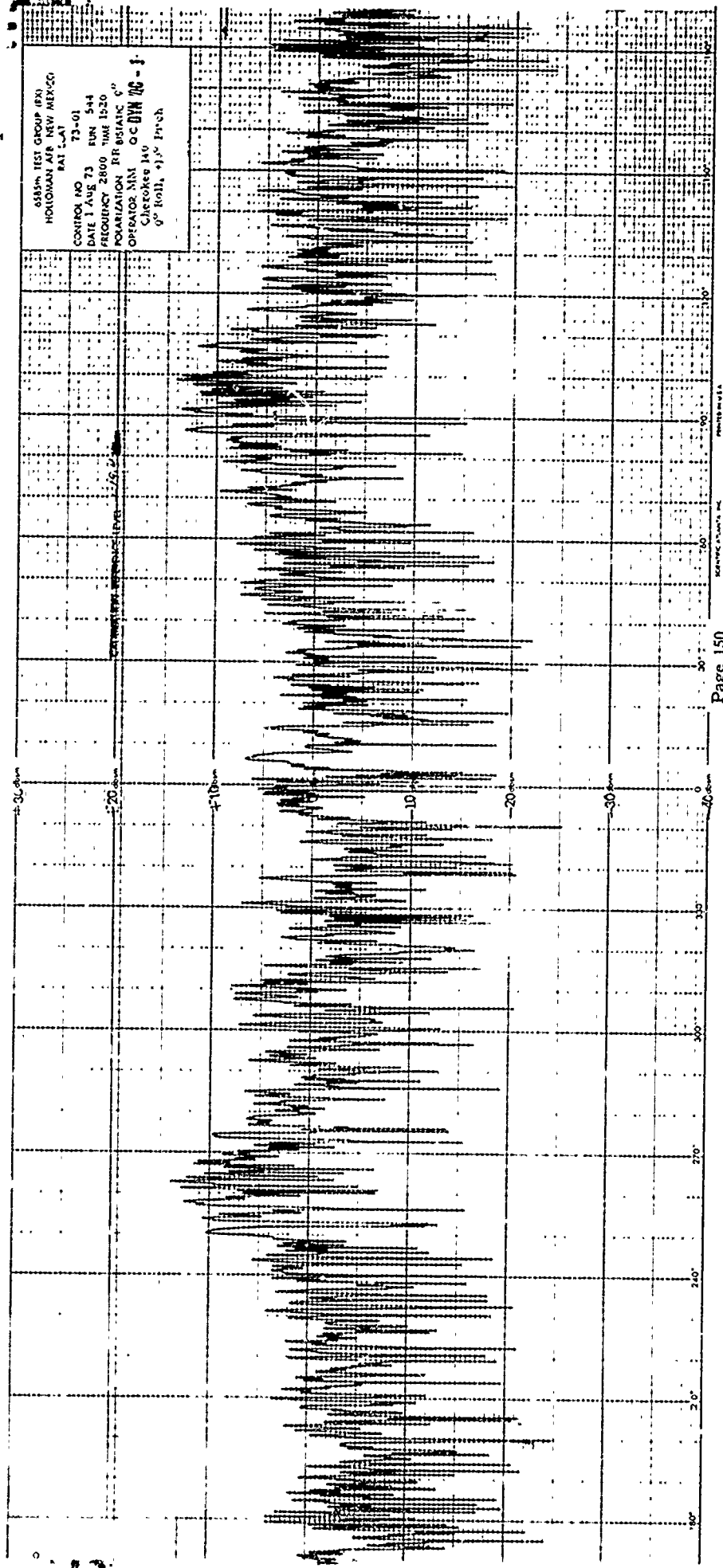
0° Roll, -3° Pitch

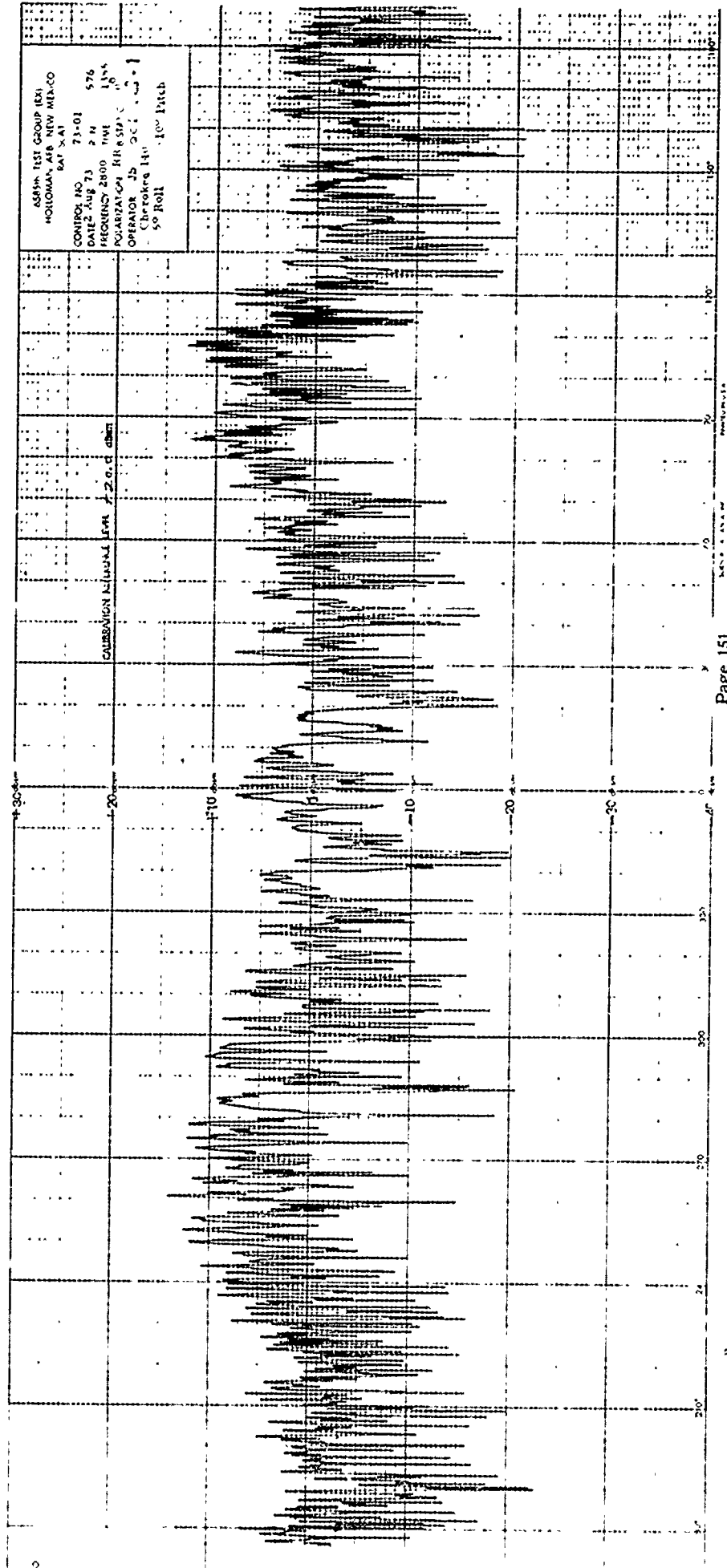
CAUSEWAY - 15.1 EL

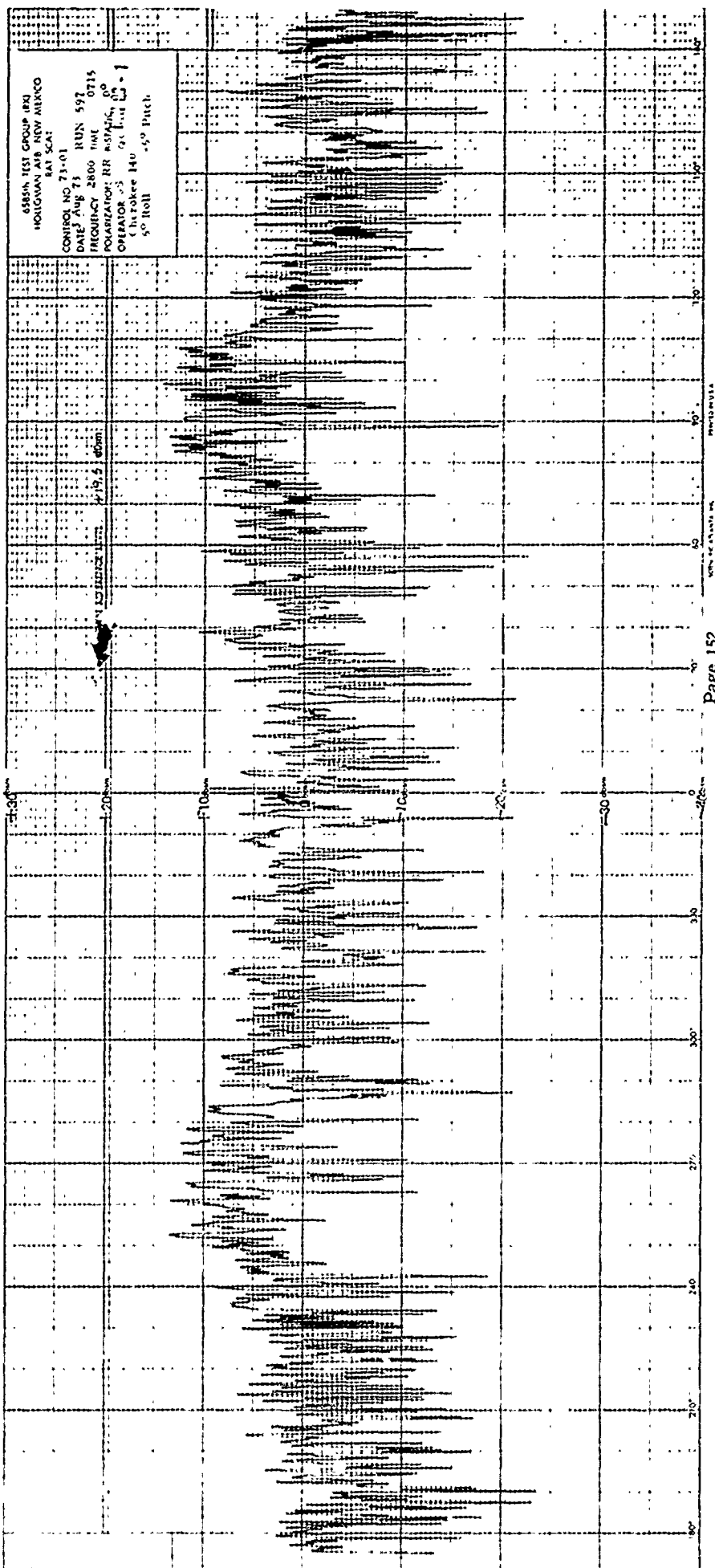


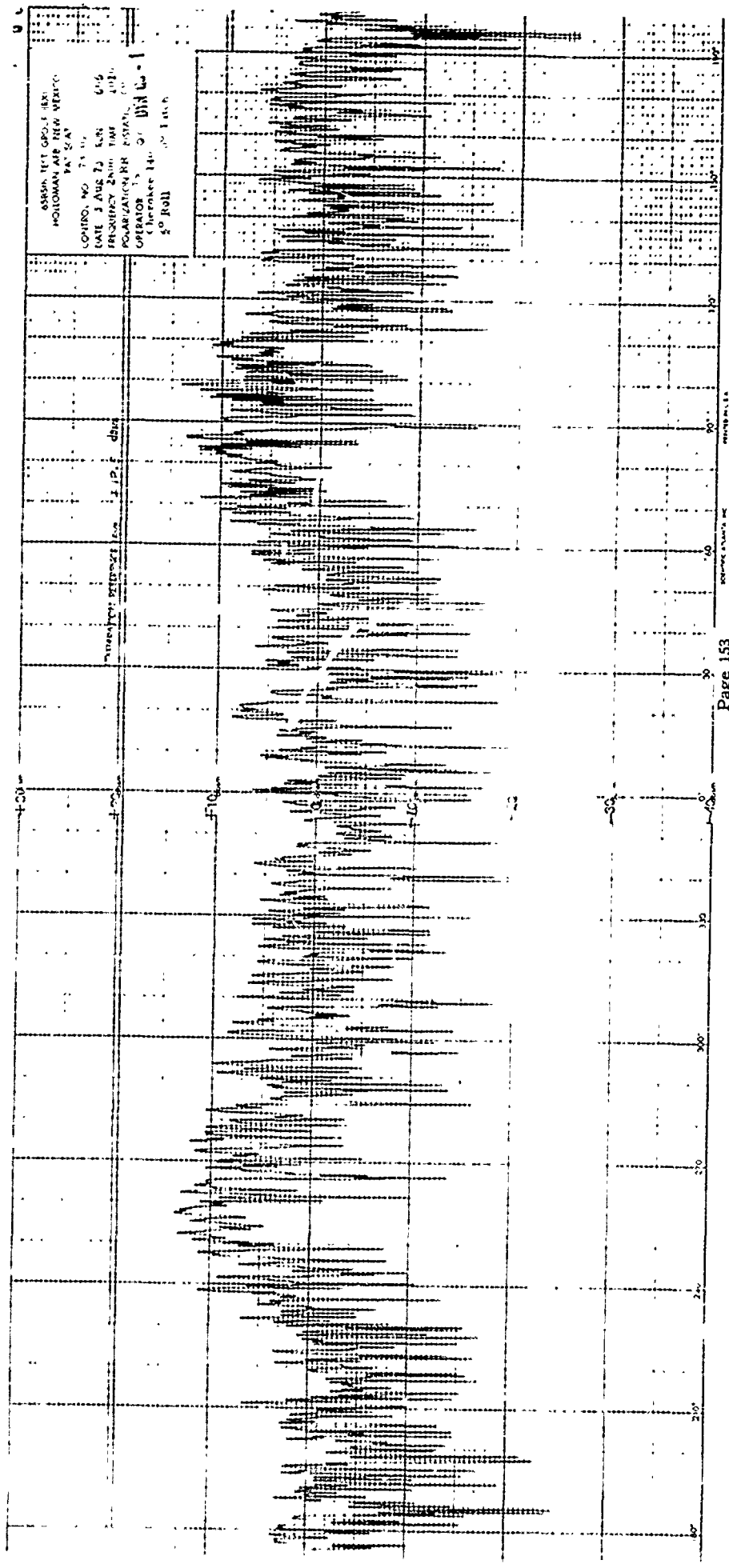




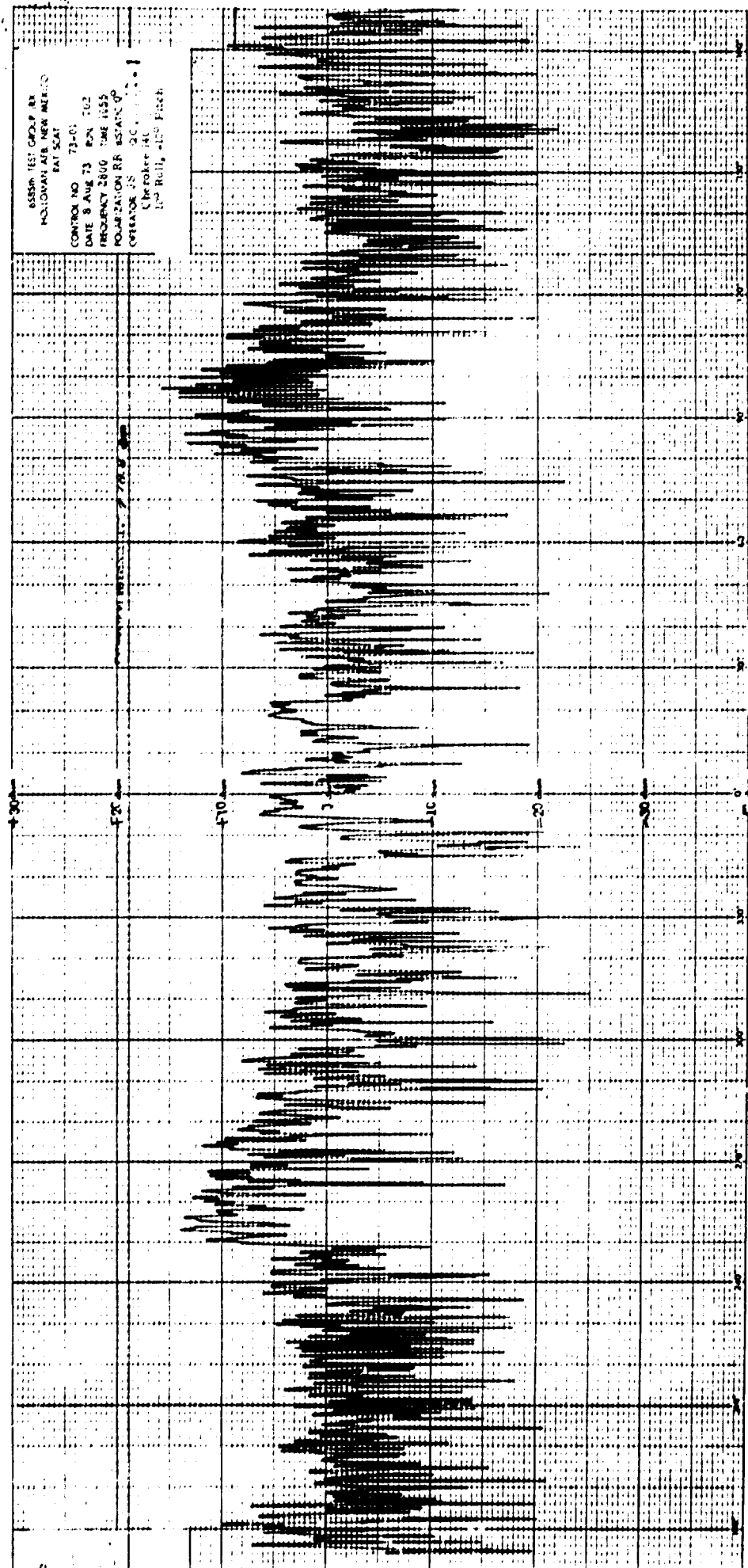


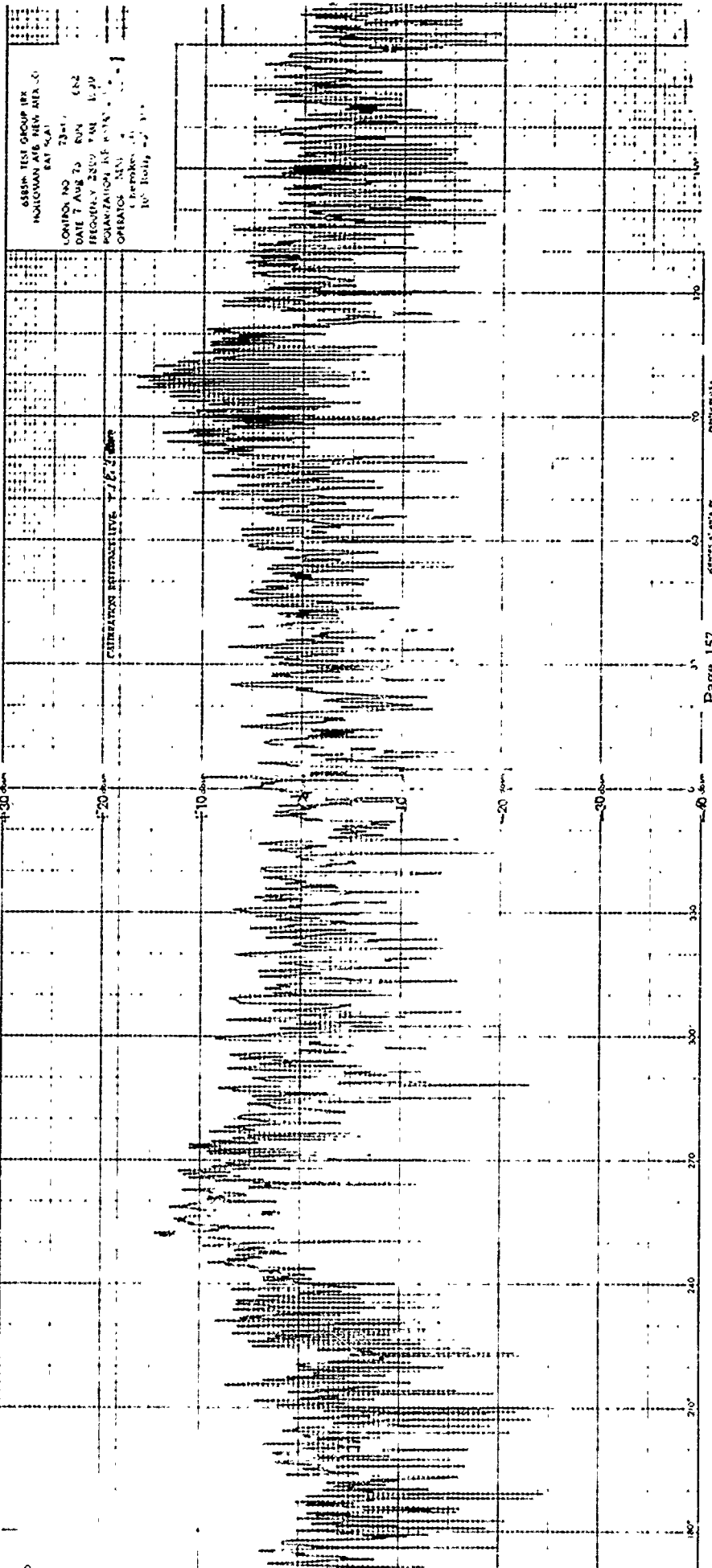






65589 TEST GROUP IN
HOLLOMAN AFB, NEW MEXICO
BAT SCAT
CONTROL NO 73-01
DATE 8 AUG 73 PUN 102
FREQUENCY 2800 TIME 1055
POLARIZATION RR 45° ANGLE 0°
OPERATOR JS 2C
C. P. ROBERT 461
1000 Roll, 4000 Pitch





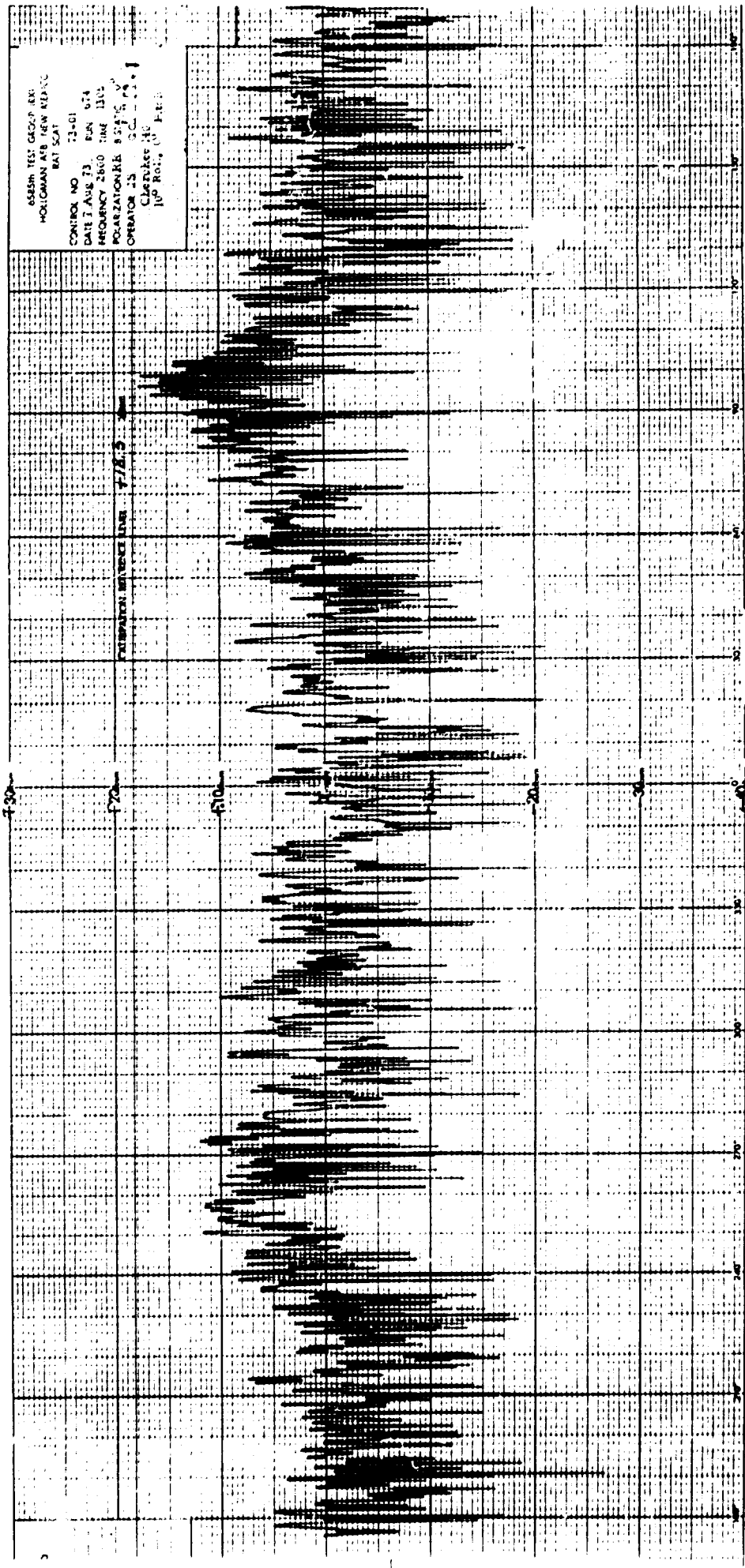
6585m TEST GROUP BR
HOLCOMMAN AFB NEW MEX CO
EAT 541

CONTROL NO 73-1
DATE 7 AUG 75 RUN 652
RECORDING 2500 YAL 11:20

REMARKS: 1. 10" Bolt
2. 10" Bolt

CHIRAZON EARTHQUAKE 7:15 PM

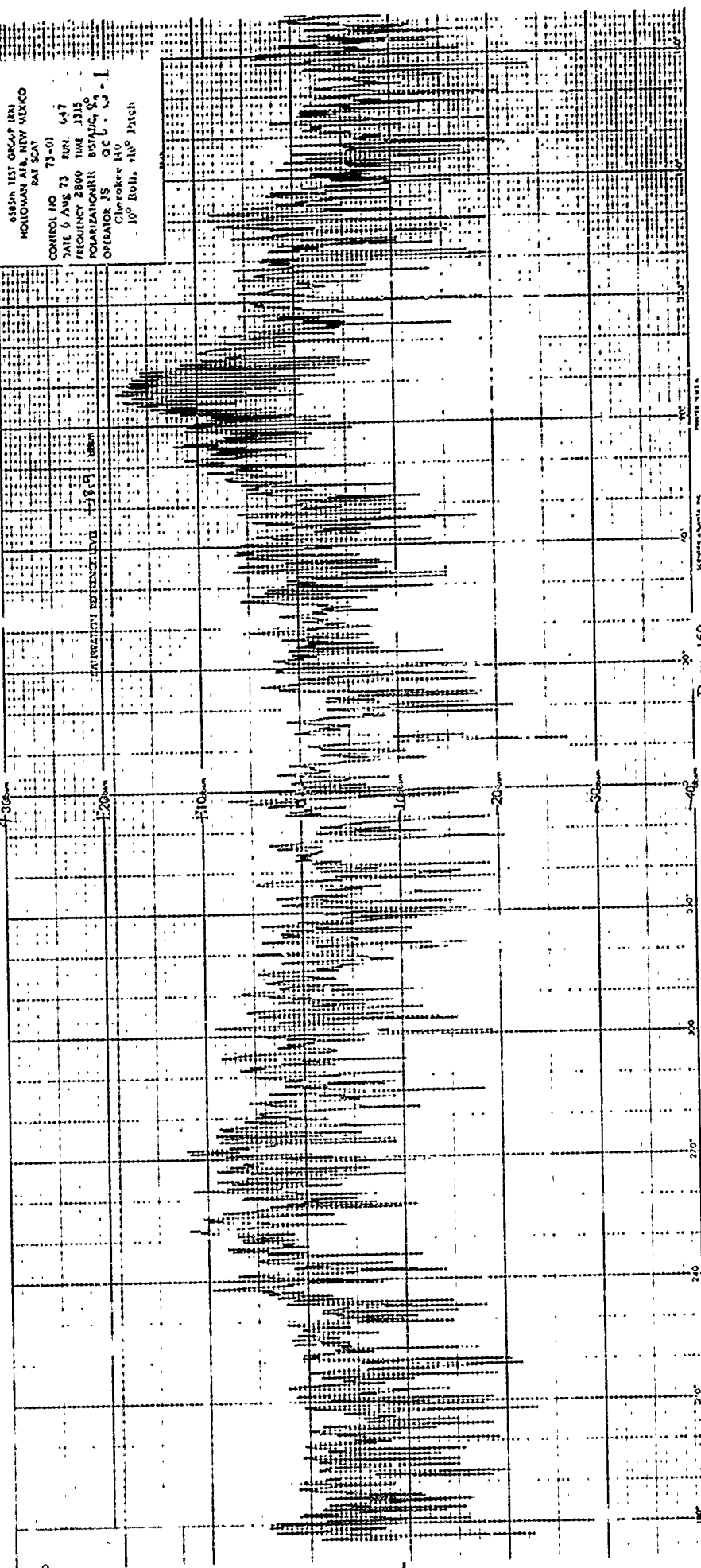
5555H TEST GROUP EXG
HOLCOMAN AFB NEW MEXICO
BAT SCAT
CONTROL NO 22-01
DATE 1 AUG 73 RUN 674
FREQUENCY 2800 MHz 1105
POLARIZATION R 57.5°
OPERATOR JS CC
Clerk: JAC
100 Rpt. 10 F.R. 2

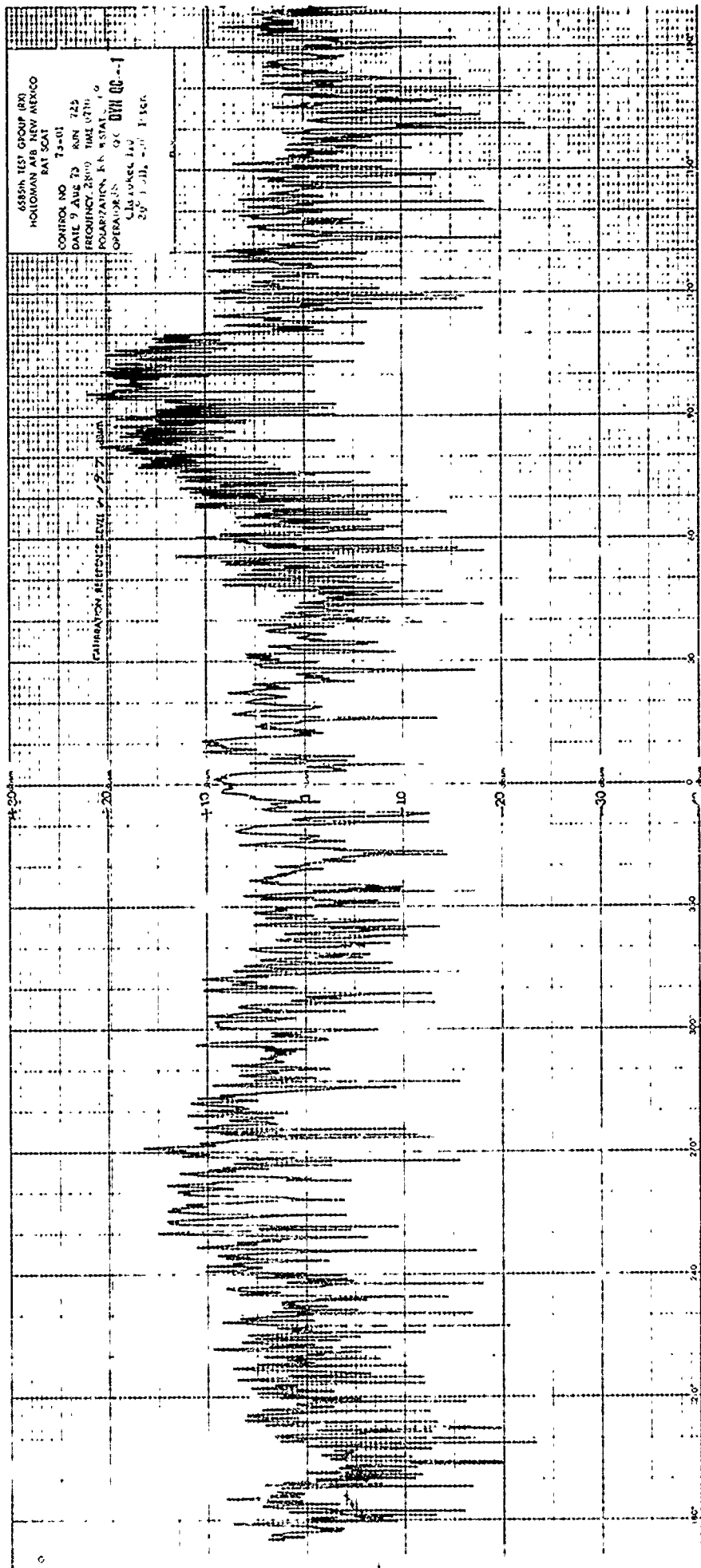


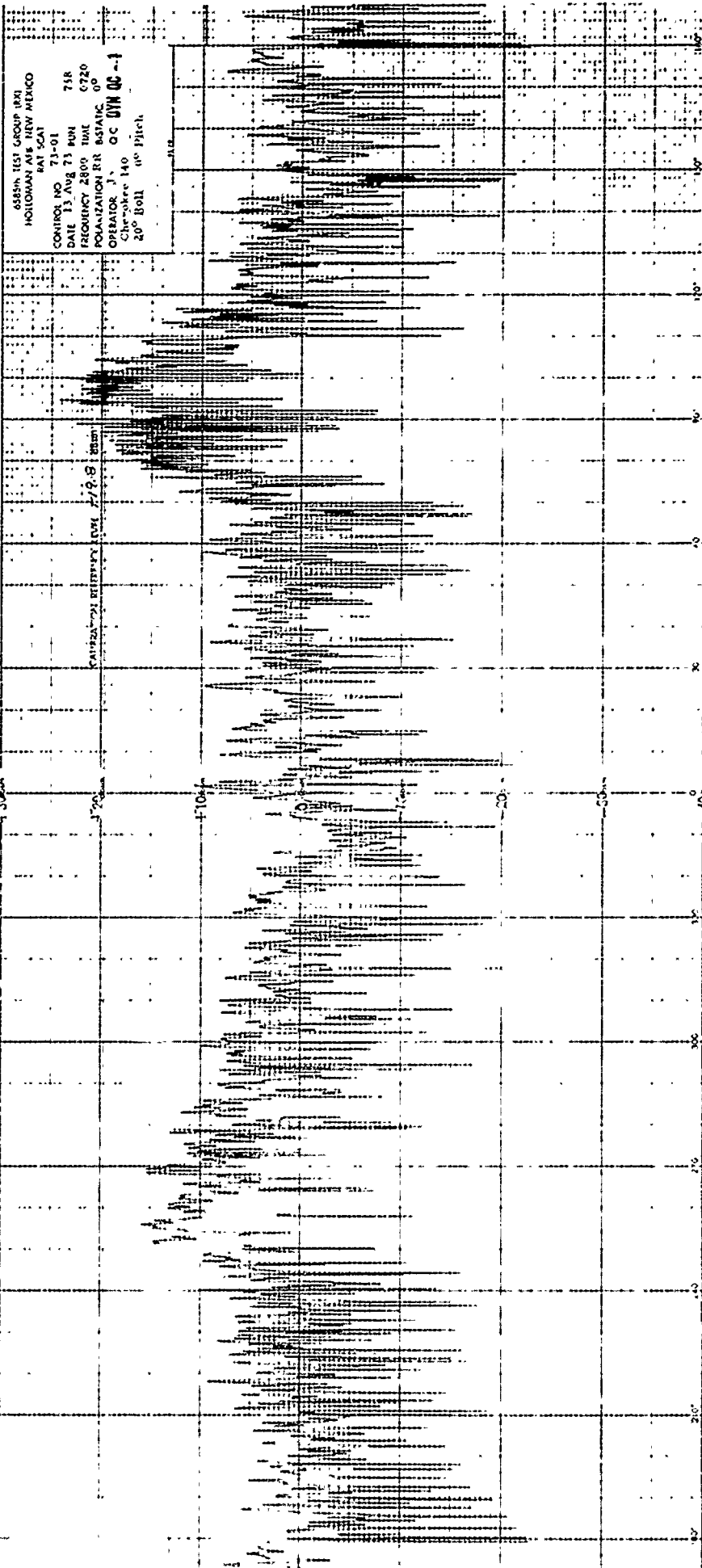
6585m 1ST GROUP 1241
HOLLANDIAN AIR NEW MEXICO
BAT SCAT

CONTOUR 10 73-00
DATE 7 Aug 73
FREQ 2500
PULSE 100
PULSE RATE 100
PULSE WIDTH 100
PULSE HEIGHT 100
PULSE AREA 100
PULSE PERIOD 100
PULSE DUTY 100
PULSE RISE 100
PULSE FALL 100
PULSE SLOPE 100
PULSE CURV 100
PULSE PHASE 100
PULSE MOD 100
PULSE AM 100
PULSE FM 100
PULSE PM 100
PULSE CM 100
PULSE DM 100
PULSE SM 100
PULSE TM 100
PULSE LM 100
PULSE HM 100
PULSE VM 100
PULSE WM 100
PULSE YM 100
PULSE JM 100
PULSE IM 100
PULSE OM 100
PULSE UM 100

6555H 11ST GRAP IRI
 HOLLAND AFB, NEW MEXICO
 BAT SCAT
 CONTROL NO 73-01
 DATE 6 AUG 73 RUN 647
 FREQUENCY 2800 TIME 1315
 FORMATION 00
 OPERATOR JS
 Cherokee Mo
 10° Roll, 110° Pitch

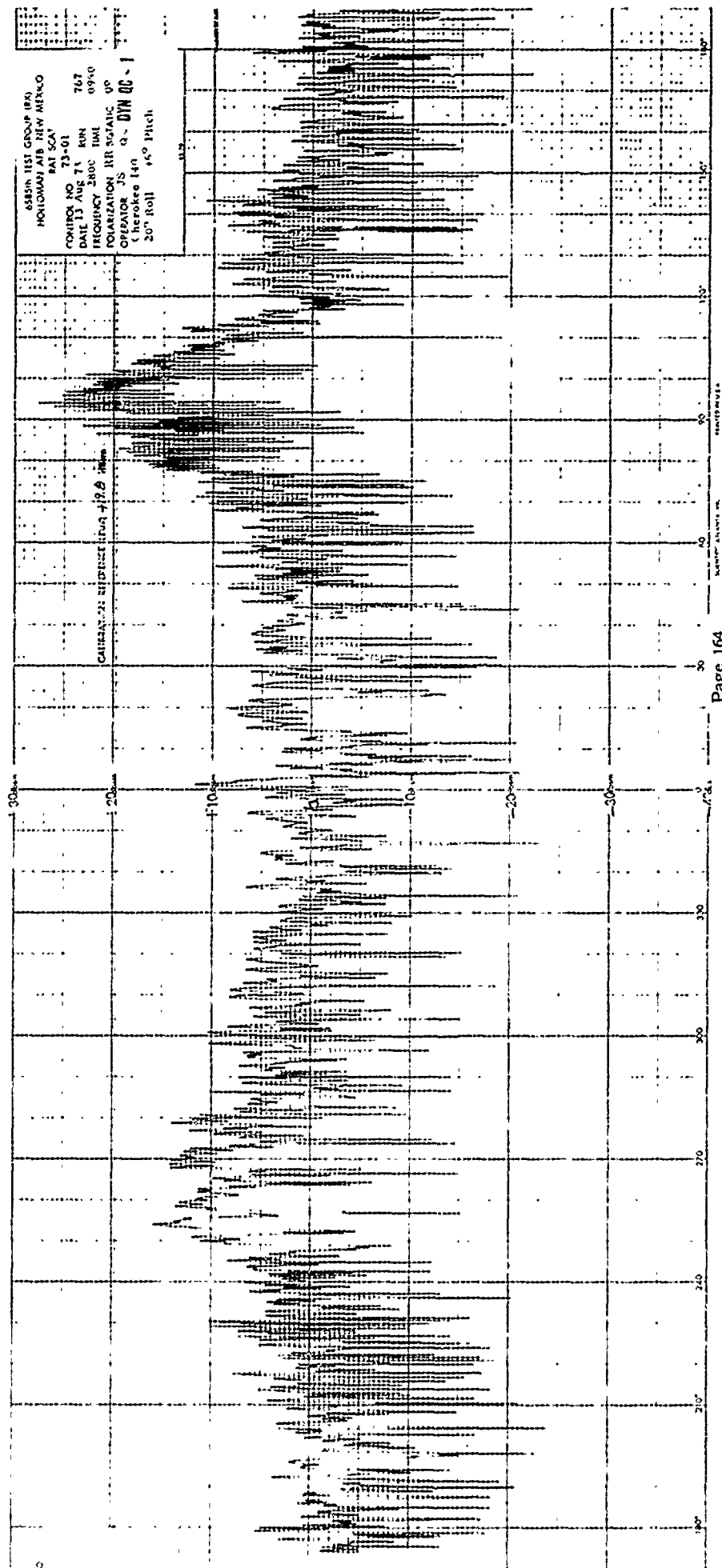


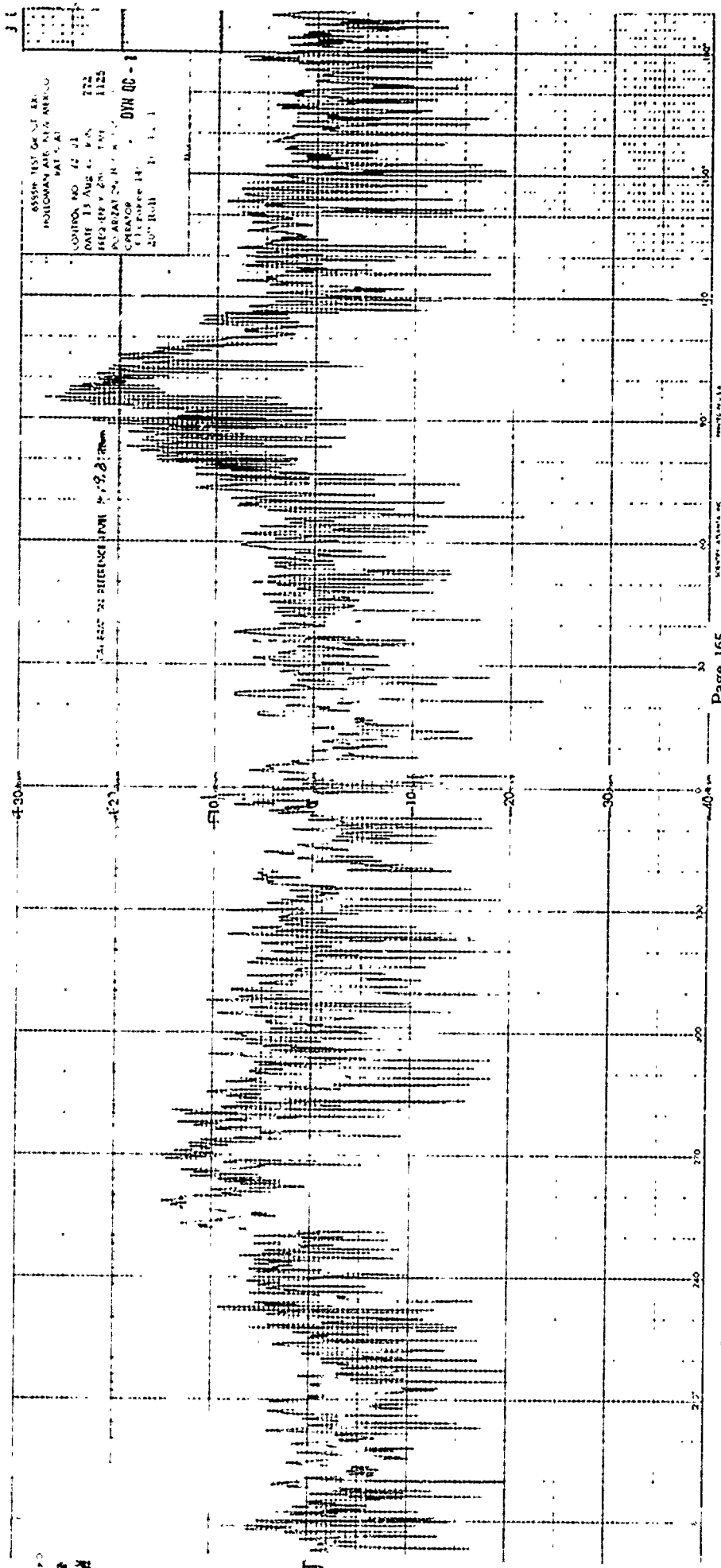




ASSIN TEST GROUP 121
HOLLOMAN AFB NEW MEXICO
BAT CAT
CONTRON NO. 73-01 758
DATE 13 Aug 73 PMH 0720
FREQUENCY 2800 HZ
POLARIZATION RR B3A1C 0°
OPERATOR J. GC DYN QC -1
Chamber 140
20" Roll 40" Pitch

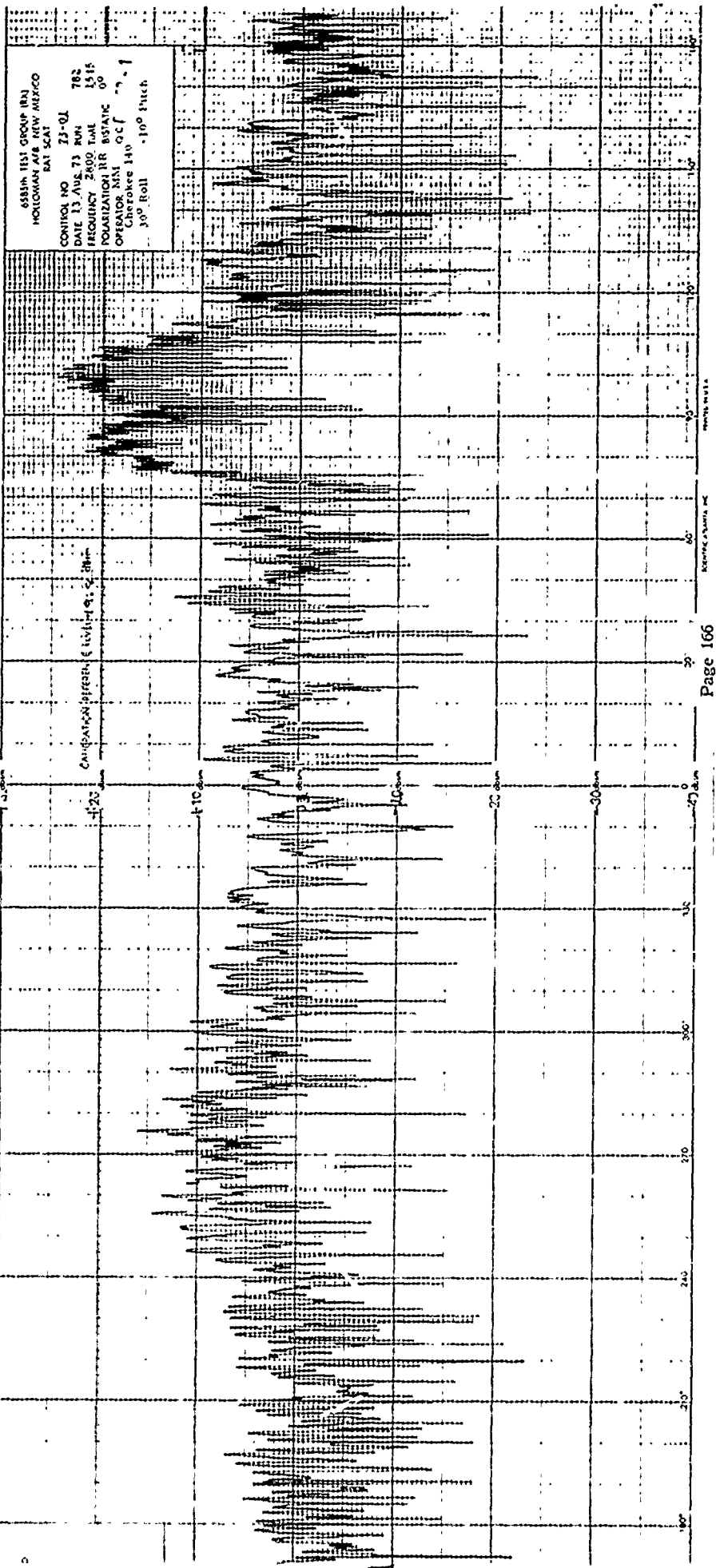
ASESIN TEST GROUP (PA)
 HOLLANDAY AFB NEW MEXICO
 PAT SCA
 CONTROL NO. 73-01 767
 DATE 13 AUG 73 MON 09:40
 FREQUENCY 2800 KHz
 POLARIZATION RH SIGALIC UP
 OPERATOR JS 0 - DYN QC - 1
 Character 14n 0 Pitch
 20° Roll 45° Pitch

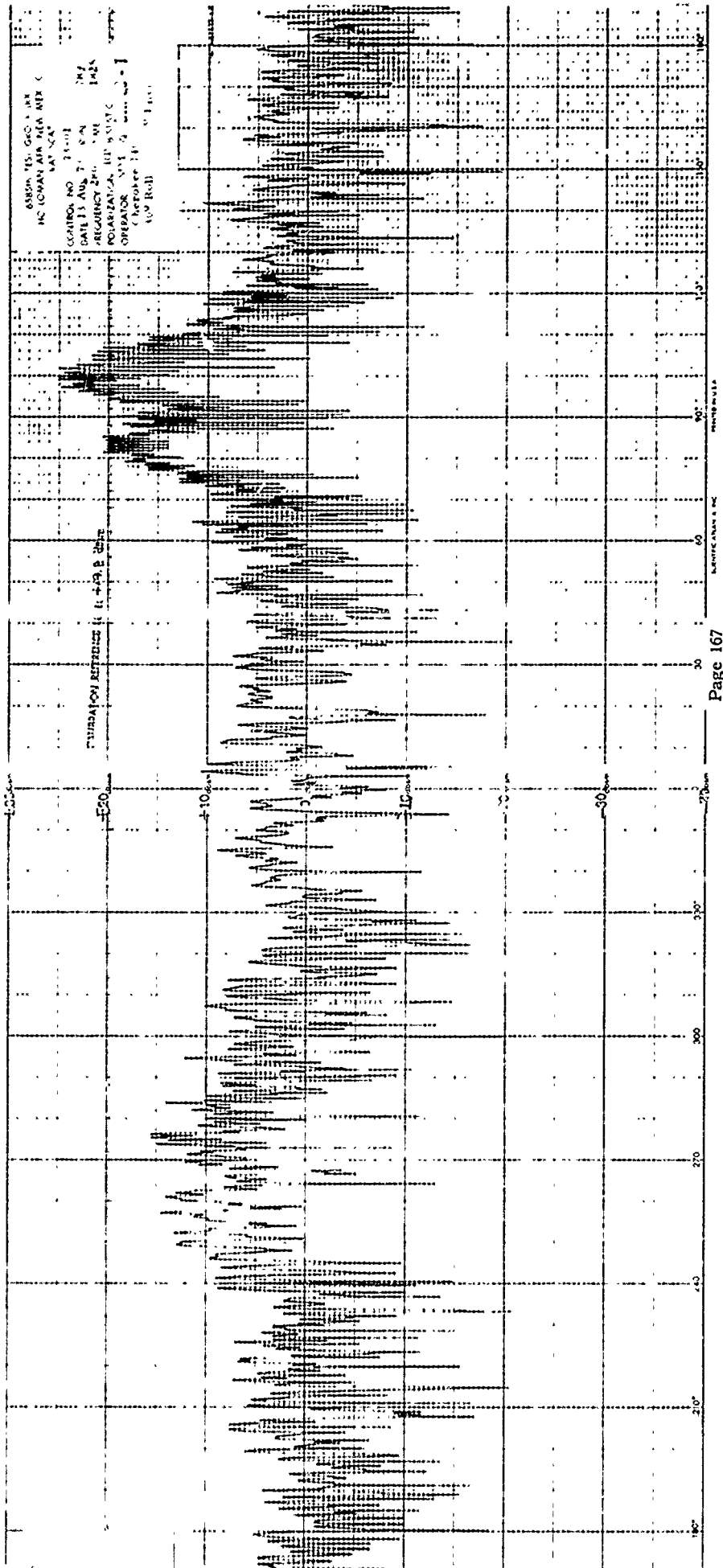


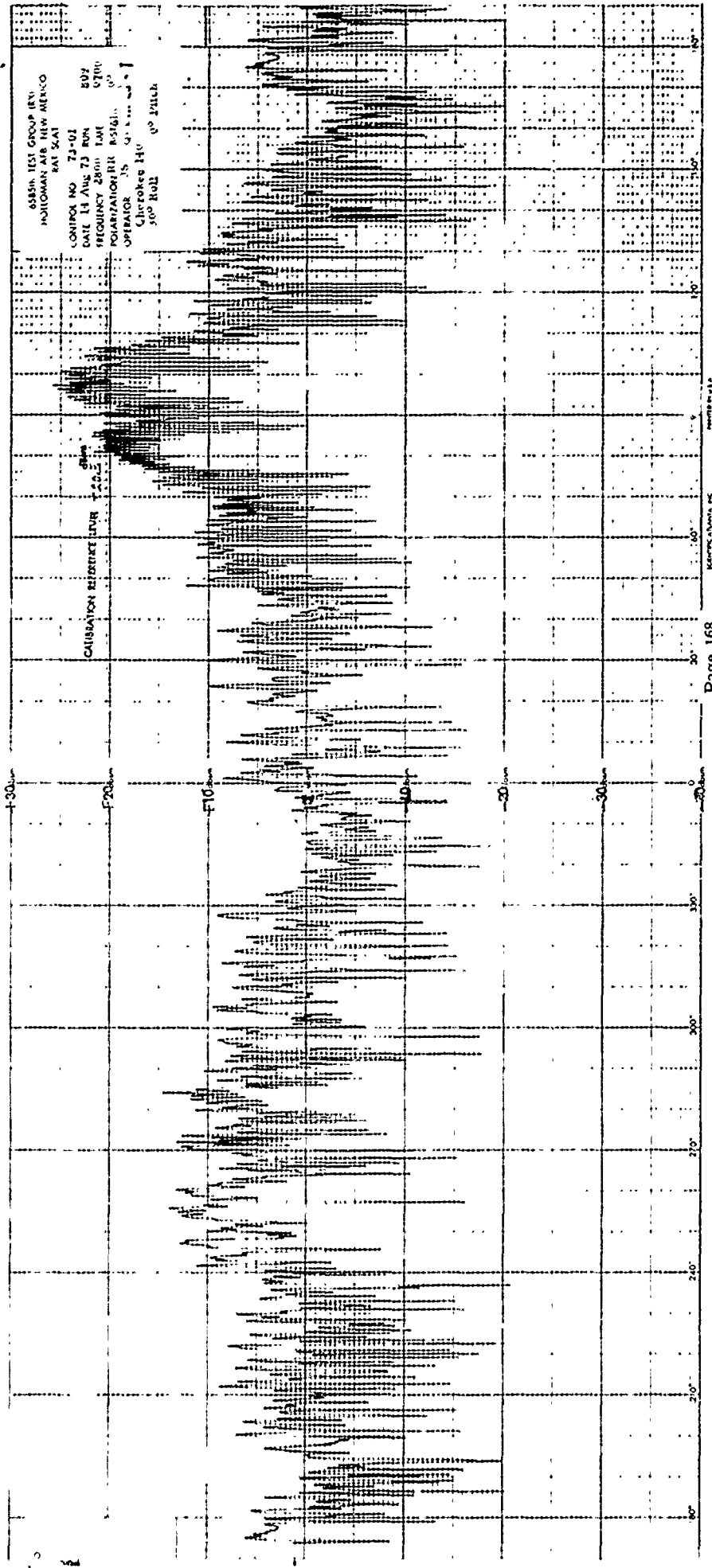


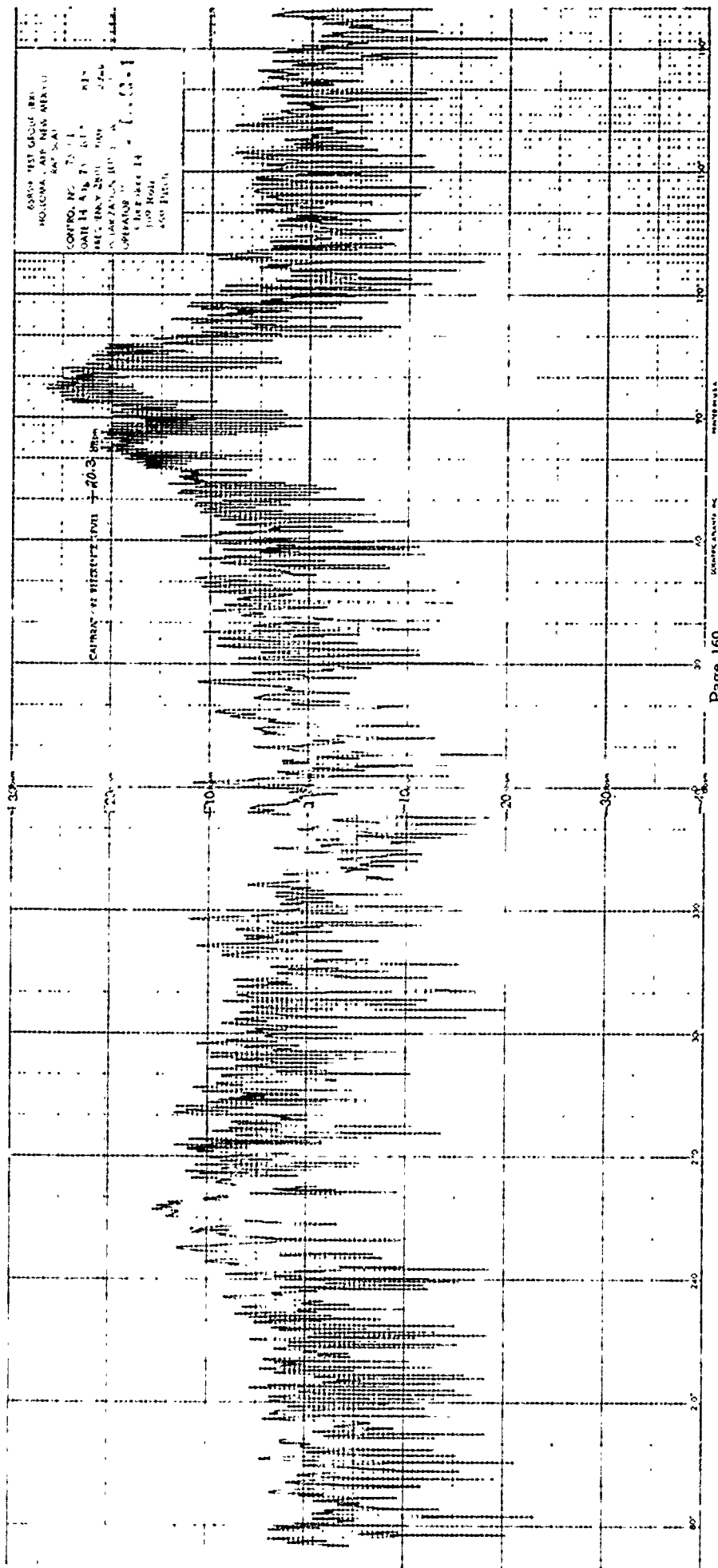
ASSISTANT TEST OF THE AIR
POLICEMAN ARE IN A MIND
DATE 11 JUL 62
CONTROL NO 1125
DATE 11 JUL 62
RECEIVED 11 JUL 62
CREATOR
207 Roll 11 11 11

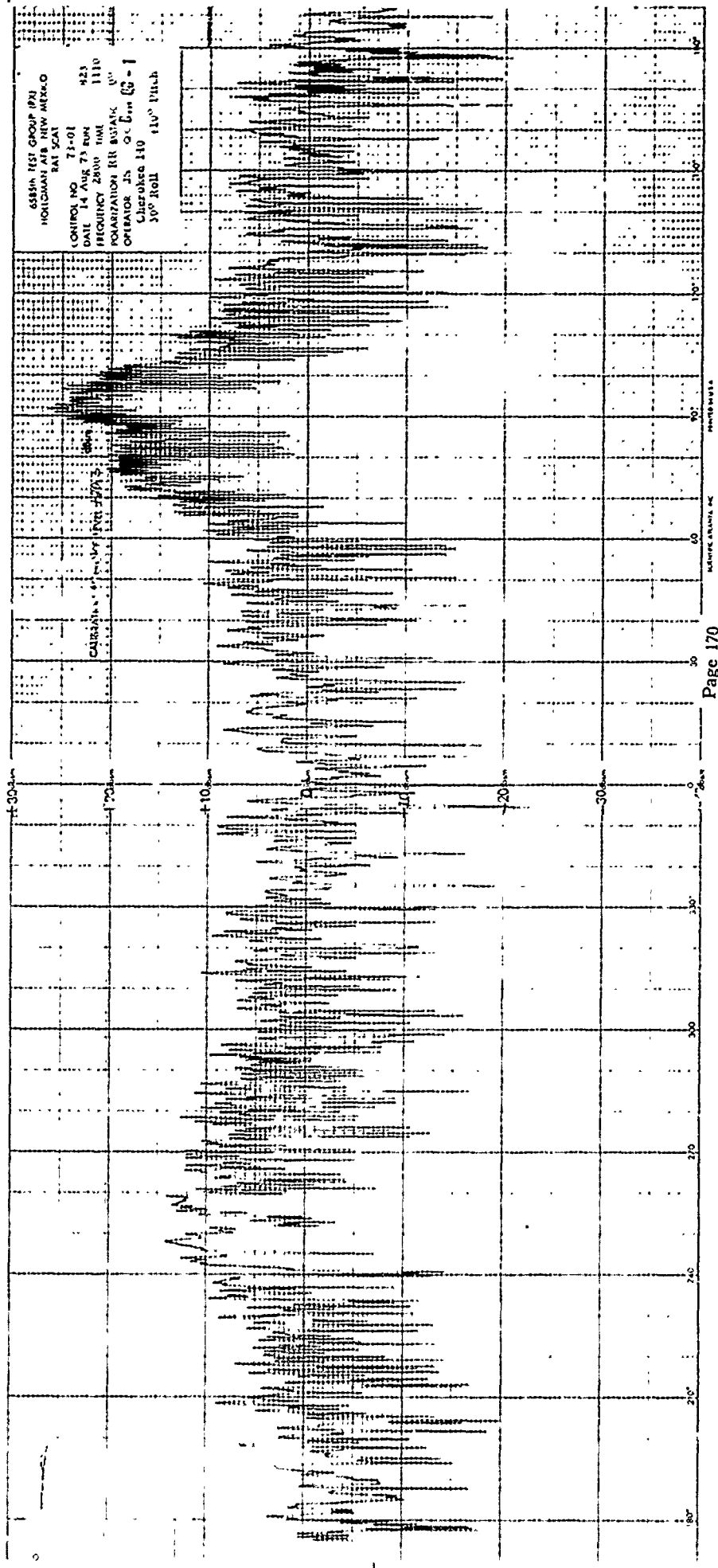
DATE 11 JUL 62

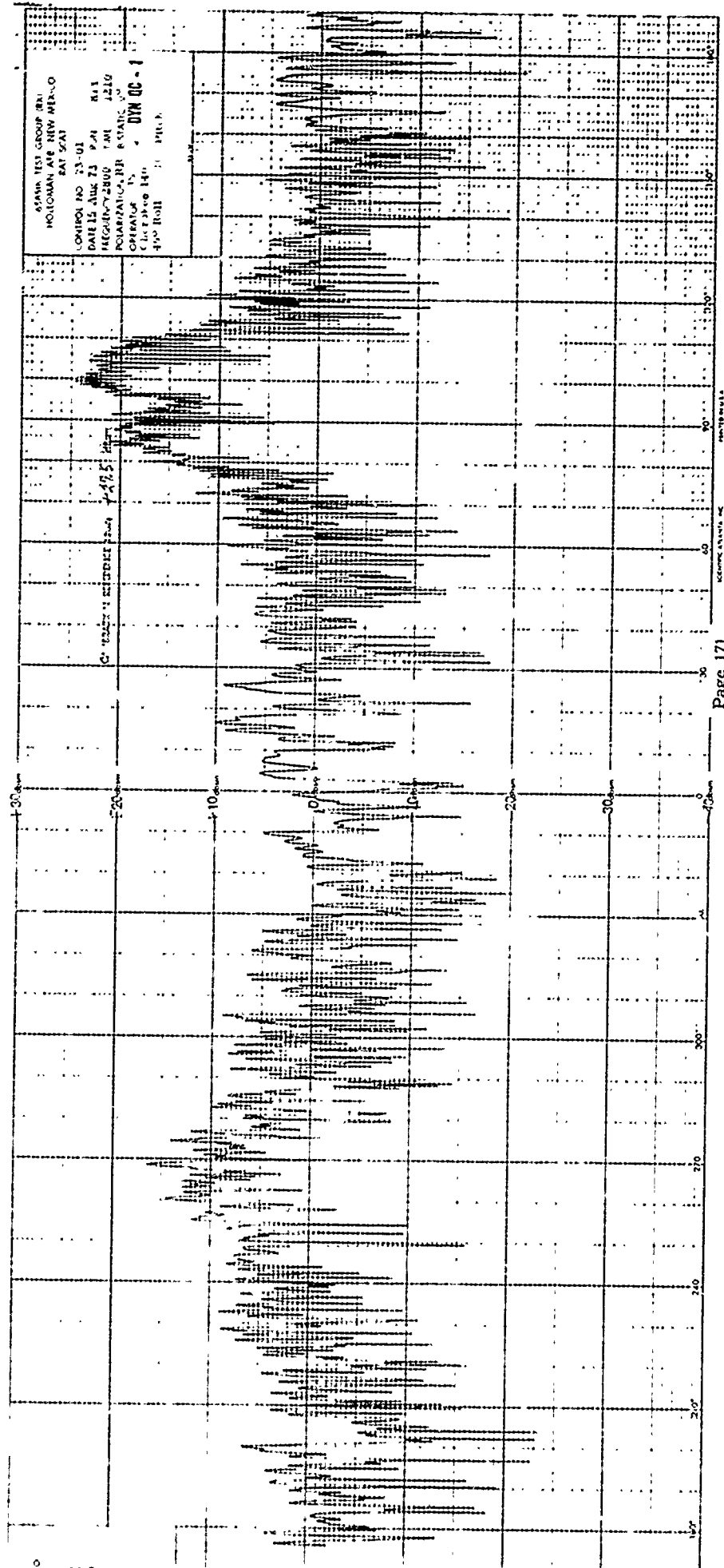




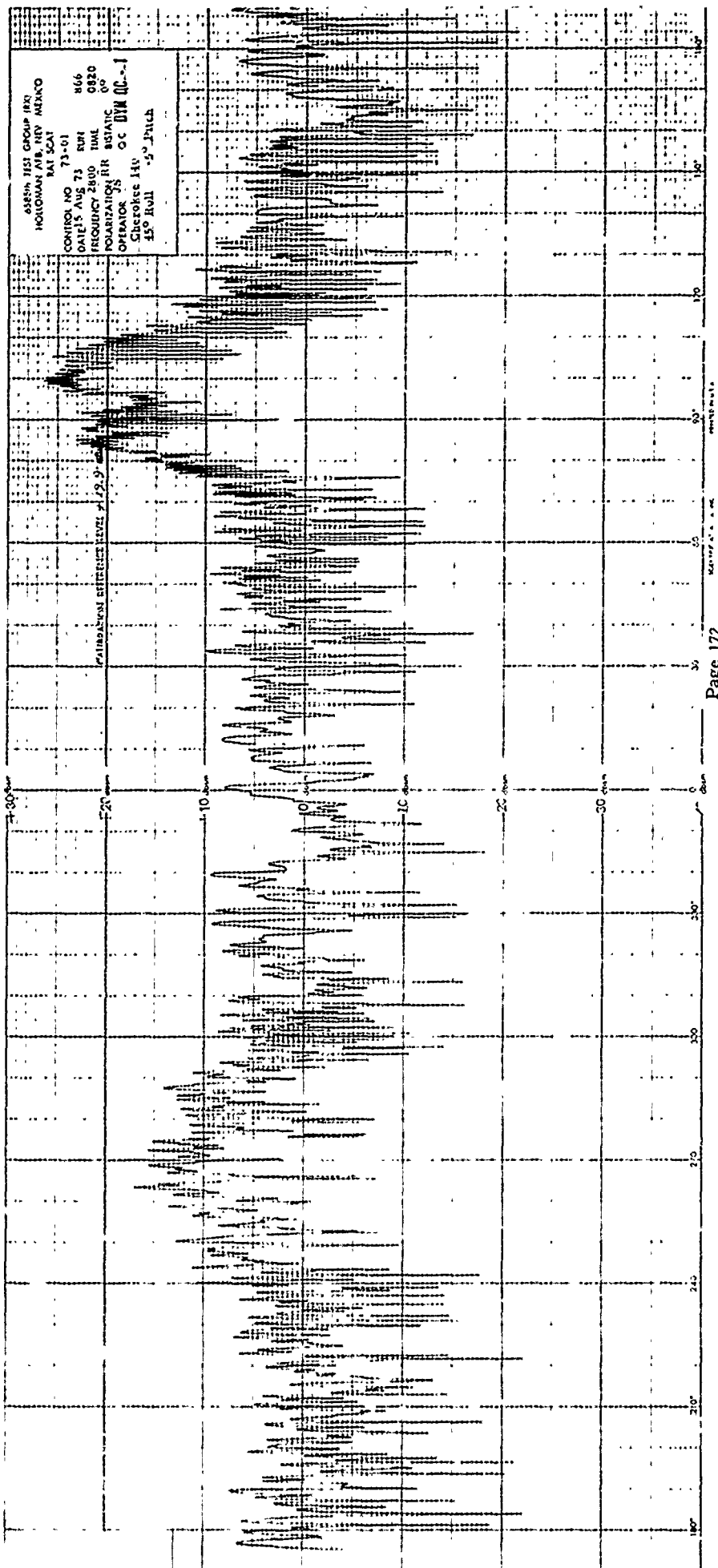


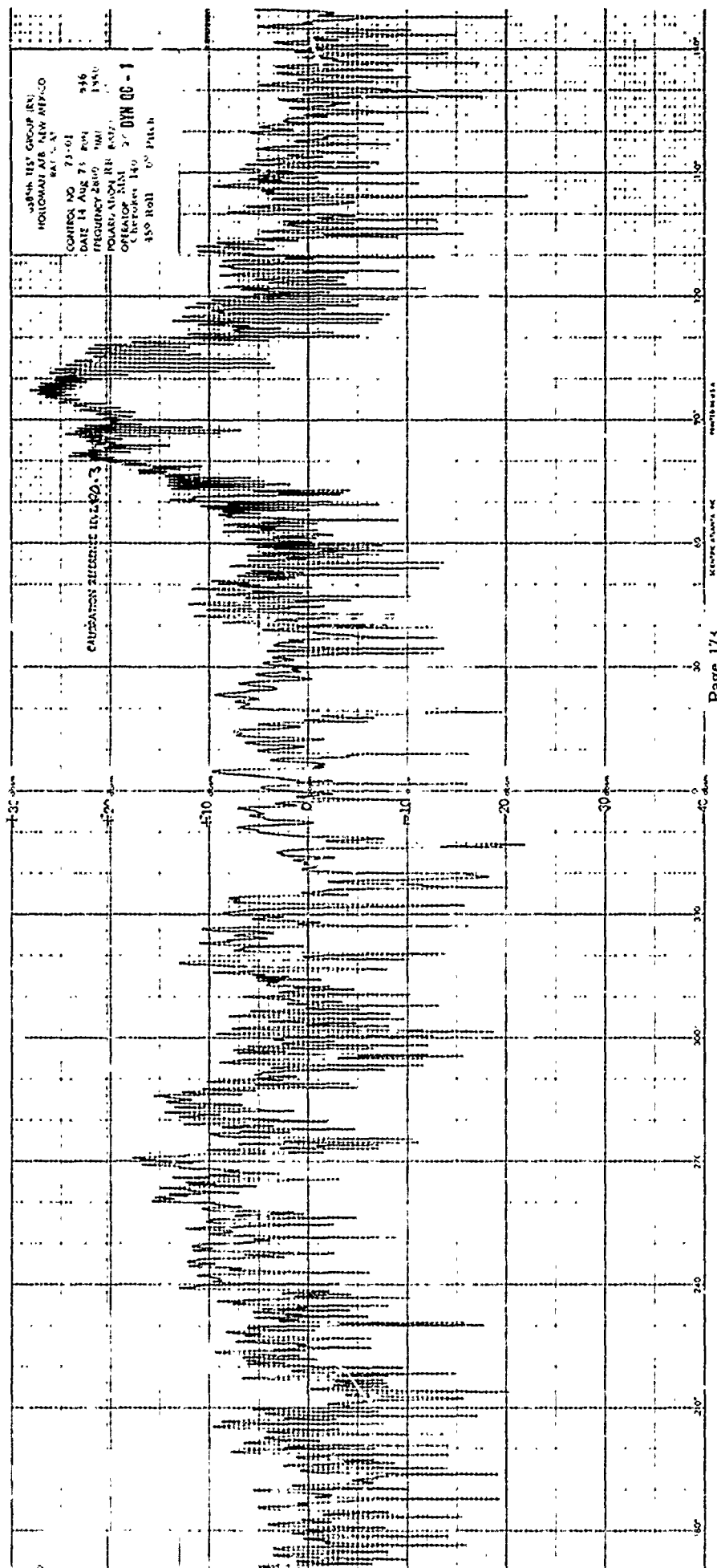


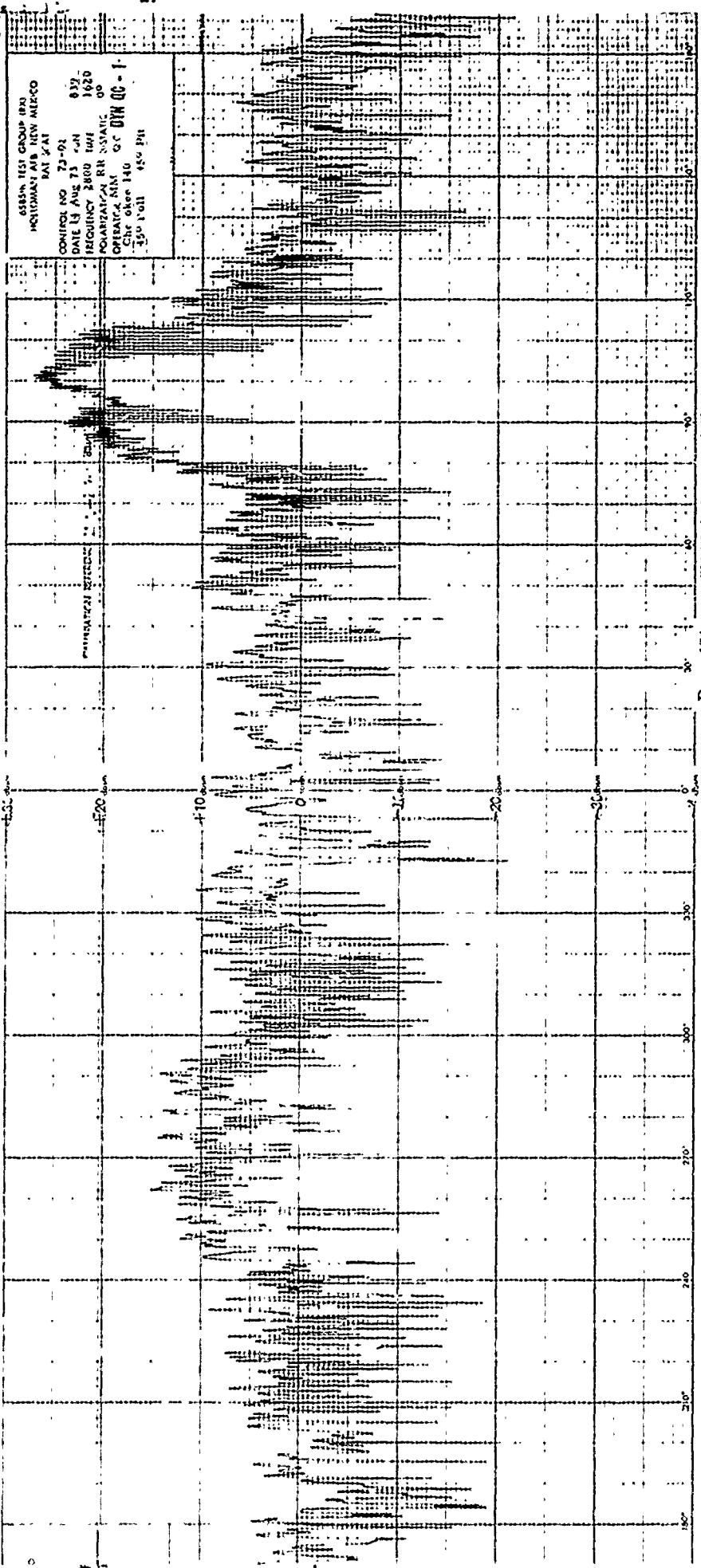




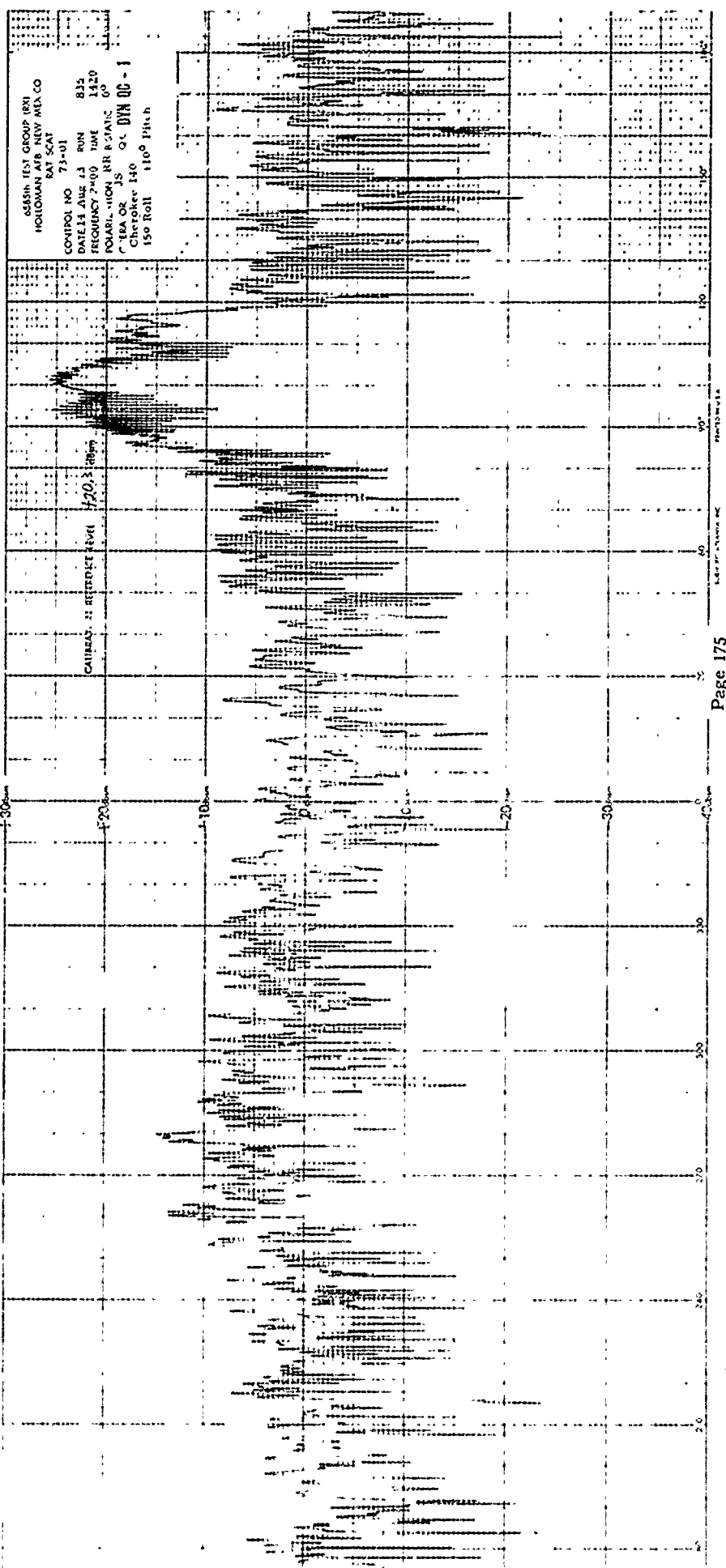
ATANK TEST GROUP (B)
HOLLOMAN AIR NEW MEXICO
EAT 5/21
CONTROL NO 23-01
DATE 15 AUG 73 P.M. 11.1
RECORDING UNIT 1210
POLARIZATION UNIT 1210
OPERATOR 1210
450 Roll 1210

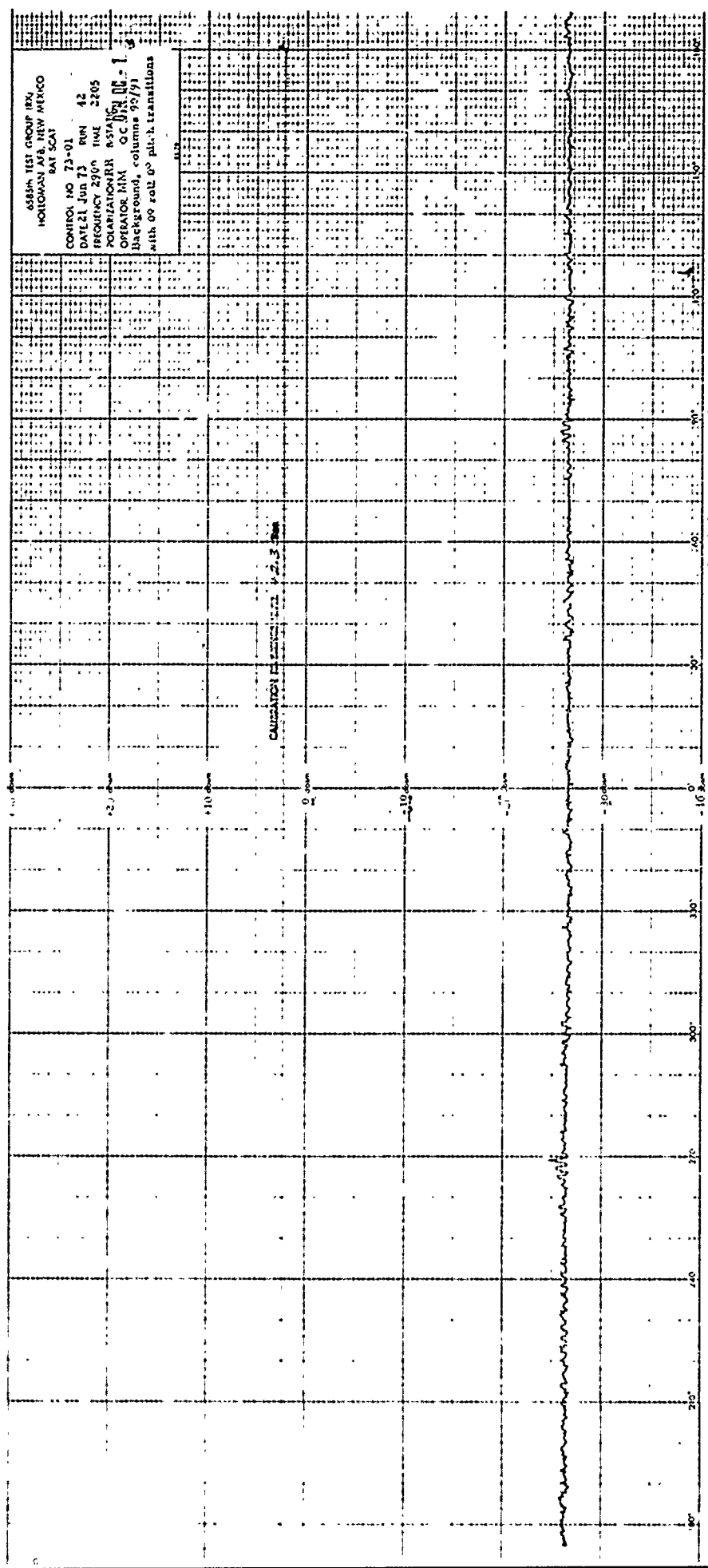


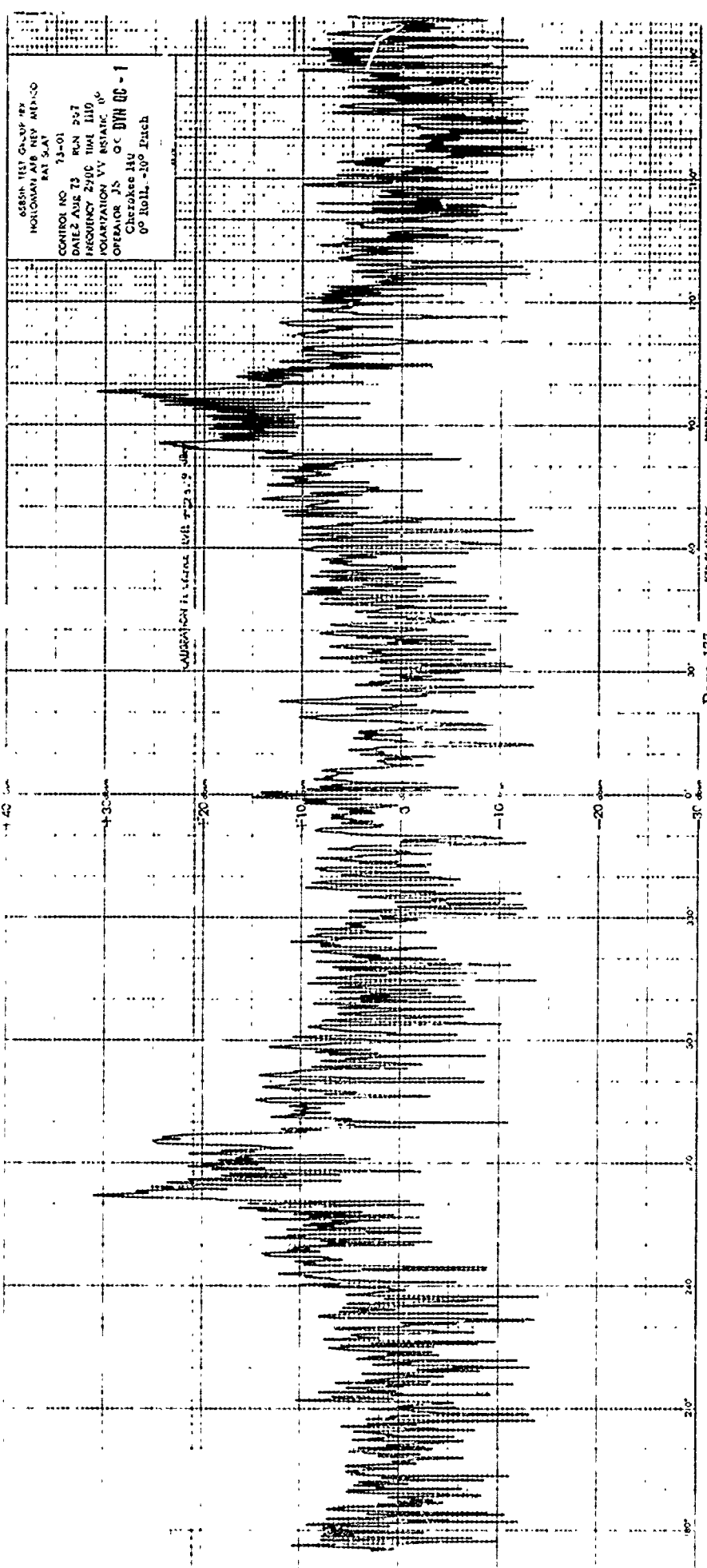


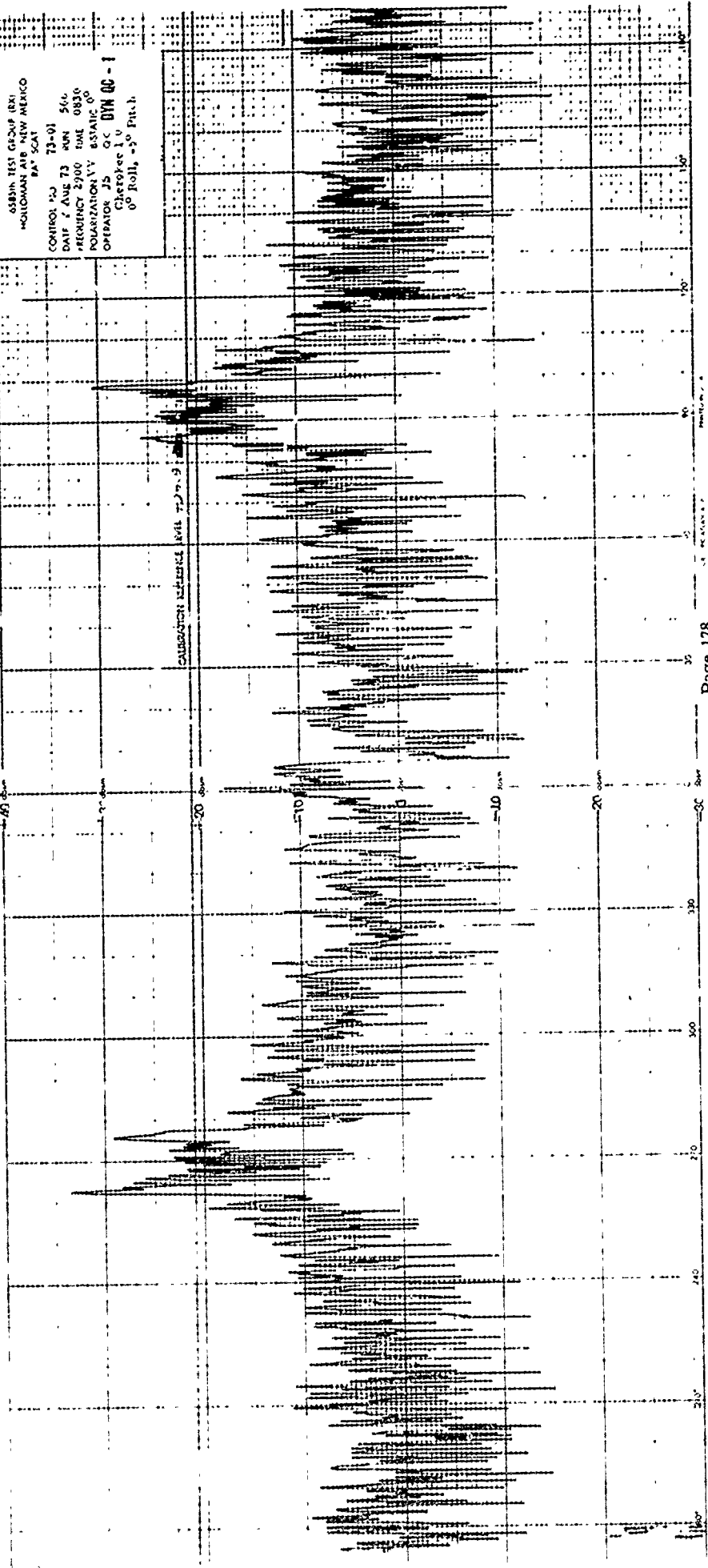


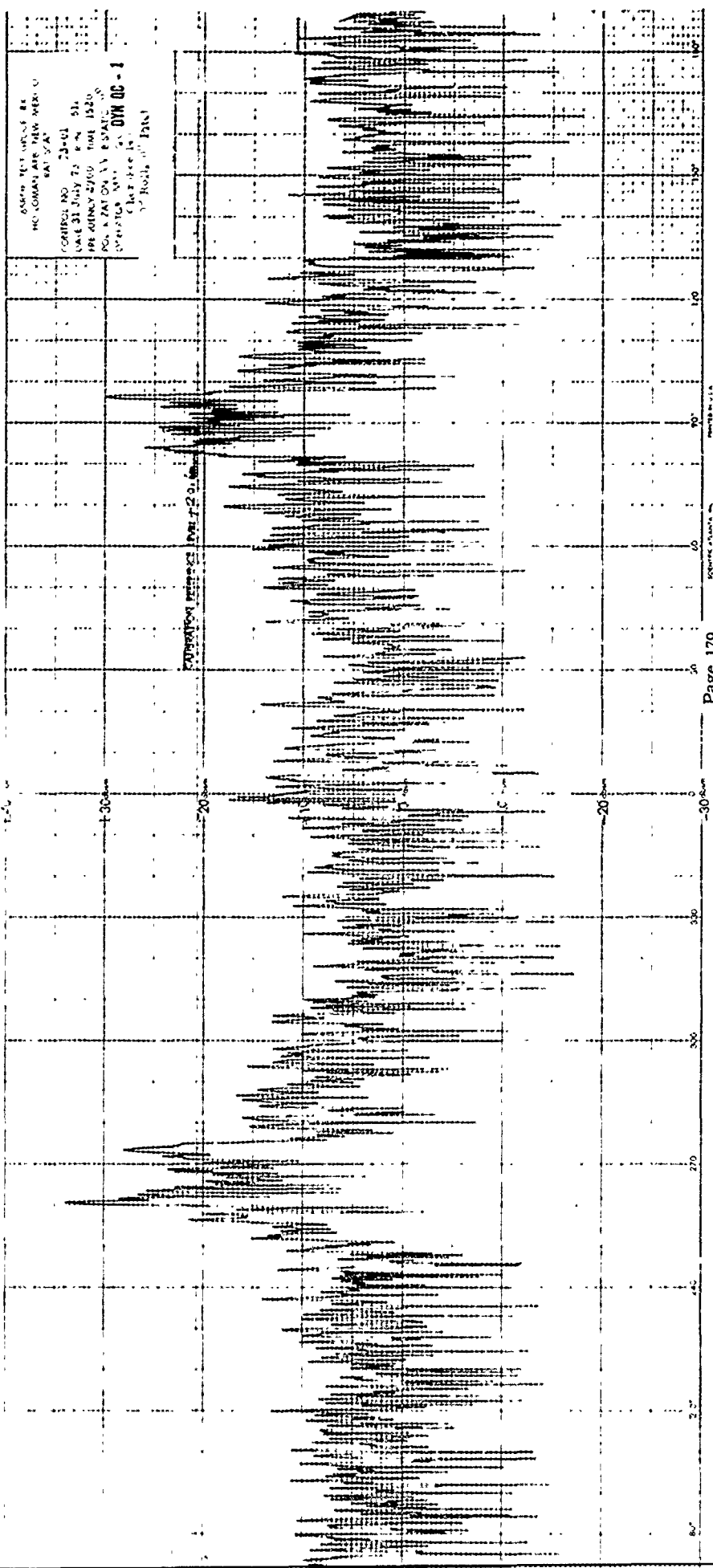
6553N TEST GROUP 183
HOLCOMB AFB NEW MEXICO
DATE 13 AUG 73
CONTROL NO 73-01
FREQUENCY 2800 MHz
POLARIZATION RR
OPERATOR MMT
Cite other 140
350 Pct 450 Pct
QC DYN QC - 1

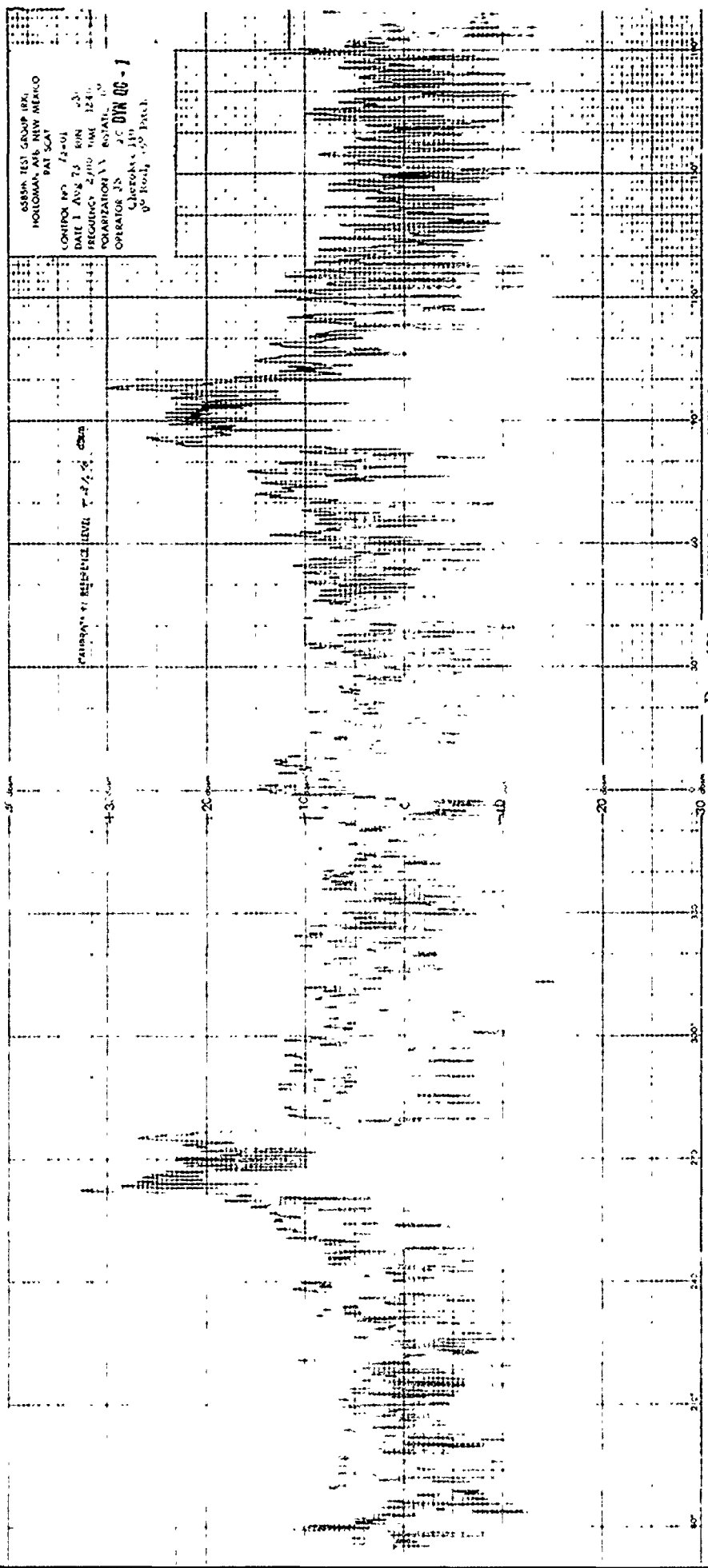


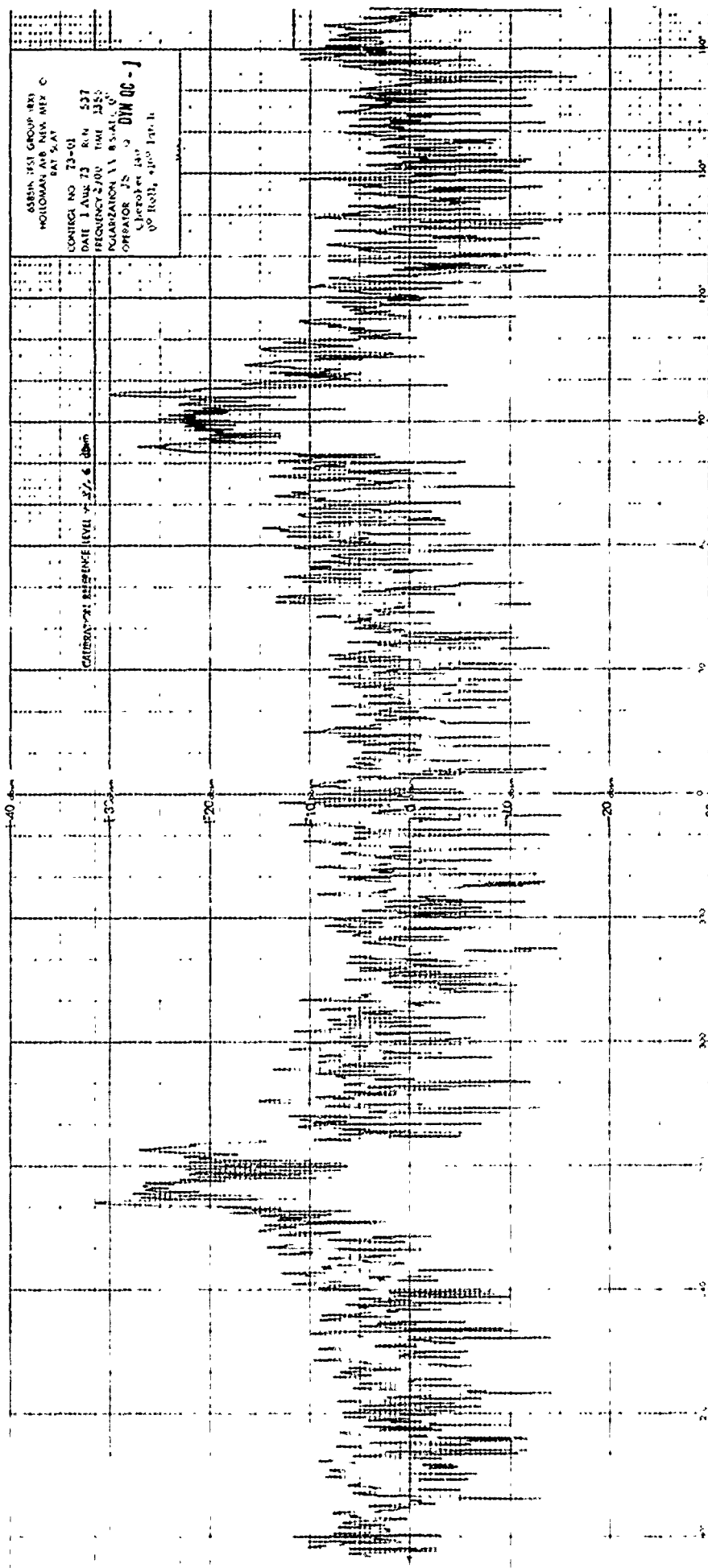


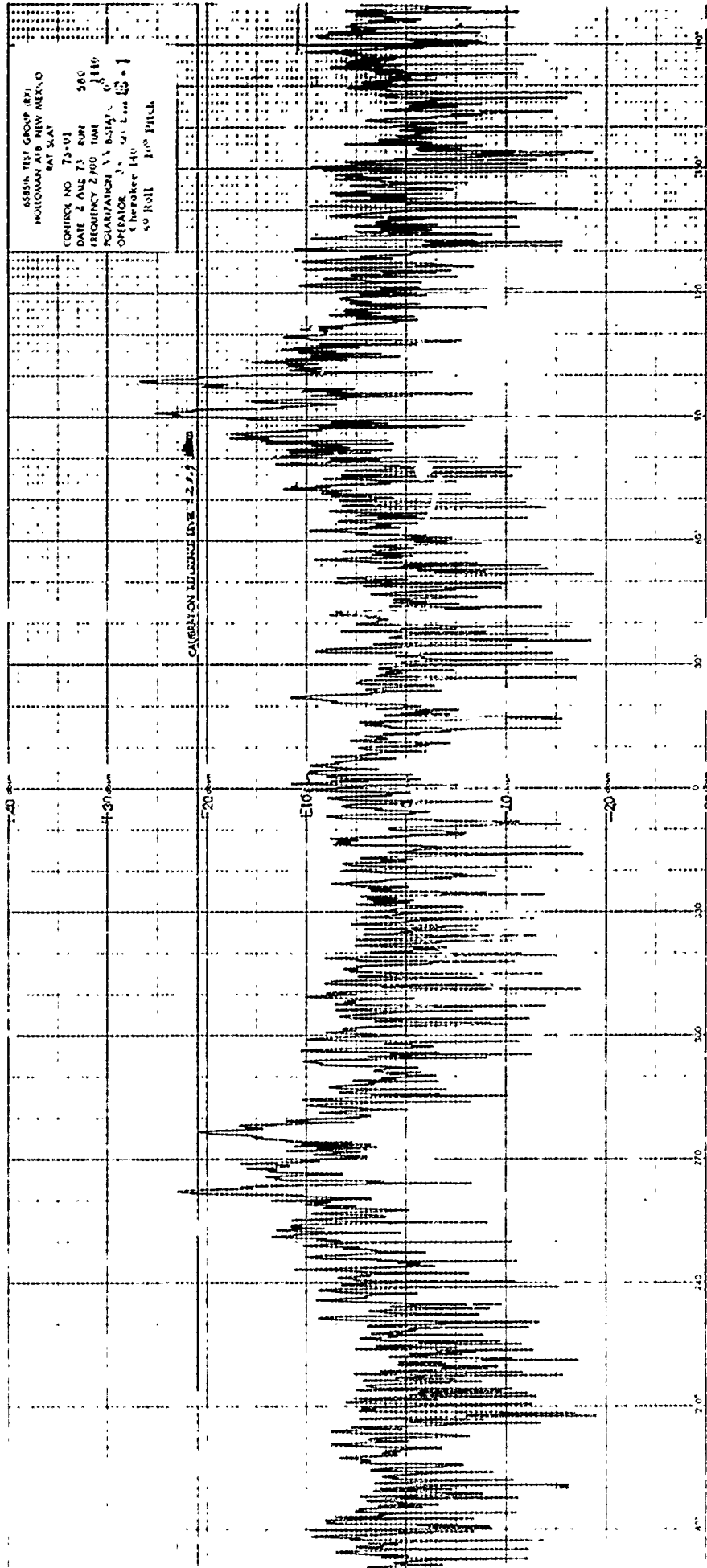












6555N TEST GROUP (B7)
HOLCOMB, NEW MEXICO
DATE 2-2-01
CONTRACT NO. 73-01
DATE 2 AUG 73 RUN 560
FREQUENCY 2.000 HZ
POLARIZATION VV B3145
OPERATOR J. S. G. L. M. 65-1
Character 140
50 Roll 100 Pitch

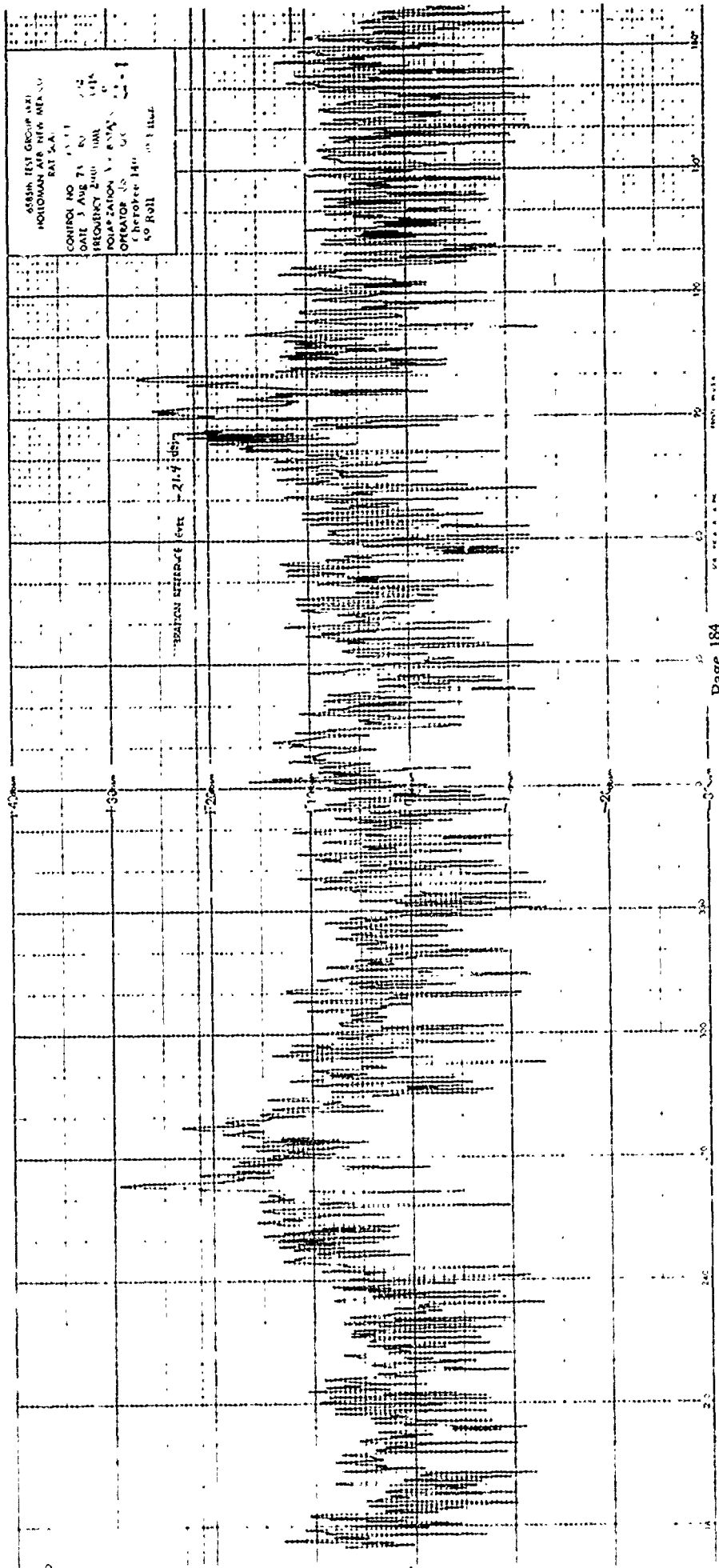
CALIBRATION REFERENCE LINE

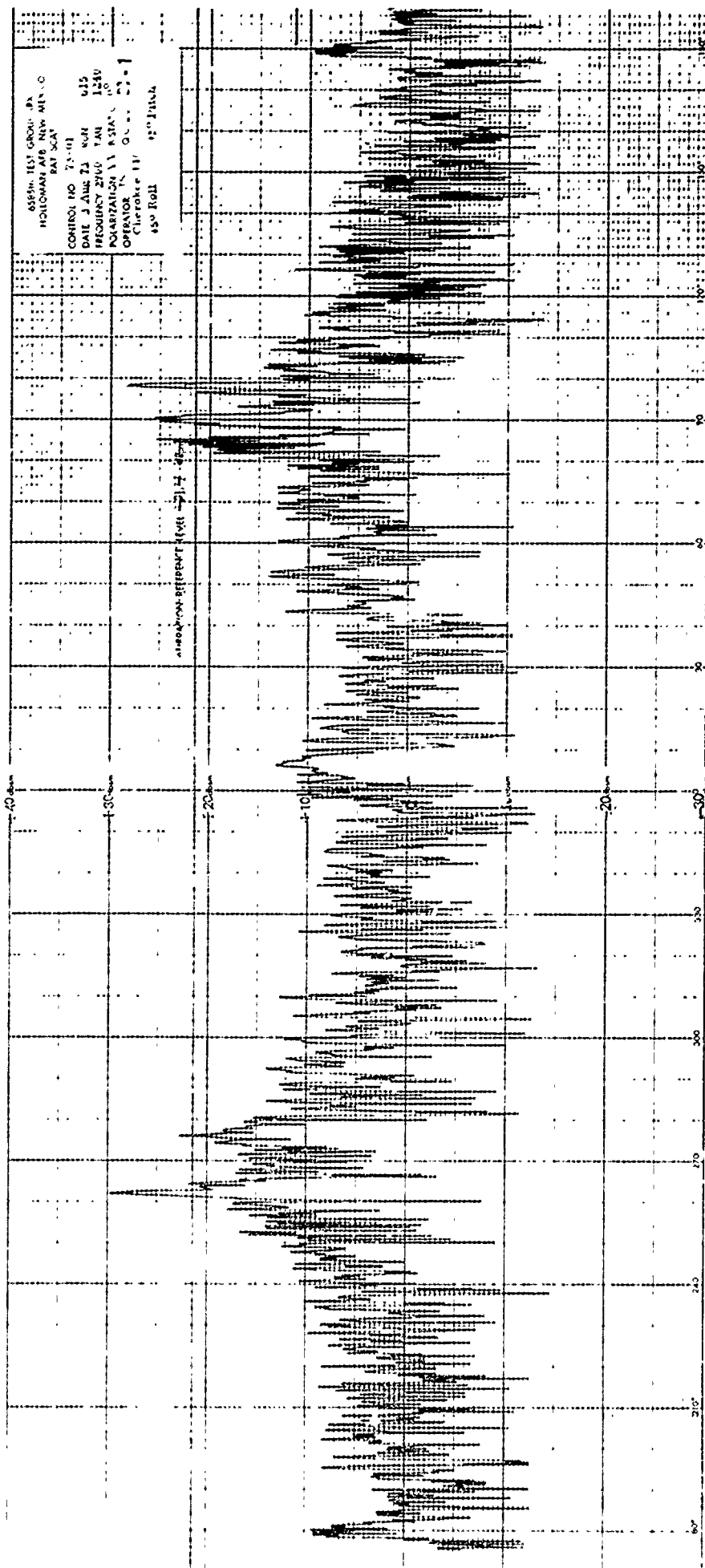
ASSN TEST GROUP (AK)
HOUQUAN, ALA
RAT 5, AT
CONTROL NO 73-01
DATE AUG 73
FREQ 2700
POLARIZATION VV
OPERATOR TS
Cherokee 140
50 Roll
50 Pitch

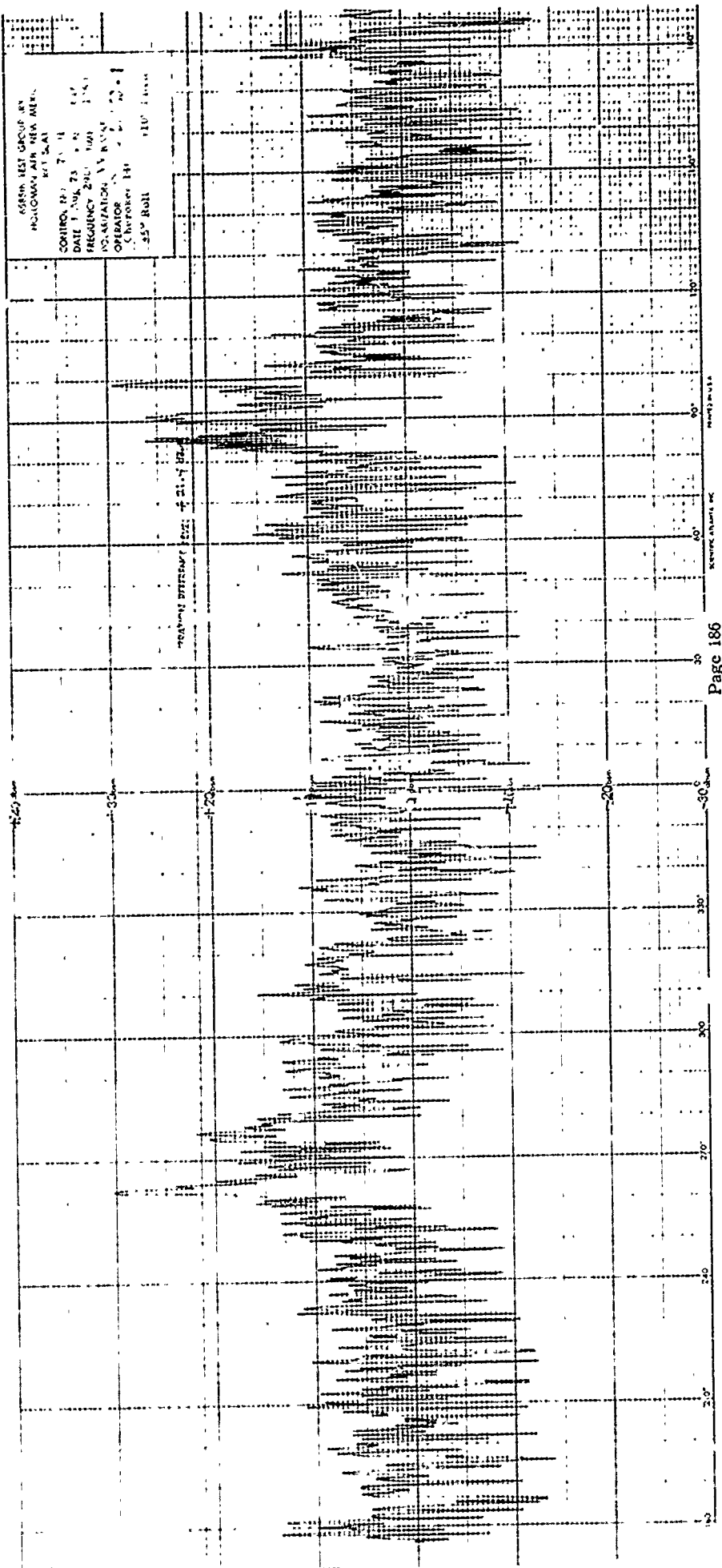
STIMULATION REFERENCE LEVEL 5.21: 4.00 dBm

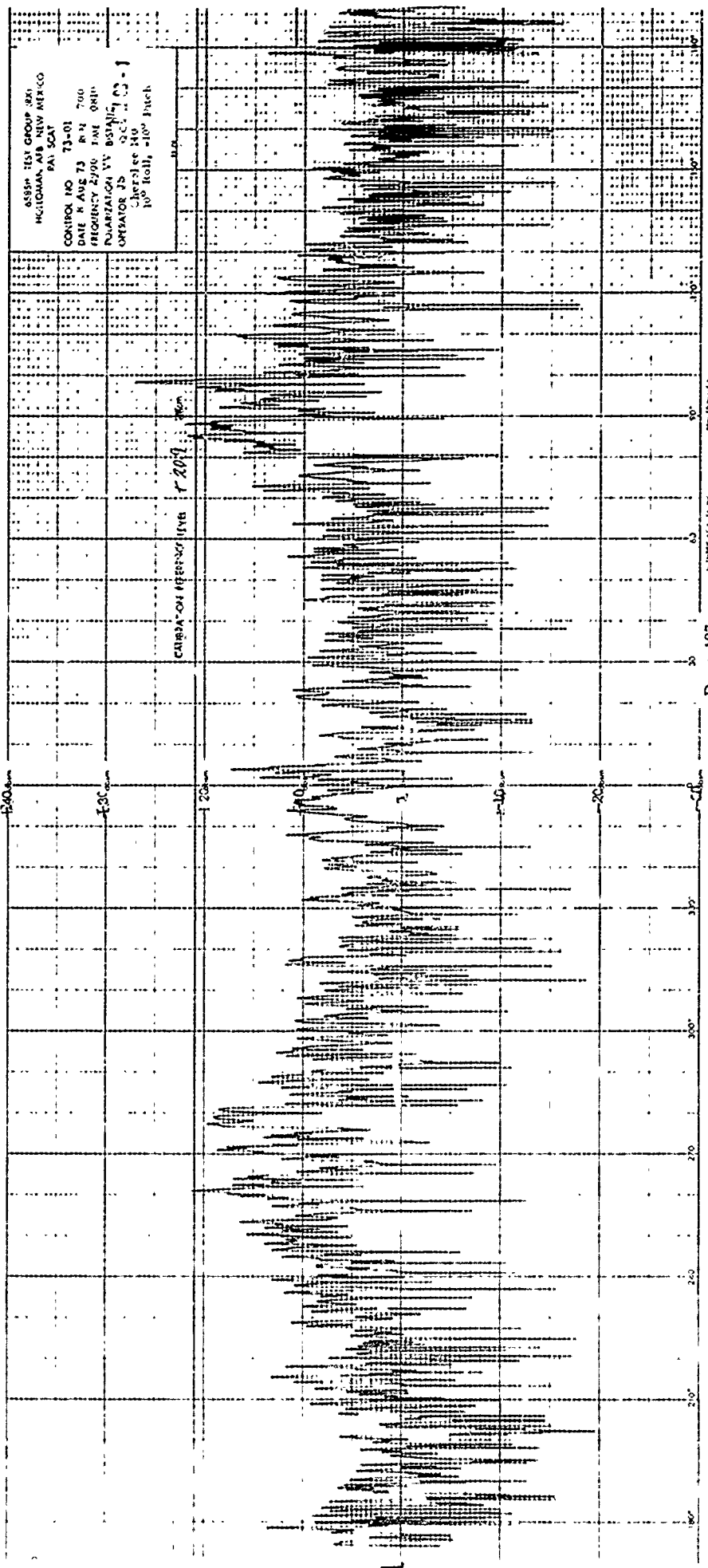
4.00 dBm
3.00 dBm
2.00 dBm
1.00 dBm
0.00 dBm

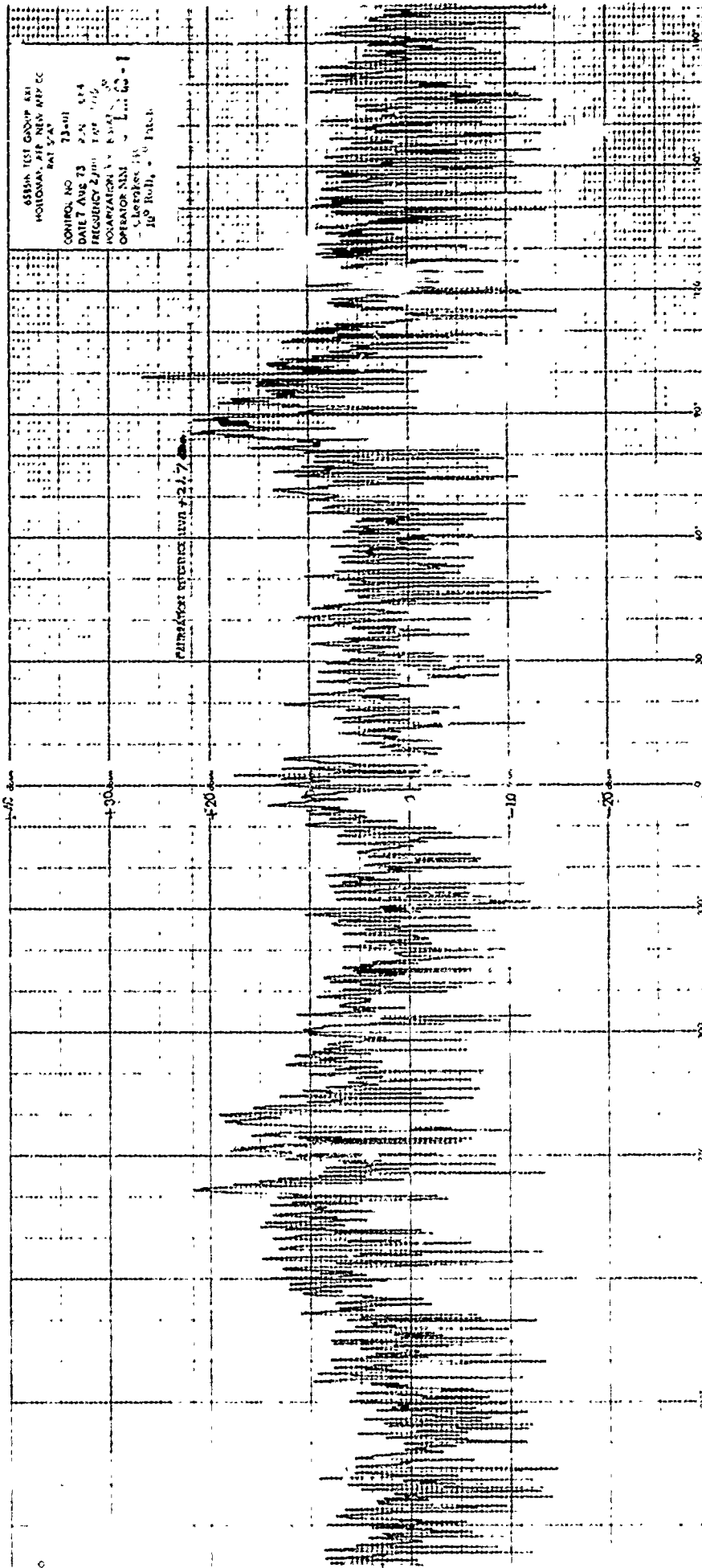
210° 220° 230° 240° 250° 260° 270° 280° 290° 300° 310° 320° 330° 340° 350° 360°

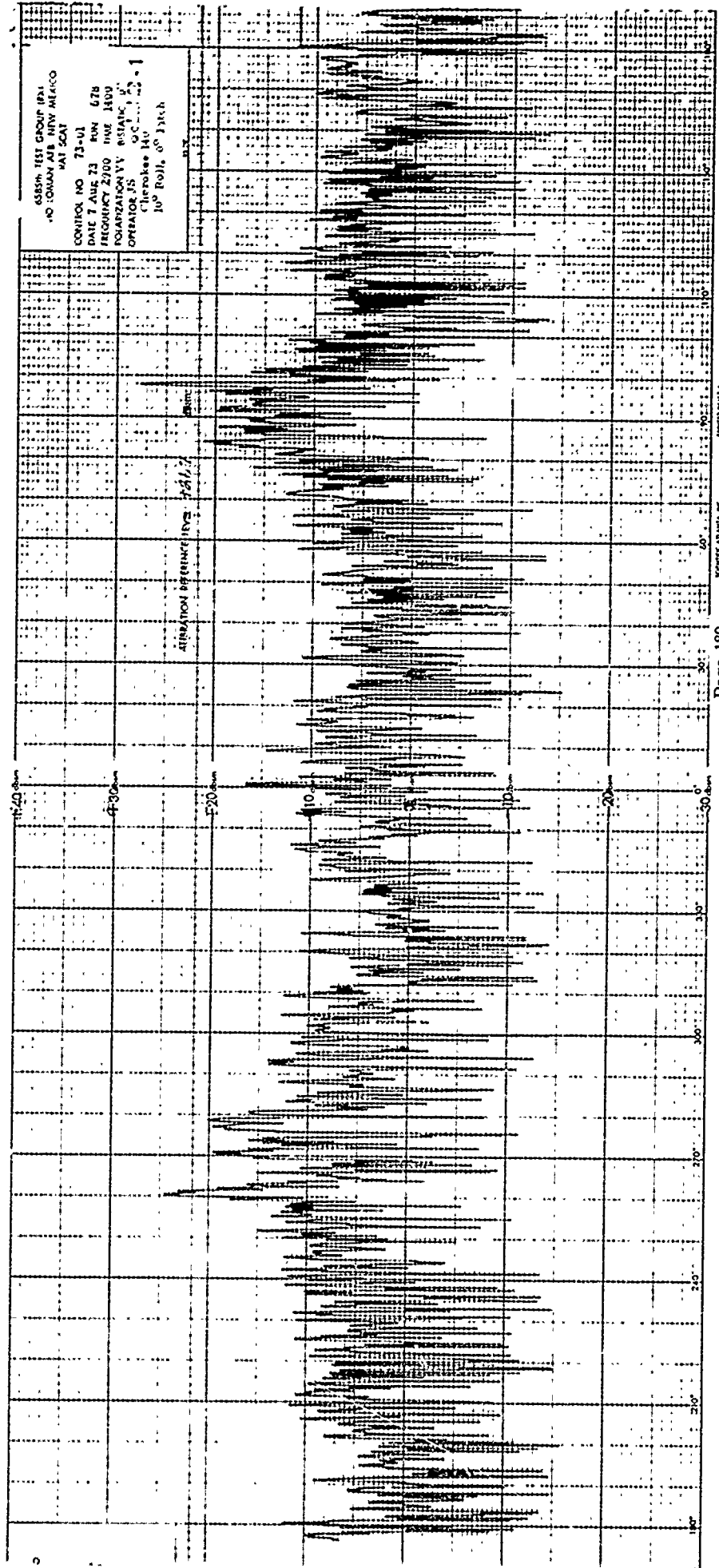


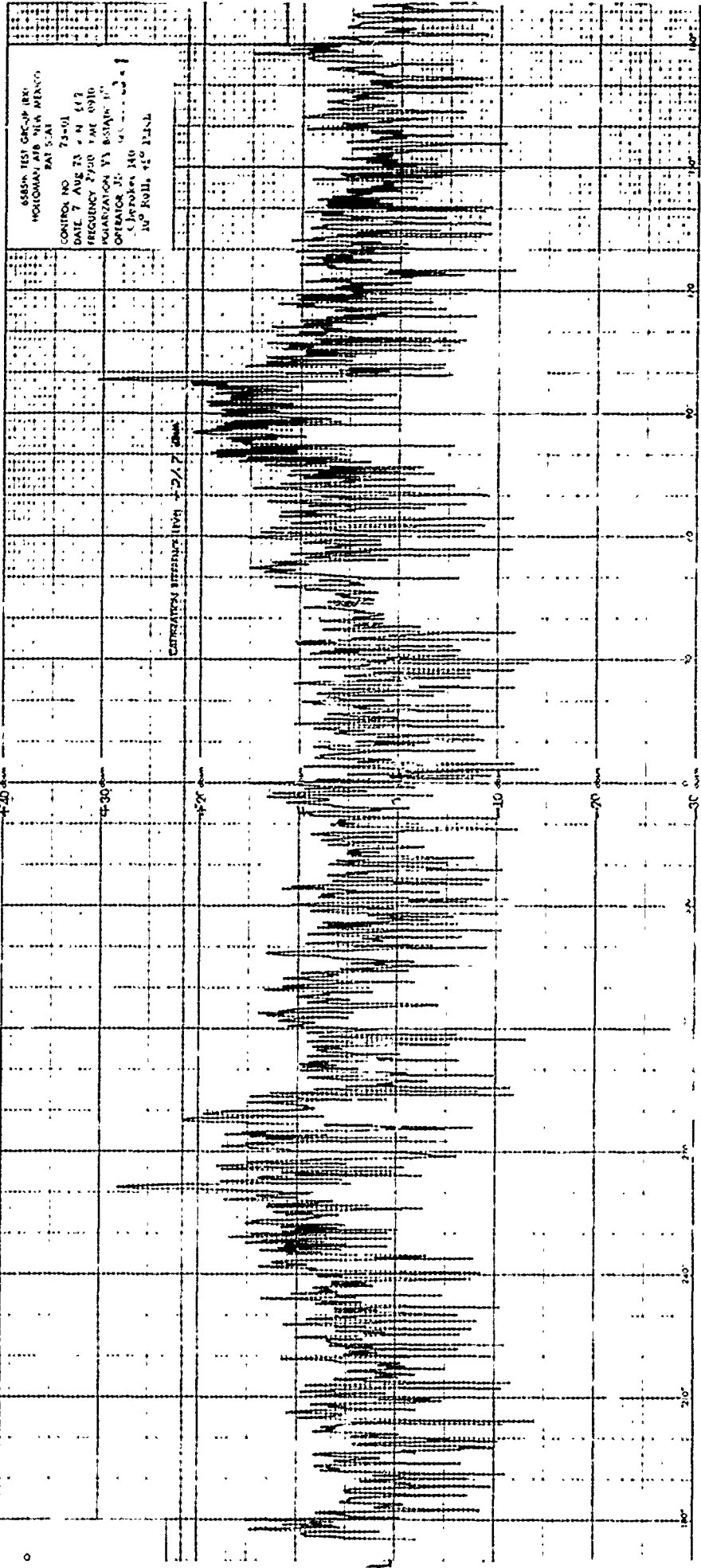












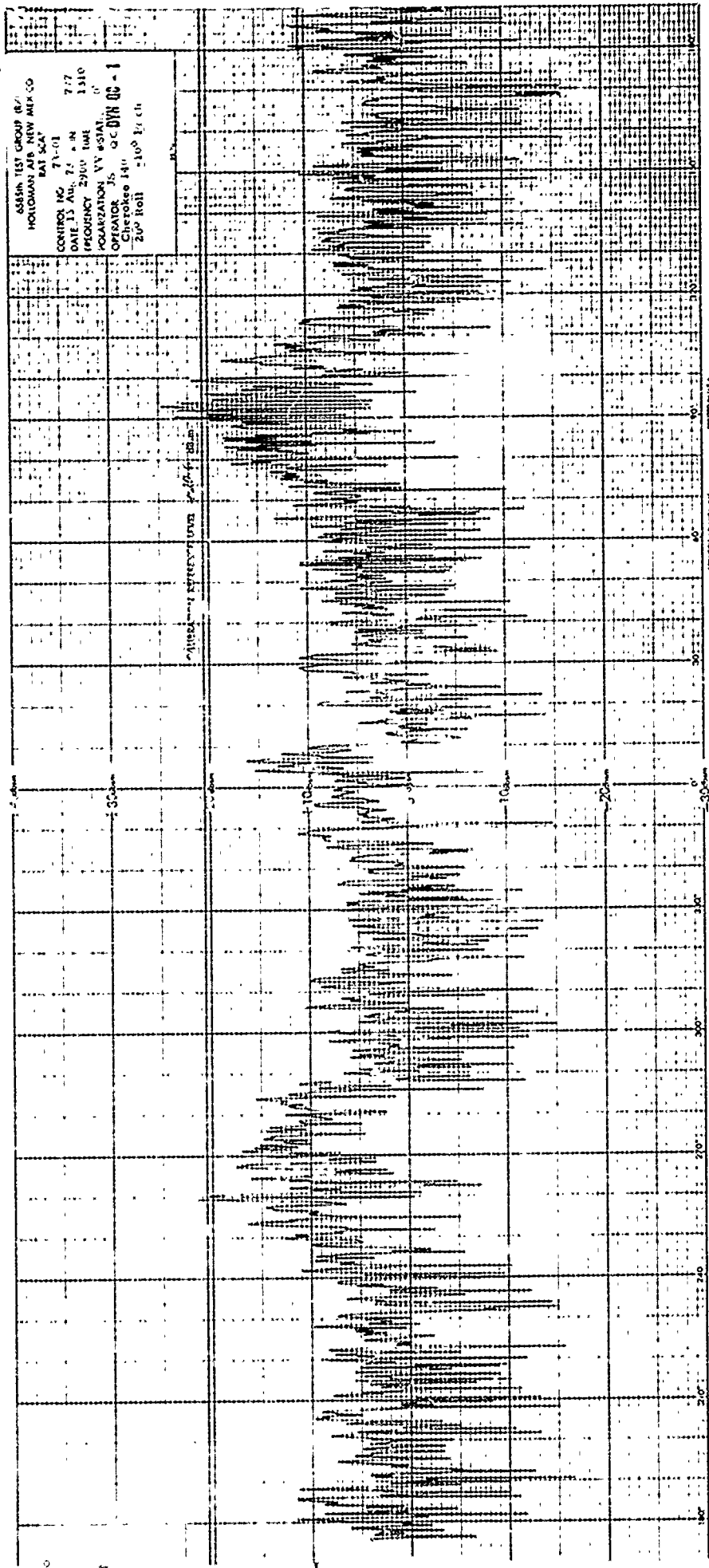
ASSTN TEST GROUP (RT)
HOLLOMAN AFB NEW MEX CO
RAT SCAL

CONTROL NO 73-01
DATE AUG 73 FOR 53
FREQUENCY 27.0 MHz 1355
POLARIZATION VV ESW
OPERATOR JS G.C. 13-1
Clemente dlt
10° Roll, 10° Pitch

TIME (min) DISTANCE (km) -9.4 min

1.40 min
1.20 min
1.00 min
0.80 min
0.60 min
0.40 min
0.20 min
0.00 min

240 260 280 300 320 340 360 380 400 420 440 460 480 500 520 540 560 580 600 620 640 660 680 700 720 740 760 780 800 820 840 860 880 900 920 940 960 980 1000



68. TEST GROUP (R)

HOLMAN ATB NEW MEXICO

FAT 300

CONTROL NO 73-11

DATE 10 AUG 73 4:41

741

REQUENCY 2300 MHz

0920

ORGANIZATION VV 2141C

00

OPERATOR 14

00

01W 00-1

110

200 Roll

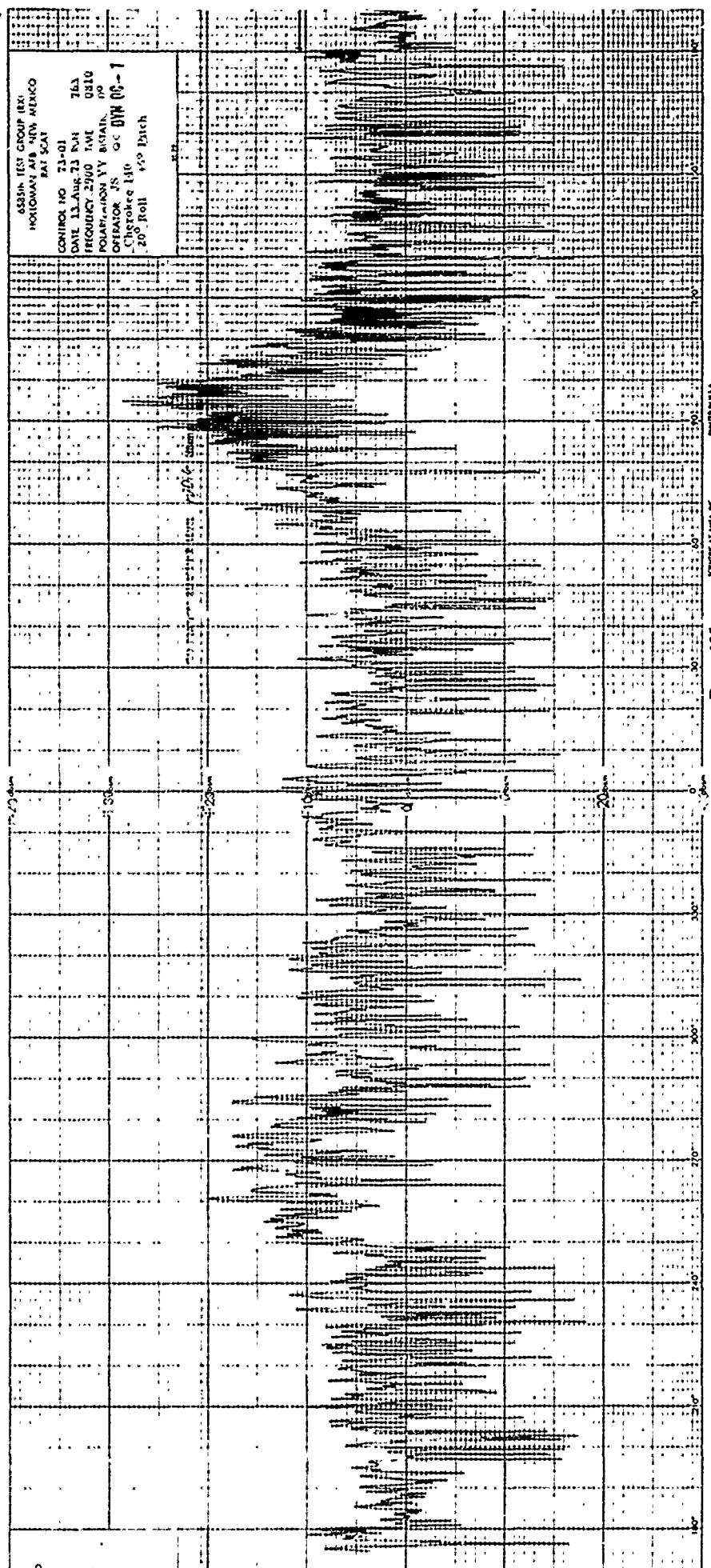
200 Pitch

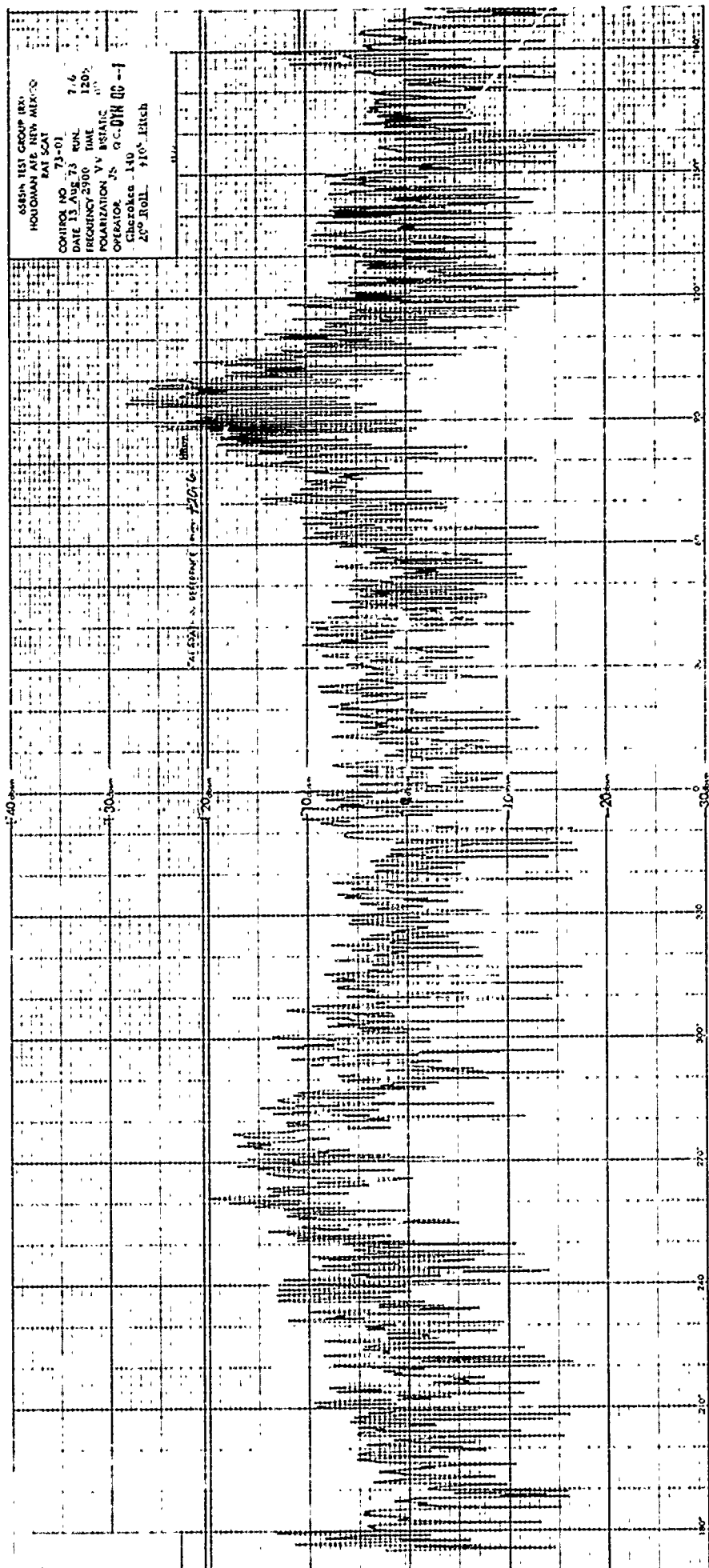
CAPTAIN'S INTEREST 100-100

TIME

0 10 20 30 40 50 60 70 80 90 100

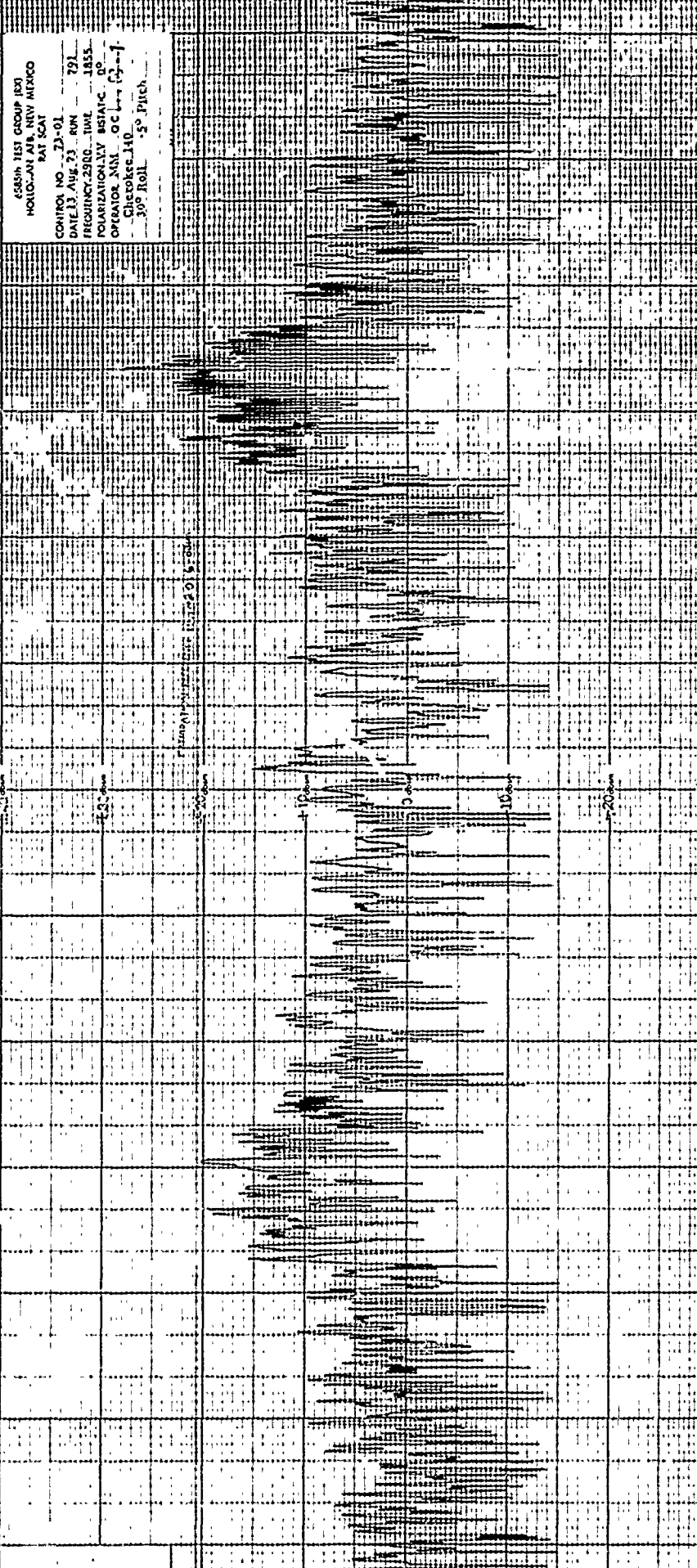
0 10 20 30 40 50 60 70 80 90 100





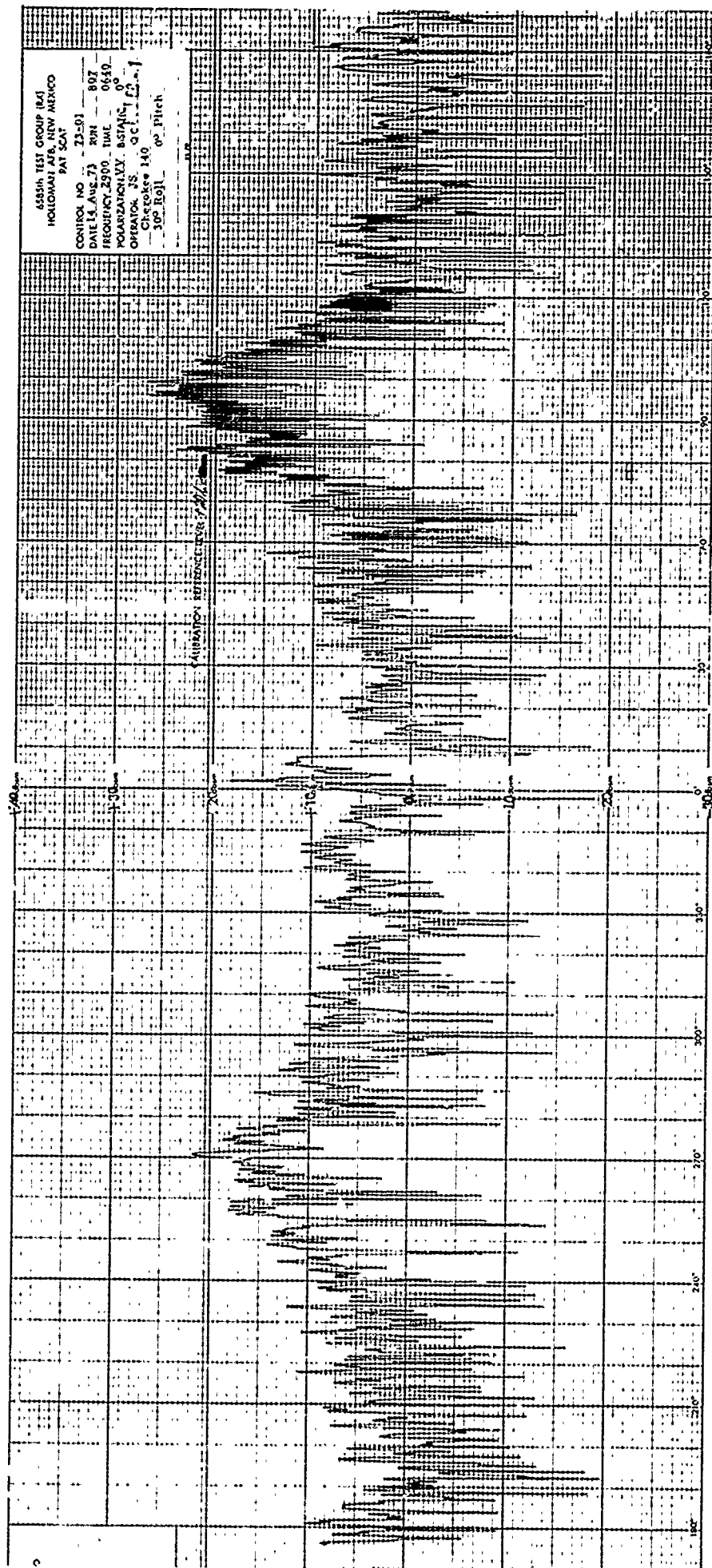
6585% TEST GROUP (B3)
HOLIO-AI AIR, NEW MEXICO
PAT SCAT

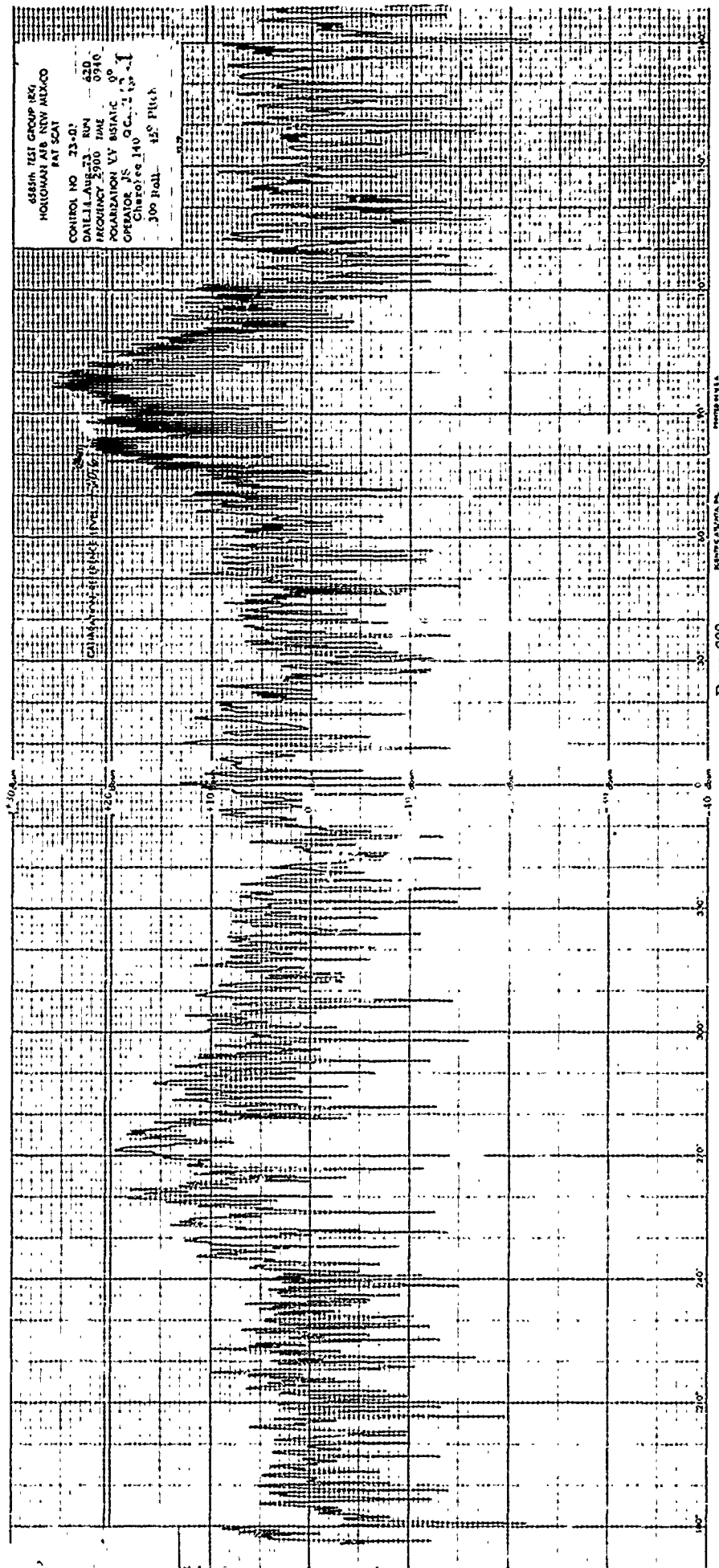
CONTROL NO. 73-01
DATE 13 Aug 73 RUN 791
FREQUENCY 2300 MHz 1855
POLARIZATION LINEAR
OPERATOR NNA OC 1-1-73
Chetoket 140
30° Roll 5° Pitch



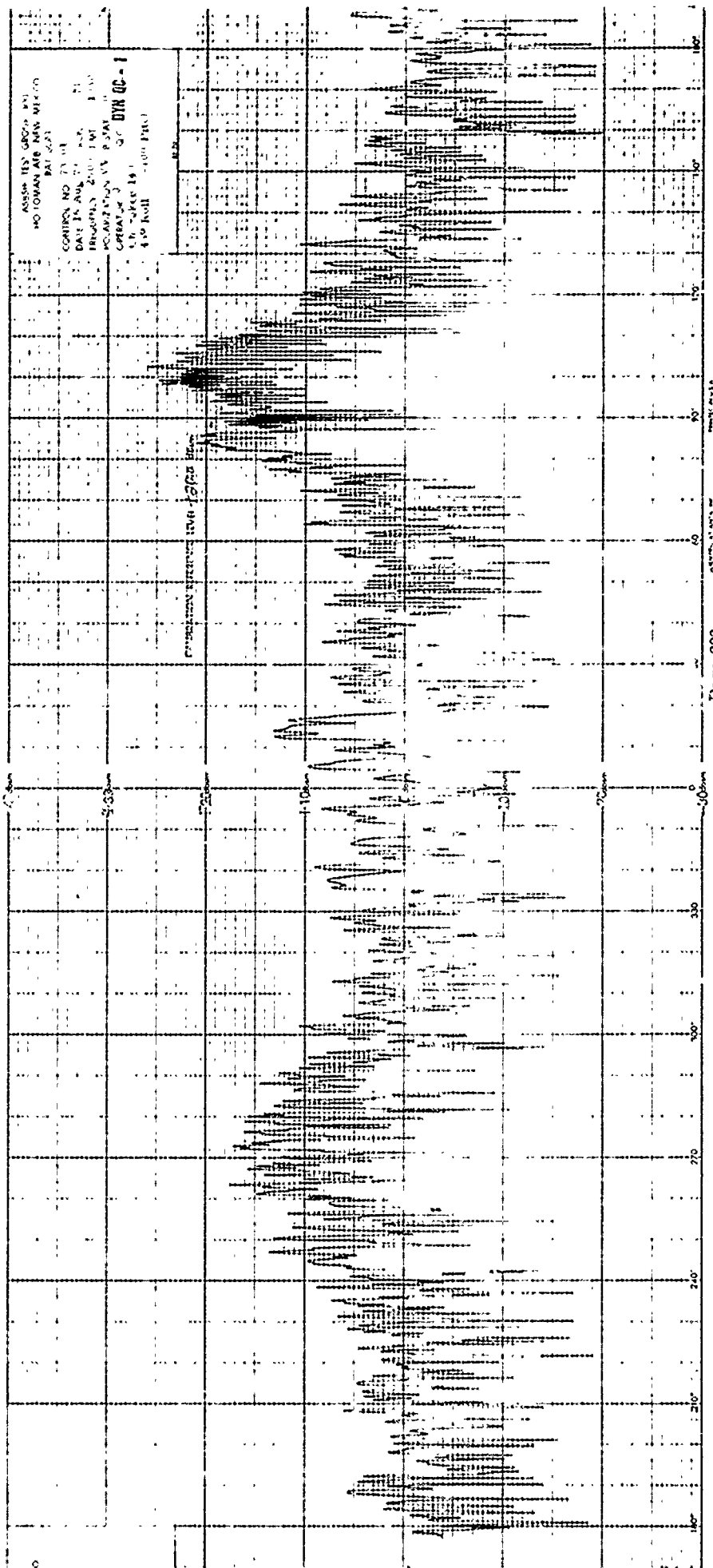
6585th TEST GROUP (R&D)
HOLLANDIAJ AFB, NEW MEXICO
PAT SCA?

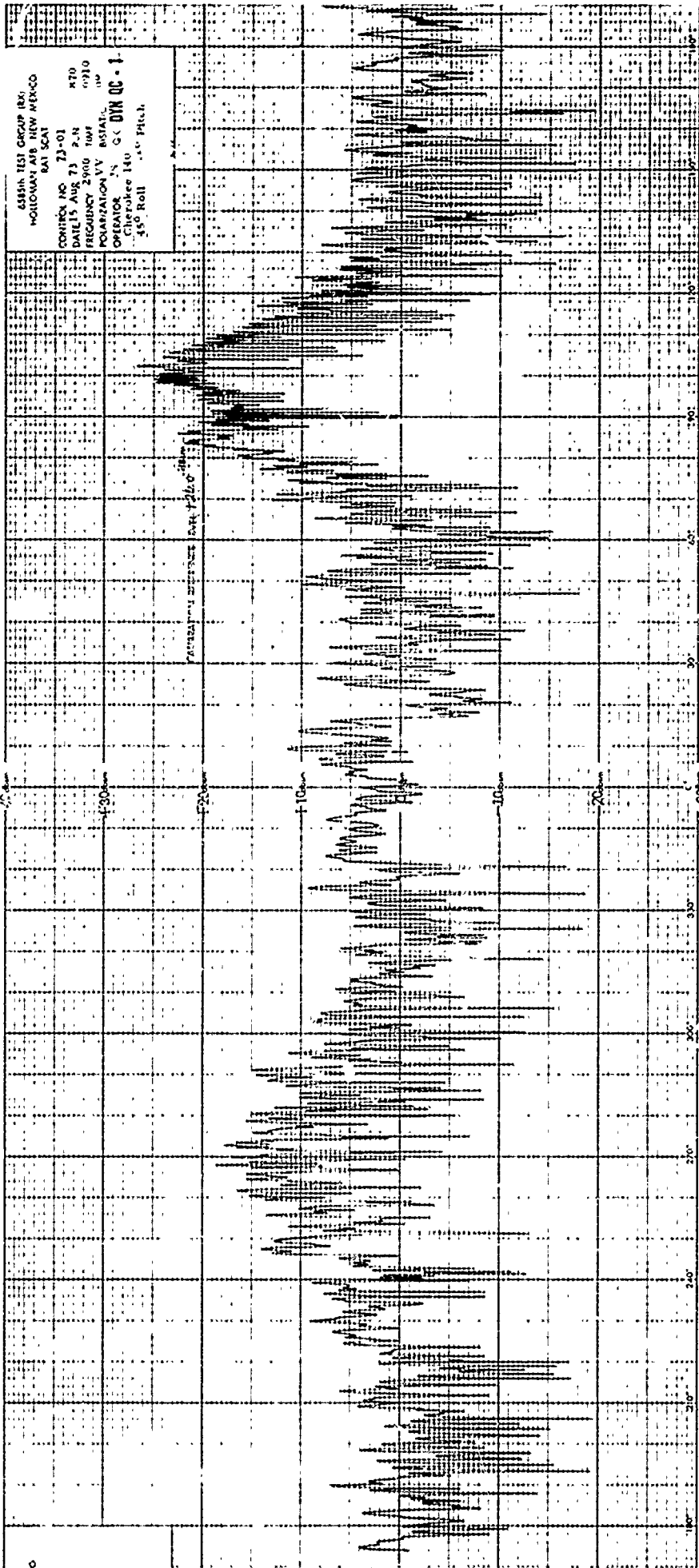
CONTROL NO. 23-01
DATE 14 AUG 73 20H 807
FREQUENCY 2900 TIME 061
POLARIZATION VV 85 ANGLE 0°
OPERATOR JS QC 173
Cherokee 140 00 plich
300 Roll

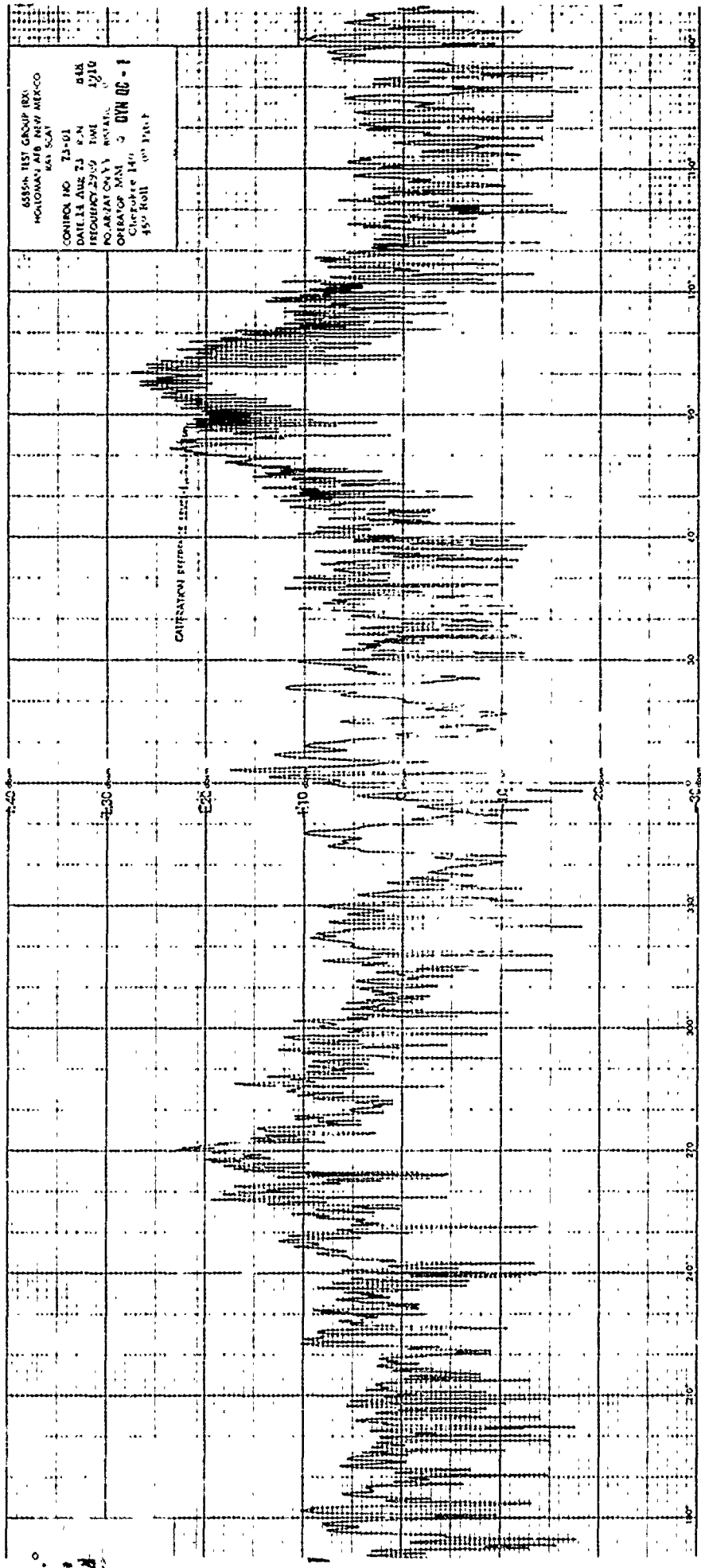




[illegible]







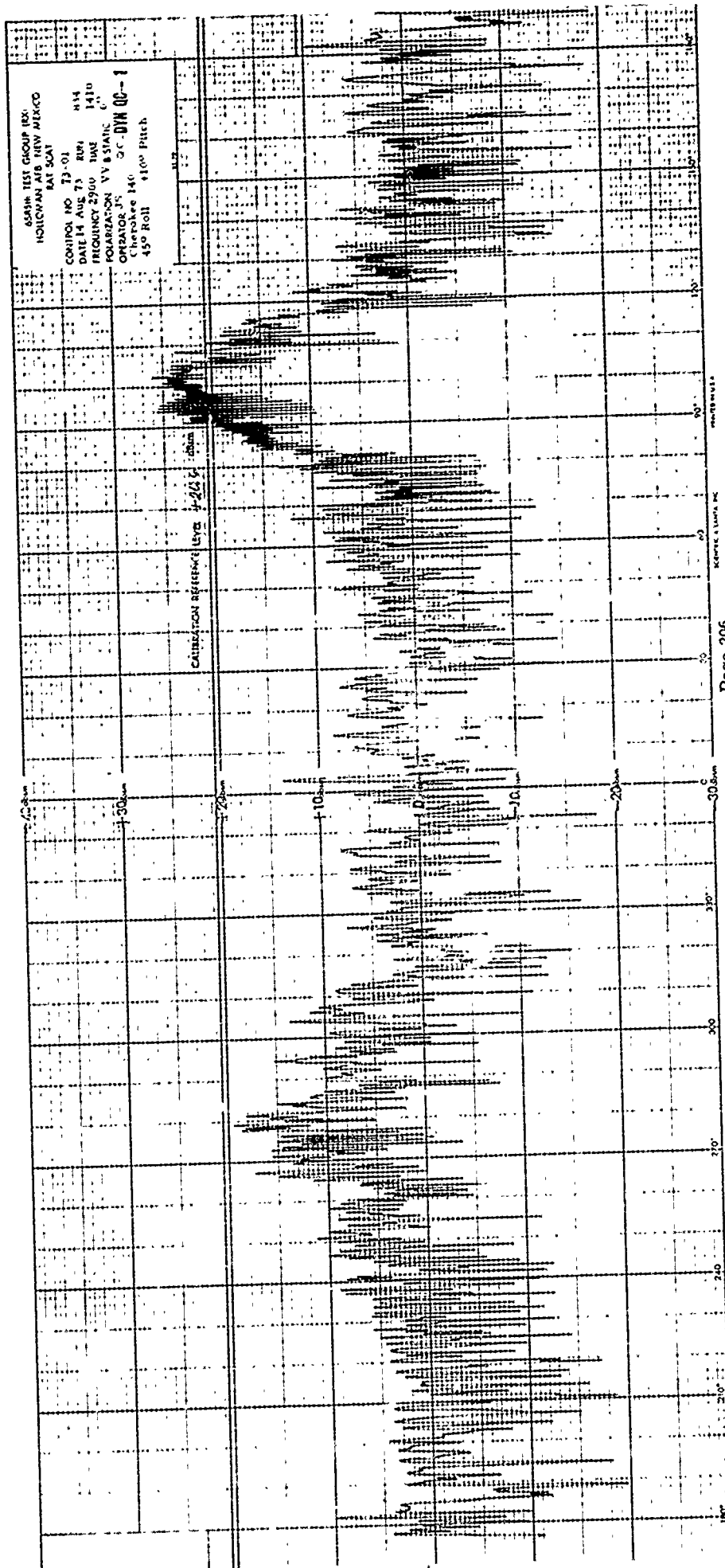
6855m TEST GROUP (BA)
MILLONARY AIR NEW MEXICO
BAT 501

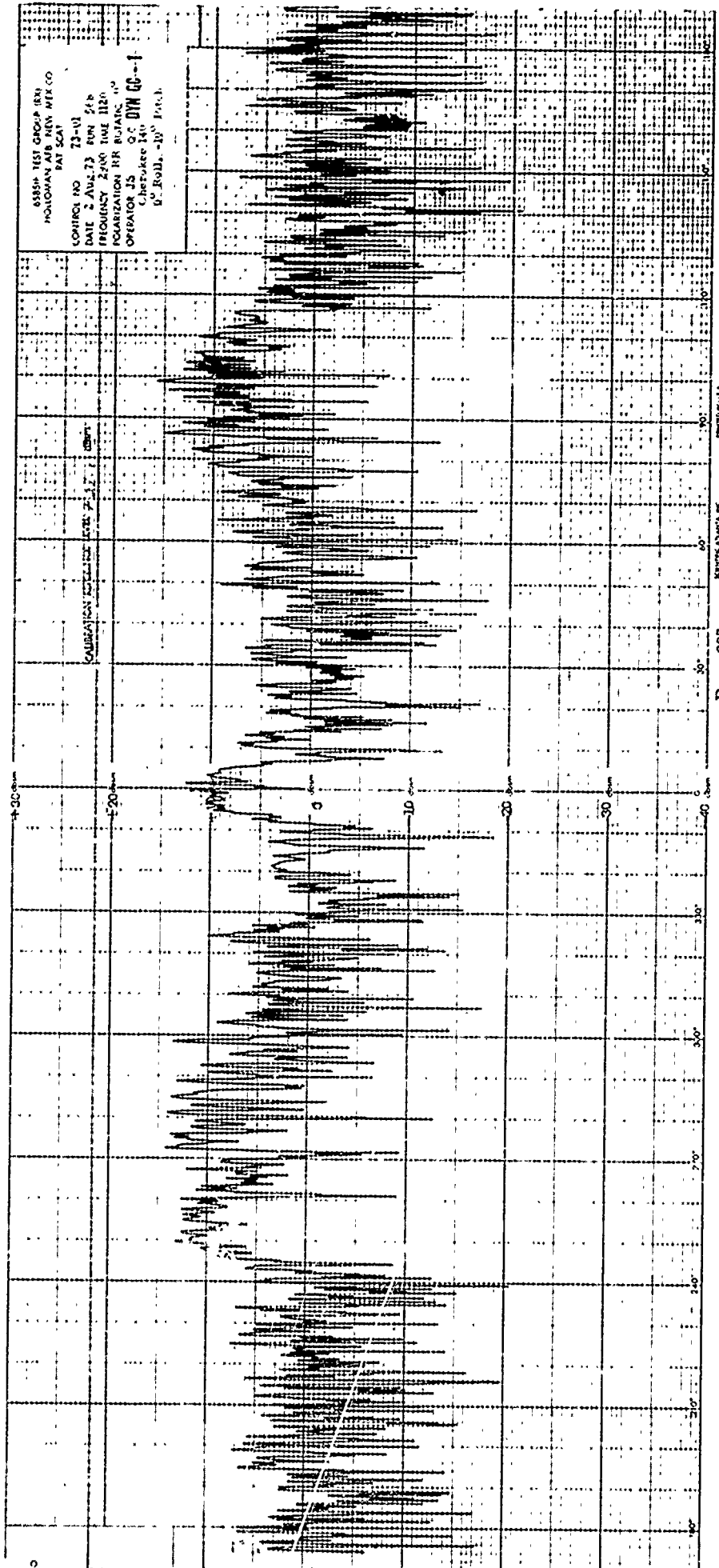
CONTOUR NO. 23-01 841
DATE 14 AUG 73 PM 1610
FREQUENCY 2500 TAD 1610
POSITION VYBASTAC
ORDERS NIN 00
Crew Chief 140 00
450 Roll 450 Pitch

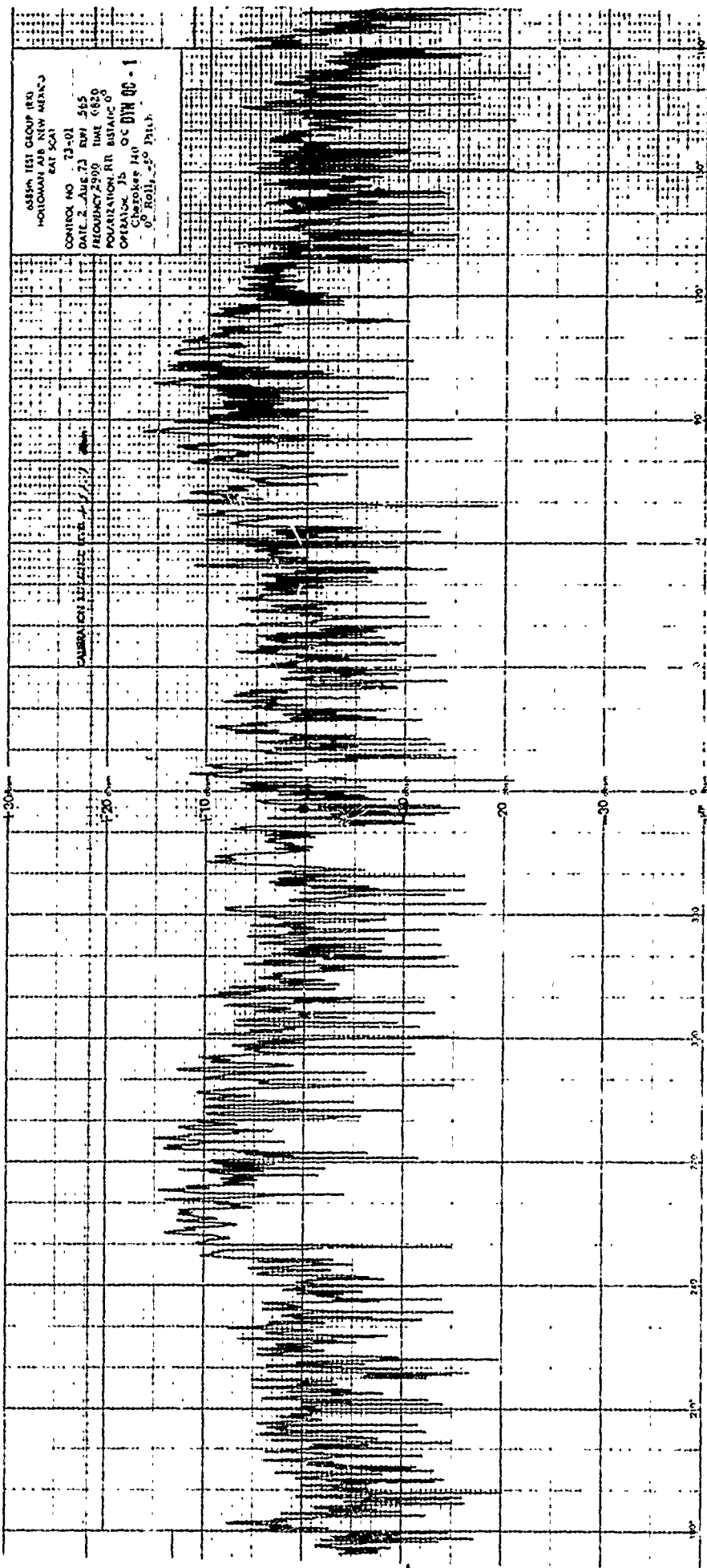
CONTINUATION OF TEST GROUP 23-01

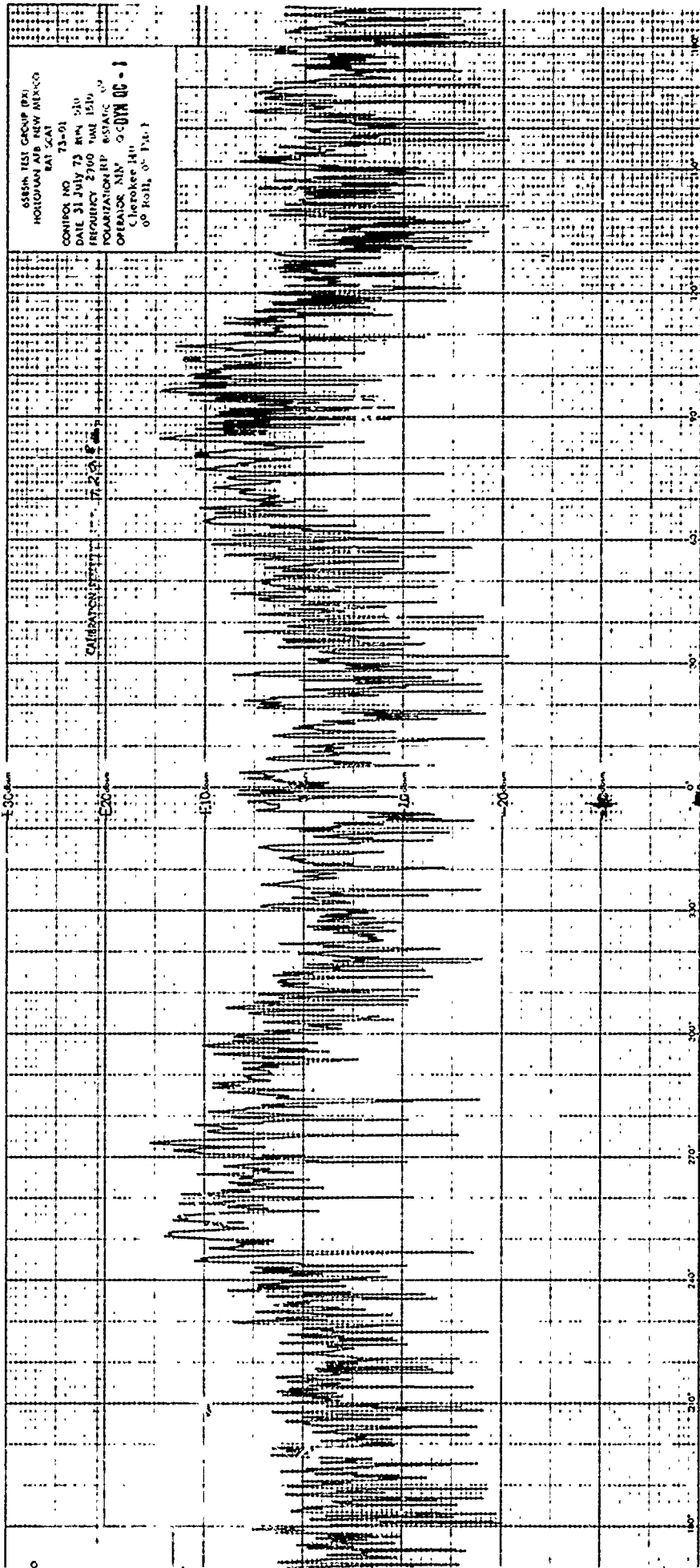
2500
2000
1500
1000
500
0

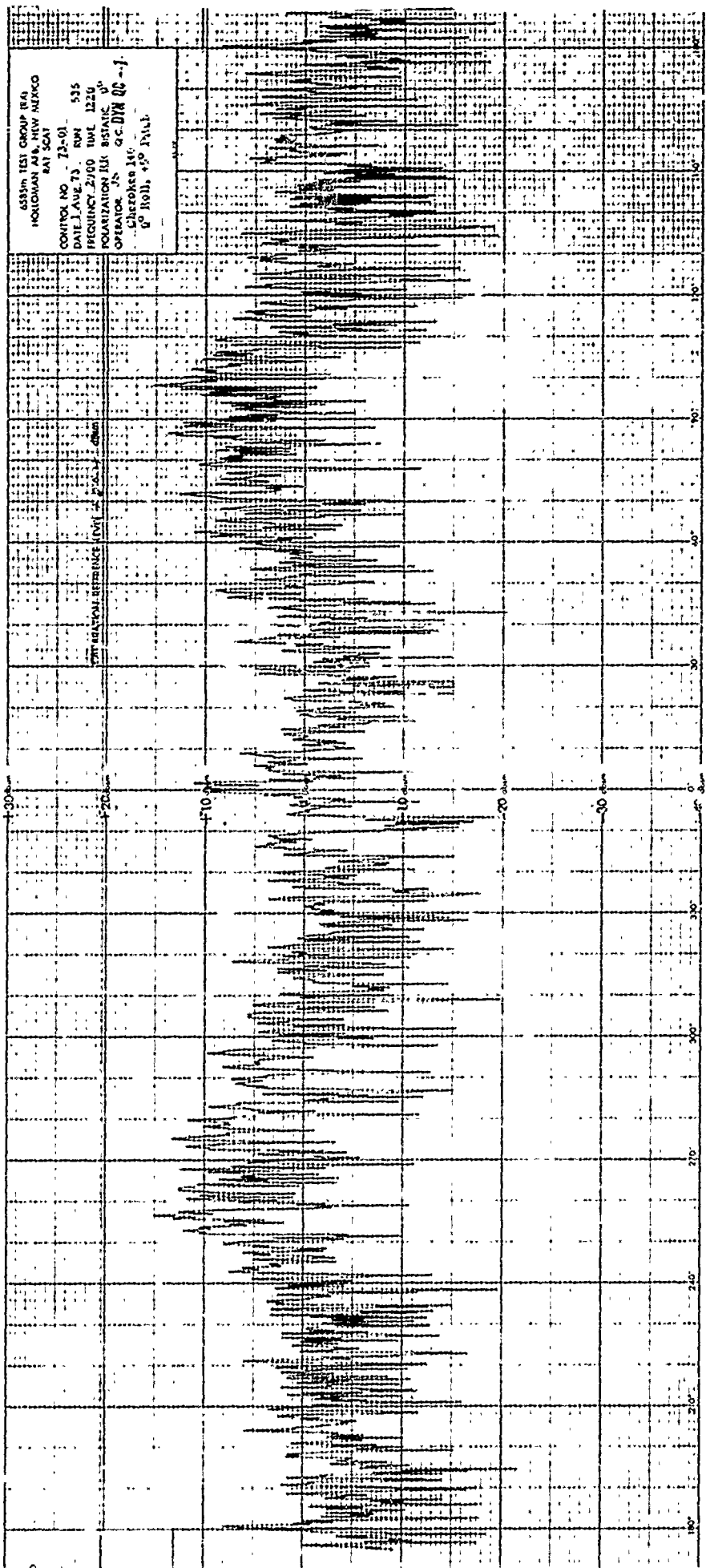
180° 210° 240° 270° 300° 330° 360°

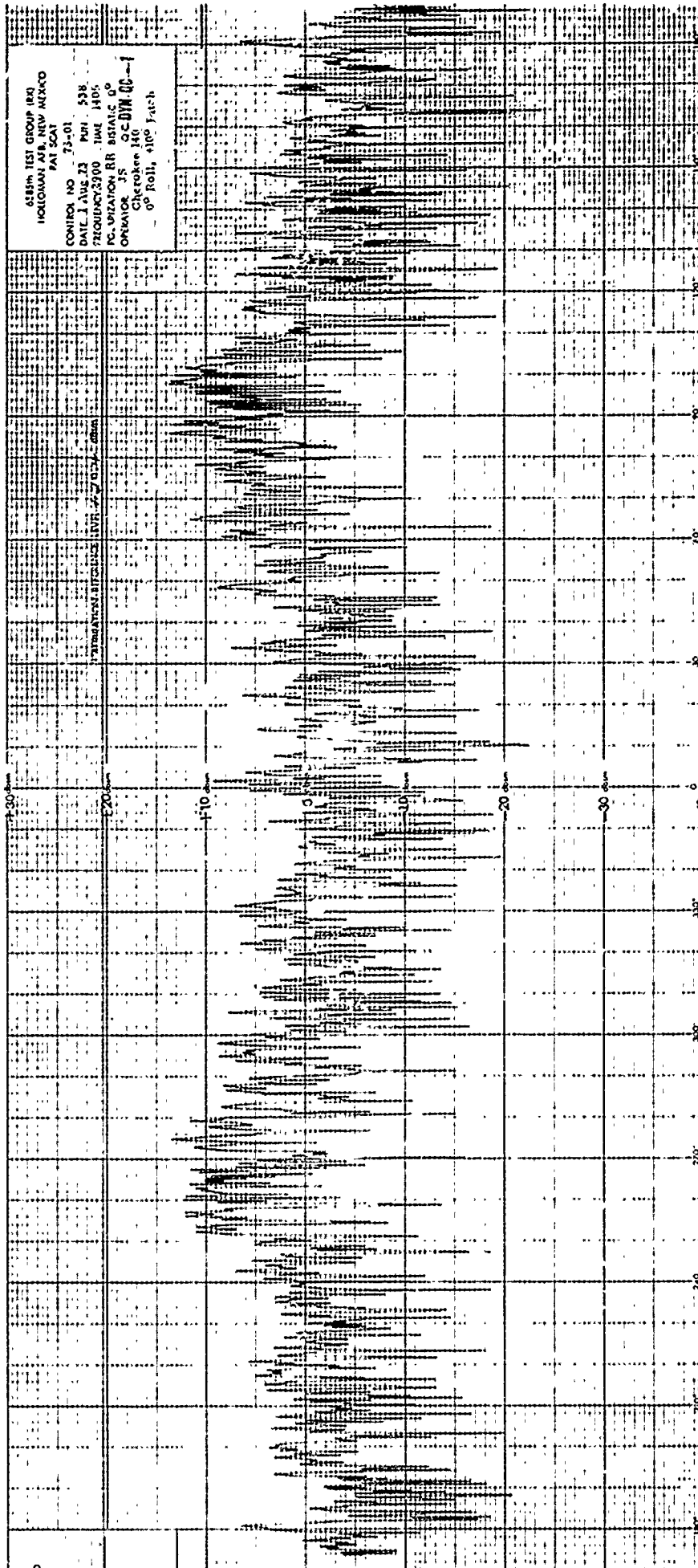


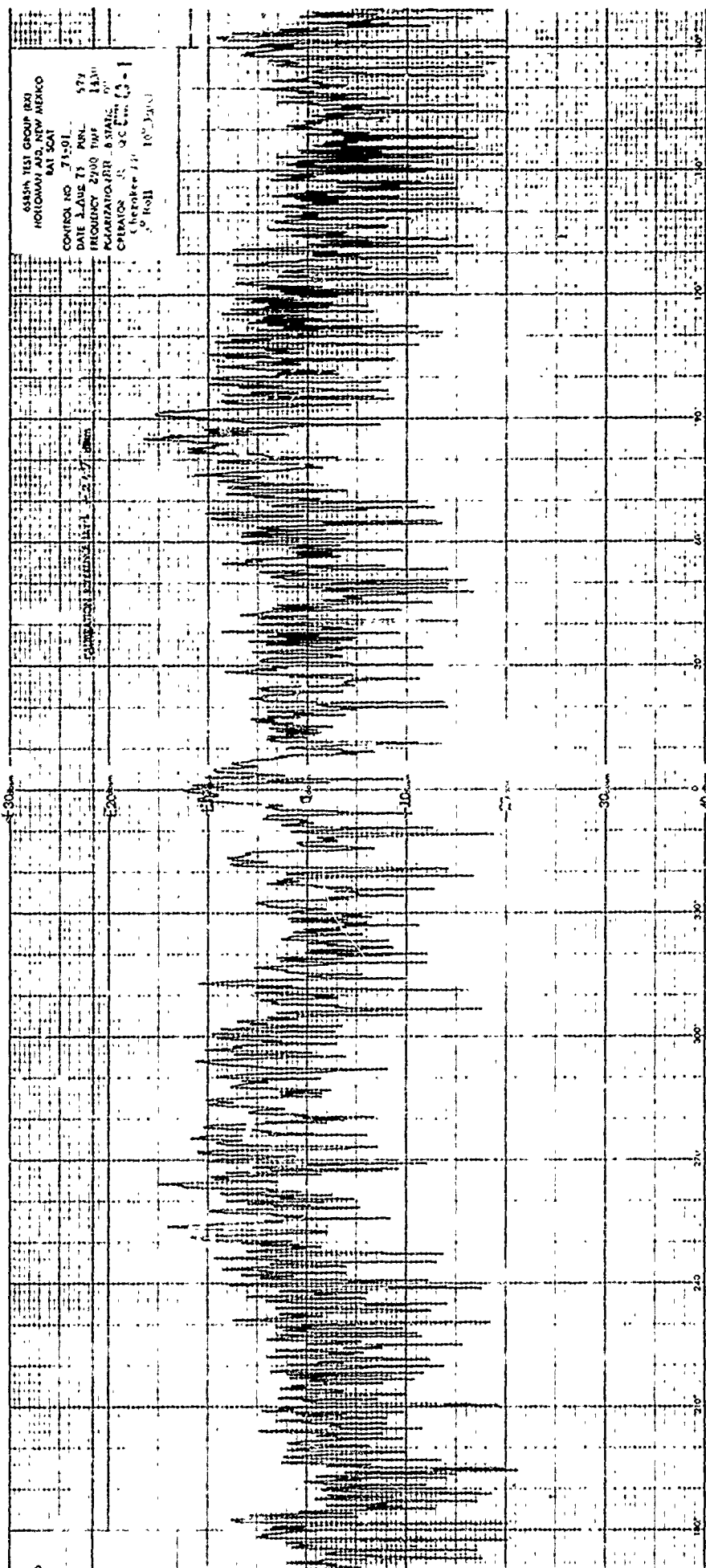


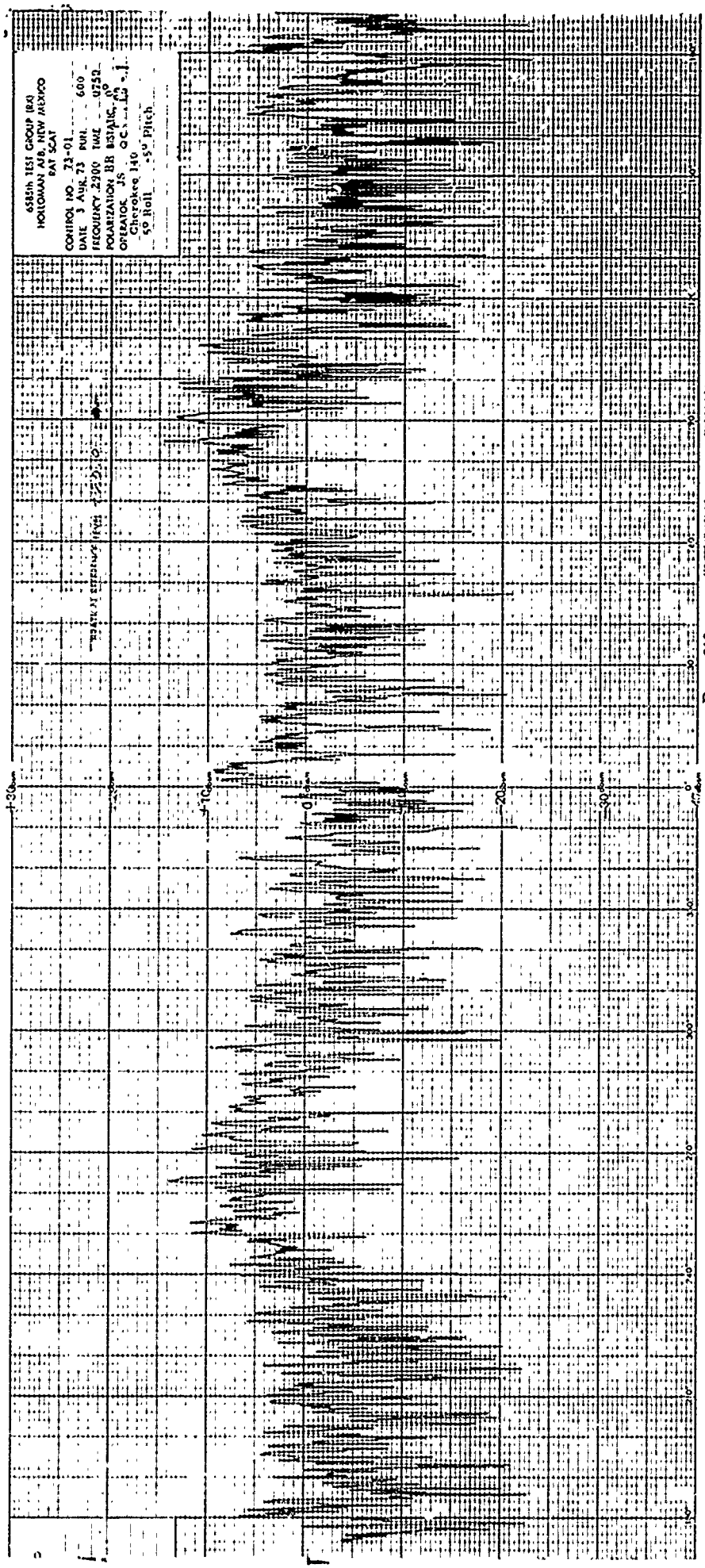






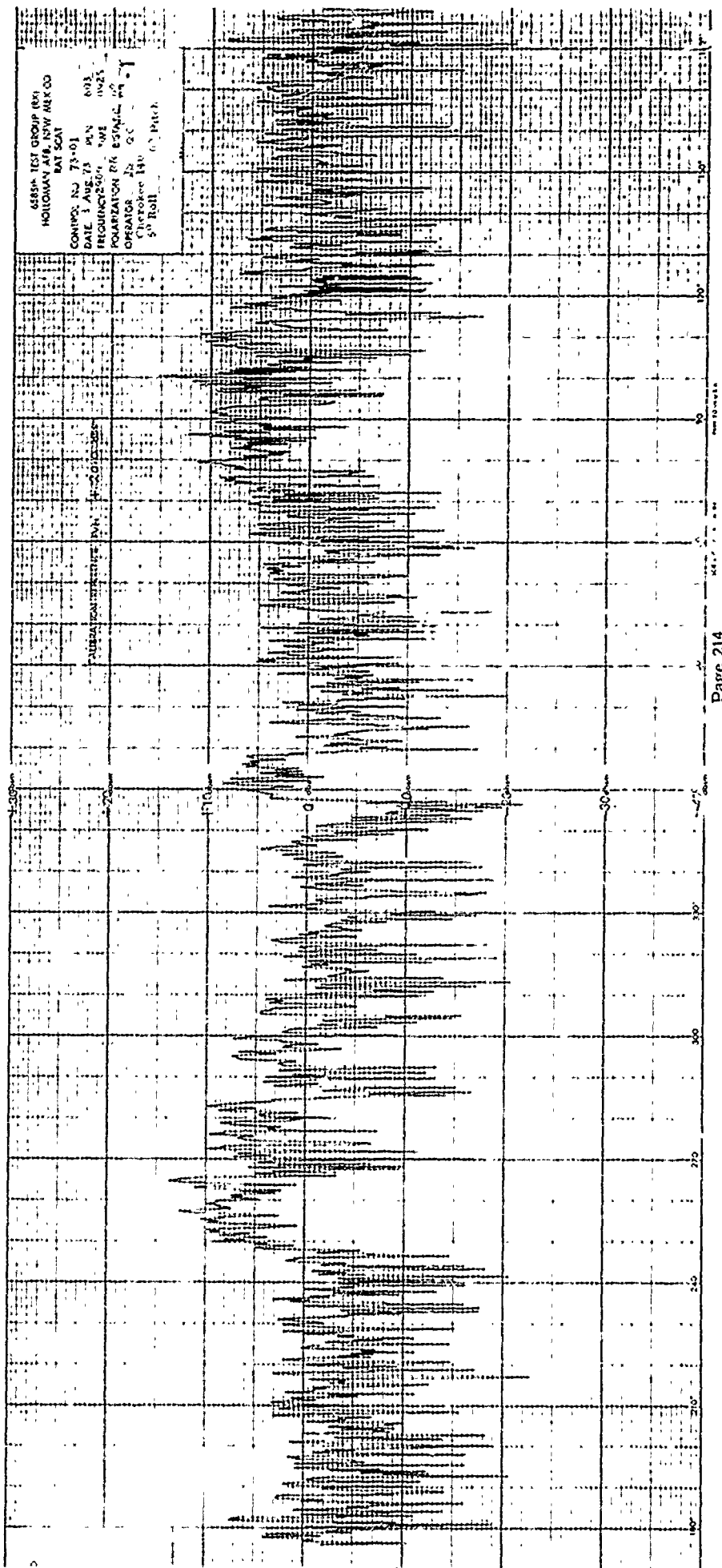


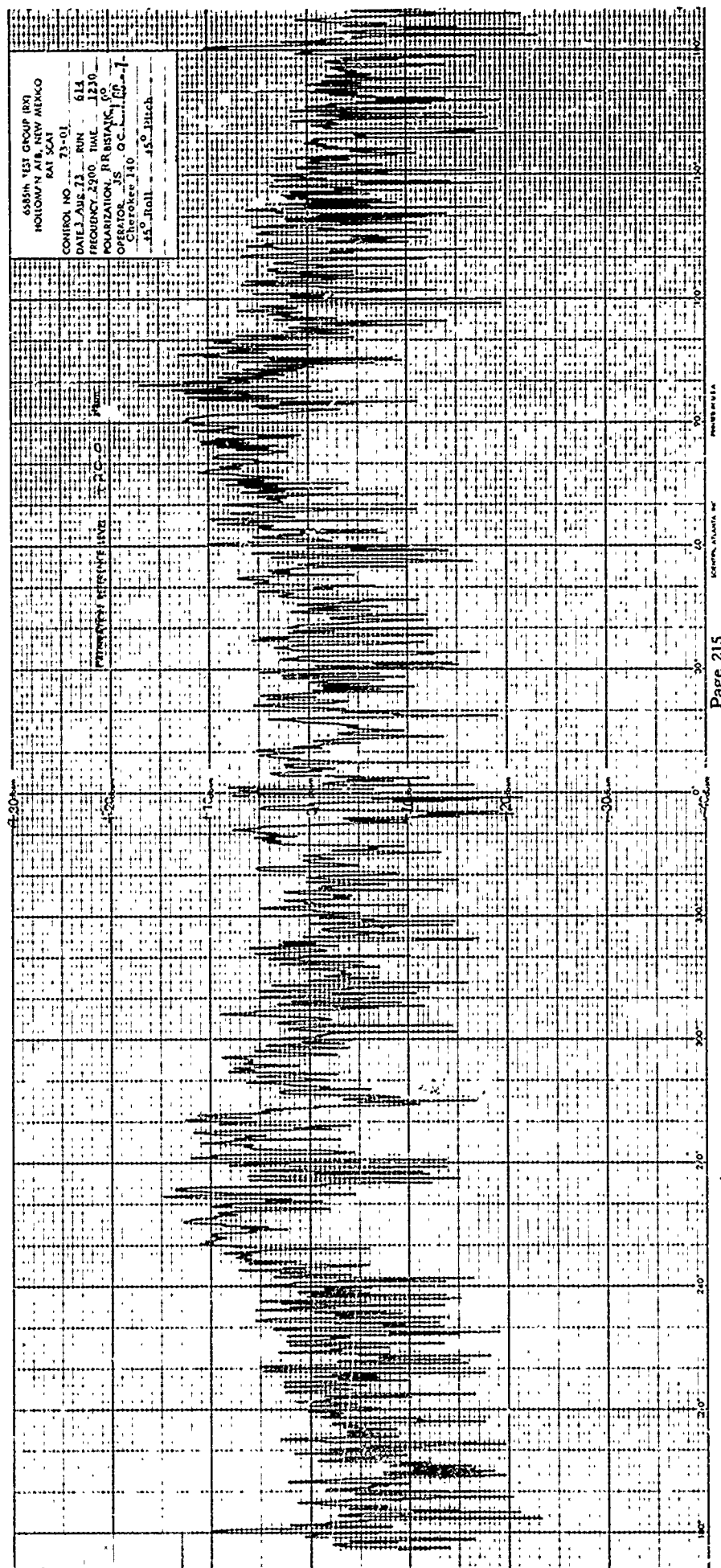


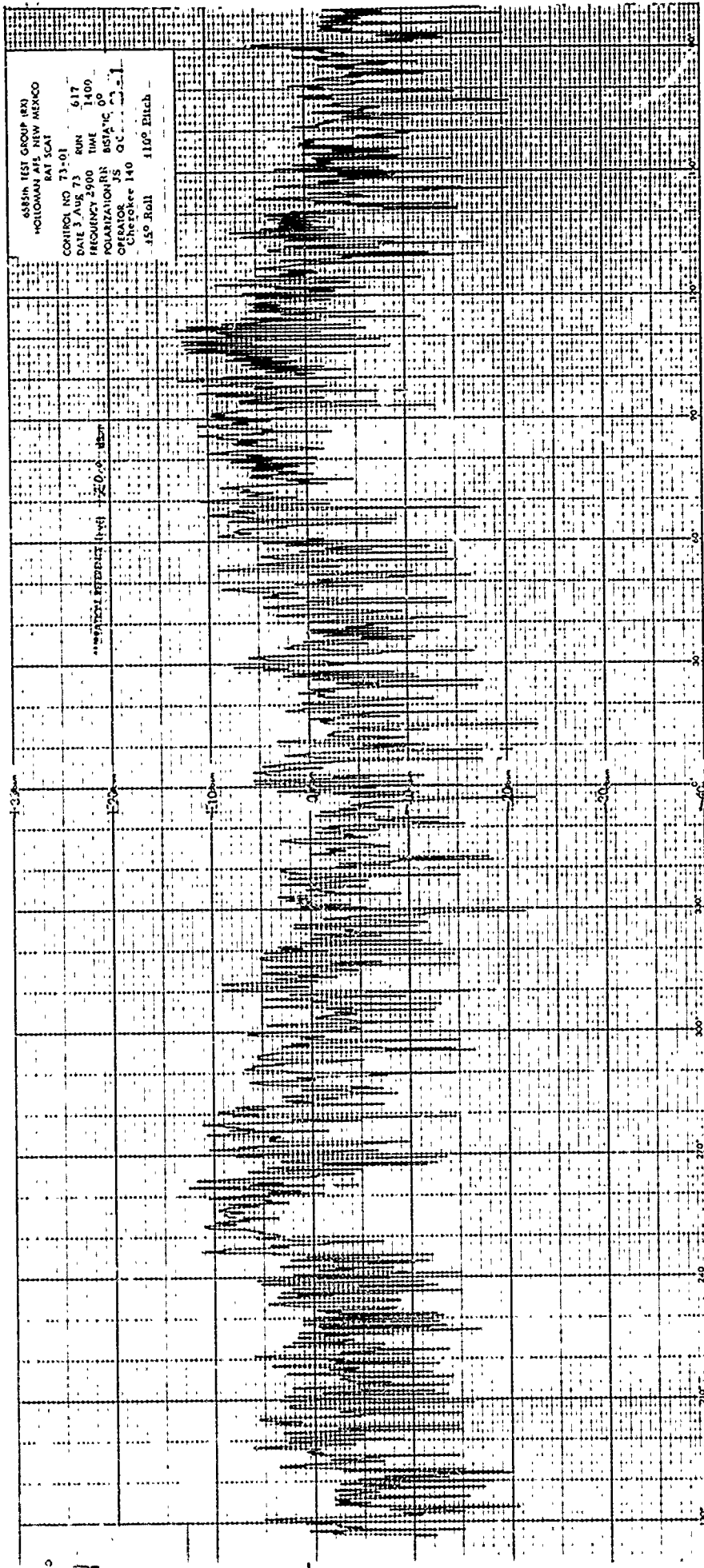


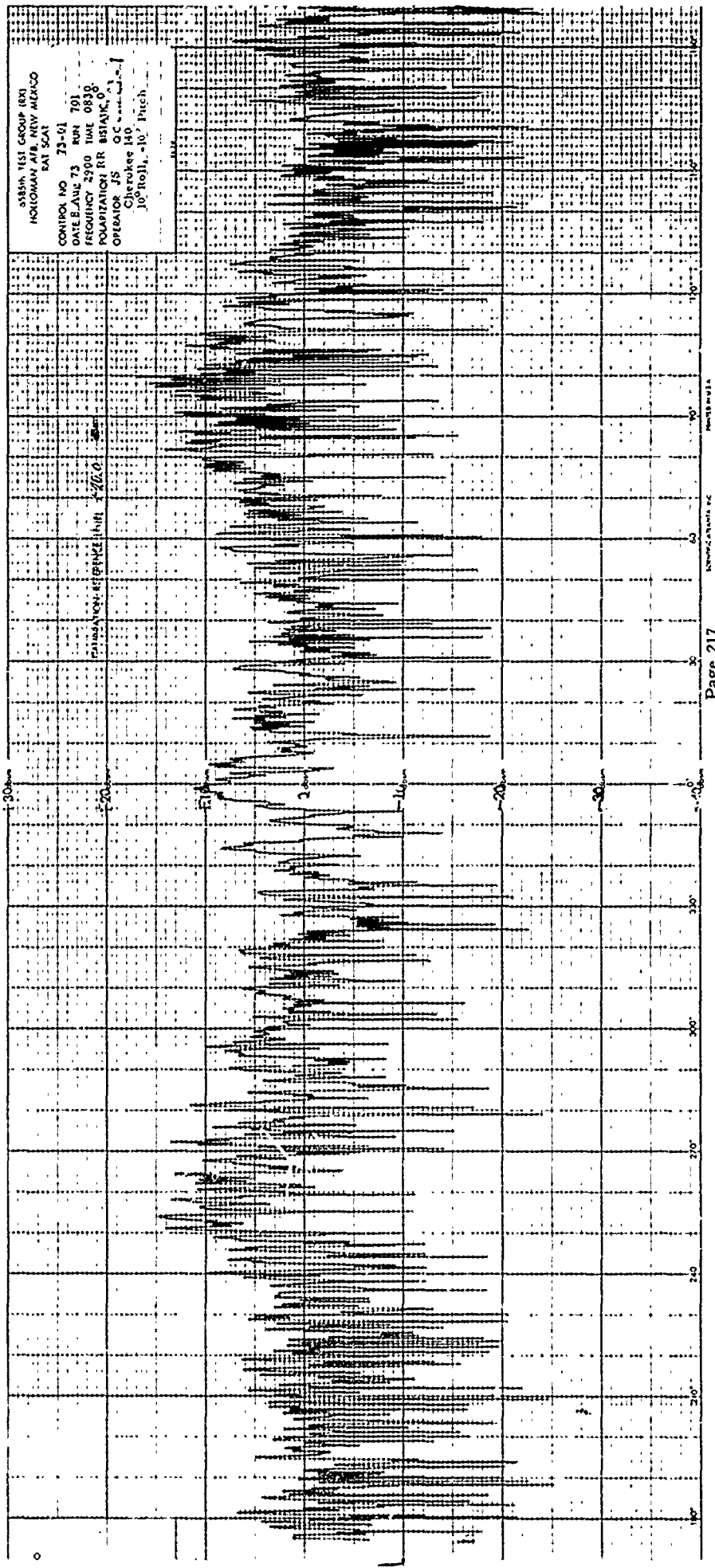
SEISMIC TEST GROUP (BA)
HOLCOMB, KANSAS, NEW MEXICO
BAT SCAT

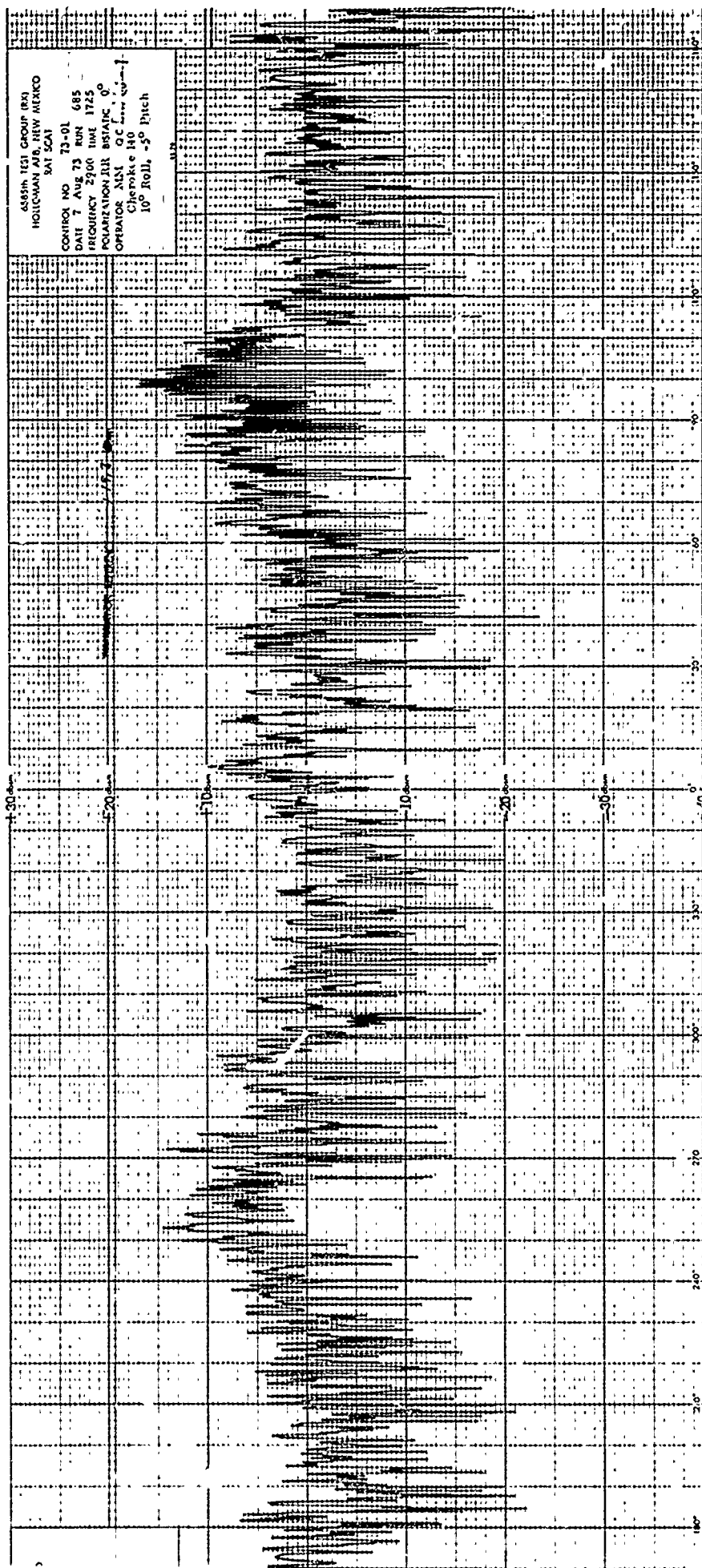
CONTROL NO. 73-01
DATE 3 APR 73 PUL 600
FREQUENCY 2300 HZ 0750
POLARIZATION BR EASTING, 0°
OPERATOR JS QC
Cherokee 140
50 Roll 55° Pitch

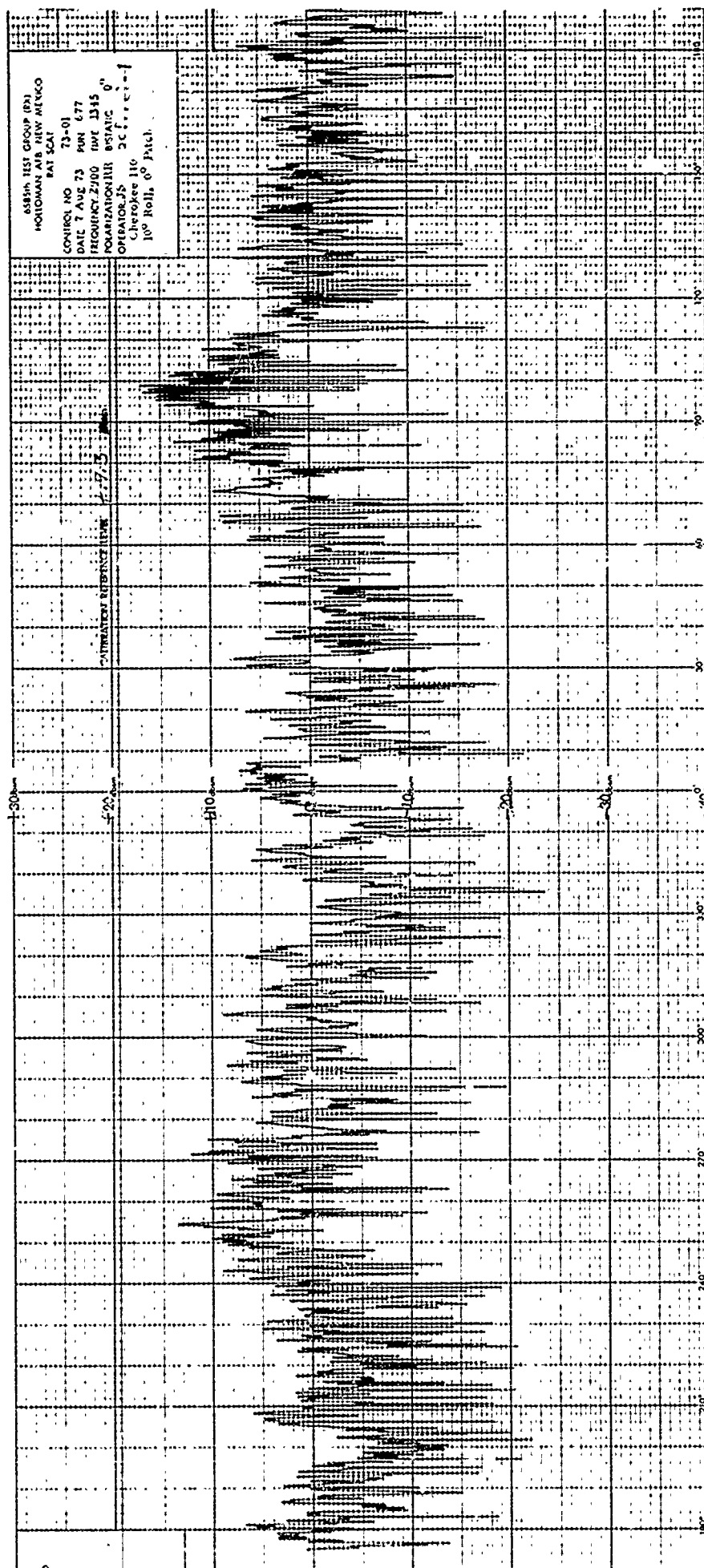


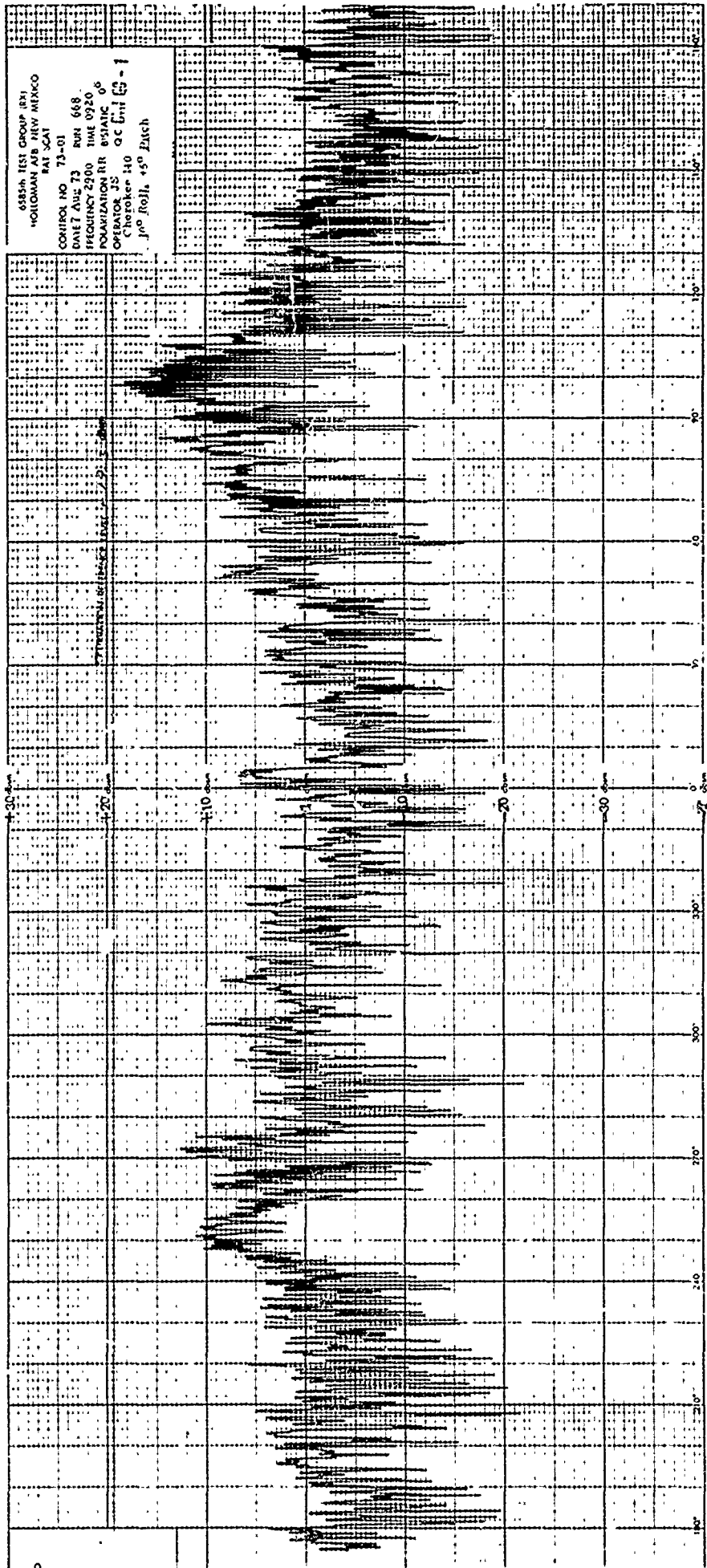


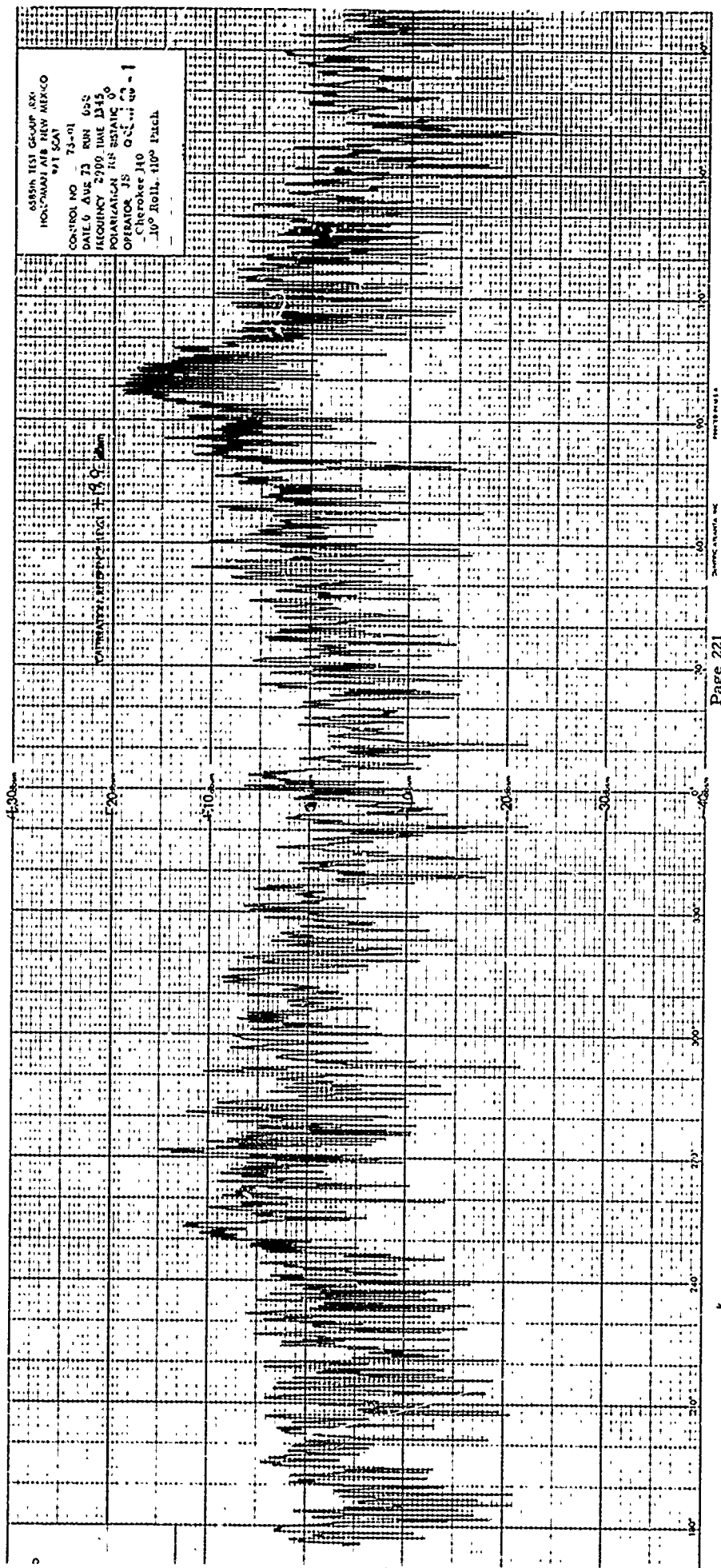












6585th TEST GROUP (RX)
MOLLOMAN AFB NEW MEXICO
RAT SCAT

CONTROL NO. 73-01

778

1320

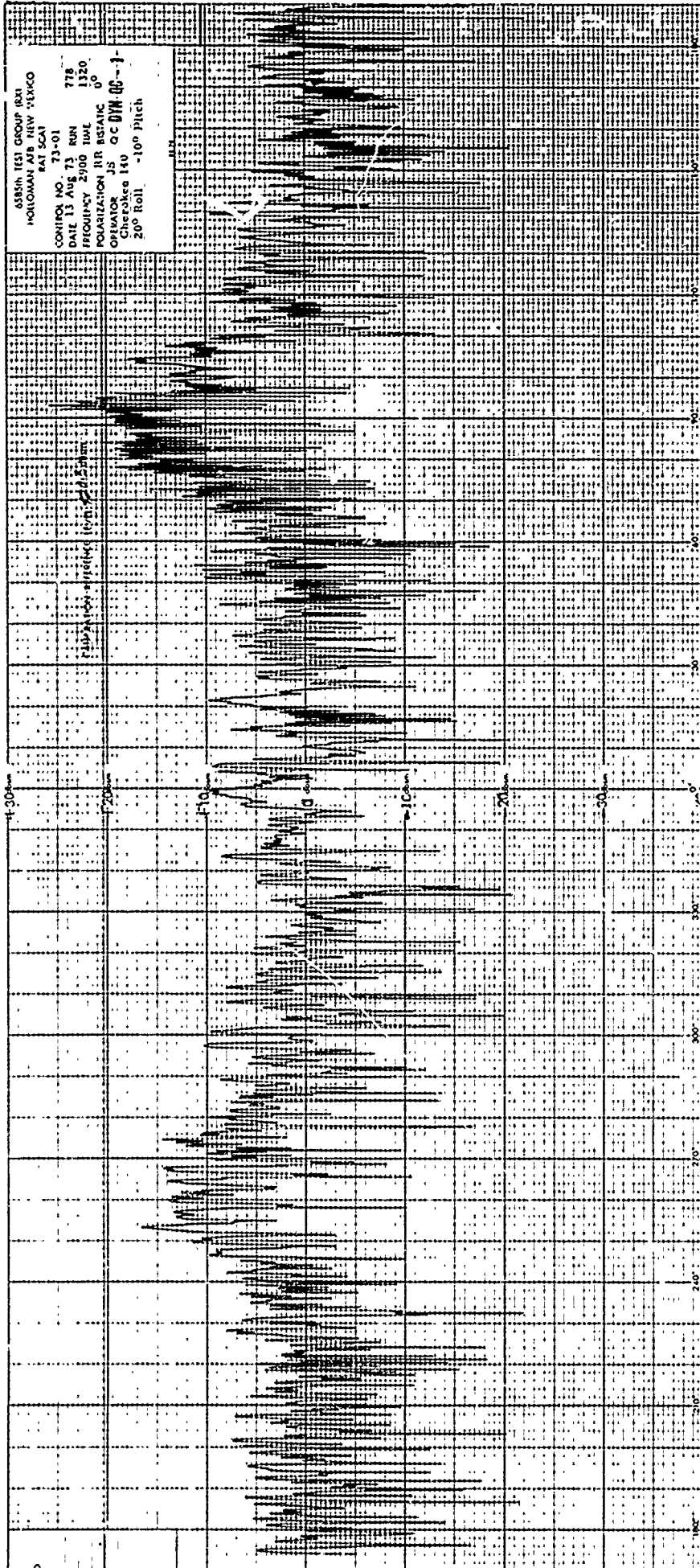
30

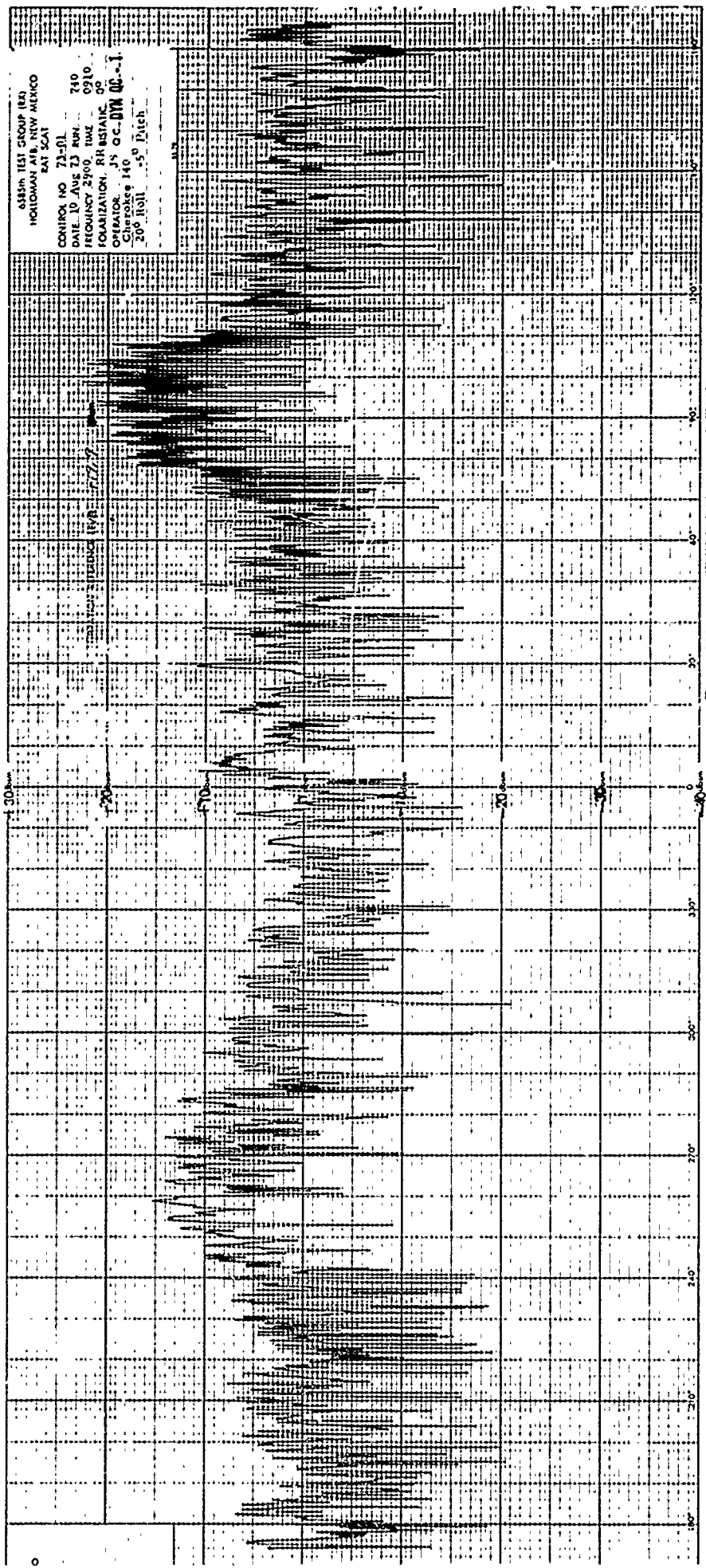
●

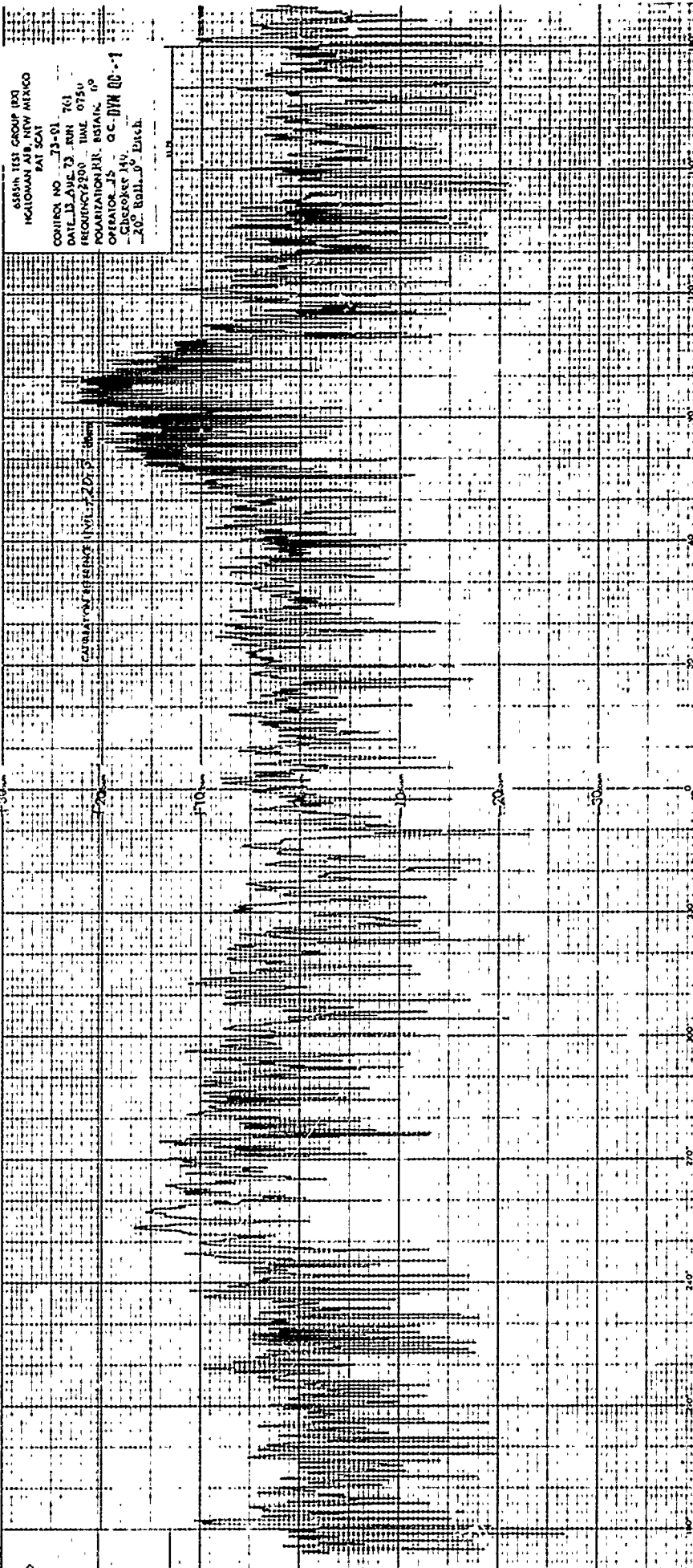
2

1

•

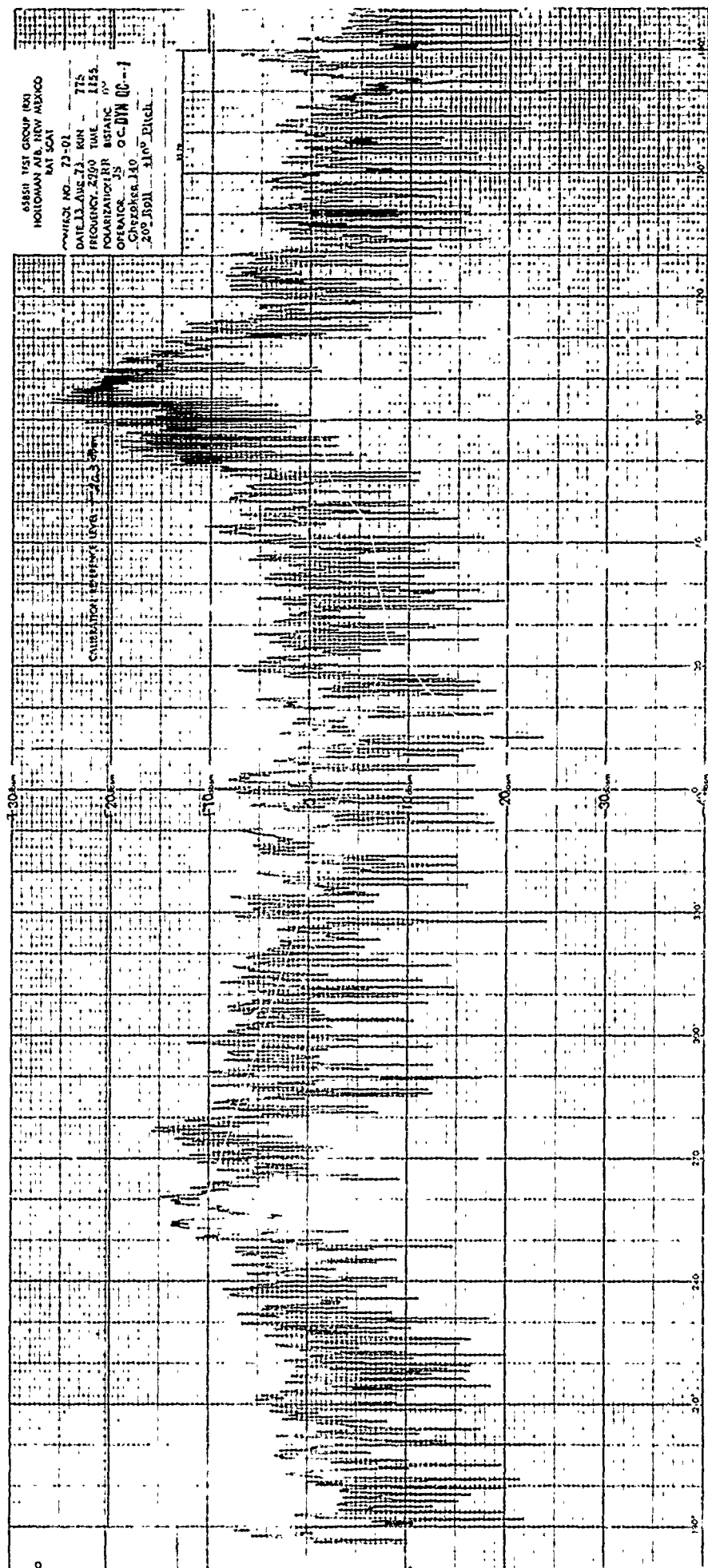


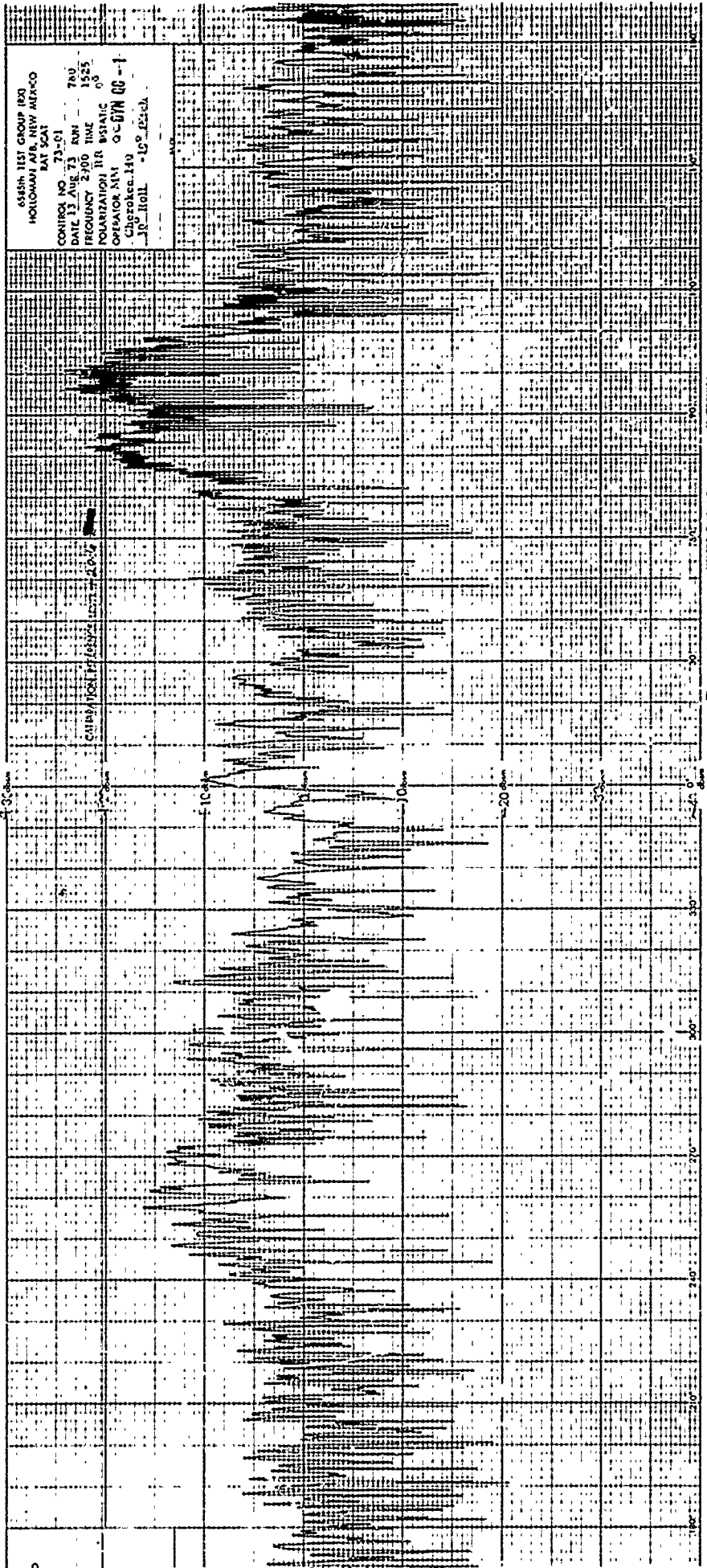


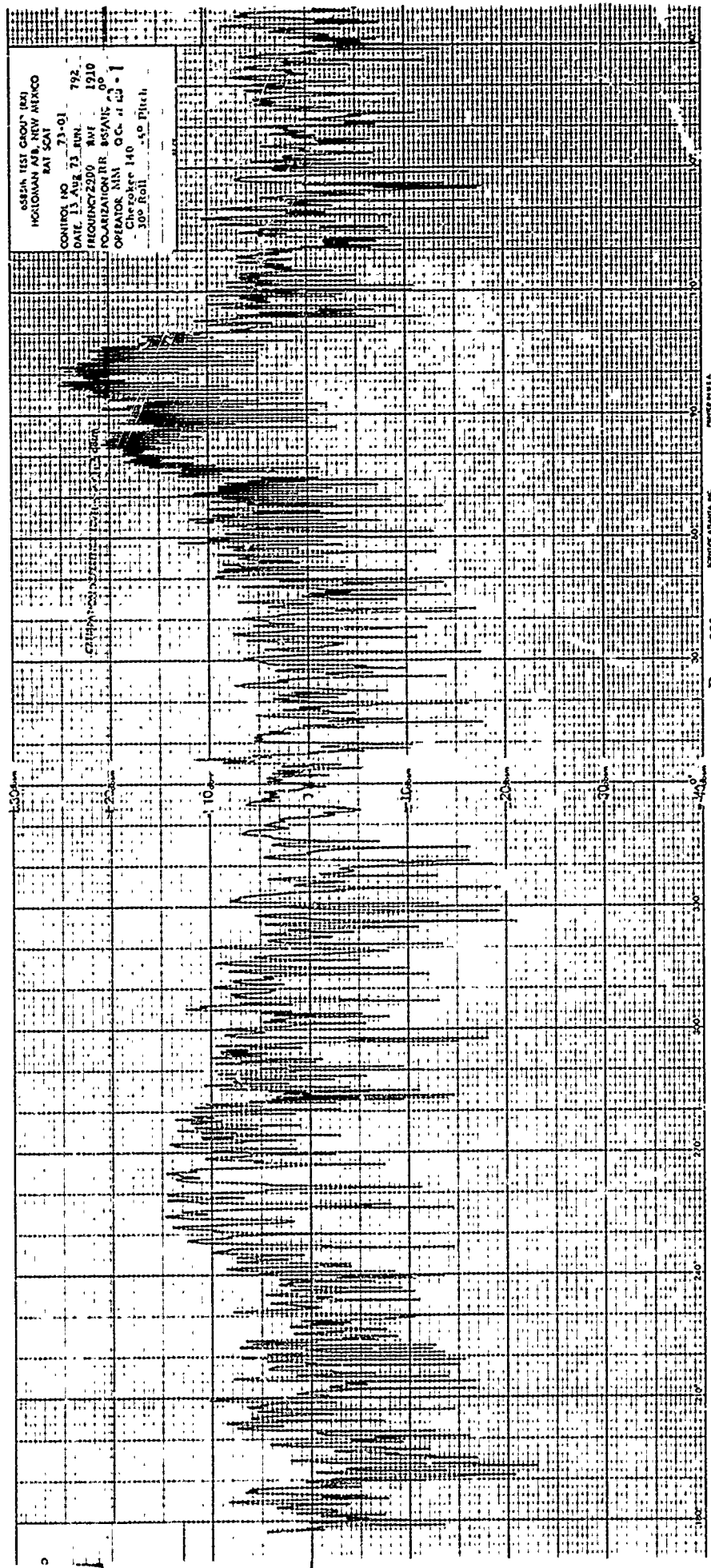


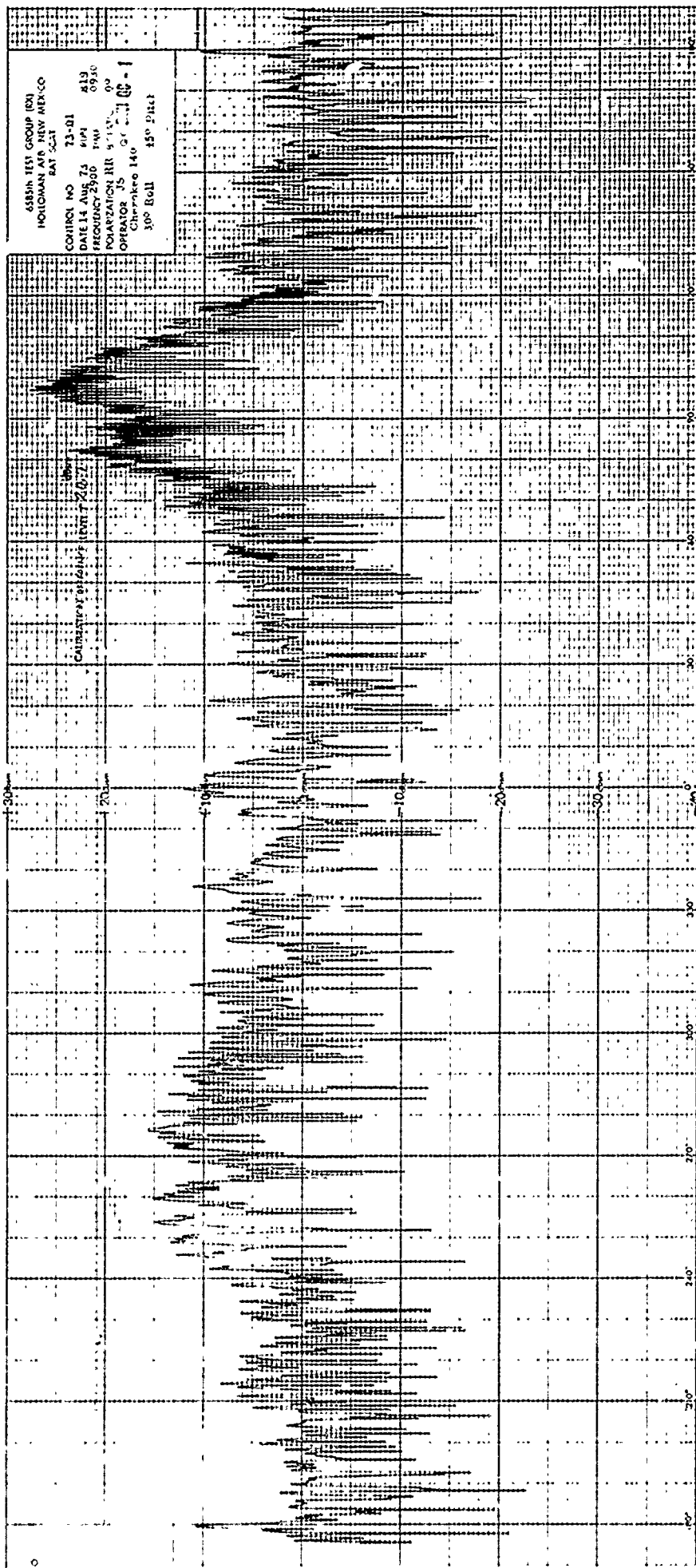
55551 151 GROUP IN
HOLCOMB, NEW ADICO
LAT 50N

WIRE NO. 72-01
DATE 13 AUG 72 RUN 775
FREQUENCY 2700 KHz 1153
POLARIZATION RSTARS 0°
OPERATOR JS OC.DYN 00-1
Chevrolet J40
200 Roll 1100 Pitch









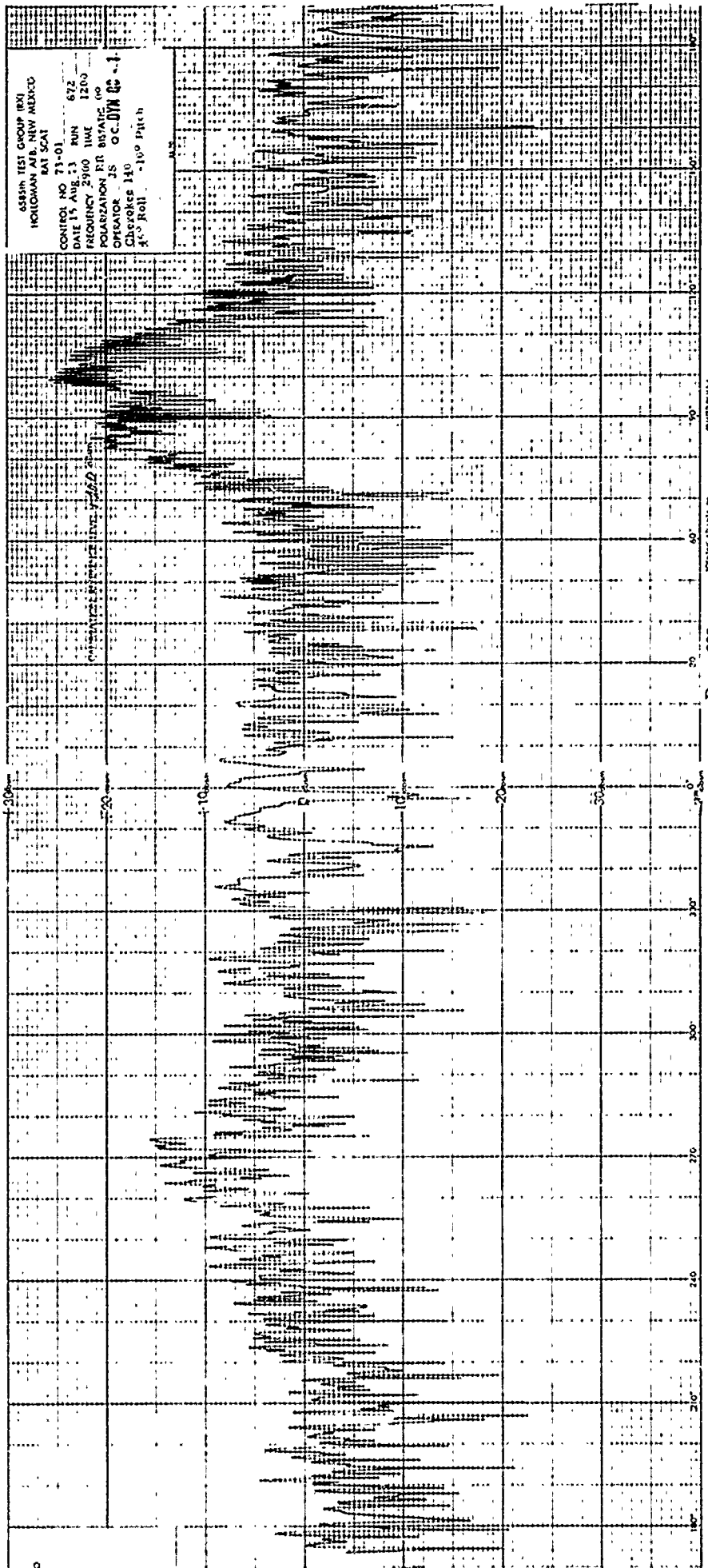
6557h TEST GROUP (R)
HOLOWARE AIR NEW MEXICO
EAT 504

CONTROL NO 71-01 822
DATE 14 AUG 73 RUN
FREQUENCY 2900 TIME 1100
POLARIZATION REGISTRATION
OPERATOR JS Q.C. 00-1
Cherokee 11V
30° Roll *10° Pitch

CHIRP ON AIRCRAFT FOR 10.57

10
20
30

100
110
120
130
140
150
160
170
180
190
200



ASBURY TEST GROUP (EX)
HOLLAND AIR NEW MEXICO
PAT 504
CONTROL NO 11-01
DATE 15 AUG 73
FREQUENCY 2700 kHz
POLARIZATION RR 55A/C
OPERATOR JN QC
Chevrolet 180
450 Roll 450 Pitch

CONTROL NO 11-01

DATE 15 AUG 73

FREQUENCY 2700 kHz

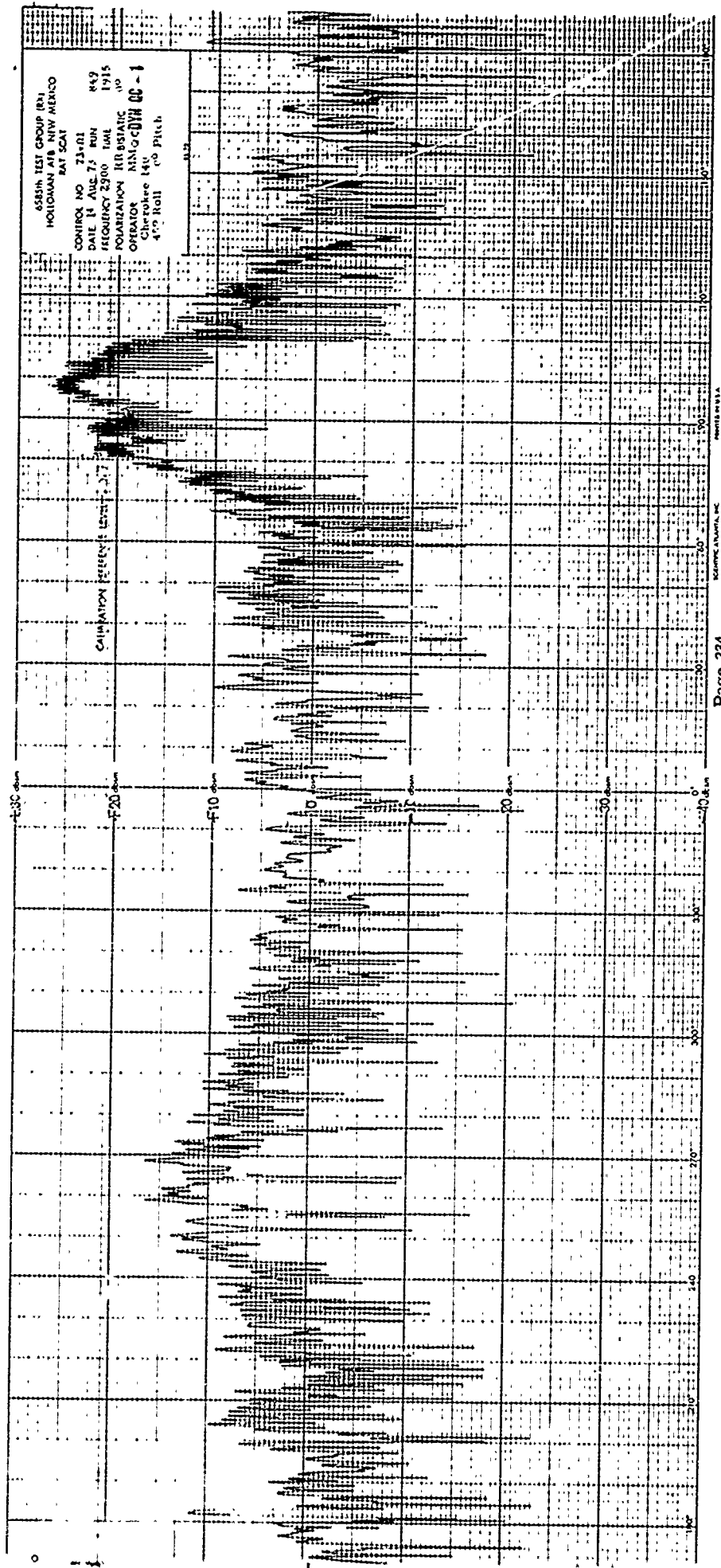
POLARIZATION RR 55A/C

OPERATOR JN QC

Chevrolet 180

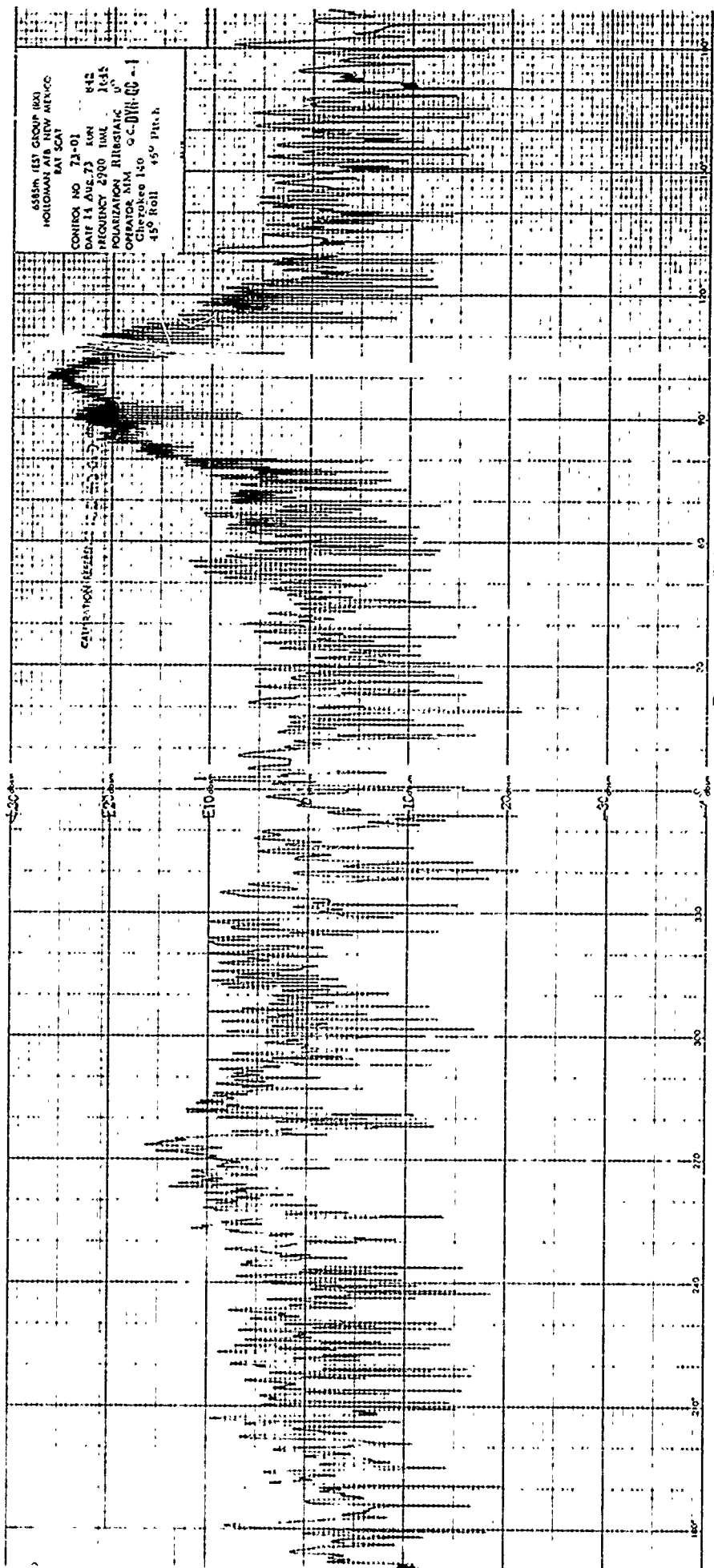
450 Roll

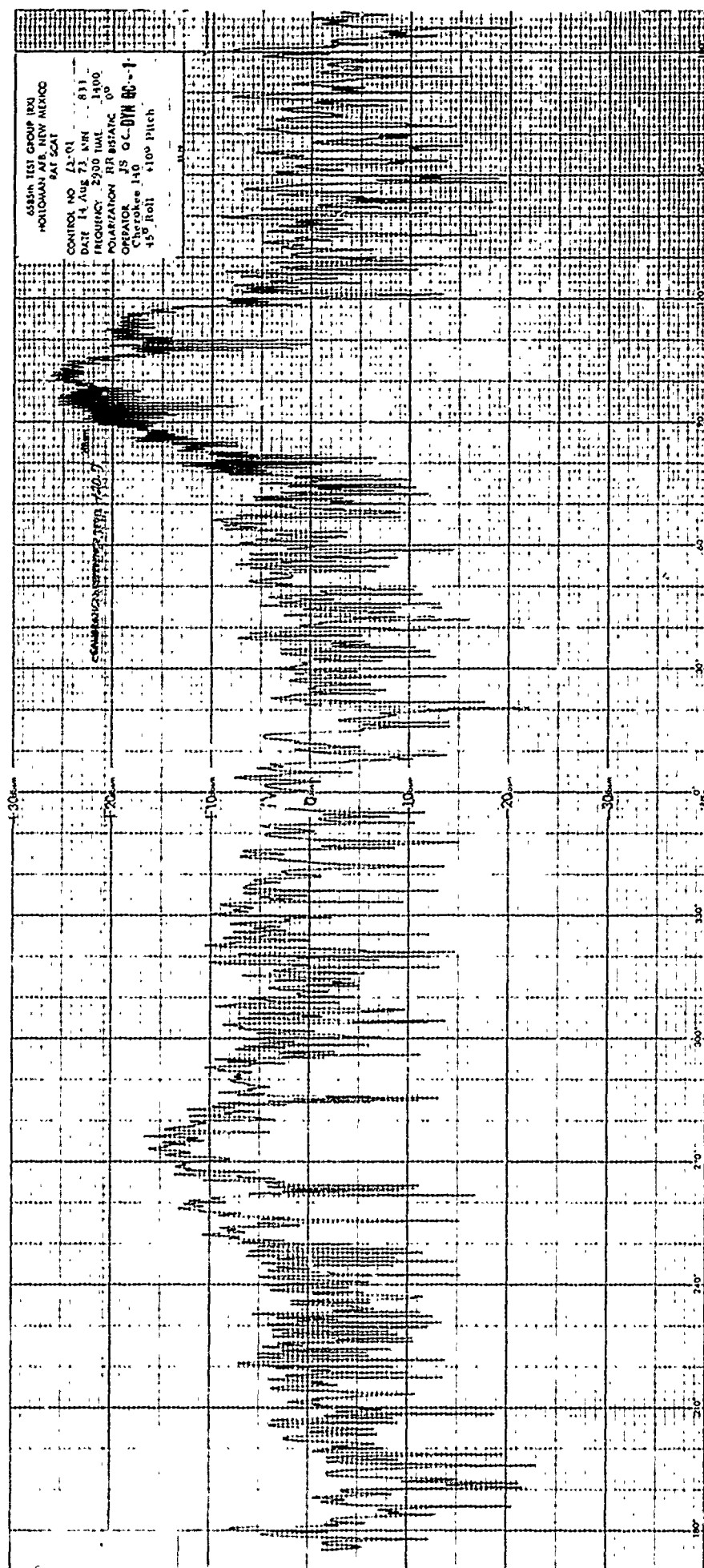
450 Pitch



6533th TEST GROUP (R1)
HOLLAND AFB NEW MEXICO
EAT SCAT
CONTROL NO 73-01
DATE 14 AUG 75 RUN 849
FREQUENCY 2300 TAIL 1915
POLARIZATION HORIZONTAL
OPERATOR NINOCORON 00-1
Cherokee 140 100 Pitch
400 Roll

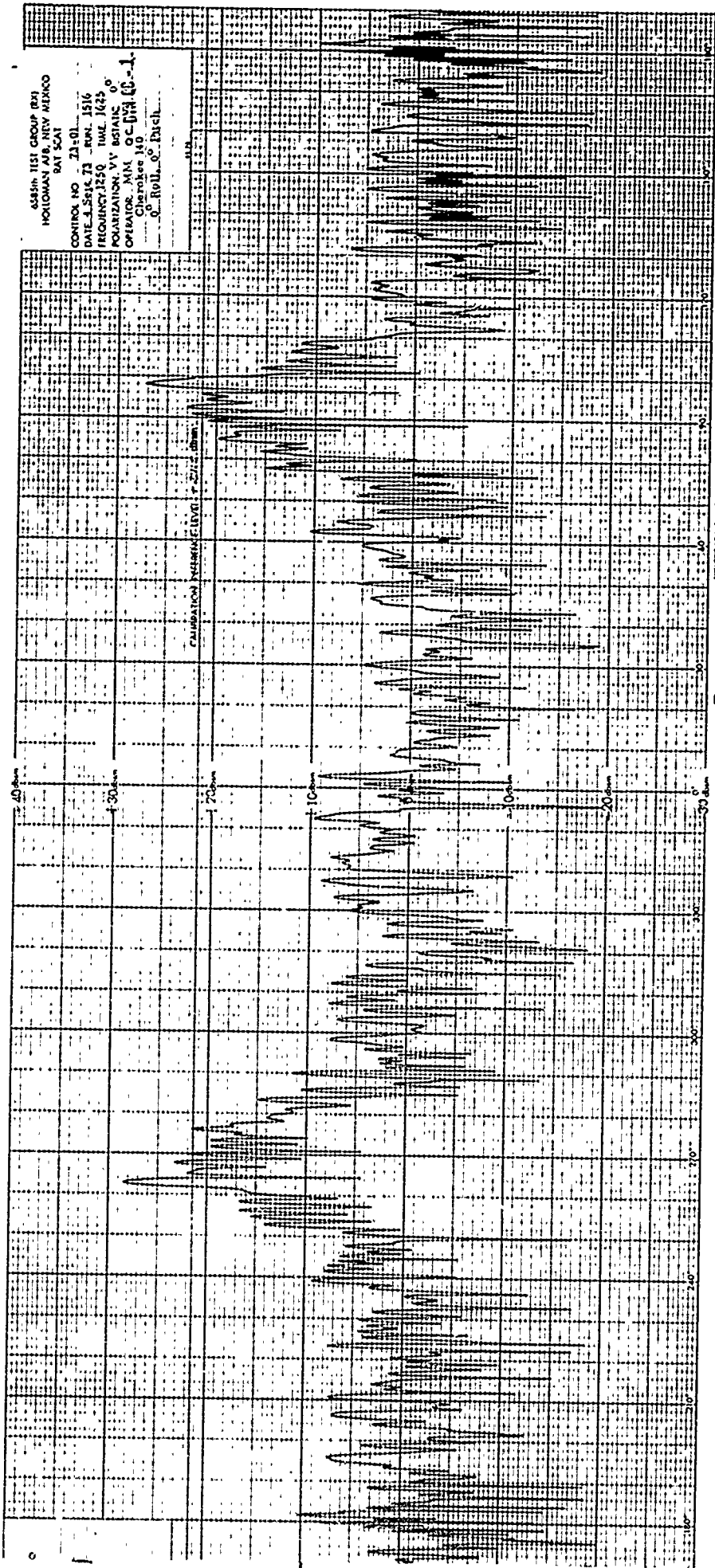
COMPUTATION SYSTEMS UNIT

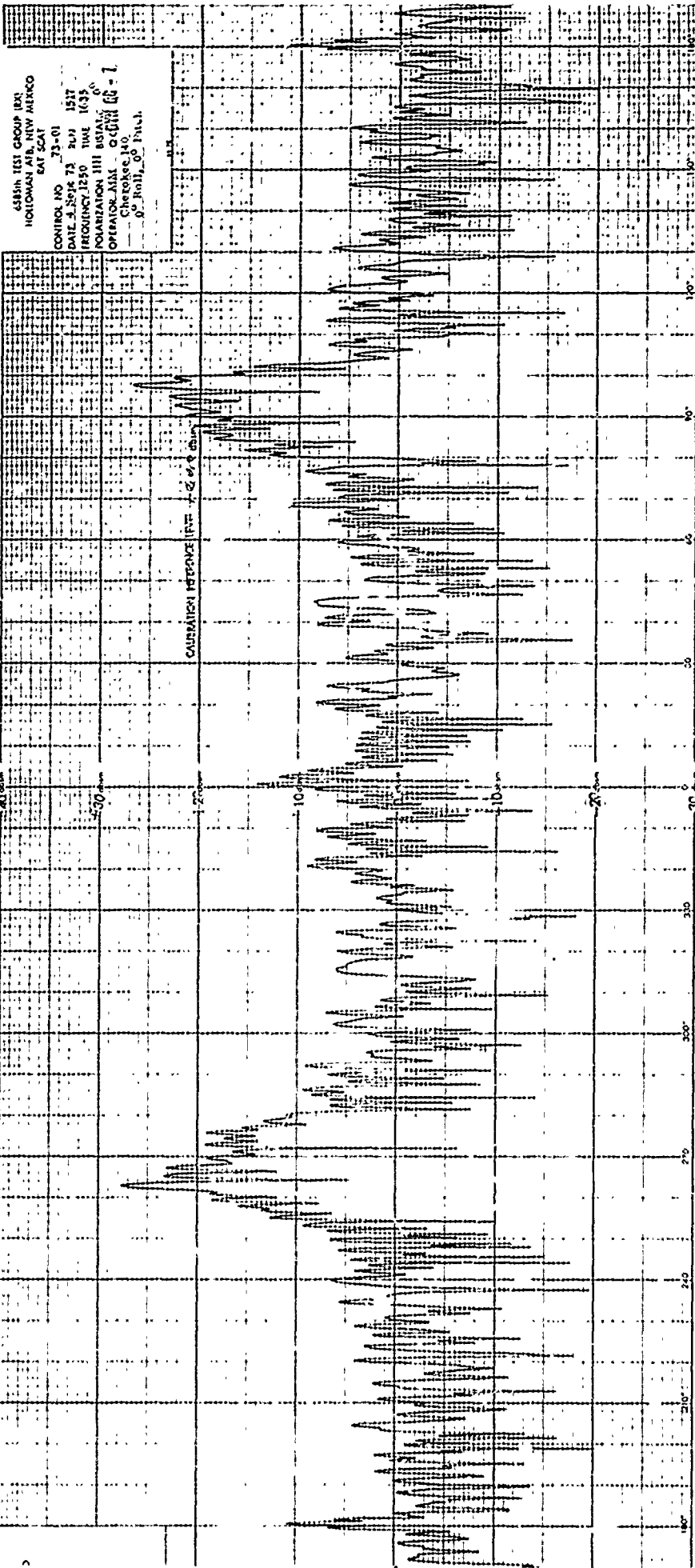


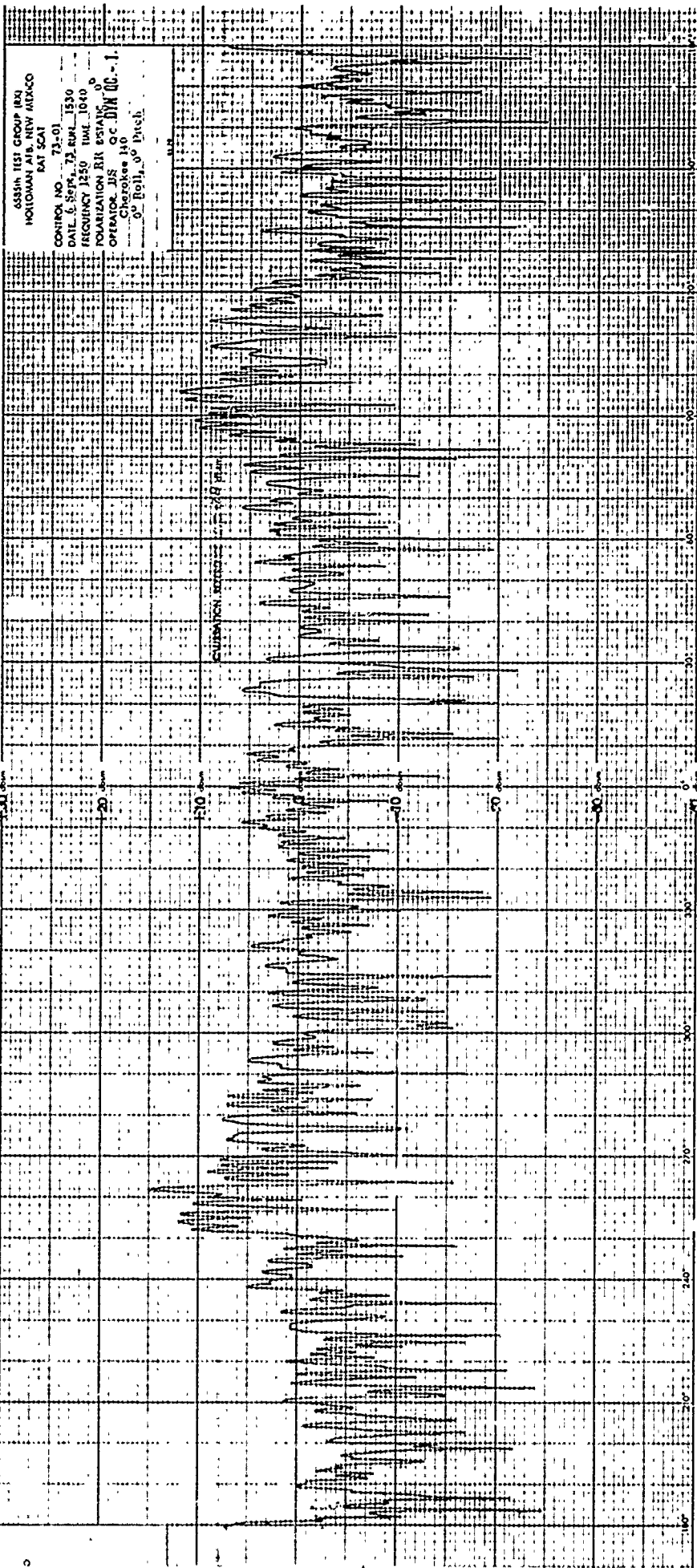


655th TEST GROUP (B1)
HOLLAMAN AFB, NEW MEXICO
BAT SCAT

CONTRACT NO. 21-01
DATE 1 Sep 73 RUN 1516
FREQUENCY 1250 TIME 1625
POLARIZATION VV DISTANCE 0°
OPERATOR MM GC-1
Cherokee 10
0° Roll 0° Pitch







6585th TEST GROUP (EX)
HOLLOWMAN AFB, NEW MEXICO
RAT SCAT

CONTROL NO. _____

DATE 4 Sept 73 RUN 1519

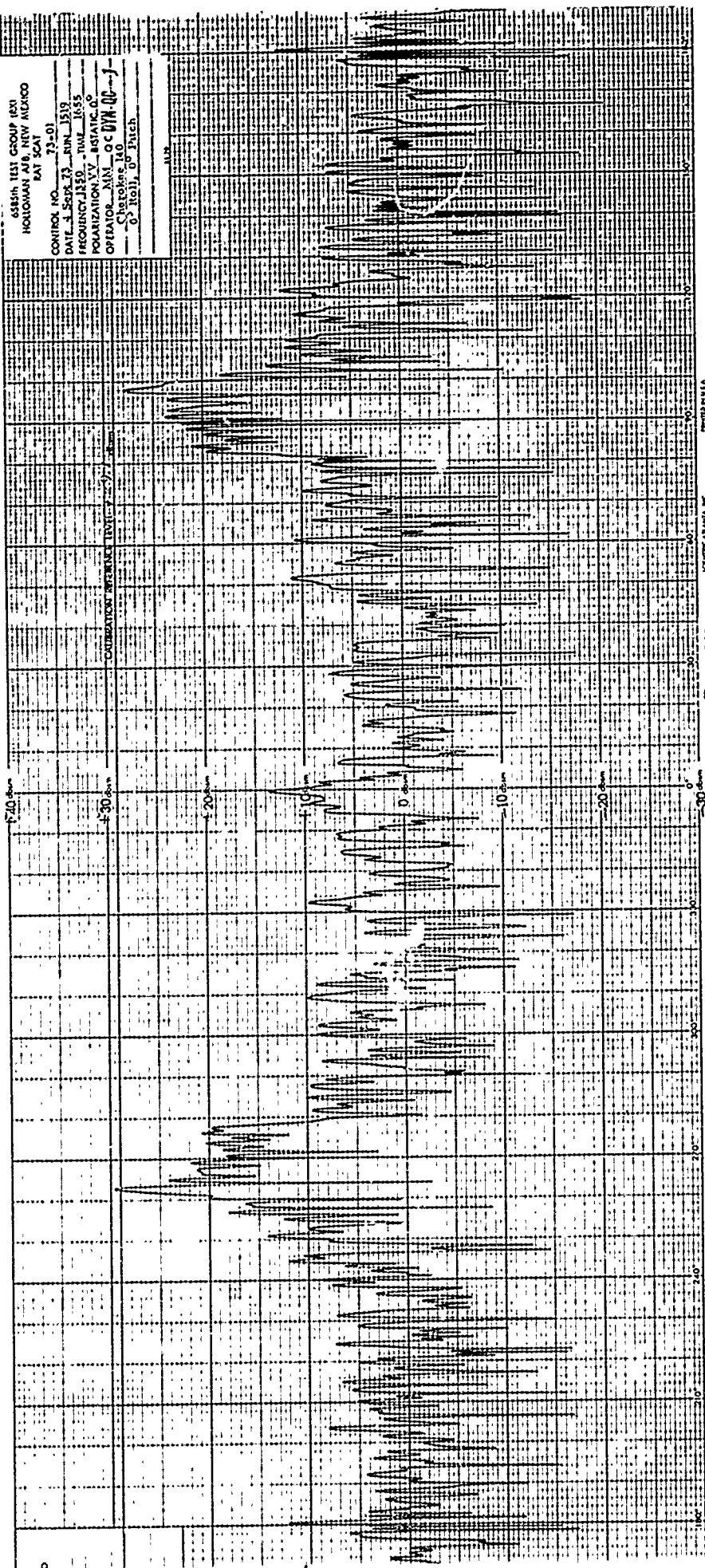
5591 7mu 033725NNOJ 1655

POLARIZATION, V.V. DISTANCE, D.

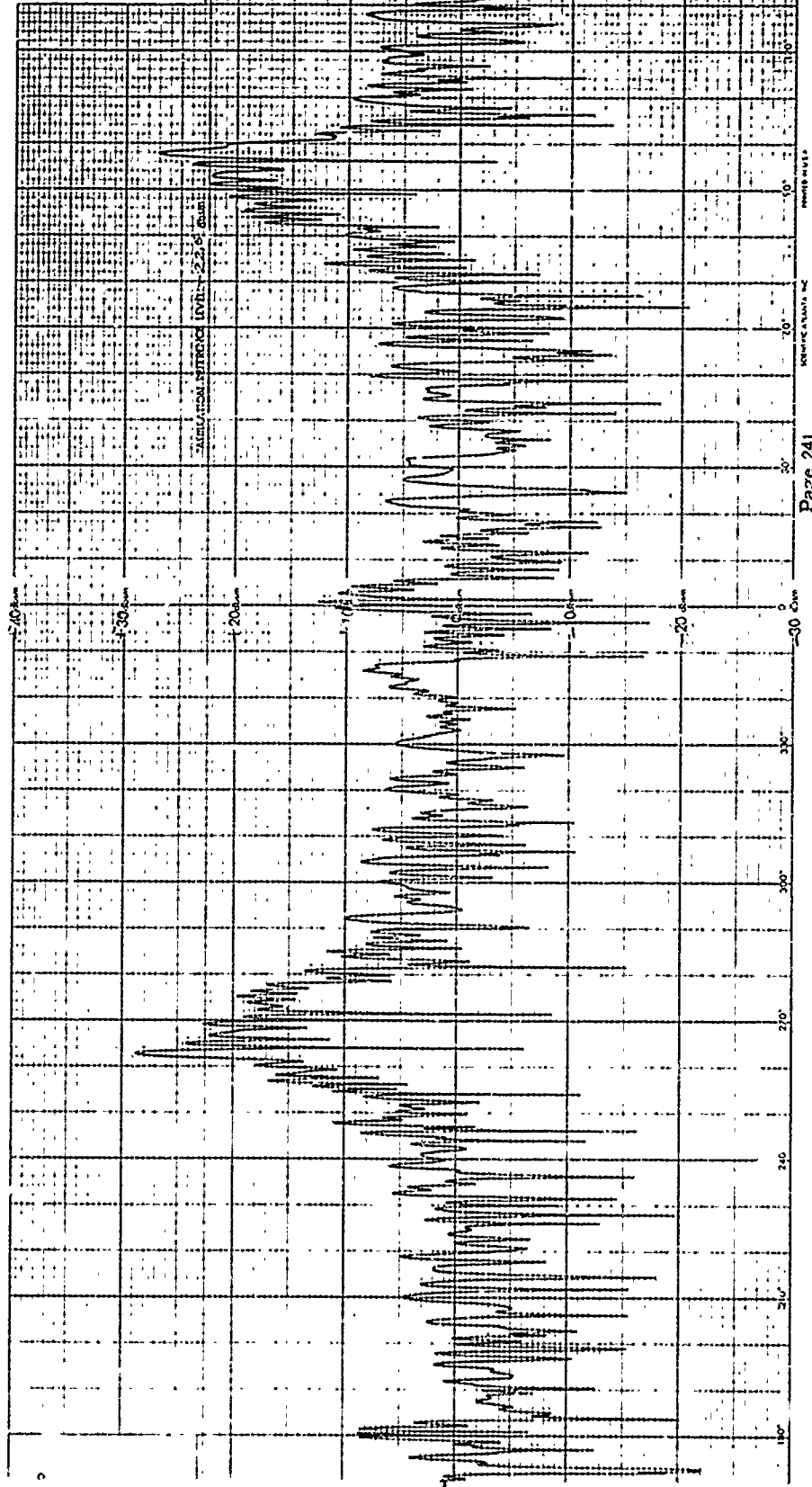
OPERATOR M41-96 DVM-OC-1-

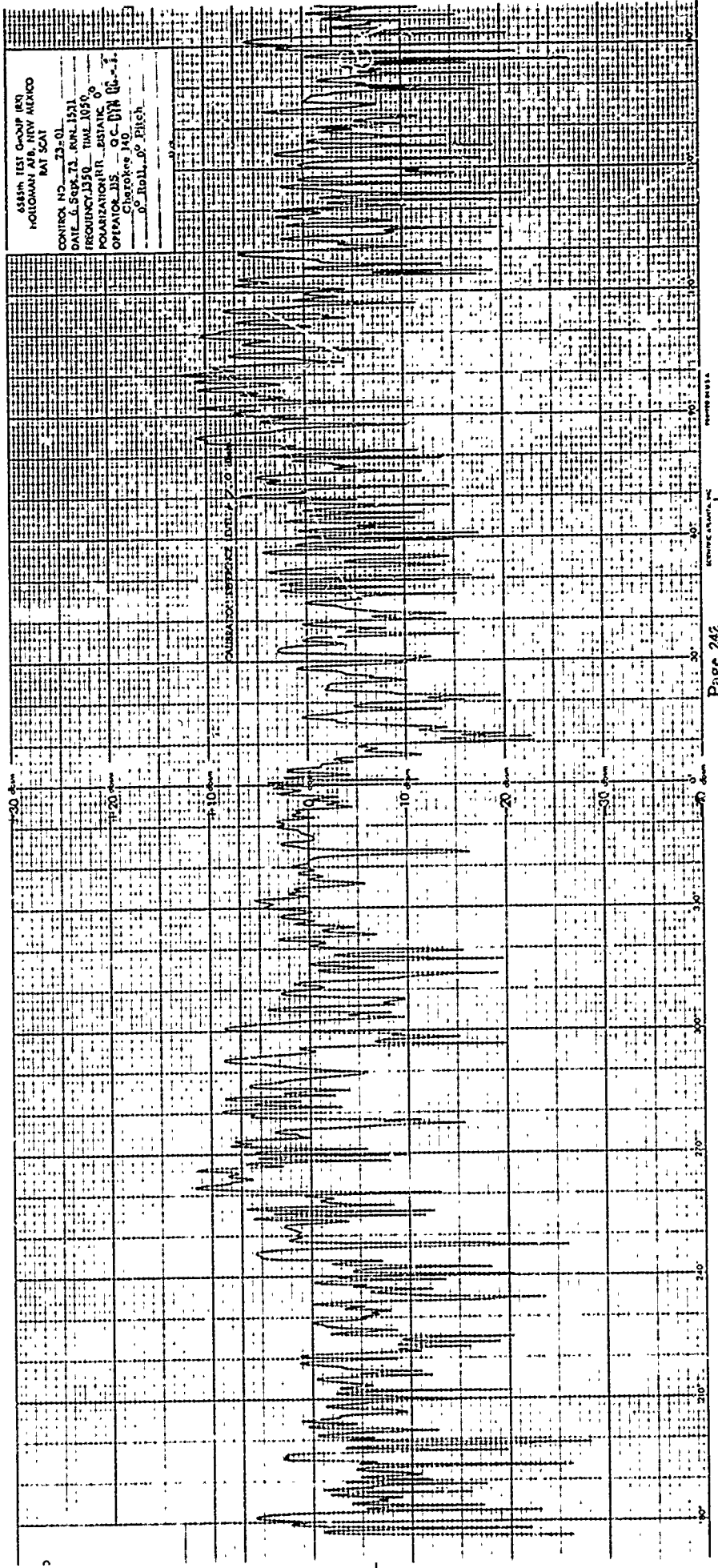
Cherokee 140

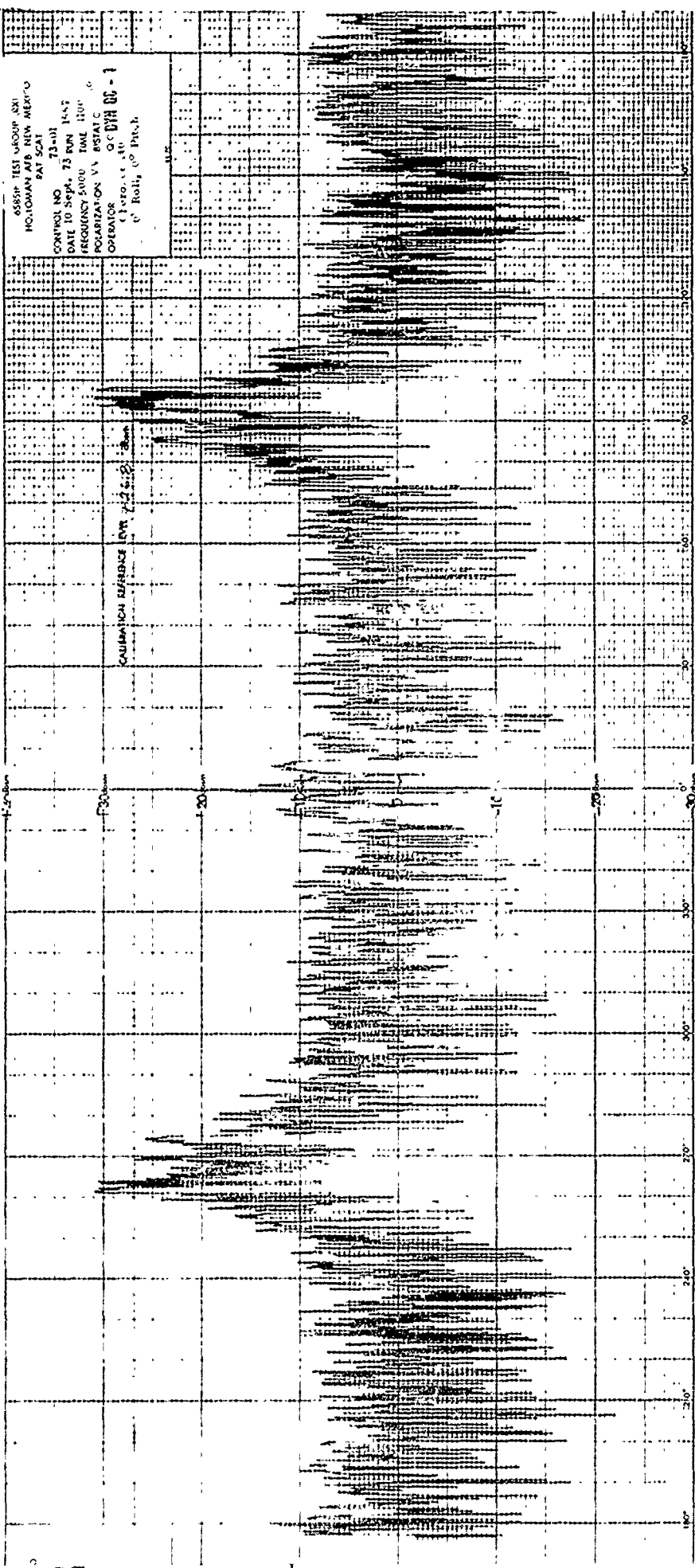
0° Roll, 0° Pitch

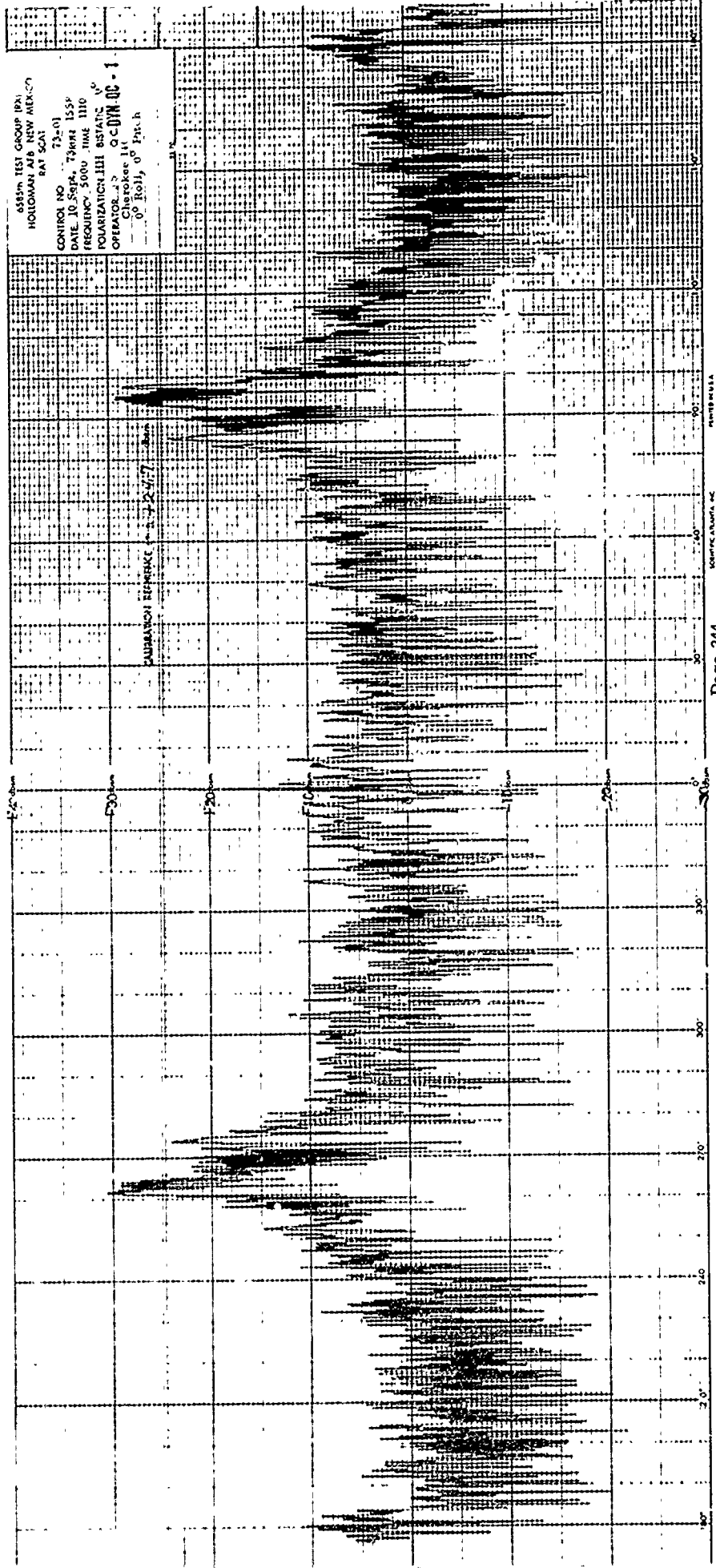


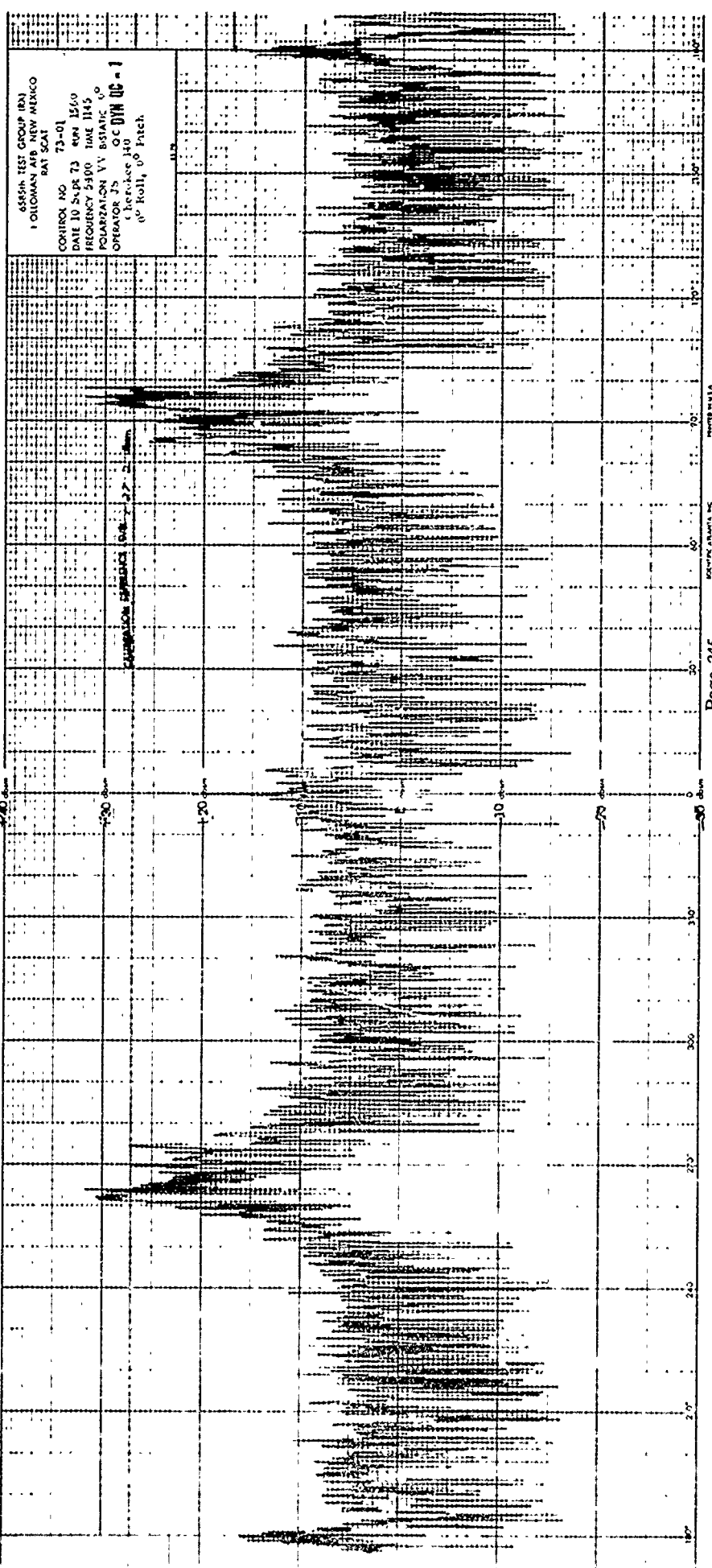
ASSUR TEST GROUP DIV
 HOLLOMAN AFB, NEW MEXICO
 RAT SCAT
 CONTROL NO. 79-01
 DATE 3 SEP 73 RUN 1518
 FREQUENCY 1350 THAL 1645
 POLARIZATION III ROTARY 0
 OPERATOR MM. oc. DVA 40-1
 Cycles 110
 0° Roll 0° Pitch











6155th TEST GROUP '301
HOLLAMAN AFB, NEW MEXICO
LAT 34°N

CONTROL NO 73-01

DATE 10 Sept 73 RUN 1159

FREQUENCY 5400 TIME 1125

POLARIZATION IIII BISTATIC

OPERATOR JS CC 1111 1125

Capacitance 140

Roll, 0° Pitch

ORIGINATOR: 6155th TEST GROUP '301

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APPENDIX A
SITE INTRODUCTION

1. GENERAL

RAT SCAT is a static ground plane radar cross section measurement site, located on Alkali Flats near Holloman Air Force Base, New Mexico. It is authorized by the DOD for use by governmental agencies. It is under the auspices of the 6585th Test Group, Air Force Special Weapons Center, Kirtland Air Force Base, New Mexico.

A ground plane range utilizes radar energy reflected from the earth as well as radar energy traveling directly to the target through the atmosphere. When the antennas and target are adjusted to proper heights, coherent phase addition of these electromagnetic waves into a flat wave front, enhances the system sensitivity. Radar returns from objects near the earth's surface are reduced thus suppressing target area interference. Target area interference is reduced further through the use of special polyfoam support columns, radar absorptive materials (RAM), and rotators located below the earth's surface (in pits).

Pulsed transmitters are employed to enable utilization of the range gated receiving system, which can selectively measure radar returns from the target area or the range displaced transfer standard. Background interference outside the target range is eliminated by range gating. Operation without background cancellation is therefore practical.

2. CAPABILITIES

The RAT SCAT electronic equipment and controls are housed in a permanent building. Three separate range lengths (458 feet, 1158 feet, and 2458 feet) are provided for range variation as shown in Figure A-1. This allows the use of convenient antenna and target heights while satisfying the far field criterion for most targets. (Special 40-foot antenna towers are attached to the building for antenna height positioning.) Further versatility is provided by two mobile equipment vans, one for monostatic range length variation and one for bistatic measurements. A duplicate set of control and data consoles in the main building enables simultaneous operation of any two of the three ranges. A summary of the RAT SCAT characteristics is contained in Table A-1.

3. CALIBRATION

The normal method of calibration at RAT SCAT is to mount a primary standard (precision sphere) scatterer with a known radar cross section and record the corresponding signal level. Then the return from another secondary standard (corner or Luneberg lens) scatterer

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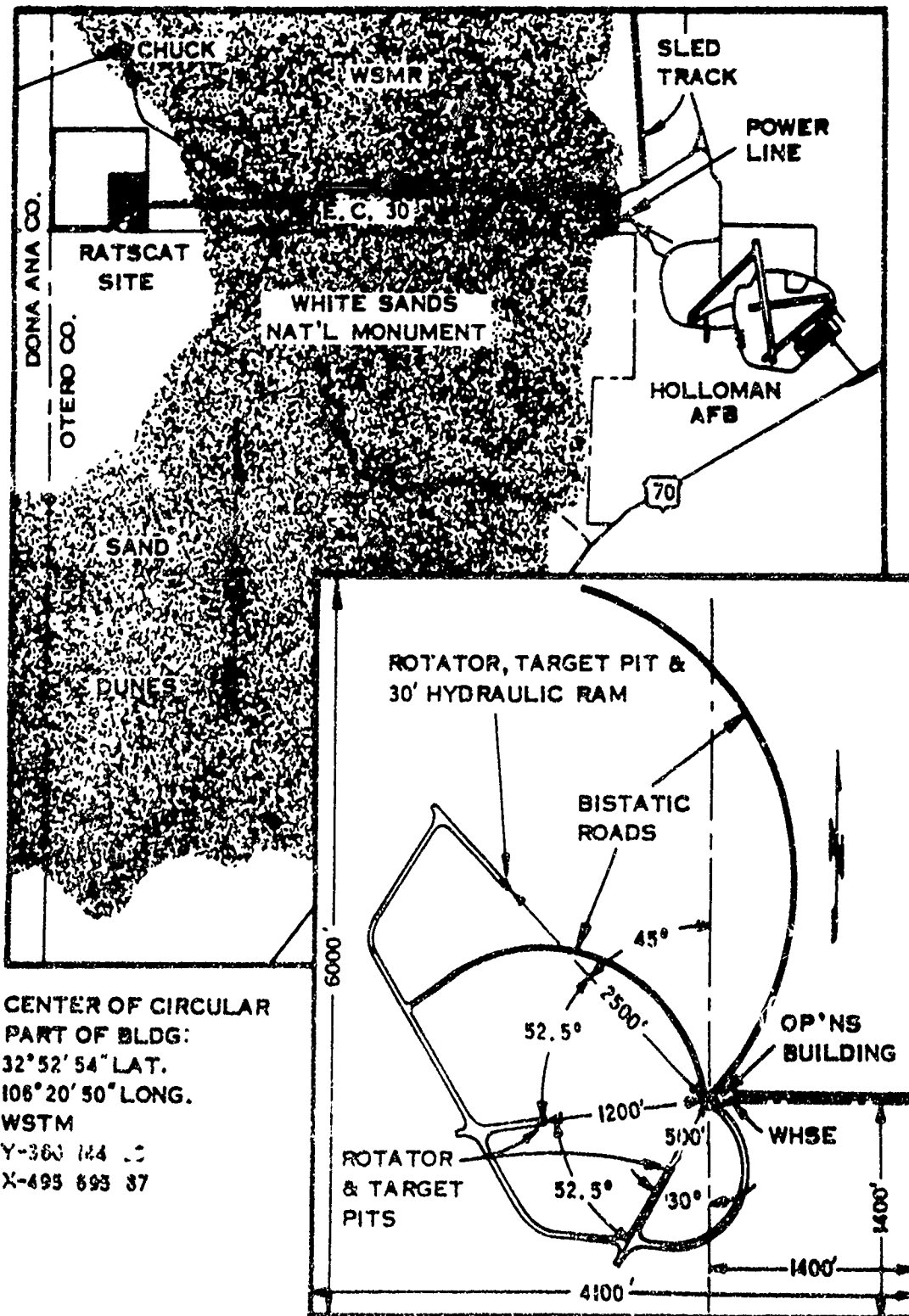


Figure A-1 MAP OF RAT SCAT SITE

A-2

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TABLE A-1

RAT SCAT CHARACTERISTICS OF ELECTRONIC EQUIPMENT

Power Output	1 KW nominal bands 1 through 8, 25 KW nominal Ku, Ka bands
Pulse Width	0.1 to 1.0 microsecond
Pulse Repetition Frequency	500 to 5000 pps
No. of Receiving Systems	Two per band, (one monostatic and one bistatic)
Receiver Minimum Detectable Signal	-94 dbm nominal
Receiver Bandwidth	2 or 10 Mhz (selectable)
Range Gate Width	0.1 to 1.0 microsecond (50 to 500 feet)
Dynamic Range	70 db
Linearity	± 0.5 db
Equipment Stability	0.1 db/hour (Average)
Analog Data Format	Polar and rectangular plots of cross section, glint and phase vs aspect angle
Digital Data Format	7 or 9 track magnetic (see Appendix C)
Antennas	1, 2, 3, 4, 6, 10, and 16 foot parabolic dishes (smaller and larger dishes available for special tests)
Antenna Feeds	Linear and circular horns with VSWR less than 2.0 to 1.0
Polarization	Horizontal, vertical, circular, elliptical in any transmitting and receiving configuration.
Background Level	As low as -80 dbsm (frequency dependent)
Background Reduction	Tuned columns and vector subtraction by using phase and amplitude measurements to reduce background by 20 db
Phase Measurement	Unique RAT SCAT capability for vector subtraction or scattering matrix applications
Azimuth Resolution	0.1 or 0.01 degree as applicable
Maximum Target Weight	40,000 pounds
Target Size	Greater than 60-foot length
Bistatic Capability	Primary ranges of 458 , 1158 , and 2458 feet for 0 to 160 degree bistatic angle
Frequency Coverage	100 to 18,000 MHz continuous, Ku, Ka bands and 95 GHz

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Band 1 - 100 to 250 MHz
Band 2 - 250 to 500 MHz
Band 3 - 500 to 1000 MHz
Band 4 - 1000 to 2000 MHz
Band 5 - 2000 to 4000 MHz
Band 6 - 4000 to 8000 MHz
Band 7 - 8000 to 12,000 MHz
Band 8 - 12,000 to 16,000 MHz

Ku, Ka bands;
95 GHz

Range Length 300 feet minimum

Building/Pit 1 - 458 ft
Building/Pit 2 - 1158 ft
Building/Pit 3 - 2458 ft
Monostatic Van/Pits 1, 2, or 3 - variable range length

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displaced in range is recorded as a transfer standard. Both the precision standard return and the transfer standard return are recorded on the same plot. Thereafter, radar cross section calibration is determined by referencing the transfer standard return for every run. Thus every run is recalibrated. The comparisons of primary and transfer standards accomplished before and after each measurement series are identified respectively as calibration and post-calibration. If the direct ratio of primary to secondary readings is not maintained before and after the measurement series, then all runs between are invalid and must be repeated.

The calibration reference level marked on each data plot is related to the transfer standard level. This reference level may under controlled conditions differ from the actual transfer standard signal level since precision calibrated attenuation is sometimes inserted in the receiver line. When such attenuation is inserted, returns from the transfer standard are reduced to a level compatible with the scale used for the target measurements. The 70 db dynamic range of the plot is placed to include the range of returns expected from the vehicle being measured. In some cases two runs are necessary to be plotted for direct overlay to include the dynamic range of the vehicle if it exceeds 70 db. Calibration plots are included with the target data when requested by the user.

The sphere calibration plots will not necessarily be straight lines. If the background return is within 20 db of the sphere return, for example, a variation in sphere return of approximately ± 1 db can result. For calibration the sphere is intentionally placed at least $1/2$ wavelength off the center of table rotation to insure sufficient phasing with the background return. The average sphere return is then chosen for a calibration level. This avoids the peak errors involved with coherent addition of sphere return and background return and allows the minimum errors involved with non-coherent addition of the returns. This is indicated in Figure A-2.

4. OPERATING PROCEDURES

The following step-by-step procedure is standard in obtaining monostatic radar cross section measurements after frequency, feeds, antennas, antenna height, target height, and pit (range length) have been chosen:

1. Calibration - As described in previous section.
2. Horizontal and vertical probes (field strength measurements at the target area) - Horizontal probes at the target area have been shown to be redundant for azimuthal boresighting. For this reason, these probes are taken only upon request for examination of near field effects.

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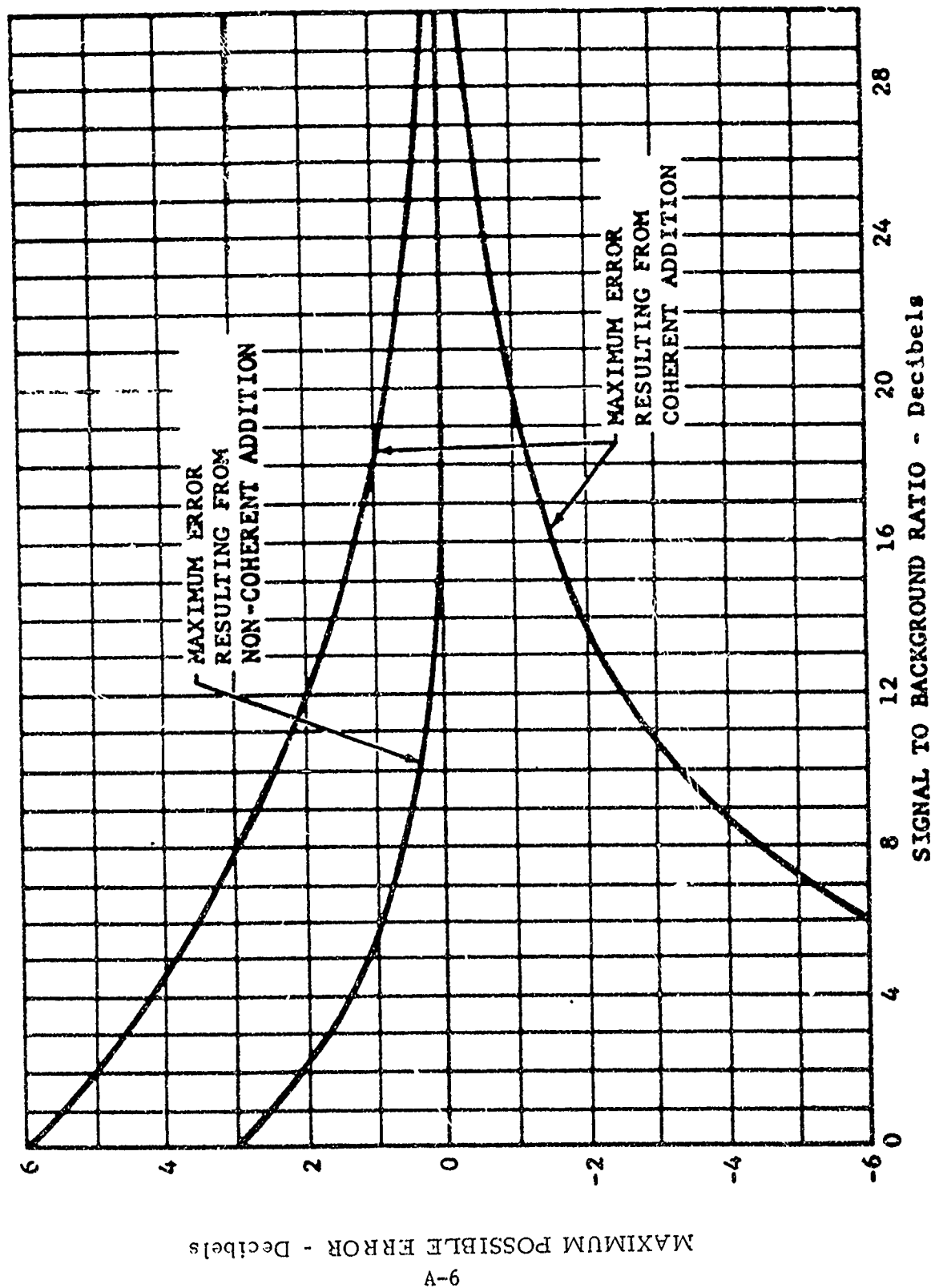


Figure A-2 PLOT OF ERROR INDUCED BY BACKGROUND INTERFERENCE

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Vertical probes are taken at the target area to determine power variation as a function of target height. If necessary, antenna height is varied to obtain an acceptable vertical probe which then necessitates a new calibration.

3. Background - The background level with the target mount in place is measured in each polarization to be used.
4. Measurement - The measurement is made with the vehicle in the position previously occupied by the primary standard.
5. Calibration - The primary calibration is repeated to verify calibration (post calibration).

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APPENDIX B
TARGET ORIENTATION AND DATA FORMAT

1. COORDINATE SYSTEM

The coordinate system described herein has been adopted as a standard for RAT SCAT operations. The system is referenced both to the vehicle being measured and to the measurement site.

a. Vehicle Reference

A three-axis system, referenced to an arbitrary vehicle, is illustrated in Figure B-1. In this system three mutually perpendicular planes (yaw, pitch, and roll) are passed through the vehicle so that the pitch and yaw planes mutually intersect on the longitudinal axis of the vehicle. These planes remain fixed with respect to the vehicle, regardless of vehicle rotation with respect to the radar or ground plane. The yaw plane, which includes the pitch axis and the roll axis, is numbered from 0 degrees to 360 degrees in a clockwise direction when the vehicle is viewed from the above. The nose-on aspect corresponds to 0 degrees, the starboard side of the vehicle corresponds to 90 degrees, and the port side to 270 degrees. The pitch plane, which contains the roll axis and the yaw axis is numbered from 0 degrees to ± 180 degrees; the + 90 degree point is below the center line, and the - 90 degree point is above the center line. The roll plane contains the yaw axis and the pitch axis. It is numbered from 0 degrees to 360 degrees, and the numbers increase in a counterclockwise direction when the vehicle is viewed from the rear.

b. Site Reference

As previously stated the coordinate system is fixed with respect to the vehicle. It is referenced to the site by means of three index marks. The exact value of any of the three angles is determined by noting the value of the vehicle coordinate opposite the index marks. Index marks come from such devices as bubble levels, inclinometers and transits.

As illustrated in Figure B-2, the index for roll angles is normal to the axis of rotation. As illustrated in Figure B-3, the index for pitch angles is normal to the axis of rotation and in line with the apparent source of radiation. For measurements at the RAT SCAT Site, targets can be mounted to provide desired pitch and roll angles.

c. Coordinate System Tilt

For small targets another angle, tilt, can be utilized in recording useful data. This angle, equipment-limited to less than 15 degrees, is formed by the axis of rotation and the normal to the line of sight to the

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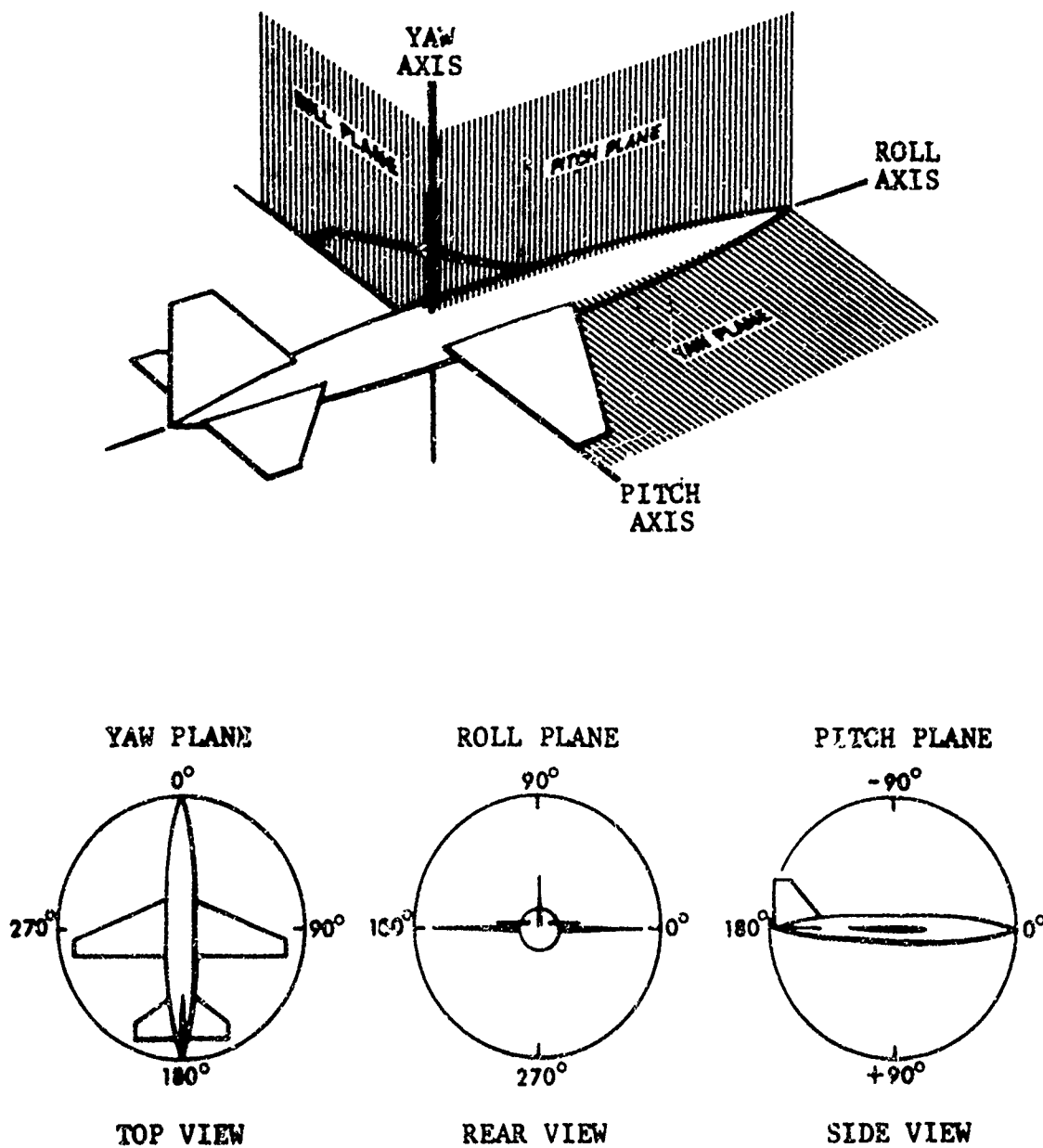
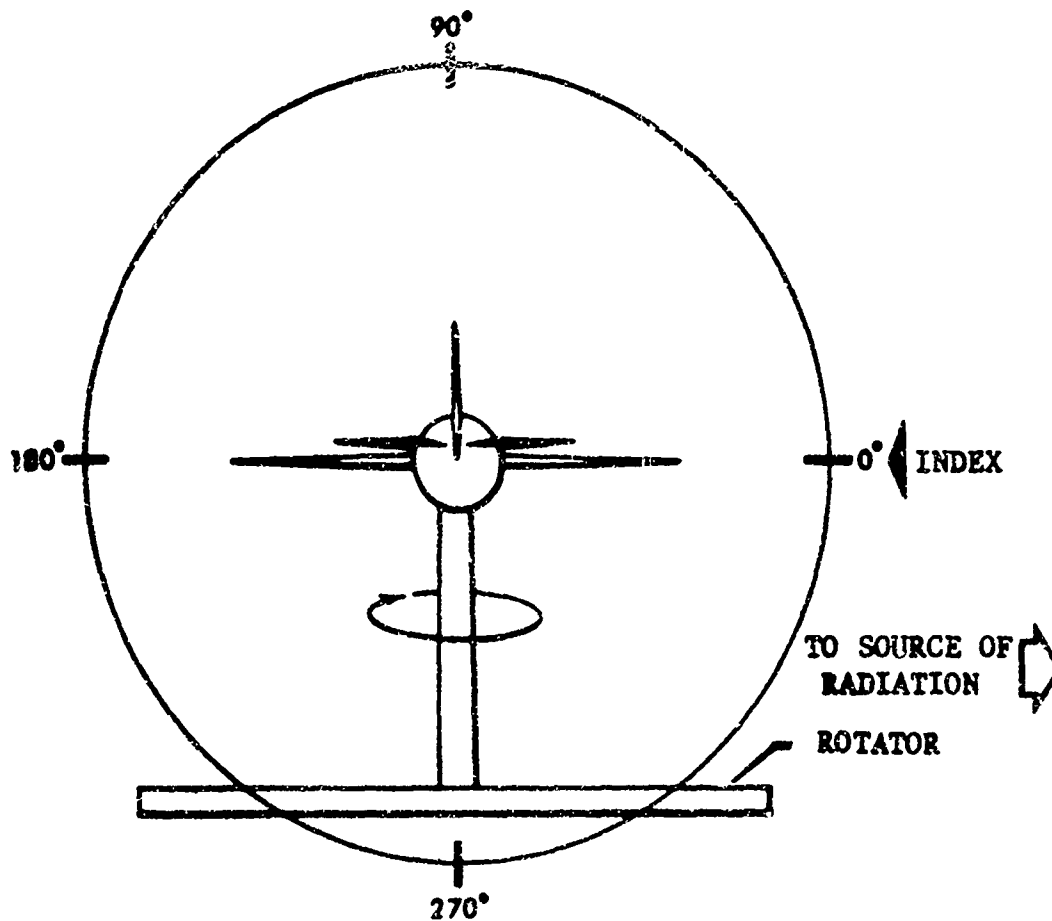


Figure B-1 VEHICLE COORDINATE SYSTEM

B-2

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NOTE: The roll scale is fixed to the vehicle. The amount of roll is determined by noting the number of degrees opposite the index. Clockwise rotation of the target (when viewed from the rear) increases the roll angle.

Figure B-2 TARGET ORIENTATION - ROLL

B-3

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The diagram illustrates a goniometer setup for measuring the angular distribution of radiation. It features a large circular scale with a horizontal plate (INDEX) at its center. The scale is marked with angles: -90° at the top, 0° on the right, $+90^\circ$ at the bottom, and 180° on the left. An arrow labeled "TO SOURCE OF RADIATION" points towards the right. The base of the assembly is labeled "ROTATOR".

Figure B-3 TARGET ORIENTATION - PITCH

apparent source of radiation. Since, in a ground plane range, radiation can be considered to emanate from a point with zero height directly beneath the antennas, a zero-degree tilted axis of rotation is slightly off the geometrical vertical. This small deviation from the geometrical vertical is neglected in the following discussions.

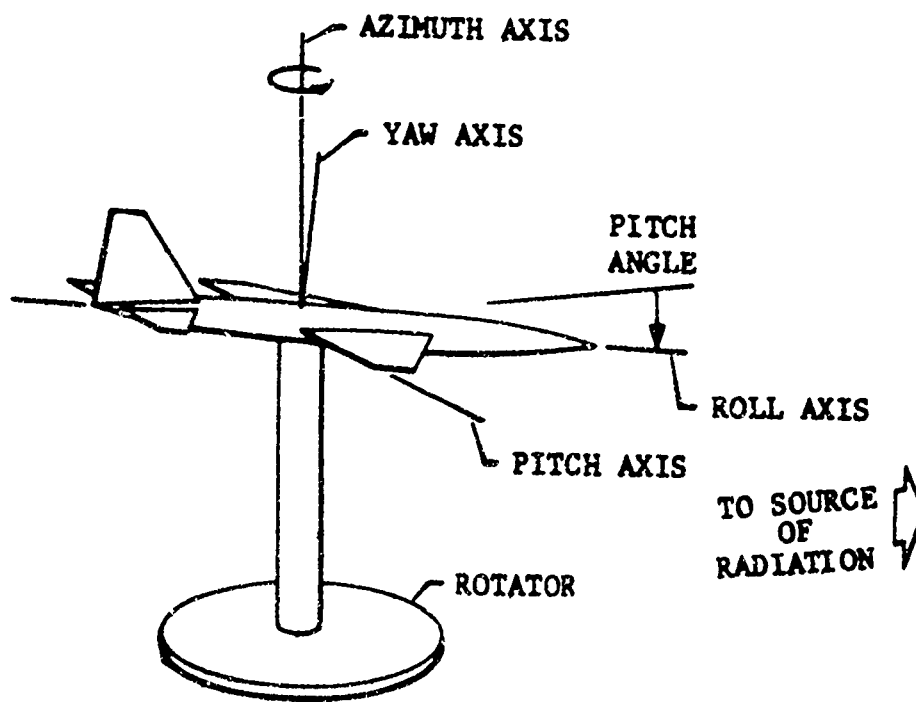
A target mounted with a pitch angle other than zero displaces the yaw axis from the vertical, but not the axis of rotation. The axis of rotation is displaced from the vertical only when non-zero tilt is employed. Tilting toward the radar is considered positive tilt and away from the radar is negative tilt. For monostatic measurements tilt will be measured in the vertical plane containing the line of sight between the radar and the target. The difference between pitch and tilt is shown in Figure B-4.

2. DATA FORMAT

Data recorders obtain azimuth angle information by means of precision synchro signals from the position of the rotating table. The line of sight from the antennas to the center of the rotator, as illustrated in Figure B-5, indexes azimuth angles. As used here the term azimuth refers to the position of the target rotator table. With zero degrees of pitch and roll, azimuth and yaw are identical. It is standard practice to turn the rotator in a clockwise (cw) direction as viewed from above. Consequently, the azimuth angle varies, for example from 180 degrees (tail-on) to 90 degrees (starboard-side) to 0 degrees (nose-on) to 270 degrees (port-side).

a. Polar and Rectilinear Plots

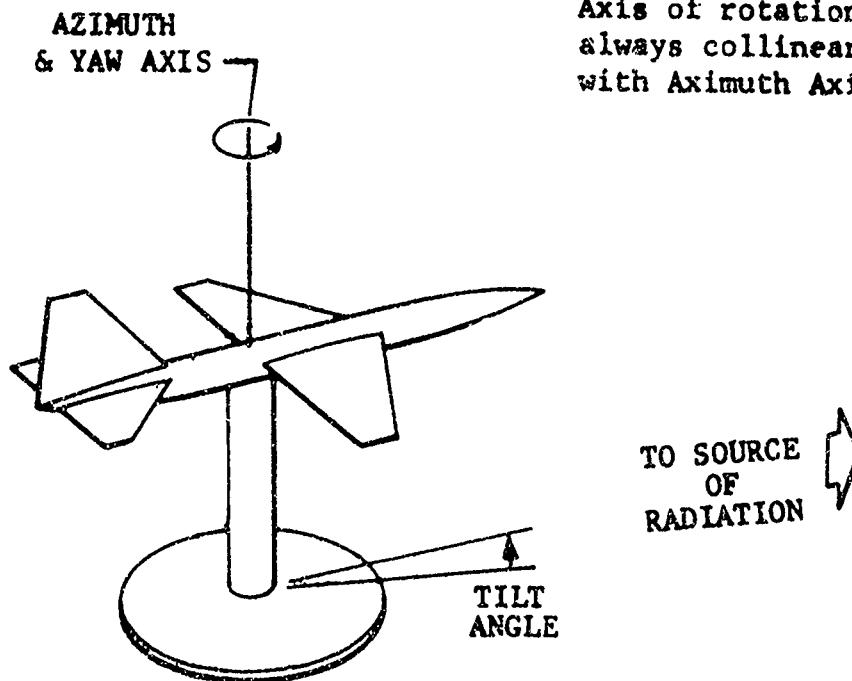
Essential information pertinent to each plot is contained in the information block located in the upper right hand corner of the rectilinear plots and in the second quadrant of the polar plots. Each rectilinear plot has the recording of the return from the left side of the vehicle on the left side of the plot, 0 degrees at the center, and the recording of the return from the right side of the vehicle on the right side of the plot; 180 degrees (tail-on) appears at the right and left extremities of the plot, as shown in Figure B-6. Since the paper moves from left to right under the recorder pen, it should be noted that measurements are limited at 180 degrees in order to obtain continuous measurements on the recorder paper. The table on the polar recorder is rotated in the same directions as the target so the 90-degree point appears on the right side of the polar plot, the 270 degree point on the left, and the zero or 360 degree point at the top of the plot.



PITCH

NOTE:

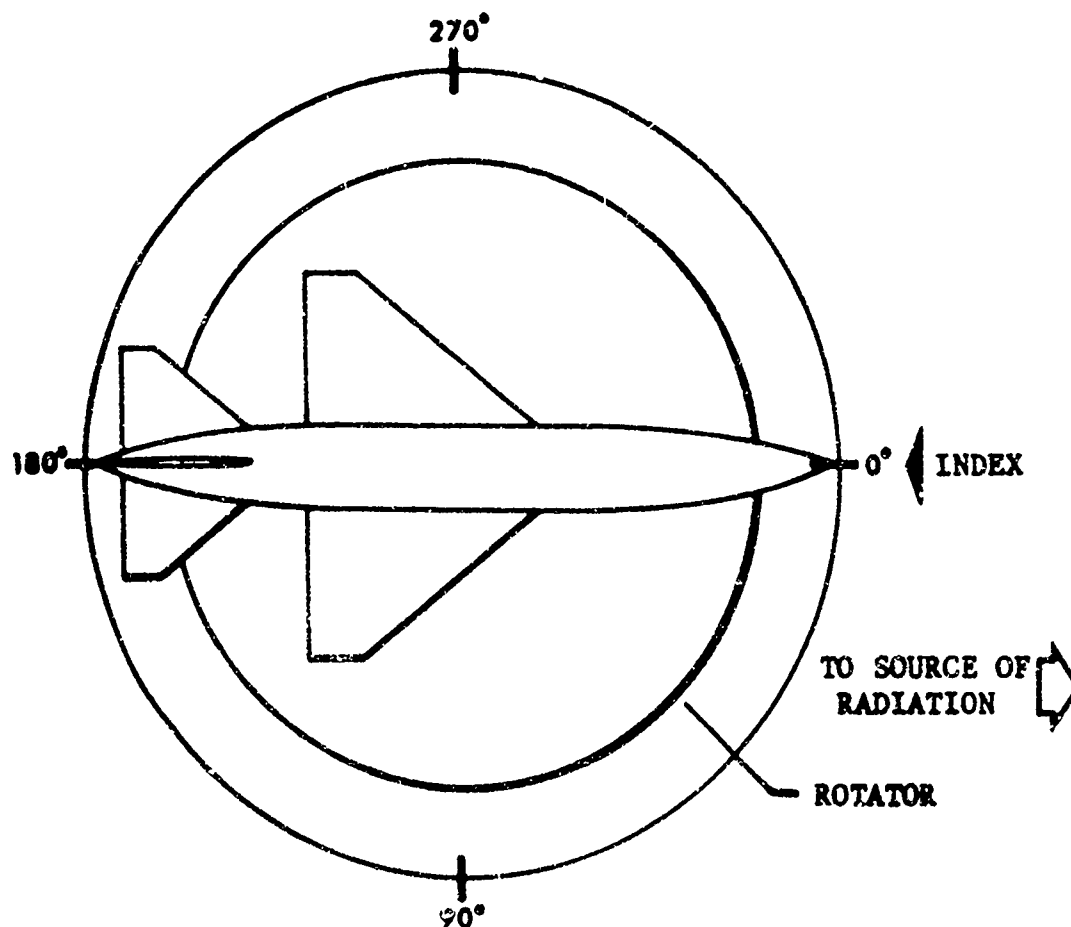
Axis of rotation is
always collinear
with Azimuth Axis



TILT

Figure B-4 COMPARISON OF PITCH AND TILT ORIENTATIONS

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NOTE: The azimuth scale is fixed to the target rotator. The azimuth value is determined by noting the value of the scale opposite the index mark as the rotator and scale revolve. The index is the line-of-sight from the radar antennas to the center of the rotator. (Azimuth angle data are transmitted to the data recorders by means of synchro signals.) The standard direction of rotation will be clockwise.

Figure B-5 TARGET ORIENTATION - AZIMUTH

B-7

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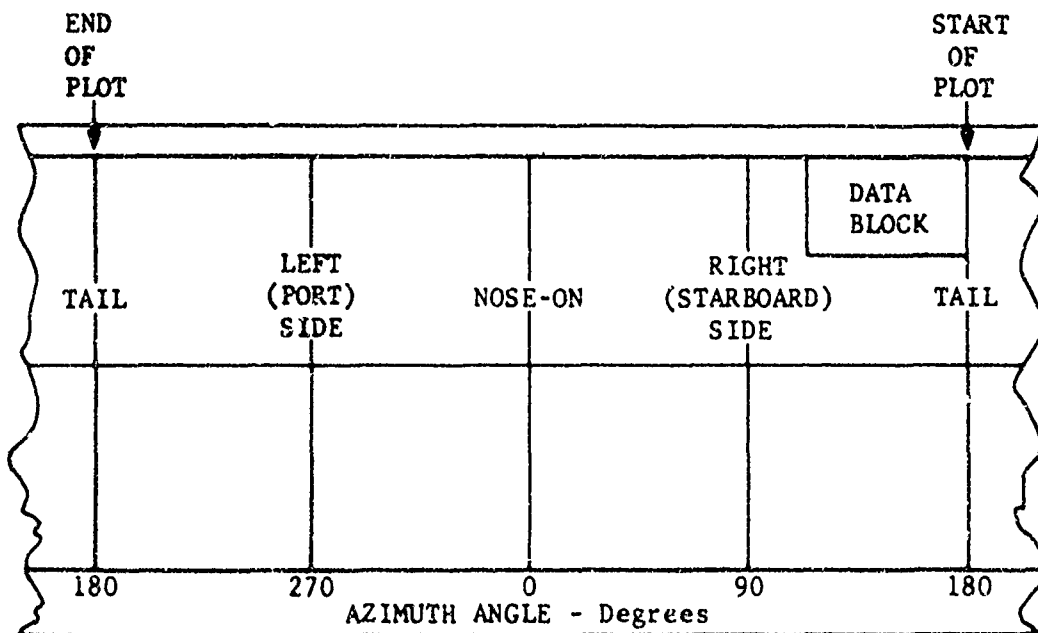
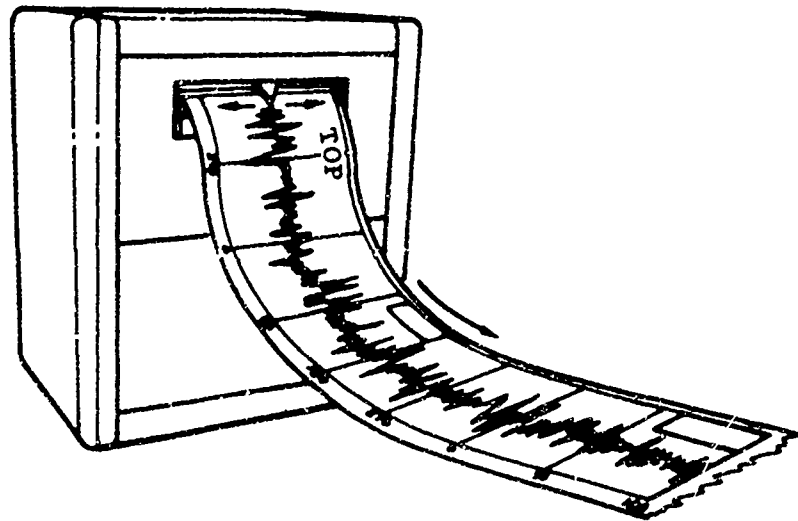
RECTILINEAR
RECORDER

Figure B-6 FORMAT FOR RECTILINEAR PLOTS